

ENGINEERING TRIPOS PART IB JUNE 2012

CHAIRMAN PROF. P DAVIDSON

MONDAY 4TH JUNE 2012 2 TO 4

PAPER 2 SOLUTIONS – STRUCTURES

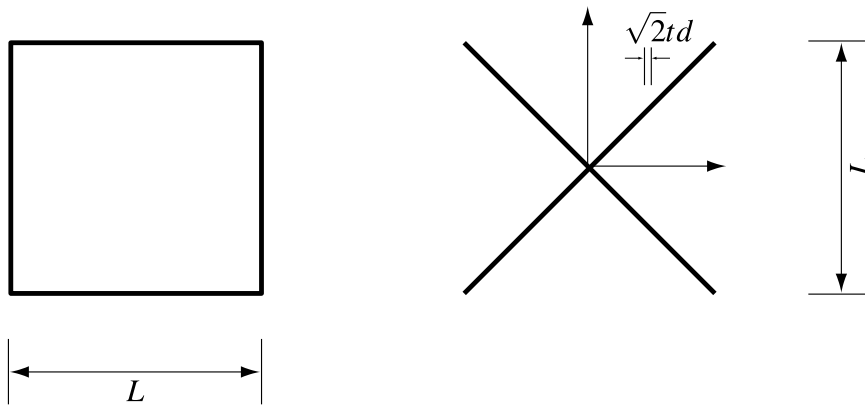
AUTHORS :

DR. F CIRAK

DR. D MCCONNELL

Part IB, Paper 2 – Structures, 2011/12

1. (a)



The second moment of area for the thin-walled square is

$$I_{xx}^s = 2 \frac{L^3 t}{12} + 2Lt \left(\frac{L}{2} \right)^2 = \frac{L^3 t}{6} + \frac{L^3 t}{2} = \frac{2}{3} L^3 t$$

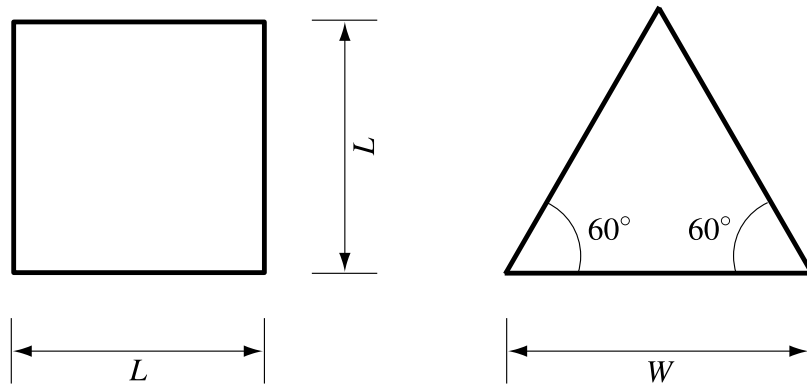
The second moment of area for the cruciform is

$$I_{xx}^c = 2 \int_{-L/2}^{L/2} \sqrt{2} t x^2 dx = 2\sqrt{2} t \frac{x^3}{3} \Big|_{-L/2}^{L/2} = \frac{\sqrt{2}}{6} L^3 t$$

The ratio of the second moments of both cross-sections has to be equal

$$\frac{I_{xx}^s}{I_{xx}^c} = \frac{2L^3 t}{3} \frac{6}{\sqrt{2}L^3 t} = \frac{4}{\sqrt{2}} = 2\sqrt{2}$$

(b)



The torsion constant for a thin-walled closed cross-section is defined as

$$J = \frac{4A_e^2}{\oint \frac{ds}{t}}$$

The torsion constant of the square is

$$J^s = \frac{4L^4 t}{4L} = L^3 t$$

The torsion constant of the equilateral triangle is

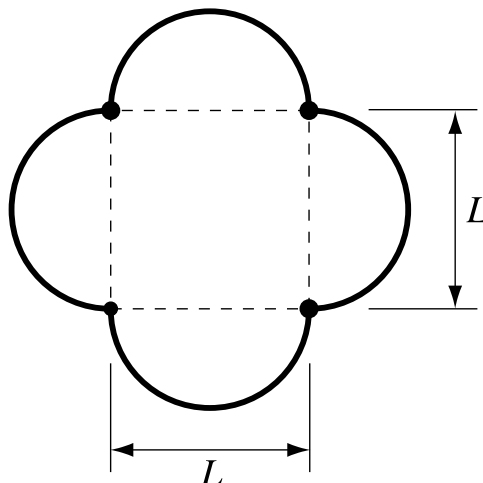
$$A_e = \frac{1}{2} W \frac{\sqrt{3}}{2} W = \frac{\sqrt{3}}{4} W^2$$

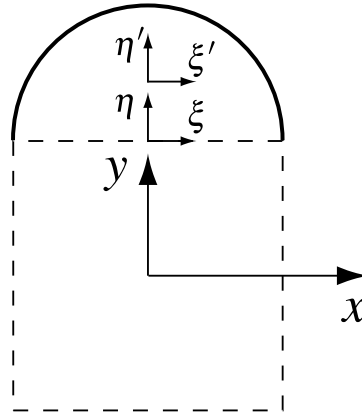
$$\Rightarrow J^t = \frac{3W^4 t}{4 \cdot 3S} = \frac{1}{4} W^3 t$$

The ration of the torsion constants is

$$\frac{J^s}{J^t} = \frac{4L^3}{W^3}$$

(c)





Parameters for one (horizontal) semi-circle

Centroid

$$\begin{aligned}\pi \frac{L}{2} t \eta_s &= \int_0^\pi \left(\frac{L}{2} \sin \theta \right) \frac{Lt}{2} d\theta = -\frac{L^2}{4} t \cos \theta \Big|_0^\pi = \frac{L^2}{2} t \\ \Rightarrow \eta_s &= \frac{L}{\pi}\end{aligned}$$

Second moment of area

$$\begin{aligned}I_{\xi\xi}^c &= \int_0^\pi \eta^2 \frac{Lt}{2} d\theta = \int_0^\pi \left(\frac{L}{2} \sin \theta \right)^2 \frac{Lt}{2} d\theta = \frac{L^3 t}{8} \int_0^\pi \sin^2 \theta d\theta \\ \Rightarrow I_{\xi\xi}^c &= \frac{L^3 t \pi}{16}\end{aligned}$$

$$\begin{aligned}I_{\eta\eta}^c &= \int_0^\pi \xi^2 \frac{Lt}{2} d\theta = \int_0^\pi \left(\frac{L}{2} \cos \theta \right)^2 \frac{Lt}{2} d\theta = \frac{L^3 t}{8} \int_0^\pi \cos^2 \theta d\theta \\ \Rightarrow I_{\eta\eta}^c &= \frac{L^3 t \pi}{16}\end{aligned}$$

Second moments of area around the centroid

$$I_{\xi'\xi'}^c = \frac{L^3 t \pi}{16} - \frac{L^3 t}{2\pi}$$

Second moment of area for the entire cross-section

$$\begin{aligned}I_{xx} &= 2 \left(I_{\xi'\xi'}^c + \frac{\pi L t}{2} \left(\frac{L}{2} + \frac{L}{\pi} \right)^2 \right) + 2 I_{yy}^c \\ &= 2 \left(\frac{L^3 t \pi}{16} - \frac{L^3 t}{2\pi} + \frac{\pi L^3 t}{8} + \frac{L^3 t}{2} + \frac{L^3 t}{2\pi} + \frac{L^3 t \pi}{16} \right) \\ &= \frac{L^3 t \pi}{2} - \frac{L^3 t}{\pi} + L^3 t + \frac{L^3 t}{\pi} = 2.5708 L^3 t\end{aligned}$$

Alternatively, the second moments of area for the semi-circle can be determined with the equations for the curved rod given in Mechanics Data Book

$$I_{\xi'\xi'}^c = \frac{\pi L t}{2} k_y^2 \quad I_{\eta'\eta'}^c = \frac{\pi L t}{2} k_x^2$$

with $a = \frac{L}{2}$ and $\alpha = \frac{\pi}{2}$

$$k_y^2 = \frac{1}{2} \frac{L^2}{4} \left(1 - 2 \left(\frac{2}{\pi} \right)^2 \right) = \frac{L^2}{8} \left(1 - \frac{8}{\pi^2} \right) \quad \Rightarrow I_{\xi'\xi'}^c = \frac{\pi L^3 t}{16} \left(1 - \frac{8}{\pi^2} \right)$$

$$k_x^2 = \frac{1}{2} \frac{L^2}{4} = \frac{L^2}{8} \quad \Rightarrow I_{\eta'\eta'}^c = \frac{\pi L^3 t}{16}$$

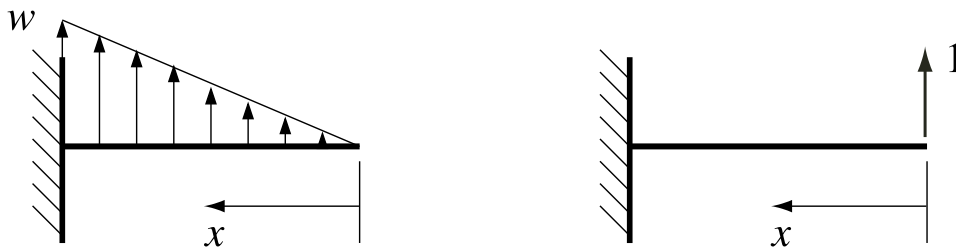
Torsion constant of the entire cross-section

$$A_e = L^2 + 2\pi \left(\frac{L}{2} \right)^2 = L^2 + \frac{\pi L^2}{2}$$

$$\oint \frac{ds}{t} = \frac{2\pi L}{t}$$

$$\begin{aligned} \Rightarrow J &= \frac{4A_e^2}{\oint \frac{ds}{t}} = \frac{t}{2\pi L} (4L^4 + 4\pi L^4 + \pi^2 L^4) \\ &= \frac{2L^3 t}{\pi} + 2L^3 t + \frac{\pi L^3 t}{2} = 4.21 L^3 t \end{aligned}$$

2. (a) Maximum displacement using virtual work



Virtual moment distribution

$$\bar{M}(x) = 1 \cdot x$$

Moment distribution due to w

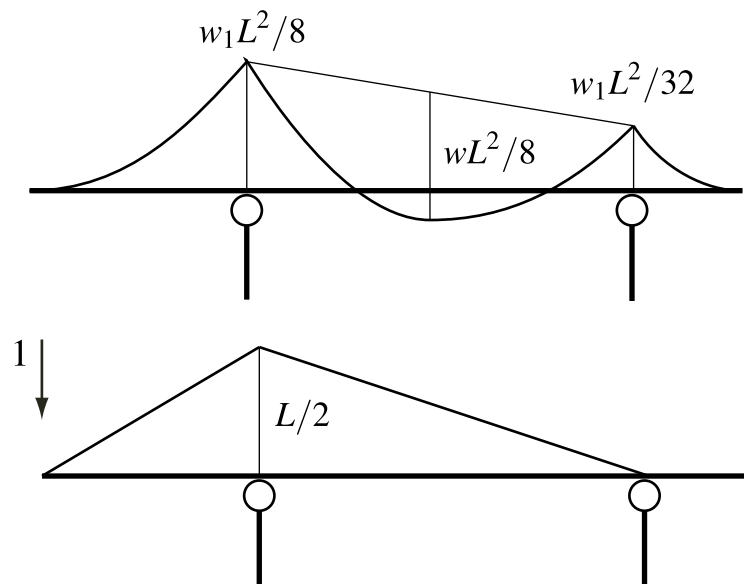
$$M(x) = \frac{1}{2} x w(x) \frac{1}{3} x = \frac{1}{6} x^2 \frac{w}{L} x = \frac{1}{6} \frac{x^3}{L} w$$

Principle of virtual work

$$1 \cdot \delta = \frac{1}{EI} \int_0^L x \frac{1}{6} \frac{x^3}{L} w dx = \frac{1}{EI} \frac{w}{6L} \int_0^L x^4 dx = \frac{1}{EI} \frac{wL^4}{30}$$

- (b) i. Although the system is statically indeterminate the bending moments in the beam can be determined without making use of the force method.

Vertical deflection at Point A



$$\begin{aligned}
 1 \cdot \delta_v EI &= \int M(x) \bar{M}(x) dx \\
 &= \frac{L}{8} \frac{w_1 L^2}{8} \frac{L}{2} + \frac{L}{6} \left(2 \frac{w_1 L^2}{8} + \frac{w_1 L^2}{32} \right) \frac{L}{2} - \frac{L L w_1 L^2}{3 \cdot 2 \cdot 8} \\
 &= \frac{w_1 L^4}{128} + \frac{3 w_1 L^4}{128} - \frac{w_1 L^4}{48} = \frac{w_1 L^4}{96}
 \end{aligned}$$

Alternatively, the vertical deflection can be obtained by superposing Data Book cases.

Rotation of the left support

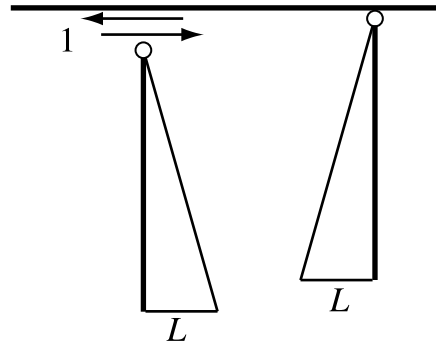
$$\frac{w_1 L^3}{24EI} - w_1 \frac{L^2}{32} \frac{L}{6EI} - \frac{w_1 L^2}{8} \frac{L}{3EI} = -w_1 \frac{L^3}{192EI}$$

Vertical deflection

$$\delta_v = w_1 \frac{L^4}{128EI} + w_1 \frac{L^3}{192EI} \frac{L}{2} = w_1 \frac{L^4}{96EI}$$

ii. Horizontal deflection at Point A

To determine the horizontal deflection, first the moments in the columns need to be determined. To this end, consider a statically determinate system where the joint between the left column and beam is released.



The compatibility of the displacements at the joint requires

$$\alpha_1 X + \alpha_0 = 0$$

where X is the yet unknown horizontal shear force at the joint.

Difference in the horizontal displacement between the beam and column caused by the virtual unit load pair is

$$\alpha_1 = \frac{2L^3}{3EI} \quad (\text{computed with principle of virtual work})$$

Difference in the horizontal displacement between the beam and column caused by w_2 is

$$\alpha_0 = \frac{w_2 L^4}{30EI} \quad (\text{given in part (a)})$$

Hence, the horizontal shear force at the joint is

$$\Rightarrow X = -\frac{w_2 L}{20}$$

After knowing the shear force at the top of the left column its horizontal deflection can be computed with

$$\delta_h = \frac{1}{EI} \frac{L}{3} L \frac{w_2 L^2}{20} = \frac{1}{EI} \frac{w_2 L^4}{60}$$

Note, this deflection is the same as for a single cantilever beam with twice the EI.

3. (a)

Axial stress

$$\sigma_a = \frac{pR}{2t} = \frac{250 \cdot 10^3 \cdot 1}{2 \cdot 3 \cdot 10^{-3}} = 41.667 \text{ MPa}$$

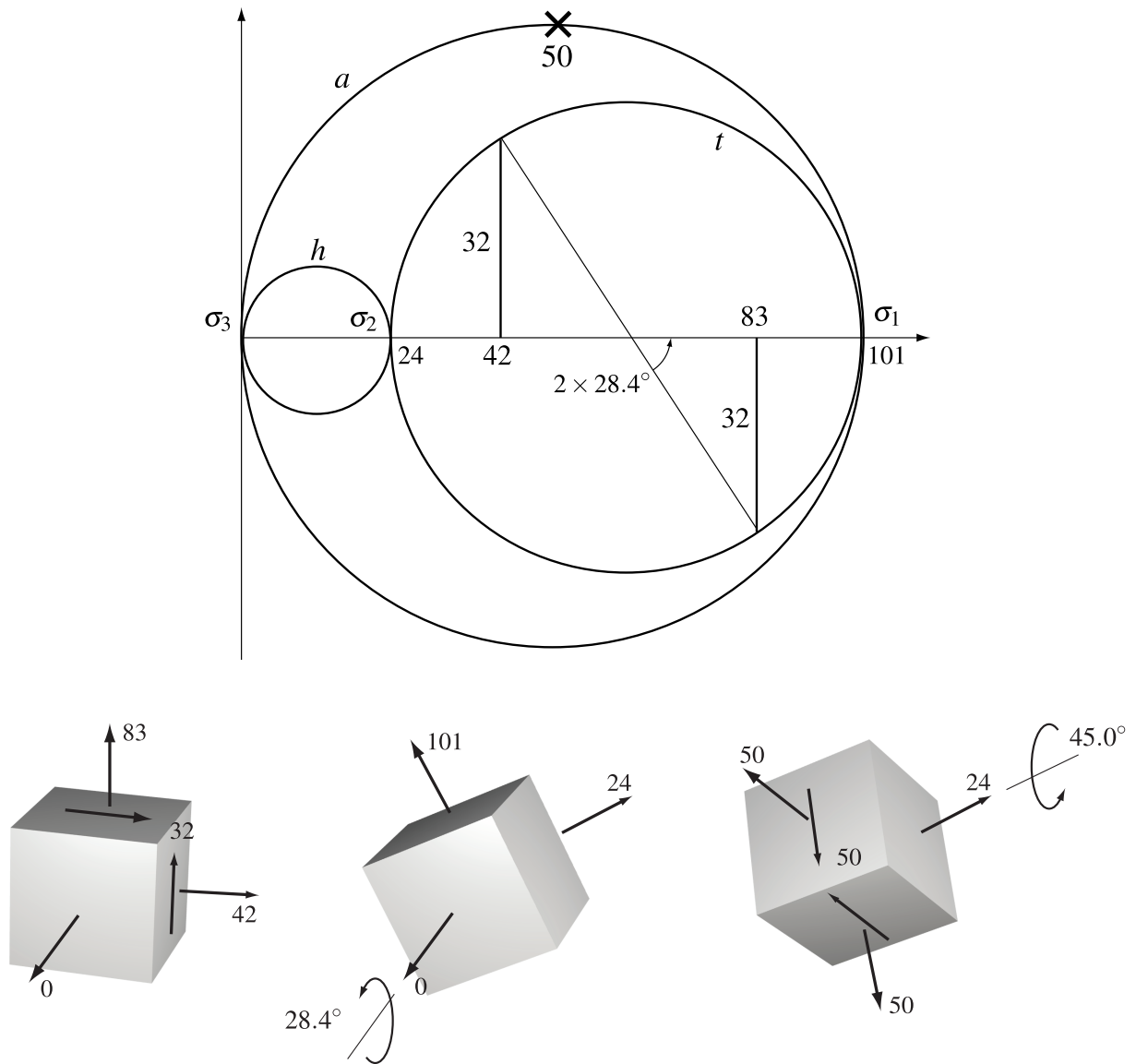
Hoop stress

$$\sigma_h = \frac{pR}{t} = 2 \cdot 41.667 \text{ MPa} = 83.33 \text{ MPa}$$

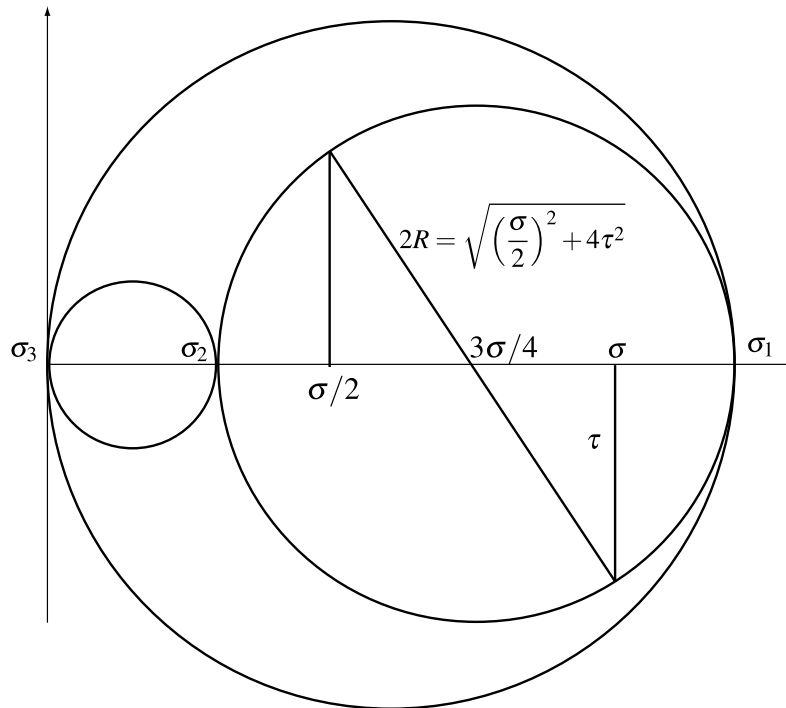
Shear stress

$$\tau = \frac{T}{2\pi R^2 t} = \frac{600 \cdot 10^3}{2\pi \cdot 1 \cdot 3 \cdot 10^{-3}} = 31.83 \text{ MPa}$$

(b)



(c)



Von Mises criterion

$$(\sigma_1 - \sigma_2)^2 + (\sigma_2 - \sigma_3)^2 + (\sigma_3 - \sigma_1)^2 = 2Y^2$$

This gives for plain stress ($\sigma_3 = 0$)

$$\sigma_1^2 - \sigma_1\sigma_2 + \sigma_2^2 = Y^2$$

From the Mohr's circle we obtain

$$\sigma_1 = \frac{3}{4}\sigma + \sqrt{\tau^2 + \frac{1}{16}\sigma^2}$$

$$\sigma_2 = \frac{3}{4}\sigma - \sqrt{\tau^2 + \frac{1}{16}\sigma^2}$$

The square root expression in the previous equations is abbreviated with R .

$$\frac{9}{16}\sigma^2 + \frac{3}{2}\sigma R + R^2 + \frac{9}{16}\sigma^2 - \frac{3}{2}\sigma R + R^2 - \frac{9}{16}\sigma^2 + R^2 = Y^2$$

$$\frac{9}{16}\sigma^2 + 3R^2 = Y^2$$

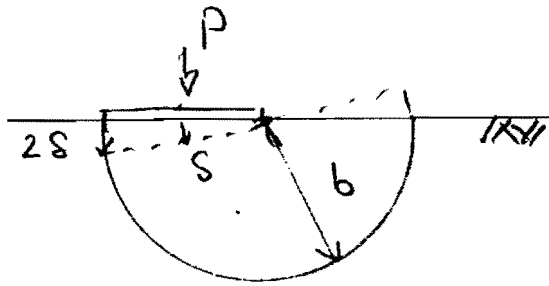
$$\frac{9}{16}\sigma^2 + 3\tau^2 + \frac{3}{16}\sigma^2 = Y^2$$

$$\frac{3}{4}\sigma^2 + 3\tau^2 = Y^2$$

B/1

IB Paper 2 Structures 2011/12

4(a)

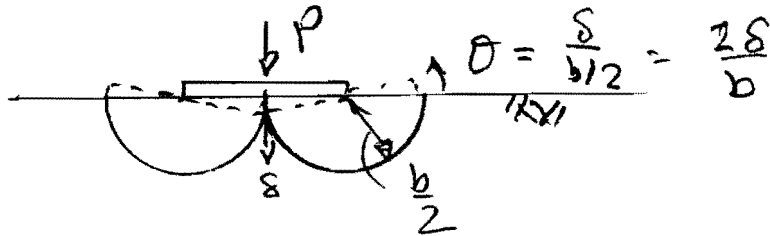


NOTE:
All calculations
per unit length
into page.

Work: $P_a s = \pi b \times 2s \times k$

$\therefore \underline{P_a = 2\pi b k}$

(b)



Work: $P_b s = 2 \left(\pi \frac{b}{2} \times s \times k \right) + 2E \times m$

$\underline{P_b = \pi b k + \frac{4m}{b}}$

(c) To prevent mode (b), $P_a \leq P_b$

$\therefore 2\pi b k \leq \pi b k + \frac{4m}{b}$

or $m \geq \frac{\pi b^2 k}{4}$

$m = Z_p \sigma_y$, where $Z_p = \frac{t^2}{4}$

4(c) Cont'd

and $\sigma_y = 2ck$ (see Materials Data Book.)

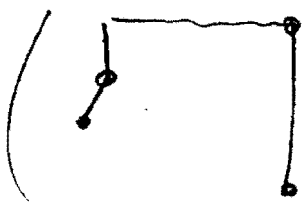
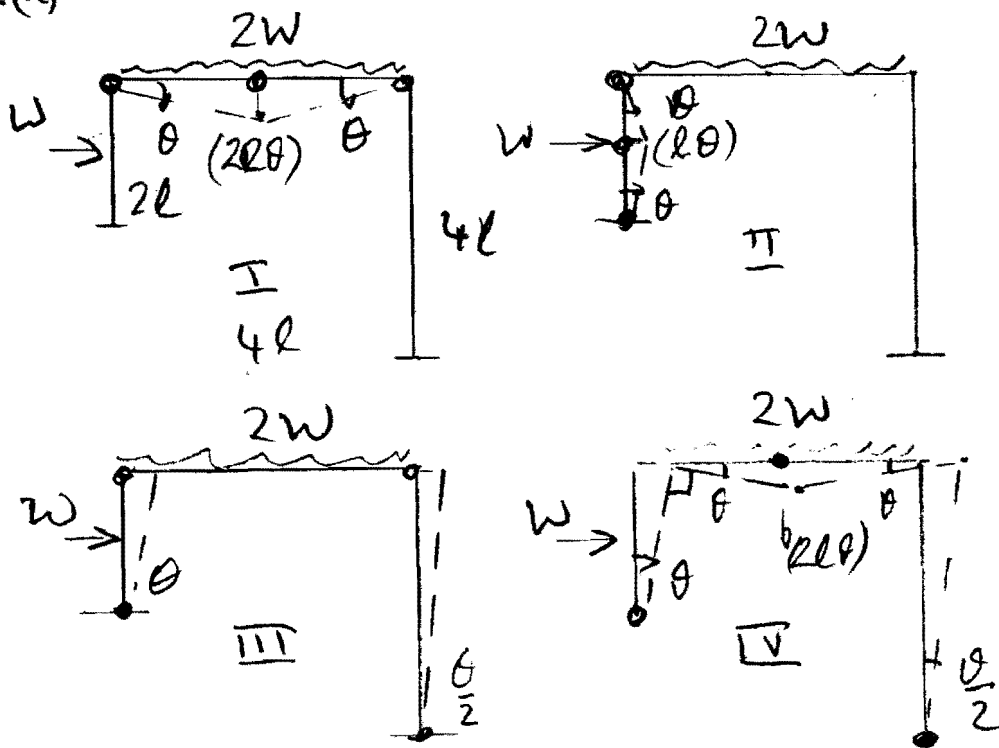
$$\therefore m = \frac{E^2 ck}{2} \geq \frac{\pi b^2 k}{4}$$

so for (a) to control,

$$\text{require } t^2 \geq \frac{\pi b^2}{2c}; \quad t \geq b \sqrt{\frac{\pi}{2c}}$$

(Most common error was to set $\sigma_y = ck$.)

5.(a)



Very unlikely because of large span (4l).

5(b) Work equations

$$\text{I: } 2W_1 \times \frac{2l\theta}{2} = 4m_p\theta; \quad W_1 = \frac{2m_p}{l}$$

$$\text{II: } W_2 \times l\theta = 4m_p\theta; \quad W_2 = \frac{4m_p}{l}$$

$$\text{III: } W_3 \times l\theta = m_p(2\theta + 2\frac{l\theta}{2})$$

$$\therefore W_3 = \frac{3m_p}{l}$$

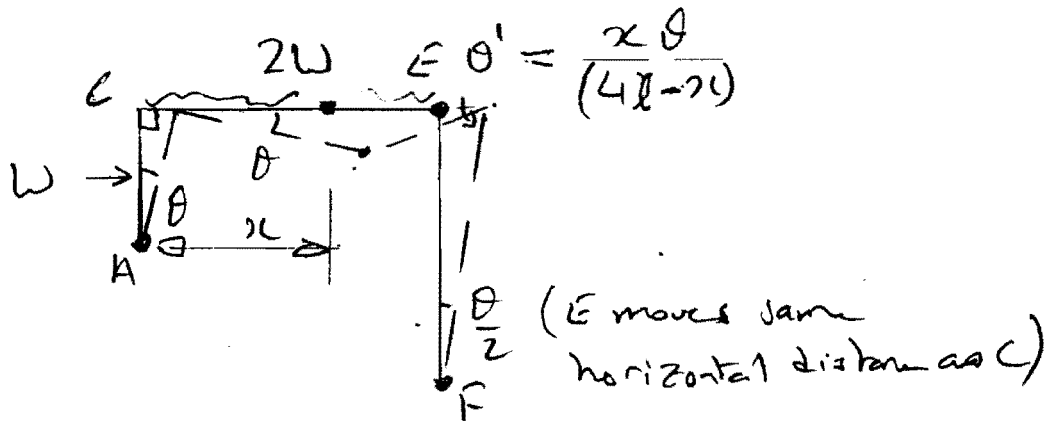
$$\text{IV: } W_4 \times l\theta + 2W_4 \times \frac{2l\theta}{2} = m_p(4\theta + 2\frac{l\theta}{2})$$

$$\therefore W_4 = \frac{5m_p}{3l}$$

5(c) Mode IV is critical, so optimize this
(Modes I and III have no plastic pin
in beam, so no improvement possible.)

In: Mode I, the beam mechanism
is symmetrical, so considering a
variable pin position will confirm
the critical form of Mode I
indeed has the pin in the
middle! (Many candidates "proved"
this for little reward.)

5(c) cont



$$\text{Work: } W \times l \theta + 2W \times \frac{x\theta}{2} = m_p \left(2\theta + \frac{2x\theta}{(4l-x)} + \frac{2\theta}{2} \right)$$

(load \times average drop)

$$\therefore W(l+x) = m_p \left(3 + \frac{2x}{(4l-x)} \right)$$

$$\therefore W = m_p \frac{(12l-x)}{(4l^2 + 3lx - x^2)}$$

$$\text{Extreme value } \left(\frac{d}{dx} \left(\frac{v}{u} \right) = \frac{v u' - u v'}{v^2} = 0 \right)$$

$$\therefore (4l^2 + 3lx - x^2) \times -1 - (12l-x)(3l-2x) = 0$$

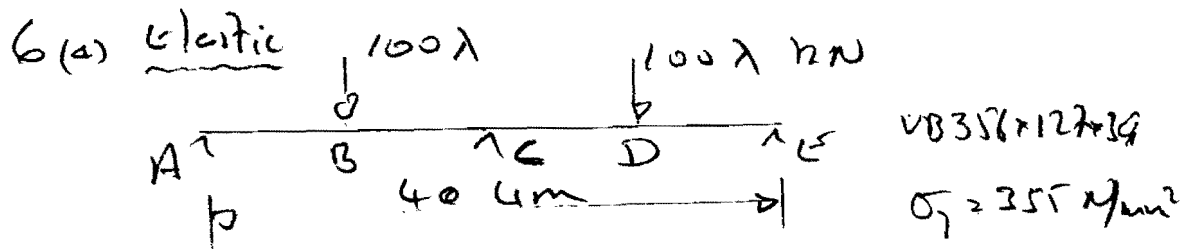
$$\Rightarrow x^2 - 24lx + 40l^2 = 0$$

$$\Rightarrow x = 12l (\pm 10.197l)$$

$$\therefore x = 1.802l$$

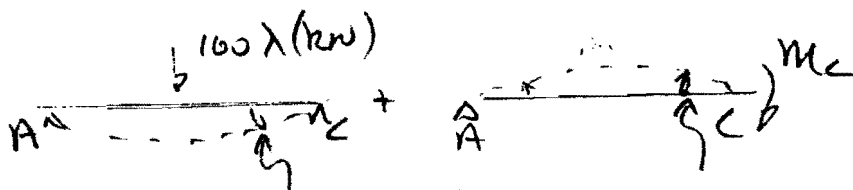
$$\Rightarrow W = m_p \frac{(12 - 1.802)}{l (4 + 5.41 - 3.25)} = 1.66 \frac{m_p}{l}$$

$$(\text{cf } 5/3 = 1.667 \text{ (w central pin)})$$



First yield at $M_{max} = M_y = Z_e \sigma_y$

Need elastic analysis for M_{max}



Compat. bility $\frac{100\lambda \times (8^2)}{16EI} = \frac{M_C \times 8}{3EI}$

$\Rightarrow M_C = \frac{3}{2} \times 100\lambda = 150\lambda \text{ kNm}$

(Other elastic methods give same result)

100λ
 $M_C = 150\lambda$
 $M_B = \frac{150\lambda}{2} - \frac{100\lambda \times 8}{4} = -125\lambda \text{ kNm}$

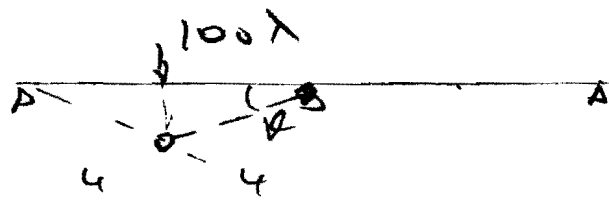
$\therefore M_{max} = 150\lambda \text{ kNm}$

Now $M_y = Z_e \sigma_y = \frac{576 \times 10^3 \times 355}{(\text{cm}^3)} \text{ Nmm}$
 $= 204 \text{ kNm}$

So $M_{max} = M_y \Rightarrow 150\lambda = 204, \lambda = \underline{1.36}$

(NOTE: Ans for major axis bending; full marks were also given for minor-axis).

6(b) Upper Bound (UB)



(Symmetric mode gives same ans)

$$\text{Work} = 100\lambda \times 4\theta = 3M_p\theta$$

$$\therefore \lambda = \frac{3}{400} M_p$$

$$M_p = Z_p \sigma_y = \frac{659 \times 355}{10^3} = 234 \text{ kNm}$$

$$\therefore \lambda = \frac{3 \times 234}{400} = \underline{\underline{1.76}}$$

6(c) Lower Bound (LB). λ checked for "a" LB,

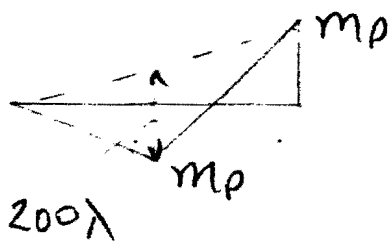
not best, so any BMD in equilibrium with load, and \$M_p\$ to match \$M_{max}\$ is OK.

eg $M_{max} = \frac{PL}{4} = 200\lambda \text{ kNm}$

The diagram shows a beam with a central downward load of \$100\lambda\$ and two supports on either side, with a distance of \$4\$ units between the supports. The maximum moment is indicated as \$M_{max} = \frac{PL}{4} = 200\lambda \text{ kNm}\$.

$$\Rightarrow \lambda = \frac{234}{200} = \underline{\underline{1.17}} \text{ OK etc}$$

Must candidates looked for best, i.e



$$\text{so } \frac{M_p}{2} = 200\lambda = -M_p$$

$$\therefore 200\lambda = \frac{3}{2} M_p, \Rightarrow \underline{\underline{\lambda = 1.76}}$$

(\equiv UB)

Rev 20/6/12.