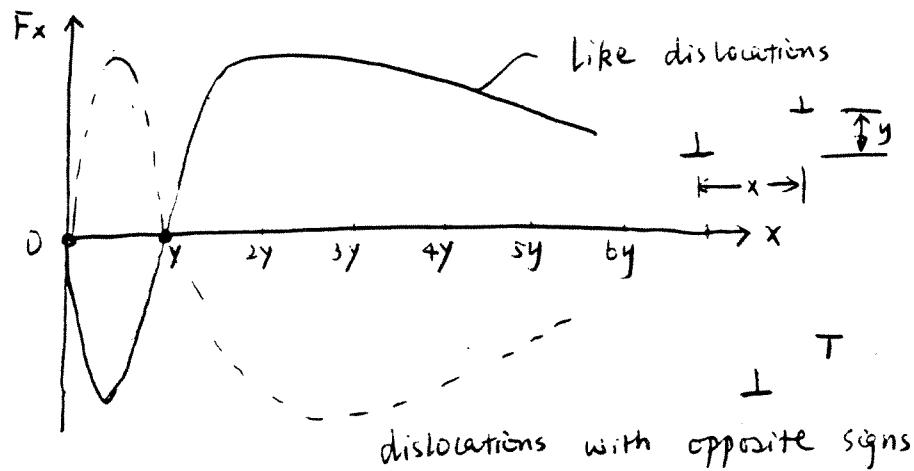
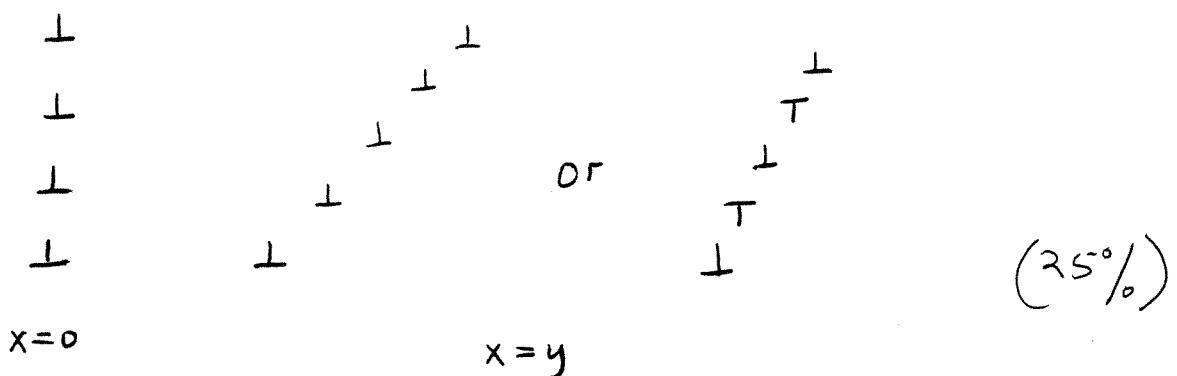


1. (a) For two screw dislocations, the <sup>4C1</sup> interaction force is independent of the orientation angle  $\theta$ , and inversely proportional to the distance  $r$ . For two edge dislocations, the interaction force is a function of both  $r$  and  $\theta$ , and is also dependent upon the sign of the Burger's vector.



From the above figure, it can be seen that there are two possible equilibrium configurations for edge dislocations:



$$(b) \quad \sigma_{xx} = -\frac{Gb}{2\pi(1-\nu)r} \sin\theta (2 + \cos 2\theta), \quad \sigma_{yy} = \frac{Gb}{2\pi(1-\nu)r} \sin\theta \cos 2\theta,$$

$$\sigma_{zz} = \nu(\sigma_{xx} + \sigma_{yy}), \quad \sigma_{xy} = \frac{Gb}{2\pi(1-\nu)r} \cos\theta \cos 2\theta.$$

$$\sigma_{xz} = \sigma_{yz} = 0 \quad (20\%)$$

$$1. (c) \quad \sigma_m = \frac{\sigma_{xx} + \sigma_{yy} + \sigma_{zz}}{3} = - \frac{Gb(1+\nu)}{3\pi(1-\nu)r} \sin\theta$$

$$\Rightarrow \frac{d\sigma_m}{d\theta} = 0 \Rightarrow (\sin\theta)' = \cos\theta = 0 \Rightarrow \theta = \pm \frac{\pi}{2}$$

At  $\theta = -\frac{\pi}{2}$ ,  $\sigma_m = \frac{Gb(1+\nu)}{3\pi(1-\nu)r}$  is the maximum

$$\text{For } r = 10b, \quad \sigma_m = \frac{1+\nu}{1-\nu} \frac{G}{30\pi} \approx 0.02G$$

This is a ~~large~~ large stress compared to  $\sigma_y$  (30%)

(d) For a mixed dislocation whose Burger's vector makes an angle  $\theta$  with the dislocation line, its energy is equal to that associated with a screw dislocation having a Burger's vector  $b \cos\theta$  plus that due to an edge dislocation having a Burger's vector  $b \sin\theta$ :

$$U = \frac{Gb^2}{4\pi} \left( \frac{\sin^2\theta}{1-\nu} + \cos^2\theta \right)$$

$$\frac{dU}{d\theta} = 0 \Rightarrow \sin 2\theta - (1-\nu) \sin 2\theta = 0 \Rightarrow \theta = \frac{\pi}{4} \text{ or } -\frac{\pi}{4}$$

This result implies that a dislocation loop tends to be circular.

(25%)

2(a) Mean stress - needed to describe brittle failure of rocks, crazing & shear failure of polymers etc.

Mises equivalent stress - needed to describe yielding of metals

Max. shear stress - "

Max. principal strain - crazing of polymers

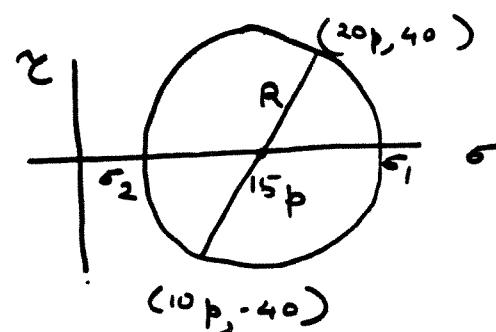
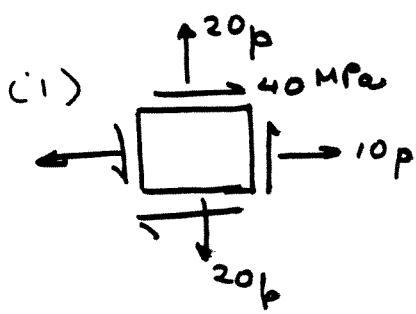
Max. principal stress - crack propagation: cleavage failure.

(25%)

$$(b) \sigma_h = \frac{pt}{t}, \quad \sigma_e = \frac{pt}{2t}, \quad \tau = \frac{T}{2A_e t}$$

$$r = 200 \text{ mm}, \quad t = 10 \text{ mm}, \quad T = 100 \text{ kNm}$$

$$\sigma_h = 20 \text{ p}, \quad \sigma_e = 10 \text{ p}, \quad \tau = \frac{100 \times 10^3 \times 10^3}{2\pi(200)^2 \times 10} \approx 40 \text{ MPa}$$



$$R = \sqrt{25p^2 + 40^2}$$

(30%)

$$\sigma_1 = 15p + \sqrt{25p^2 + 40^2}, \quad \sigma_2 = 15p - R$$

$$\text{Trueca criterion} \Rightarrow \sqrt{25p^2 + 40^2} = \frac{\sigma_1}{2} = 100 \Rightarrow p = 18.3 \text{ MPa}$$

2(b)

(iii) Mohr principal stress criterion

$$\sigma_1 = \sigma_f$$

$$\Rightarrow 15p + \sqrt{25p^2 + 40^2} = 200$$

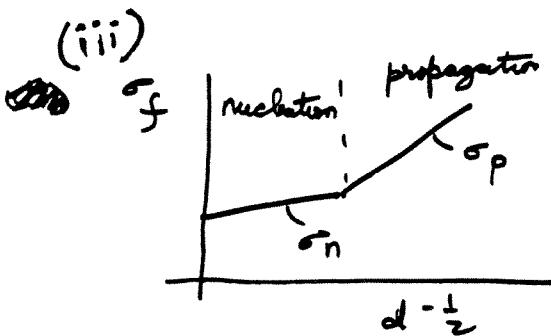
$$p^2 - 30p + 192 = 0$$

$$p = \frac{30 - \sqrt{30^2 - 4 \times 192}}{2}$$

$$p = 9.26 \text{ MPa}$$

$$\tan 2\phi = \frac{40}{5 \times 9.26} \Rightarrow \phi = 20.4^\circ$$

∴ crack propagation at  $69.6^\circ$  from longitudinal axis. (30%)



In coarse grained mtl, nucleation controlled

In fine grained mtl, yielding occurs first & then failure after a certain amt. of plastic flow  $\Rightarrow$  propagation controlled

(15%)

3(a)  $G$  is the rate of change of potential energy with crack length

$$G = -\frac{\partial U}{\partial a}$$

where  $U$  is the potential energy &  $a$  the crack length.

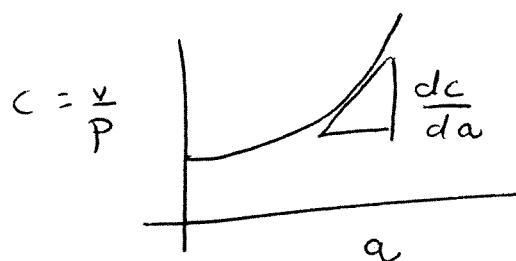
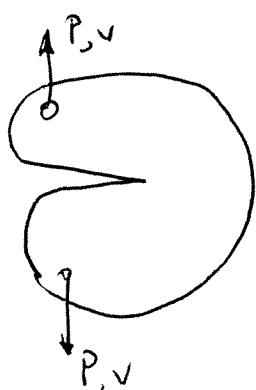
Fracture occurs when  $G = G_c$ , where  $G_c$  is material dependent.

$K$  is a measure of the magnitude of the stress singularity in the vicinity of a sharp crack in a nominally elastic material. Fracture occurs when  $K = K_c$ , where  $K_c$  is a material specific value (20%)

(b)  $G = \frac{1}{2B} P^2 \frac{dc}{da}$

where  $c$  is the compliance of the specimen  $c = \frac{v}{P}$

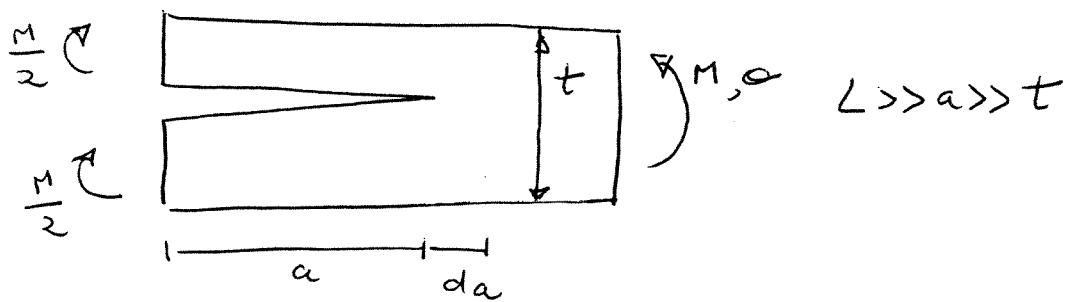
&  $a$  the crack length.



Perform a compliance measurement as a function of crack length & extract  $G$  from above formula. (20%)

C(i)

Consider  $1/2$  specimen



Consider crack growth at constant  $\sigma$

$$U = \frac{1}{2} \frac{M^2 L}{EI}, \quad I = \frac{Bt^3}{12}$$

For crack growth  $da$

$$dU = \left[ \frac{1}{2} \left( \frac{M}{2} \right)^2 \frac{12 \times 8 \times 2}{EBt^3} - \frac{1}{2} \frac{M^2}{E} \frac{12}{Bt^3} \right] da$$

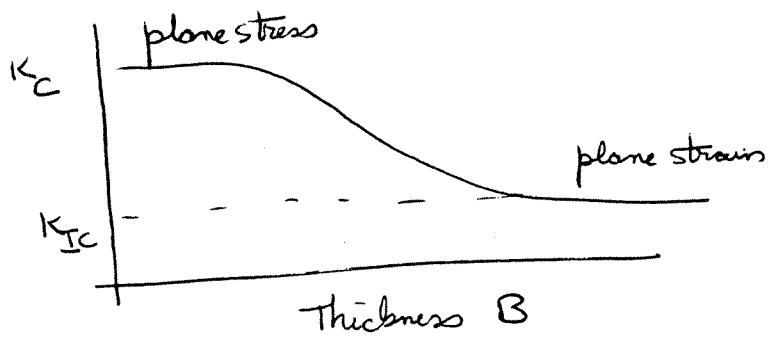
$$G_I = \frac{18 M^2}{E t^3} \quad (40\%)$$

$$(ii) G_{Ic} = \frac{1}{2} \sigma_u s_u$$

$$M = \sqrt{\frac{\sigma_u s_u E t^3}{36}} \quad (20\%)$$

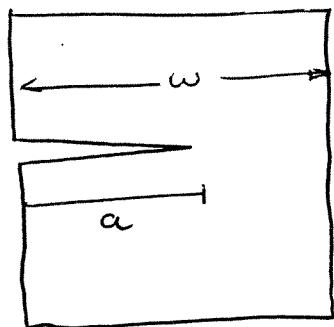
$$= \frac{1}{6} \sqrt{\sigma_u s_u E t^3}$$

4(a)



Usually plane stress  $K_C \approx 2K_{IC}$  for ductile metals.  
(20%)

(b)



Consider for example edge cracked specimen shown above.  
The conditions for a valid  $K_{IC}$  test are.

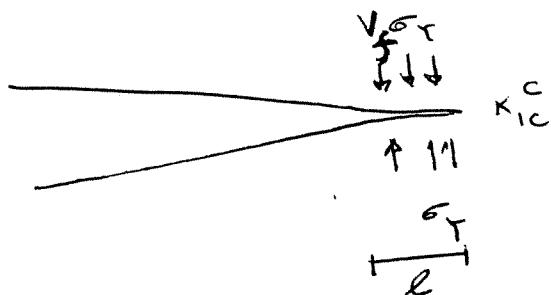
$$a, w-a > 2.5 \left( \frac{K_{IC}}{\sigma_Y} \right)^2 \rightarrow \text{plastic zone smaller than } a, w$$
$$B > 2.5 \left( \frac{K_{IC}}{\sigma_Y} \right)^2, \text{ where } B = \text{thickness of specimen}$$

→ so that plane strain  $K_{IC}$  is measured  
(see fig above) (20%)

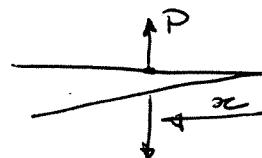
(c) Crack bridging occurs with stretching of the fibres in  
(i) the wake of the crack. Load is transmitted  
across the crack faces by the fibres. The bridged  
zone increases in length with increasing crack

size and this results in a increase in the remote measured  $K_c$  or  $G_c$ . Therefore the reinforced ceramic exhibits R-curve effects. (20%)

(ii)



From data sheet



$$K = \sqrt{\frac{2}{\pi}} \frac{P}{\sigma_c}$$

$$K^\infty = K_{IC}^C + \sqrt{\frac{2}{\pi}} \int_0^l \frac{V_f \sigma_T}{\sqrt{x}} dx$$

$$K^\infty = K_{IC}^C + \sqrt{\frac{2}{\pi}} V_f \sigma_T l$$

$$K^\infty = K_{IC}^C + \frac{2\sqrt{2}V_f \sigma_T}{\sqrt{\pi}} \sqrt{l}.$$

(40%)