

EGT3  
ENGINEERING TRIPOS PART IIB

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Tuesday 4 May 2021 9 to 10.40

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**Module 4A15**

**AEROACOUSTICS**

*Answer not more than **three** questions.*

*All questions carry the same number of marks.*

*The **approximate** percentage of marks allocated to each part of a question is indicated in the right margin.*

*Write your candidate number **not** your name on the cover sheet and at the top of each answer sheet.*

**STATIONERY REQUIREMENTS**

Write on single-sided paper.

**SPECIAL REQUIREMENTS TO BE SUPPLIED FOR THIS EXAM**

CUED approved calculator allowed

Attachment: 4A15 Aeroacoustics data sheet (6 pages)

You are allowed access to the electronic version of the Engineering Data Books.

**10 minutes reading time is allowed for this paper at the start of the exam.**

**The time taken for scanning/uploading answers is 15 minutes.**

**Your script is to be uploaded as a single consolidated pdf containing all answers.**

1 The acoustic pressure on the surface of a vibrating sphere of radius  $a$  is given by

$$p' = A \cos \theta \cos(\omega t)$$

where  $A$  and  $\omega$  are constant and  $\theta$  is the polar angle in spherical coordinates. Assume that the mean density and speed of sound in surrounding fluids are constant and denoted by  $\rho_o$  and  $c_o$ , respectively.

(a) Find the time averaged acoustic power generated by the source. [85%]

(b) If  $\lambda$  is the wavelength of sound produced, explain why the amplitude of vibration  $A$  needs to be much higher to get the same power when  $\lambda \gg a$  compared to when  $\lambda \ll a$ . [15%]

2 A 330 ml open Coke can is to be used as a Helmholtz resonator. The Helmholtz resonance frequency is given by

$$f = \frac{c_0}{2\pi} \sqrt{\frac{A}{lV}}$$

where  $c_0$  is the speed of sound,  $A$  and  $l$  are the area and the effective length of the neck of the resonator, respectively and  $V$  is the volume of the cavity.

(a) State all the assumptions necessary to derive the above expression. [15%]

(b) By assuming that the can has a thickness of  $10^{-4}$  m throughout, a height of 0.1 m and a circular opening of area  $4 \times 10^{-4}$  m<sup>2</sup>, derive the Helmholtz resonance frequency of the can. [25%]

(c) The measured frequency is a bit lower than the value calculated in part (b). Without doing detailed calculations, explain why this would be the case. [15%]

(d) If, by using a can opener, we remove the top of the can completely, what would be the new lowest acoustic resonance frequency? [15%]

(e) A 330 ml Coke bottle has a resonance frequency of 300 Hz. Explain why the frequency is so different from the can resonance frequency. [15%]

(f) We want the Coke bottle to produce a C4 note (262 Hz) when we blow over its top. How would you accomplish this with the 330 ml bottle? [15%]

3 If  $\hat{S}_n$  are the strengths of a series of monopole sources located close to the origin at  $\mathbf{x}_n$ , then the acoustic pressure field is given by,

$$\hat{p}(\mathbf{x}) = \sum_n \hat{S}_n f(\mathbf{x} - \mathbf{x}_n) \quad \text{where} \quad f(\mathbf{x}) = \frac{1}{|\mathbf{x}|} e^{-ik|\mathbf{x}|}.$$

(a) Find the multipole series expansion for the acoustic field  $\hat{p}(\mathbf{x})$  about the origin. Include the first two terms of the series. [20%]

(b) Under what circumstances will the pressure field  $\hat{p}(\mathbf{x})$  far away from the origin have significant directional dependence? [10%]

(c) 4 monopole sources are placed at  $(1, 0)$ ,  $(0, 1)$ ,  $(-1, 0)$ , and  $(0, -1)$ , with strengths  $\hat{S}_1$ ,  $\hat{S}_2$ ,  $\hat{S}_3$ , and  $\hat{S}_4$  respectively. What strength of dipole, placed at the origin, could be used to model these 4 sources if  $\hat{S}_1 = -\hat{S}_3 = 3$  and  $\hat{S}_2 = -\hat{S}_4 = 5$ ? How will the accuracy of the dipole approximation vary with frequency? [15%]

If instead of a discrete set of sources, we have a continuous distribution  $\hat{S}(\mathbf{x})$ , the acoustic field  $\hat{p}(\mathbf{x})$  will be governed by the equation,

$$(\nabla^2 + k^2)\hat{p}(\mathbf{x}) = 4\pi\hat{S}(\mathbf{x}).$$

(d) Use the Green's function to find the integral expression for  $\hat{p}(\mathbf{x})$ . [10%]

(e) By doing a Taylor series expansion of the Green's function derive the effective monopole and dipole strengths at the origin that can be used to approximate the sound field produced by  $\hat{S}(\mathbf{x})$ . [20%]

A sphere of radius  $a$ , oscillates along the  $x_1$  axis with velocity amplitude  $U$  and frequency  $\omega$ .  $k = \omega/c_0$  is the wavenumber, and  $(r, \theta, \psi)$  are spherical polar coordinates. If the sphere is compact the acoustic field is given by,

$$\hat{p}(r, \theta) = -\frac{1}{2}i\omega\rho_0 U a^3 \cos\theta \frac{\partial}{\partial r} \left( \frac{e^{-ikr}}{r} \right).$$

(f) What are the effective monopole and dipole strengths of the oscillating sphere? [25%]

4 (a) Plane waves travel in a duct, aligned with the  $x$ -axis, that has varying cross-sectional area  $A(x)$ . Show that these waves are governed by the Webster horn equation

$$\frac{1}{c_0^2} \frac{\partial^2 p'}{\partial t^2} - \frac{1}{A} \frac{\partial}{\partial x} \left( A \frac{\partial p'}{\partial x} \right) = 0.$$

[20%]

If we make the substitutions  $\psi(x) = \hat{p}(x) \sqrt{A(x)}$  and  $A(x) = \pi a(x)^2$ , where  $a(x)$  is the local radius of the duct, then the Webster horn equation can be written

$$\frac{d^2 \psi}{dx^2} + \left( k^2 - \frac{1}{a} \frac{d^2 a}{dx^2} \right) \psi = 0. \quad (1)$$

where  $k = \omega/c_0$  is the wavenumber.

(b)  $A(x)$  takes the form  $A(x) = A_0[1 + \cos(2\alpha x)]/2$ .  $A_0$  is a real constant and  $\alpha$  is a complex constant. What form does the pressure field take? [35%]

(c) What value of  $\alpha$  could cause waves not to propagate in the duct? Sketch the corresponding shape of the duct. [20%]

(d) What does equation (1) tell us about wave propagation for:

(i) a cylindrical duct?

(ii) a conical duct?

[15%]

(e) Using an example of a musical instrument explain why it is beneficial to have a duct in which not all plane waves of all frequencies can propagate. [10%]

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# Module 4A15 Aeroacoustics Data Sheet

## USEFUL DATA AND DEFINITIONS

### Physical Properties

Speed of sound in an ideal gas  $\sqrt{\gamma RT}$ , where  $T$  is temperature in Kelvins

### Units of sound measurement

$$\text{SPL (sound pressure level)} = 20 \log_{10} \left( \frac{p'_{rms}}{2 \times 10^{-5} \text{Nm}^{-2}} \right) \text{dB}$$

$$\text{IL (intensity level)} = 10 \log_{10} \left( \frac{\text{intensity}}{10^{-12} \text{watts m}^{-2}} \right) \text{dB}$$

$$\text{PWL (power level)} = 10 \log_{10} \left( \frac{\text{sound power output}}{10^{-12} \text{watts}} \right) \text{dB}$$

### Definitions

**Surface impedance**  $Z_s$ , relates the pressure perturbation applied to a surface,  $p'$ , to its normal velocity  $v'$ ;  $p' = Z_s v'$

**Characteristic impedance** of a fluid  $\rho_0 c_0$

**Specific impedance** of a surface  $Z_s / (\rho_0 c_0)$

**Wavenumber**  $k = \omega / c_0 = 2\pi / \lambda$ , where  $\lambda$  is the wavelength

**Helmholtz number** (or compactness ratio)  $= kD$ , where  $D$  is a typical dimension of the source.

**Strouhal number**  $= \omega D / (2\pi U)$  for sound of frequency  $\omega$  (in rad/s), produced in a flow with speed  $U$ , length scale  $D$ .

## Basic equations for linear acoustics

### Conservation of mass

$$\frac{\partial \rho'}{\partial t} + \rho_0 \nabla \cdot \mathbf{v}' = 0$$

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### Conservation of momentum

$$\rho_0 \frac{\partial \mathbf{v}'}{\partial t} + \nabla p' = 0$$

### Isentropic

$$c_0^2 = \left. \frac{dp}{d\rho} \right|_s$$

### Wave equation

$$\frac{1}{c_0^2} \frac{\partial^2 p'}{\partial t^2} - \nabla^2 p' = 0$$

### Energy density

$$e = \frac{1}{2} \rho_0 v'^2 + \frac{1}{2 \rho_0 c_0^2} p'^2$$

### Intensity $\mathbf{I} = p' \mathbf{v}'$

**Velocity potential**  $\phi'$  satisfies the wave equation and  $p' = -\rho_0 \frac{\partial \phi'}{\partial t}$ ,  $\mathbf{v}' = \nabla \phi'$ .

**Autocorrelation**  $F(\xi)$ , the autocorrelation of  $f(y)$  is given by

$$F(\xi) = \overline{f(y)f(y+\xi)}$$
$$F(0) = \overline{f^2}$$

### Integral length scale, $l$

$$l \overline{f^2} = \int_{-\infty}^{\infty} F(\xi) d\xi$$

### Sound power

Sound power from a source is defined as

$$P = \int_S \bar{\mathbf{I}} \cdot d\mathbf{S} = \int_{S_\infty} \frac{\overline{p'^2}}{\rho_0 c_0} d\mathbf{S}$$

for a statistically stationary source. For an outward propagating spherically symmetrical sound field  $P = \frac{\overline{p'^2}}{\rho_0 c_0} 4\pi r^2$ , where  $p'$  is the acoustic pressure at radius  $r$ .

For a sound field, which is a function of spherical polar coordinates  $r, \theta$  only, and is independent of the azimuthal angle,

$$P = 2\pi r^2 \int_0^\pi \frac{\overline{p'^2}}{\rho_0 c_0} \sin \theta d\theta$$



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## Simple wave fields

### 1D or plane wave

The general solution of the 1D wave equation is  $p'(x,t) = f(t - x/c_0) + g(t + x/c_0)$ , where  $f$  and  $g$  are arbitrary functions. In a plane wave propagating to the right  $p' = \rho_0 c_0 u'$ ; in a plane wave propagating to the left  $p' = -\rho_0 c_0 u'$ ,  $u'$  being the particle velocity.

### Spherically symmetric sound fields

The general spherically symmetric solution of the 3D wave equation is

$$\phi'(r,t) = \frac{f(t - r/c_0)}{r} + \frac{g(t + r/c_0)}{r},$$

where  $r$  is the distance from the source;  $f$  and  $g$  are arbitrary functions.

### $\cos \theta$ dependence

The general solution of the 3D wave equation with  $\cos \theta$  dependence is

$$p'(\mathbf{x},t) = \frac{\partial}{\partial x} \left[ \frac{f(t - r/c_0)}{r} + \frac{g(t + r/c_0)}{r} \right] = \cos \theta \frac{\partial}{\partial r} \left[ \frac{f(t - r/c_0)}{r} + \frac{g(t + r/c_0)}{r} \right]$$

## Useful mathematical formulae

### Spherical polar coordinates $(r, \theta, \psi)$

#### Gradient

$$\nabla p' = \left( \frac{\partial p'}{\partial r}, \frac{1}{r} \frac{\partial p'}{\partial \theta}, \frac{1}{r \sin \theta} \frac{\partial p'}{\partial \psi} \right)$$

#### Divergence

$$\nabla \cdot \mathbf{v}' = \frac{1}{r^2} \frac{\partial}{\partial r} (r^2 v'_r) + \frac{1}{r \sin \theta} \frac{\partial}{\partial \theta} (\sin \theta v'_\theta) + \frac{1}{r \sin \theta} \frac{\partial v'_\psi}{\partial \psi}$$

#### Laplacian

$$\nabla^2 p' = \frac{1}{r^2} \frac{\partial}{\partial r} \left( r^2 \frac{\partial p'}{\partial r} \right) + \frac{1}{r^2 \sin \theta} \frac{\partial}{\partial \theta} \left( \sin \theta \frac{\partial p'}{\partial \theta} \right) + \frac{1}{r^2 \sin^2 \theta} \frac{\partial^2 p'}{\partial \psi^2}$$

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## Delta functions

### Kronecker Delta

$$\delta_{ij} = \begin{cases} 1 & i = j \\ 0 & i \neq j \end{cases}$$

**1D  $\delta$ -function**  $\delta(x) = 0$  for  $x \neq 0$  and  $\int_{-\infty}^{\infty} \delta(ax - b)f(x)dx = f(b/a)/|a|$

**3D  $\delta$ -function**  $\delta(\mathbf{x}) = \delta(x_1)\delta(x_2)\delta(x_3)$

## Convolution algebra

### Convolution of $f(\mathbf{x})$ and $g(\mathbf{x})$

$$(f \star g)(\mathbf{x}) = \int_{-\infty}^{\infty} f(\mathbf{y})g(\mathbf{x} - \mathbf{y})d\mathbf{y}$$

### Commutative properties

$$f \star g = g \star f$$

$$\frac{\partial}{\partial x_i}(f \star g)(\mathbf{x}) = f \star \frac{\partial g}{\partial x_i} = \frac{\partial f}{\partial x_i} \star g$$

## Green's function

### 3D Green's function for wave equation

$$\left(\frac{\partial^2}{\partial t^2} - c_0^2 \nabla^2\right) g(\mathbf{x}, t | \mathbf{y}, \tau) = \delta(t - \tau) \delta(\mathbf{x} - \mathbf{y})$$
$$g(\mathbf{x}, t | \mathbf{y}, \tau) = \frac{\delta\{|\mathbf{x} - \mathbf{y}| - c_0(t - \tau)\}}{4\pi c_0 |\mathbf{x} - \mathbf{y}|}$$

## Lighthill's Acoustic Analogy

### Lighthill's equation

$$\left(\frac{\partial^2}{\partial t^2} - c_0^2 \nabla^2\right) \rho' = \frac{\partial^2 T_{ij}}{\partial x_i \partial x_j}$$

For cold, isentropic, low Mach-number jets,  $T_{ij}$  can be approximated as:

$$T_{ij} = \rho_0 u_i u_j$$

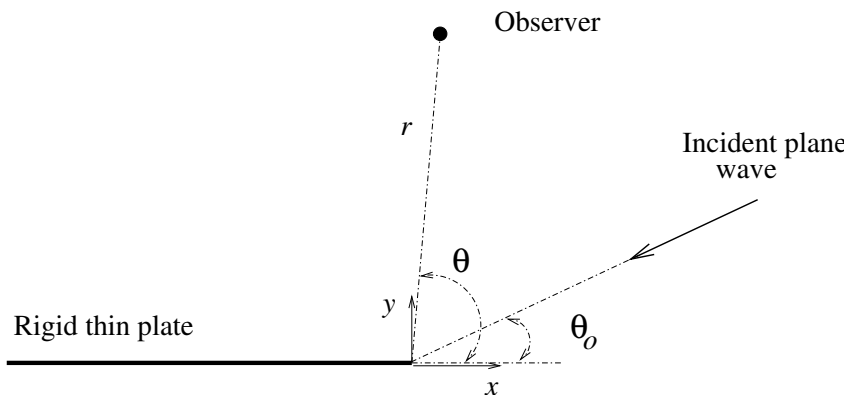


Figure 1: Geometry for edge scattering

**Lighthill eight power law** Acoustic power,

$$P_a \sim \frac{\rho_o d_j^2}{c_0^5} u_j^8,$$

where  $d_j$  and  $u_j$  are the jet exit diameter and velocity, respectively.

## Refraction

**Snell's law for determining a ray path is**

$$\frac{\sin \theta}{c_0} = \text{constant} . \quad (1)$$

## Diffraction

**Field scattered by a sharp edge** If the incident plane waves is

$$p_i(\mathbf{x}, t) = P_{\text{inc}} \exp(i\omega t + ik_0 x \cos \theta_0 + ik_0 y \sin \theta_0) , \quad (2)$$

then the diffracted pressure is

$$p_d = P_{\text{inc}} \left( \frac{2}{\pi k_0 r} \right)^{\frac{1}{2}} \frac{\sin(\theta_0/2) \sin(\theta/2)}{\cos \theta + \cos \theta_0} \exp \left( -\frac{i\pi}{4} - ik_0 r \right) . \quad (3)$$

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## In a cylindrical duct of radius $a$

The pressure field is given by

$$p'(\mathbf{x}, t) = e^{i(\omega t + n\theta)} J_n(z_{mn}r/a) (Ae^{-ikx_3} + Be^{ikx_3}),$$

where  $z_{mn}$  is the  $m$ th zero of  $J_n(z)$  and  $k = (k_0^2 - z_{mn}^2/a^2)^{1/2}$ .

For large azimuthal wavenumber,  $n$

$$z_{1n} \approx n + 1.85n^{1/3}$$