Module Leader

Dr W R Graham [1]

Lecturer

Dr W R Graham

Lab Leader

Dr W R Graham

Timing and Structure

Michaelmas and Lent Terms. 14 lectures + 2 examples classes + coursework. Assessment: Coursework/Report/week 9 Lent Term/100%

Prerequisites

A working knowledge of Part IA and IB fluid mechanics and control theory will be assumed.

Aims

The aims of the course are to:

- develop an understanding of the dynamics of an aircraft in flight, and an appreciation of how their characteristics may be improved using automatic control systems.

Objectives

As specific objectives, by the end of the course students should be able to:

- Appreciate how the equations of motion for an aircraft follow from Newton's second law, and how they may be simplified to the small-disturbance form;
- Understand how the free modes of motion follow from the equations of motion, and be aware of the approximate derivations of the modes;
- Know the factors determining the static stability of an aircraft, and understand the significance of the position of the centre of gravity;
- Have a knowledge of basic control strategies for autopilots, and their effects on aircraft stability;
- Appreciate that the dynamic characteristics of the aircraft may be improved by feedback control, and understand how this concept applies to stability augmentation systems, and command augmentation systems.

Syllabus

The flight test part of this module has a number limit of 30. If it is oversubscribed, selection will be made on a competitive basis, subject to priority being given to students in Engineering Areas 3 (Aerospace and Aerothermal...
Engineering) and 8 (Instrumentation and Control). The module can also be taken without participating in the flight tests.

**Aircraft Stability (8L, Michaelmas term, Dr W.R. Graham)**

- Aircraft equations of motion, small disturbance form, stability derivatives.
- Longitudinal motion: phugoid mode, short period oscillation and approximate forms.
- Lateral motion: roll subsidence, dutch roll, spiral mode and approximate forms.
- Static stability of aircraft: longitudinal stability, directional stability, lateral stability.

**Automatic Control Systems (6L, Lent term, Dr W.R. Graham)**

- Root locus plots and their use in designing feedback control systems.
- Response to control inputs.
- Autopilots: pitch and roll angle control, effect on aircraft dynamic response and stability.
- Stability augmentation systems: pitch rate SAS & yaw damper as means of improving dynamic stability characteristics, relaxed static stability.
- Command augmentation systems: C-star criterion as basis for longitudinal CAS in fly-by-wire aircraft.

**Coursework**

Flight tests on Cranfield Jetstream 31 flying laboratory. Assessment of static and dynamic stability based on flight test data. Design study for an automatic control system for the aircraft. A report on the stability assessment and design study forms the basis for module assessment. The flight tests will take place at the end of the Michaelmas term.

**Booklists**

Please see the Booklist for Group A Courses [2] for references for this module.

**Assessment**

Please refer to Form & conduct of the examinations [3].

**UK-SPEC**

The UK Standard for Professional Engineering Competence (UK-SPEC) [4] describes the requirements that have to be met in order to become a Chartered Engineer, and gives examples of ways of doing this.

UK-SPEC is published by the Engineering Council on behalf of the UK engineering profession. The standard has been developed, and is regularly updated, by panels representing professional engineering institutions, employers and engineering educators. Of particular relevance here is the 'Accreditation of Higher Education Programmes' (AHEP) document [5] which sets out the standard for degree accreditation.

The Output Standards Matrices [6] indicate where each of the Output Criteria as specified in the AHEP 3rd edition document is addressed within the Engineering and Manufacturing Engineering Triposes.

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