

## Question 1.

(a) Show that the pressure in the pipe calculated from the tank and from the reservoir are the same

$$\begin{aligned} \text{From the tank } P &= P_{\text{atm}} - \rho g (h_1 - H) = P_{\text{atm}} - 0.1 \rho g \\ \text{From the reservoir } P &= P_{\text{atm}} - \rho g h_2 = P_{\text{atm}} - 0.1 \rho g \end{aligned} \quad \left. \vphantom{\begin{aligned} \text{From the tank } P &= P_{\text{atm}} - \rho g (h_1 - H) = P_{\text{atm}} - 0.1 \rho g \\ \text{From the reservoir } P &= P_{\text{atm}} - \rho g h_2 = P_{\text{atm}} - 0.1 \rho g \end{aligned}} \right\} \text{same value}$$

$$P - P_{\text{atm}} = -981 \text{ Pa}$$

(b) The level in the pipe must reach  $-0.1 \text{ m}$  on the reservoir side, hence pressure in the pipe must be

$$P - P_{\text{atm}} = +981 \text{ Pa}$$

Moreover, to conserve the volume of air,  $h_1$  must have gone up by  $0.2 \text{ m}$ .  
hence:  $h_1 = 0.4 \text{ m}$

Since pressure in the pipe is above atmospheric,  $H$  must be higher than  $h_1$

$$P - P_{\text{atm}} = \rho g (H - h_1)$$

$$\text{So } H = h_1 + 0.1 = 0.5 \text{ m}$$

(c) Water will flow when  $h_1 = L$ , and pressure in the pipe will remain at the level calculated in (b). Hence  $H$  will still be  $0.1$  higher than  $h_1$ :

$$H = 0.6 \text{ m}$$

(d) Water is poured slowly so we expect a constant temperature.

Between the initial condition and the moment where bubbles appear, the pressure difference is

$$\Delta P = 2 \times 981 \text{ Pa}$$

$$\frac{\Delta \rho}{\rho} = + \frac{\Delta P}{P} \approx \frac{2 \times 981}{101325} \approx 2\%$$

Hence the length of the air column is reduced by 2%

$$\Delta \text{length} = 0.02 \times (0.3 + 0.3 + 0.4) = 2 \text{ cm}$$

$$\text{Hence } H \text{ is } 2 \text{ cm above (b): } H = 0.52 \text{ m}$$

## Question 2

(2) Mass conservation  $w_1 h_1 V_1 = w_2 h_2 V_2$

$w_1 = 13 \text{ m}$  width upstream

$w_2 = 8 \text{ m}$  width at the bridge

Two unknowns  
 $h_2$  and  $V_2$

Other equation: Since the flow is inviscid, Bernoulli applies between 1 and 2, for example at the surface

$$\rho g h_1 + \frac{1}{2} \rho V_1^2 = \rho g h_2 + \frac{1}{2} \rho V_2^2$$

This is another equation relating  $h_2$  and  $V_2$ . Substituting  $h_2 = \frac{w_1 h_1 V_1}{w_2 V_2}$

$$g h_1 + \frac{V_1^2}{2} = g \frac{w_1 h_1 V_1}{w_2 V_2} + \frac{V_2^2}{2} \quad \text{) equation for } V_2$$

Check  $V_2 = 5 \text{ m.s}^{-1}$

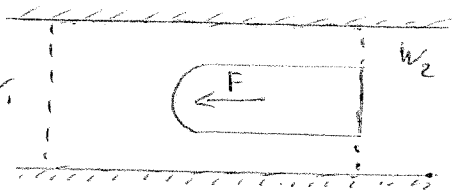
$$10 \times 5 + \frac{2.5^2}{2} = 53.125 \quad \text{and} \quad 10 \frac{13 \times 5 \times 2.5}{8 \times 5} + \frac{5^2}{2} = 53.125$$

Hence  $h_2 = \frac{w_1 h_1 V_1}{w_2 V_2} = \frac{13 \times 5 \times 2.5}{8 \times 5} = 4.0625 \text{ m}$

(b) Control Volume

$$\dot{m}_1 = \rho w_1 h_1 V_1 = 10^3 \times 13 \times 5 \times 2.5 = 162500 \text{ } w_1$$

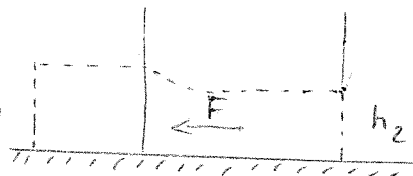
$$\dot{m}_2 = \rho w_2 h_2 V_2 = 10^3 \times 8 \times 4.0625 \times 5 = \text{''}$$



Pressure forces: hydrostatic

$$P_1 A_1 = \rho g \frac{h_1^2}{2} w_1 = 10^4 \times \frac{5^2}{2} \times 13 = 1625000 \text{ } h_1$$

$$P_2 A_2 = \rho g \frac{h_2^2}{2} w_2 = 10^4 \times \frac{4.0625^2}{2} \times 8 = 1072754$$



SFME:  $\dot{m}_2 V_2 - \dot{m}_1 V_1 = F + P_1 A_1 - P_2 A_2$

$$F = 162500 (5 - 2.5) - 1625000 + 1072754$$

$$= -145996 \text{ N} \quad (\text{to the left})$$

Question 3

[see page 11 for air properties in databook]

(a) Heat pulse in the air system  $Q = m c_v \Delta T$

Mass of air  $PV = mRT$   $m = \frac{10^5 \times 10^{-2}}{287 \times 300} \approx 0.0011614 \text{ kg}$

Hence  $\Delta T = \frac{2 \cdot 10^3}{0.0011614 \times 718} \approx 2398 \text{ K}$

$T = 2398 + 300 = 2698 \text{ K}$

$\frac{P}{P_{initial}} = \frac{T}{T_{initial}}$

$P = \frac{10^5 \cdot 2698}{300} \approx 9 \text{ bar}$

(b) Initial water jet velocity: Bernoulli  $P = P_{atm} + \rho \frac{v_i^2}{2}$

$v_i = \sqrt{\frac{2 \times 8 \times 10^5}{10^3}} = 40 \text{ m.s}^{-1}$

Inside pressure at the end of the air expansion (adiabatic)

$PV^\gamma = \text{const}$

Final pressure  $P_f = P_i \left(\frac{V_i}{V_f}\right)^\gamma = 9 \times 10^5 \left(\frac{1}{2}\right)^{1.4} \approx 3.41 \text{ bar}$

Hence  $v_f = \sqrt{\frac{2 \times 2.41 \times 10^5}{10^3}} = 21.95 \text{ m.s}^{-1}$

(c) Thrust. Control Volume

$\dot{m} v = F$  (no pressure force)



$F = \rho A v^2 = 10^3 \times 2 \times 10^{-6} \times 40^2 = 3.2 \text{ N}$  initially

$= 10^3 \times 2 \times 10^{-6} \times 21.95^2 = 0.964 \text{ N}$  at the end

(d) Between initial and final time: energy conservation

change in internal energy of air = water kinetic energy + work against atmosphere

$m_{air} c_v \Delta T$

final air temperature:  $T_f = \frac{P_f V_f}{m R} = \frac{3.41 \times 10^5 \times 2 \times 10^{-3}}{0.0011614 \times 287} = 2046 \text{ K}$

Water kinetic energy =  $0.0011614 \times 718 (2698 - 2046) - 10^5 \times 10^{-3}$

$= 543.7 - 100 = 443.7 \text{ J}$

NB: this is 1 kg of water at 29.79 m.s<sup>-1</sup>

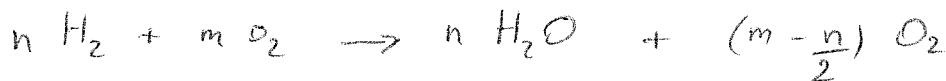
Question 4

(2) Hydrogen  $P_{H_2} \dot{V} = \dot{n} \bar{R} T \rightarrow P_{H_2} = 5 \times 10^{-4} \times 8.3143 \times 10^3 \times 298.15$

Oxygen  $P_{O_2} = 2 \times 10^{-3} \times 8.3143 \times 10^3 \times 298.15$   
 $= 4957.8 \text{ Pa}$   
 $P_{H_2} = 1239.45 \text{ Pa}$

Hence  $P = P_{O_2} + P_{H_2} = 4957.8 + 1239.4 = 6197.2 \text{ Pa}$

(b) Combustion with excess oxygen



$n = 5 \times 10^{-4} \text{ kmol} \cdot \text{s}^{-1}$  of water  $H_2O$

$m - \frac{n}{2} = 2 \times 10^{-3} - \frac{5 \times 10^{-4}}{2} = 1.75 \times 10^{-3} \text{ kmol} \cdot \text{s}^{-1}$  of oxygen  $O_2$

(c) S.F.E.E with lower enthalpy of reaction  $120,000 \text{ MJ} \cdot \text{kg}^{-1}$   
 $= 240 \text{ MJ} \cdot \text{kmol}^{-1}$  of  $H_2$

$$n h_{H_2O}(T) + \left(m - \frac{n}{2}\right) h_{O_2}(T) = n h_{H_2}(T_0) + m h_{O_2}(T_0)$$

$$n \left( h_{H_2O}(T) - h_{H_2O}(T_0) \right) + \left(m - \frac{n}{2}\right) \left( h_{O_2}(T) - h_{O_2}(T_0) \right) = n \left[ h_{H_2}(T_0) - h_{H_2O}(T_0) + \frac{1}{2} h_{O_2}(T_0) \right]$$

$$= n \times 240 \text{ MJ} \cdot \text{kmol}^{-1}$$

$$h_{H_2O}(T) - h_{H_2O}(T_0) + 3.5 \left( h_{O_2}(T) - h_{O_2}(T_0) \right) = 240 \text{ MJ} \cdot \text{kmol}^{-1}$$

Table page 12  $T = 1700 \text{ K} \quad 67.65 - 9.3 + 3.5(56.63 - 8.66) = 225.645$

$T = 1800 \text{ K} \quad 72.58 - 9.3 + 3.5(60.35 - 8.66) = 243.595$

$$T = 1800 - 100 \frac{243.595 - 240}{243.595 - 225.645} \approx 1780 \text{ K}$$

Pressure being different from 1.01324 bar plays no role since enthalpies do not depend on pressure for perfect or semi-perfect gases.

Question 5

(a) Adiabatic & fully resisted  $\therefore$  isentropic  
(reversible)

$$T_2 = T_1 \left( \frac{P_2}{P_1} \right)^{\frac{\gamma-1}{\gamma}} = 1500 \left( \frac{1}{15} \right)^{\frac{0.4}{1.4}} \approx 691.93 \text{ K}$$

S.F.E.E. :  $\dot{Q} - \dot{W}_x = \dot{m} (h_2 - h_1) = \dot{m} c_p (T_2 - T_1)$

$$\dot{W}_x = 50 \times 1005 \times (1500 - 691.93) \approx 40.61 \text{ MW}$$

(b)  $du = dq + dw = 0.01 du - p dv$

$$0.99 du = -d(pv) + v dp$$

$$0.99 c_v dT = -R dT + \frac{RT}{P} dp$$

$$(0.99 c_v + R) \frac{dT}{T} = R \frac{dp}{P}$$

$$\frac{dT}{T} = \frac{R}{0.99 c_v + R} \frac{dp}{P}$$

$$= \frac{c_p - c_v}{c_p - 0.01 c_v} \frac{dp}{P} = \frac{\gamma - 1}{\gamma - 0.01}$$

$$\frac{T}{P^{\frac{\gamma-1}{\gamma-0.01}}} = \text{const}$$

(c) New exit temperature  $T = 1500 \left( \frac{1}{15} \right)^{\frac{0.4}{1.39}} \approx 688.09 \text{ K}$

S.F.E.E.  $\dot{Q} - \dot{W}_x = \dot{m} c_p (T_2 - T_1)$

$$\uparrow$$

$$0.01 \dot{m} c_v (T_2 - T_1)$$

hence  $\dot{W}_x = 50 \times 1005 (1500 - 688.09) - 0.01 \times 50 \times 718 (1500 - 688.09)$

$$\dot{W}_x \approx 40.507 \text{ MW}$$

Very small change in  $\dot{W}_x$ , but the isentropic efficiency  $\frac{\dot{W}_x}{\dot{m} c_p (T_2 - T_1)} \approx 0.993$   
drops by nearly 1%

5

T. Alboussière

Q.6

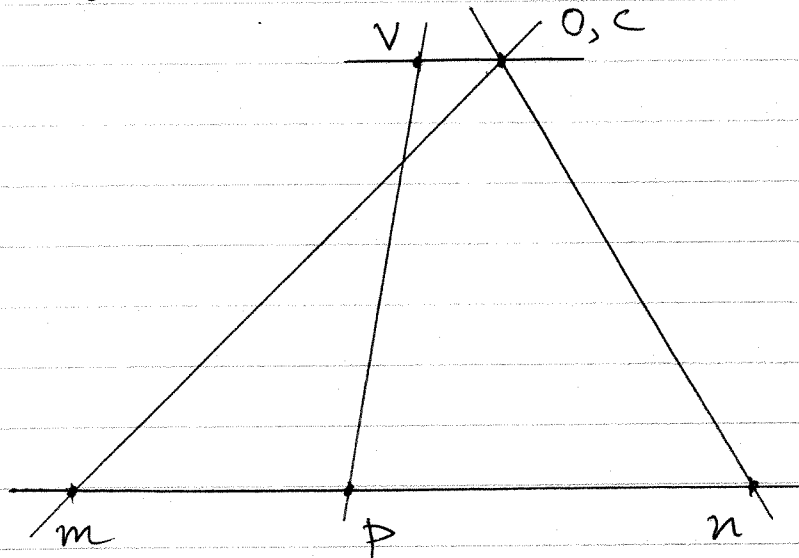
PAPER 1: MECHANICS

(a) Velocity of M :-

$$V = r\omega = 20 \text{ mm} \times 20 \text{ rad s}^{-1} = 400 \text{ mm s}^{-1}$$

→ 80 mm on diagram

$$O, c / m = 80$$



Note that  $\frac{pm}{nm} = \frac{PM}{NM} = \frac{40}{100} = \frac{pm}{90 \text{ mm diagram}}$

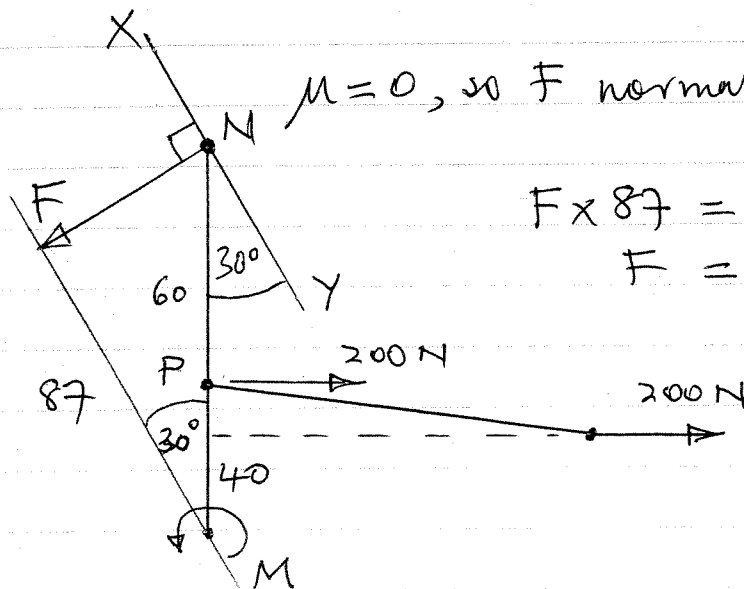
$$\therefore pm = \frac{40}{100} \times 90 \text{ mm} = 36 \text{ mm diagram}$$

Velocity  $OV = 10 \text{ mm diagram} \rightarrow \underline{50 \text{ mm s}^{-1}}$

(b)  $F \times 400 \text{ mm s}^{-1} = 200 \text{ N} \times 50 \text{ mm s}^{-1}$

$$F = 25 \text{ N}$$

$$25 \text{ N} \times 20 \text{ mm} = \underline{500 \text{ N mm}}$$



$\mu = 0$ , so  $F$  normal to  $XY$

$$F \times 87 = 200 \text{ N} \times 40$$

$$F = \underline{92 \text{ N}}$$

(6)

Q.7.

$$(a) mgh = U_{el}, \quad m = U_{el} / gh$$

$$m = \frac{(1000 \text{ N} \times 15 \text{ m} + \frac{1}{2} \times 1500 \text{ N} \times 15 \text{ m})}{9.81 \times 30} = \underline{89.2 \text{ kg}}$$

$$(b) mg(l+y) = U_{el}(y)$$

$$60 \times 9.81 \times 15 + 60 \times 9.81 y = 1000y + \frac{1}{2} \cdot \frac{y}{15} \cdot 1500y$$

$$50y^2 + 411y - 8829 = 0$$

$$y = \frac{-411 \pm \sqrt{411^2 + 4 \times 50 \times 8829}}{100} = \underline{9.80 \text{ m}}$$

(The root  $-18.02 \text{ m}$  is discarded, since  $18.02 > 15$ )

$$F = 1000 \text{ N} + \frac{9.80}{15} \times 1500 = \underline{1.98 \text{ kN}}$$

(c) Taking the maximum extension, in one cycle of loading / unloading,  $500 \text{ N} \times 15 \text{ m} = 7500 \text{ J}$  is dissipated as friction in the rope, compared to a  $U_{el}$  of  $26250 \text{ J}$ . i.e., 29% of the energy input is lost. The results in the jumper rebounding to a position which is well below the bridge deck, which is obviously essential for a safe jump.

(d) For  $m = m_{max}$ ,  $30 \text{ kN} \geq F \geq 2.5 \text{ kN}$ . This can impose a very large force on the person, which could cause injury or detachment of the person from one end of the rope. For  $m > m_{max}$ , the rope will snap.

Q.8.

$$(a) s = ut + \frac{1}{2}at^2,$$

$$h = \frac{1}{2}gt^2, \quad t = \sqrt{\frac{2h}{g}}$$

$$(b) F = ma$$

$$mg - T = m \frac{d^2h}{dt^2}$$

$$c = I\dot{\omega}$$

$$a\omega = \frac{dh}{dt}, \quad \frac{d^2h}{dt^2} = a\dot{\omega}$$

$$c = Ta = \frac{I}{a} \frac{d^2h}{dt^2}, \quad T = \frac{I}{a^2} \frac{d^2h}{dt^2}$$

$$mg - \frac{I}{a^2} \frac{d^2h}{dt^2} = m \frac{d^2h}{dt^2}$$

$$mg - \frac{ma^2}{2a^2} \frac{d^2h}{dt^2} = m \frac{d^2h}{dt^2} \quad (I \text{ value from MDS})$$

$$\frac{d^2h}{dt^2} = \frac{2g}{3} \rightarrow \int_0^h \frac{dh}{dt} = \int_0^t \frac{2g}{3} dt,$$

$$\frac{dh}{dt} = \frac{2g}{3} t, \quad \int_0^h dh = \int_0^t \frac{2g}{3} t dt, \quad h = \frac{2gt^2}{6}$$

$$t = \sqrt{\frac{3h}{g}}$$

[ Alternative energy-based method :-

$$mgh = \frac{1}{2}mv^2 + \frac{1}{2}I\omega^2$$

$$= \frac{1}{2}mv^2 + \frac{1}{2} \frac{Iv^2}{a^2} = \frac{v^2}{2} (m + I/a^2) \quad (8)$$



$$mgh = \frac{v^2}{2} (m + m/2) \quad (\text{I value from MDB})$$

$$= 3mv^2 / 4$$

$$v^2 = \frac{4gh}{3}, \quad v = \sqrt{\frac{4gh}{3}}$$

$$\frac{dh}{dt} = \sqrt{\frac{4g}{3}} h^{1/2}$$

$$\int_0^h \frac{dh}{h^{1/2}} = \int_0^t \sqrt{\frac{4g}{3}} dt, \quad 2h^{1/2} = \sqrt{\frac{4g}{3}} t,$$

$$t = \frac{2\sqrt{h} \sqrt{3}}{\sqrt{4g}} = \sqrt{\frac{3h}{g}}$$

(c) Simple disc,  $mgh \rightarrow \frac{1}{2}mv^2$ .

Yo-yo,  $mgh \rightarrow \frac{1}{2}mv^2 + \frac{1}{2}I\omega^2$ .

Since  $mgh$  is the same in both cases, the yo-yo falls more slowly than the simple disc, so the time is greater.

Q.9.

$$(a) \quad m \frac{dv}{dt} = -\lambda v$$

$$m \frac{dv}{dt} + \lambda v = 0$$

$$\frac{dv}{dt} + \left(\frac{\lambda}{m}\right)v = 0$$

---

$$(b) \quad \int \frac{dv}{v} = \int -\left(\frac{\lambda}{m}\right) dt$$

$$\ln v = -\left(\frac{\lambda}{m}\right)t + C$$

$$v = C e^{-\left(\lambda/m\right)t}$$

$v = u$  at  $t = 0$  gives  $C = u$

$$\underline{v = u e^{-\left(\lambda/m\right)t}}$$

$$\underline{T = m/\lambda}$$

$$(c) \quad v = \frac{dx}{dt} = u e^{-t/T}$$

$$\int dx = \int u e^{-t/T} dt$$

$$x = -uT e^{-t/T} + D$$

$x = 0$  at  $t = 0$  gives  $D = uT$

$$x = uT (1 - e^{-t/T})$$

$x = L$  at  $t = \infty$  gives

$$\underline{L = uT = \frac{um}{\lambda}}$$

(d)  $t = \infty$ . Friction in rear and at train wheels / bearings will result in a finite  $t$ .

(e) The characteristics of the buffer-stops are bad, because they result in maximum force at the moment of collision, and rapidly decreasing force as the train slows. Damage / injury may result on first impact. Ideally, want constant force. Use active or passive control of oil valves to obtain this. Or, use a frictional device (see King's Cross).

Q.10.

$$(a) \quad m\ddot{y} = -k(y-x) - \lambda(\dot{y}-\dot{x})$$

$$m\ddot{y} + \lambda\dot{y} + ky = \lambda\dot{x} + kx$$

$$\frac{m}{k}\ddot{y} + \frac{\lambda}{k}\dot{y} + y = \frac{\lambda}{k}\dot{x} + x$$

This is Case (c), since  $\omega_n = \sqrt{\frac{k}{m}}$ ,  $\zeta = \frac{\lambda}{2\sqrt{km}}$

$$\frac{m}{k} = \frac{1}{\omega_n^2}, \quad \frac{\lambda}{k} = \frac{1}{k} \cdot 2\zeta\sqrt{km}$$

$$= 2\zeta\sqrt{\frac{m}{k}} = \frac{2\zeta}{\omega_n}$$

$\therefore$  DE is

$$\frac{\ddot{y}}{\omega_n^2} + \frac{2\zeta}{\omega_n}\dot{y} + y = \frac{2\zeta}{\omega_n}\dot{x} + x$$

(b) For  $\zeta = 0$ , MOB gives

$$\left| \frac{y}{x} \right| = \frac{1}{1 - (\omega/\omega_n)^2} = \frac{10 \mu\text{m}}{50 \mu\text{m}} = \frac{1}{5}$$

$$1 - (\omega/\omega_n)^2 = 5, \quad \omega/\omega_n = \sqrt{5+1} = 2.45$$

$$k = m\omega_n^2 = 20 \left\{ 2\pi \times \frac{10}{2.45} \right\}^2 = \underline{13159 \text{ N/m}}$$

(c)  $\left| \frac{y}{x} \right| \leq \frac{100 \mu\text{m}}{50 \mu\text{m}} = 2$ , From the chart on

p13, a peak response of 2 occurs for  $\zeta = 0.3$

(d) From the chart on p13, for  $\omega/\omega_n = 2.45$ , adding the damping increases  $|Y/X|$  from approximately 0.2 to approximately 0.4. In other words, the amplitude of the machine increases from  $10\ \mu\text{m}$  to approximately  $20\ \mu\text{m}$ .