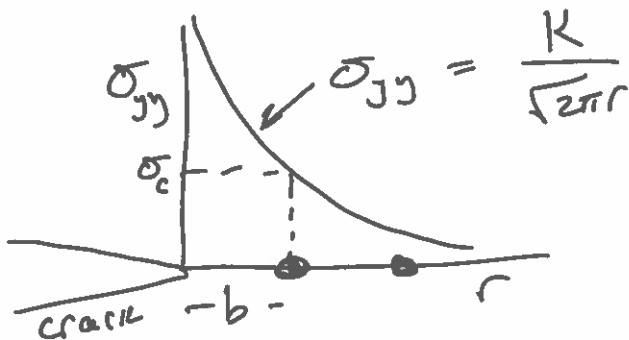


Part IIA 2023-4 crib for paper 3c9

Fracture mechanics of materials and structures

1. (a) Brittle ceramic : fails by cleavage.

Toughness $G_c \approx 2\gamma_s$ where $\gamma_s =$ surface energy by Griffith



Assume that $K = K_{Ic}$
 when $\sigma_{yy}(r=b) = \sigma_c$
 atomic spacing \nearrow
 ideal strength \nearrow

$$\Rightarrow \frac{K_{Ic}}{\sqrt{2\pi b}} = \sigma_c$$

Ideal strength σ_c is on the order of $E/10$

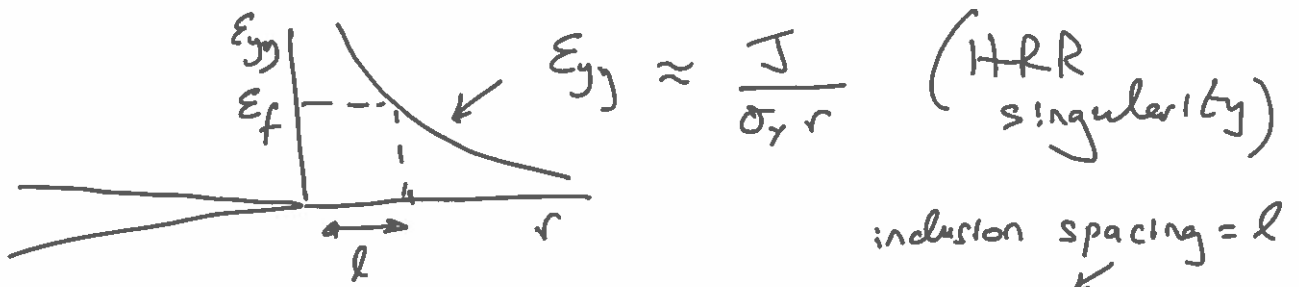
$$\Rightarrow K_{Ic} \approx \frac{E}{10} \sqrt{2\pi b}$$

$$G_{Ic} = \frac{K_{Ic}^2}{E} \text{ by Irwin relation}$$

$$\Rightarrow G_{Ic} \approx \frac{2\pi}{100} E b \approx 2\gamma_s \quad \left(\begin{array}{l} \text{consistent with} \\ \gamma_s \approx \frac{\pi}{100} E b \end{array} \right)$$

steel with inclusions

The toughness is governed by the nucleation and growth of voids directly ahead of the crack tip.



Assume $\epsilon_{yy} > \epsilon_f$ over $0 < r < l$
 \uparrow strain to failure

$\epsilon_f \approx f^{-1/2} \exp\left(-\frac{3\bar{\sigma}_h}{2\sigma_y}\right)$ by Rice-Tracey
 \uparrow volume fraction of inclusions $\bar{\sigma}_h \approx 2.5\sigma_y$
 \downarrow

Thus: $\epsilon_{yy} = \frac{J_{Ic}}{\sigma_y l} = \epsilon_f = f^{-1/2} \exp\left(-\frac{3\bar{\sigma}_h}{2\sigma_y}\right)$

So $J_{Ic} \propto \sigma_y l f^{-1/2}$

Also, inclusion spacing l scales with $f^{-1/2}$
 So a high volume fraction of inclusions
 implies a small l and a low toughness J_{Ic} .
 Under LEFM, $G_{Ic} \equiv J_{Ic}$.

$$1 \text{ (b) (i)} \quad \phi(\theta_0) = \frac{1}{2} M \theta_0^2 = \frac{1}{2} \frac{\theta_0^2}{c} \quad \text{where } \theta = CM$$

potential energy $P(M_0) = \phi - M_0 \theta$
 where $-M_0 \theta$ is the potential energy of the loading system

$$\Rightarrow P = \frac{1}{2} M \theta_0^2 - M_0 \theta = -\frac{1}{2} M_0 \theta = -\frac{1}{2} C M_0^2$$

$$(ii) \quad bG = - \left. \frac{\partial \phi}{\partial a} \right|_{\theta_0} \quad \text{and} \quad EG = K^2 \quad (\text{Irwina})$$

$$\Rightarrow bG = -\frac{1}{2} \theta_0^2 \frac{\partial}{\partial a} \left(\frac{1}{c} \right) = \frac{1}{2} \theta_0^2 \frac{1}{c^2} \frac{\partial c}{\partial a}$$

$$\text{where } c = \frac{l}{EI} + \frac{24 ab}{EI(b-a)}$$

$$\frac{\partial c}{\partial a} = ? \quad \frac{\partial}{\partial a} \left(\frac{a}{b-a} \right) = \frac{b-a+a}{(b-a)^2} = \frac{b}{(b-a)^2}$$

$$\text{So } bG = \frac{1}{2} \frac{\theta_0^2}{c^2 EI} \frac{24 b^2}{(b-a)^2} \Rightarrow G = \frac{12 M^2}{EI} \frac{b}{(b-a)^2}$$

$$K^2 = EG = \frac{12 M^2}{EI} \frac{b}{(b-a)^2} \quad I = \frac{1}{12} b^4$$

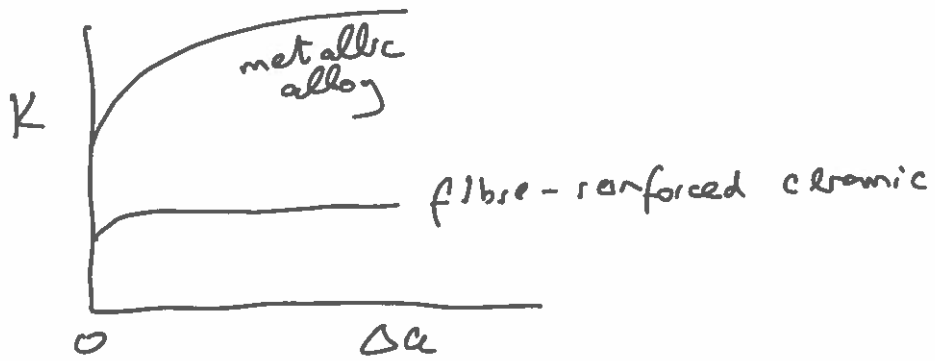
$$\Rightarrow K = \frac{12 M}{b^2} \frac{\sqrt{b}}{(b-a)}$$

$$(iii) \quad bG = - \frac{\partial P}{\partial a} = M_0^2 \frac{\partial c}{\partial a} = \frac{1}{2} \left(\frac{\theta_0}{c} \right)^2 \frac{\partial c}{\partial a}, \text{ as above}$$

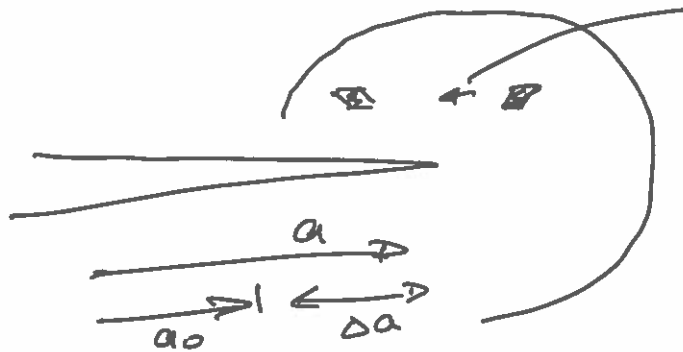
$$(iv) \quad K = \frac{12 M}{b^2} \frac{\sqrt{b}}{(b-a)} \quad \text{So } \left. \frac{\partial K}{\partial a} \right|_M = \frac{12 M}{b^2} \sqrt{b} \frac{1}{(b-a)^2} > 0$$

Hence unstable.

2. (a)

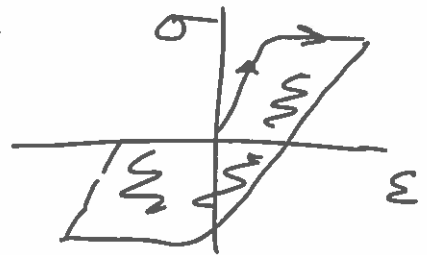


Alloy Crack growth initiates at $K = K_{Ic}$ and curve rises to a steady state value of approx. $2K_{Ic}$ due to irreversible plastic dissipation.

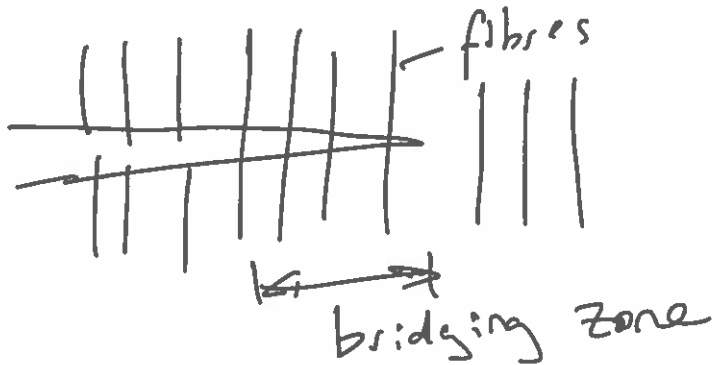


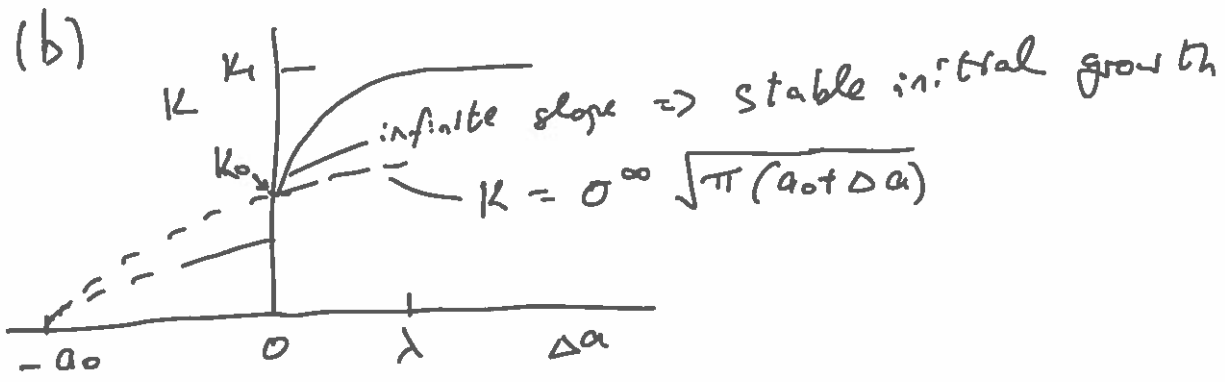
plastic hysteresis of mat^l. elements as crack grows.

Near crack tip, the $\Delta\sigma - \Delta\varepsilon$ history is non-proportional.

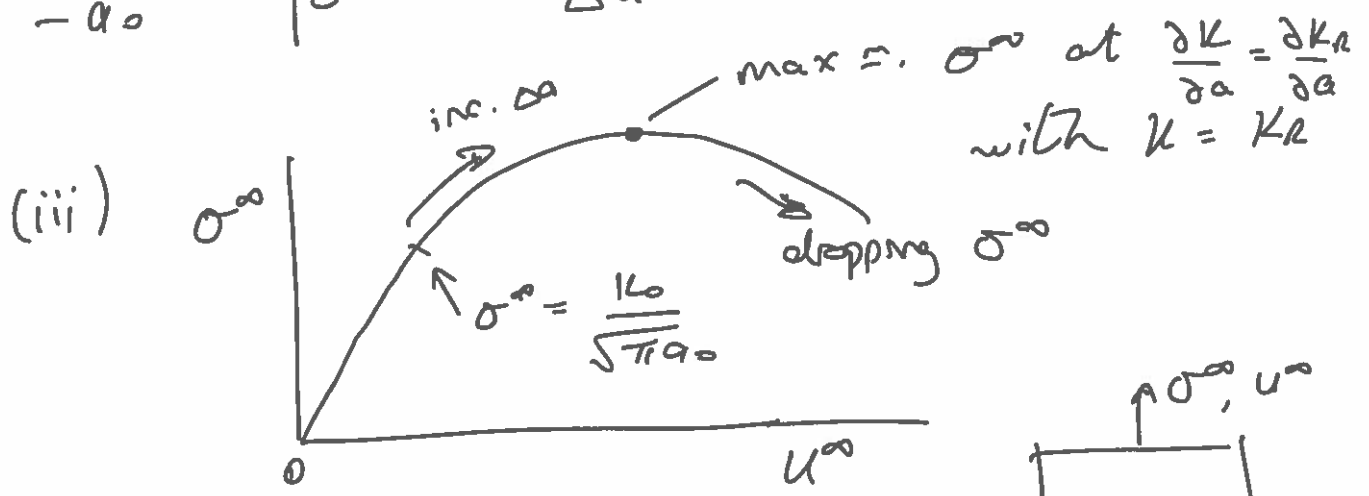
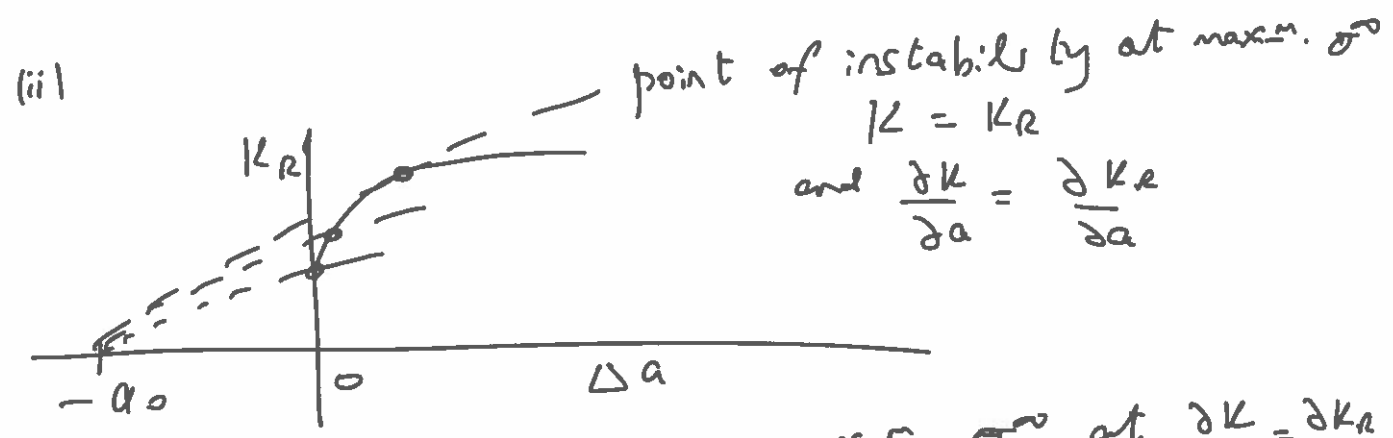


Fibre-reinforced ceramic



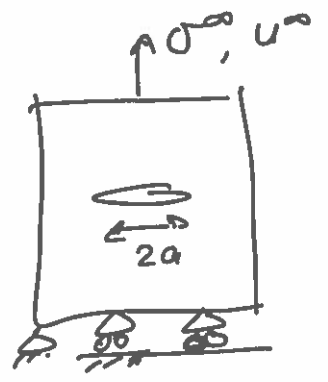


(i) $\sigma^\infty \sqrt{\pi a_0} = K_0$

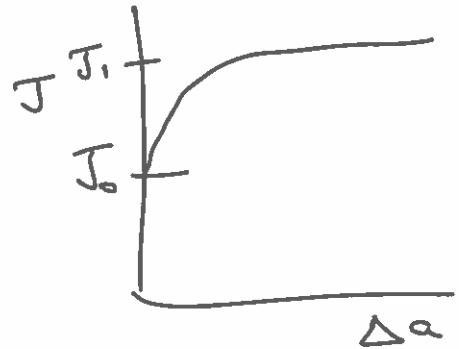
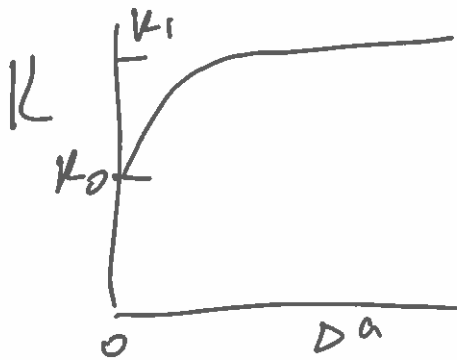


Stable under prescribed σ^∞

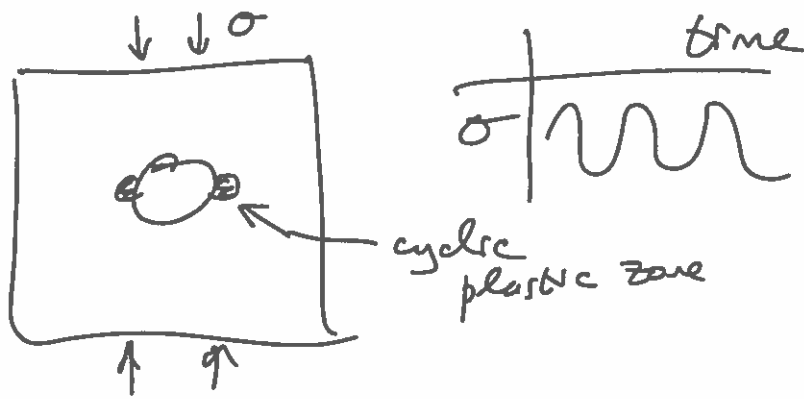
Stable under prescribed u^∞



(iv) $EJ = K^2$ so
convert $K \rightarrow J$ for the
case of small specimens
where J is valid but not K .



3. (a)



A fully reversed plastic zone exists adjacent to the hole, leading to the initiation and growth of a fatigue crack. The number of cycles to initiate the crack N_i depends upon ΔE at the edge of the hole, and ΔE is determined from $\Delta \sigma$ by Neuber's rule for example ($K_o K_E = K_T^2$).

After the crack has reached the periphery of the cyclic plastic zone, it continues to grow in a cyclic compressive stress field until the point where $\Delta K < \Delta K_{th}$.

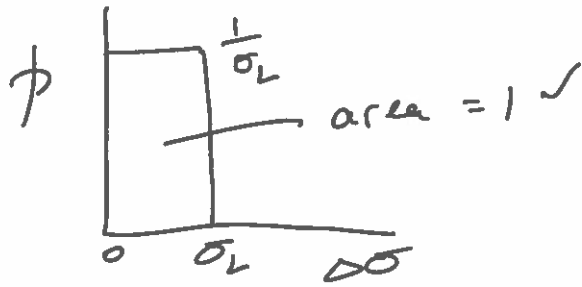
(b) (i) For a thumb nail-shaped edge crack, the 3CG data sheet gives:

$$K = \frac{1.12}{\bar{\Phi}} \sigma \sqrt{\pi a} \quad \bar{\Phi} = \bar{\Phi}(a/c)$$

For $a/c = 1 \Rightarrow \bar{\Phi} = 1.56$

$$\Rightarrow K = 0.718 \sigma \sqrt{\pi a}$$

(ii)



Consider n cycles of random loading.

The number of cycles of stress with a range of $\Delta\sigma$ to $(\Delta\sigma + \delta\Delta\sigma)$ is $n p(\Delta\sigma) \delta\Delta\sigma$.
The crack growth due to these cycles is

$$\frac{da}{dN} n p(\Delta\sigma) \delta\Delta\sigma$$

Total growth for n cycles is

$$\int_0^{\infty} \frac{da}{dN} n p(\Delta\sigma) d\Delta\sigma$$

So, total growth for $n=1$ is

$$\left\langle \frac{da}{dN} \right\rangle = \int_0^{\infty} \frac{da(\Delta\sigma)}{dN} p(\Delta\sigma) d\Delta\sigma$$

$$\text{where } \Delta K = 0.718 \Delta\sigma \sqrt{\pi a}$$

(iii) N_T cycles required to grow a crack from length a_0 to $2a_0$. $N_T = ?$

$$\left\langle \frac{da}{dN} \right\rangle = \int_0^{\sigma_L} A (0.718 \Delta\sigma \sqrt{\pi a})^n \frac{1}{\sigma_L} d\Delta\sigma$$

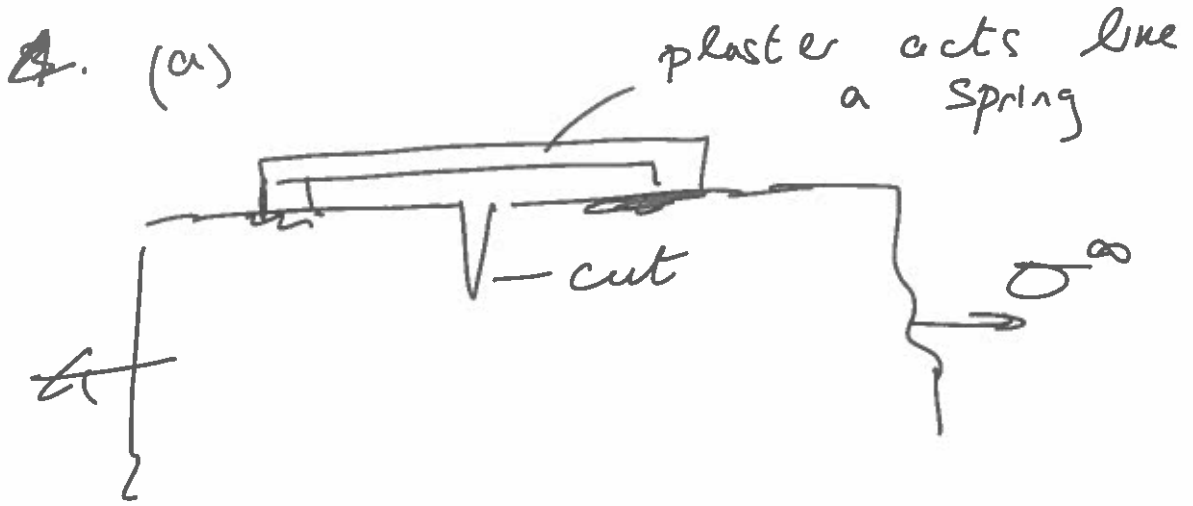
$$\Rightarrow \left\langle \frac{da}{dN} \right\rangle = A \cdot 0.718^n (\pi a)^{n/2} \frac{\sigma_L^n}{n+1}$$

$$\Rightarrow N_T = \int_{a_0}^{2a_0} \left\langle \frac{da}{dN} \right\rangle^{-1} da$$

$$= \int_{a_0}^{2a_0} \frac{(n+1)}{A \sigma_L^n} \frac{1}{(0.718)^n \pi^{n/2}} a^{-\frac{n}{2}} da$$

$$\Rightarrow N_T = \frac{(n+1)}{A (0.718)^n \pi^{n/2}} \frac{1}{\sigma_L^n} \left[a^{\frac{2-n}{2}} \right]_{a_0}^{2a_0} \frac{2}{2-n}$$

(iv) Single peak overloads can lead to a severe retardation in growth rate and so the actual fatigue life is much greater than that predicted by a linear summation of damage (i.e. by Miner's law).

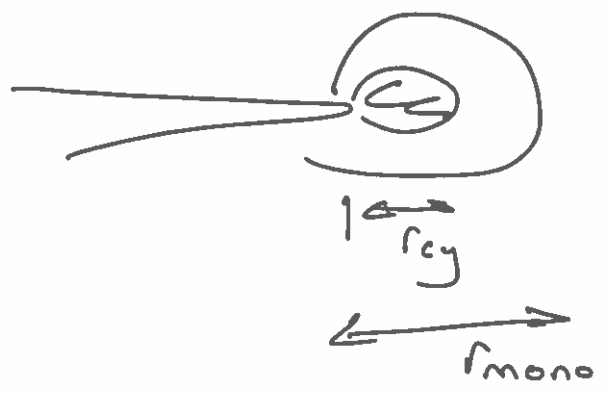


— plaster bridges the crack and develops tension to reduce the K at the crack tip.

$$K_{\text{plaster present}} = \alpha \cdot K_{\text{no-plaster}}$$

$\alpha \leq 1$

(b)



$$r_{cr} \sim \frac{1}{4\pi} \frac{\Delta K^2}{\sigma_y^2}$$

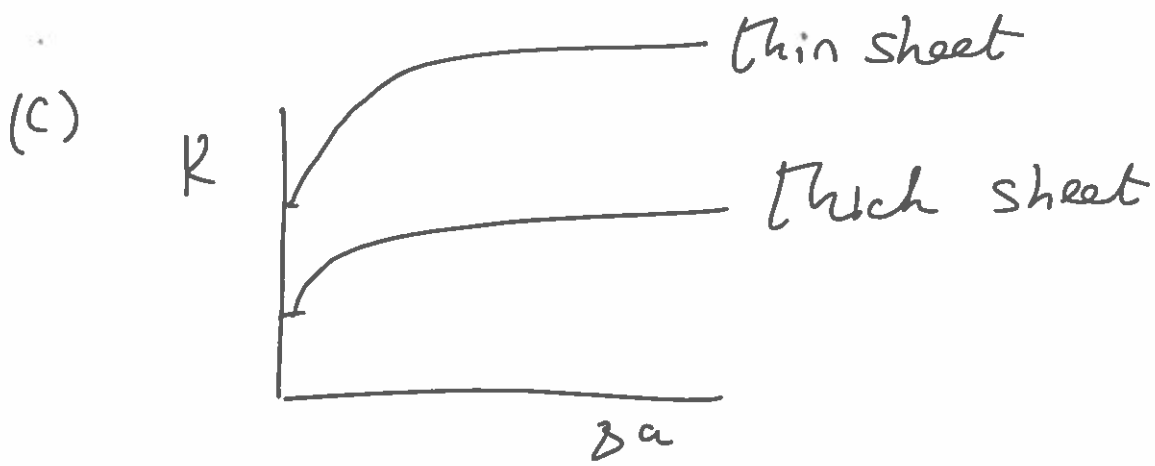
$$r_{mono} \sim \frac{1}{\pi} \frac{K_{max}^2}{\sigma_y^2}$$

Paris Law :

$$\frac{da}{dN} = A \Delta K^n$$

$$= A (4\pi r_{cr} \sigma_y^2)^{n/2}$$

$$\propto r_{cr}^{n/2}$$



Assume that crack advance is by void growth.

A thick sheet has a high tensile hydrostatic stress at the crack tip, thereby reducing ductility (Rice-Tracy) and consequently reducing the toughness. So a stack of thin sheets behave as a material of high toughness compared to a single thick sheet.

