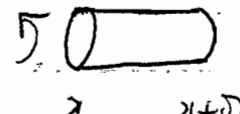


PART IIA PAPER Datasheet: 3C5/3C6 Data Sheet Dynamics & Vibration

1. (a)

$$\text{Torque } GJ \frac{\partial \theta}{\partial x} \Big|_x \quad \text{---} \quad \text{Torque } GJ \frac{\partial \theta}{\partial x} \Big|_{x+\Delta x}$$


where  $J = \frac{\pi}{2} a^4$  (from data book)

$$\text{So } \underbrace{\rho J \Delta x}_{\text{Polar moment of inertia of element}} \frac{\partial^2 \theta}{\partial t^2} = GJ \left( \frac{\partial \theta}{\partial x} \Big|_{x+\Delta x} - \frac{\partial \theta}{\partial x} \Big|_x \right)$$

Polar moment  
of inertia of element

Divide by  $\Delta x$ , let  $\Delta x \rightarrow 0$ :

$$\rho J \frac{\partial^2 \theta}{\partial t^2} = GJ \frac{\partial^2 \theta}{\partial x^2} \quad \text{as data sheet.}$$

(b) Fixed at  $x=0 \rightarrow \theta=0$

Free at  $x=L \rightarrow$  no torque  $\rightarrow \frac{\partial \theta}{\partial x} = 0$

For a mode, let  $\theta(x,t) = u(x) e^{i\omega t}$

$$\text{Then } -\rho \omega^2 u = G \frac{d^2 u}{dx^2}$$

General solution  $u = A \cos kx + B \sin kx$ ,  $k^2 = \frac{\rho \omega^2}{G}$

$$u = 0 \text{ at } x=0 \rightarrow A = 0$$

$$u' = 0 \text{ at } x=L \rightarrow \cos kL = 0$$

$$\therefore kL = (n + \frac{1}{2})\pi, \quad n = 0, 1, 2, \dots$$

So modes are  $u_n(x) = \sin \frac{(n + \frac{1}{2})\pi x}{L}$

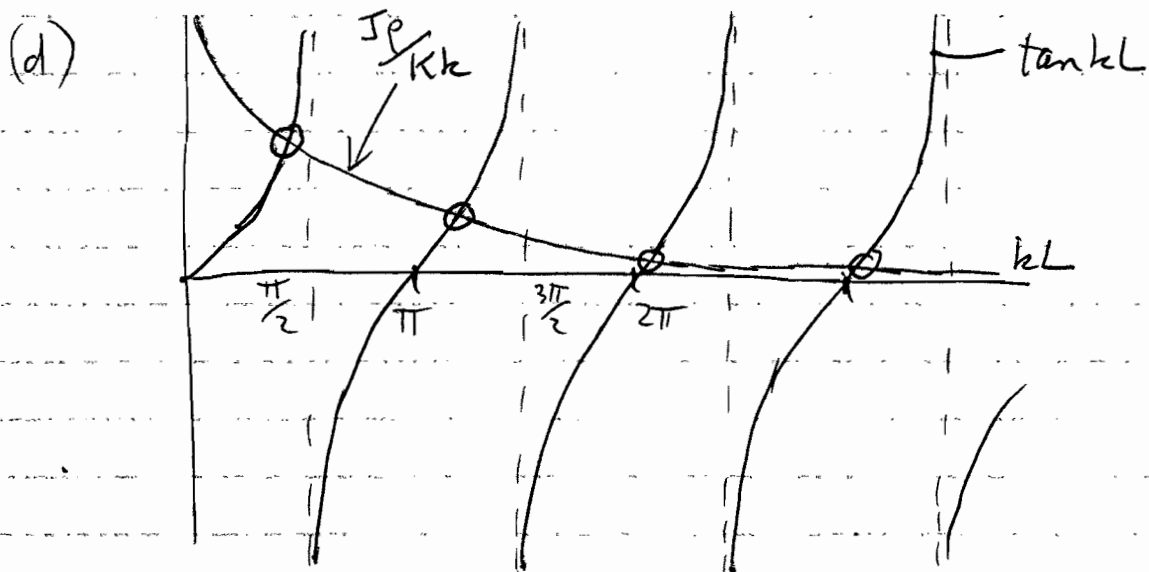
with natural frequency  $\omega_n = \sqrt{\frac{G}{\rho}} \frac{(n + \frac{1}{2})\pi}{L}$

(c) Rotor attached at  $x=L$

$$\text{Apply Newton to rotor: } K \frac{\partial^2 \theta}{\partial t^2} = -GJ \frac{\partial \theta}{\partial x} \text{ at } x=L$$

l. cont. i.e.  $-K\omega^2 u = -GJ u'$  at  $x = L$   
 so with  $u = \sin kx$ ,  $K\omega^2 \sin kL = GJ k \cos kL$

$$\therefore \tan kL = \frac{GJ}{K} \cdot \frac{1}{\omega k} = \frac{J\rho}{Kk} \quad \text{with } \omega = k \sqrt{\frac{G}{\rho}}$$



Values of  $kL$  at natural frequencies are ringed. All are lower than corresponding values without the rotor, which are the values where  $\tan kL \rightarrow \infty$ . This lowering is expected since inertia has been added without changing stiffness.

The higher values tend towards the zeros of the tangent function. These correspond to  $\sin kL = 0$ , which gives the natural frequencies of a shaft fixed at both ends. The inertia of the rotor acts as a clamp at these high frequencies.

As  $K \rightarrow \infty$  (heavy rotor) the lowest frequency  $\rightarrow 0$ . This is a slow mode with the rotor moving on the "spring" of a massless shaft.

2. (a) From data sheet, equation is

$$m \frac{\partial^2 w}{\partial t^2} + EI \frac{\partial^4 w}{\partial x^4} = f \quad (1)$$

For modes,  $f=0$  and  $w = u(x) e^{i\omega t}$ .

So require  $-m\omega^2 u + EI u'''' = 0$

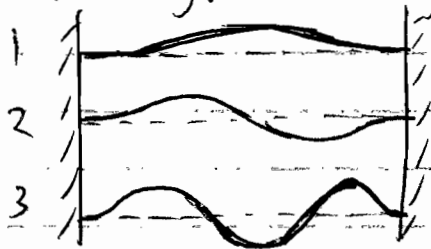
Solutions are  $u = \sin(\alpha x), \sinh(\alpha x), \cos(\alpha x), \cosh(\alpha x), \alpha^4 = \frac{m\omega^2}{EI}$

Take a general linear combination, with 4 constants.

Use the boundary conditions  $u=0, u'=0$  at  $x=0, L$  to get 4 simultaneous equations for these constants.

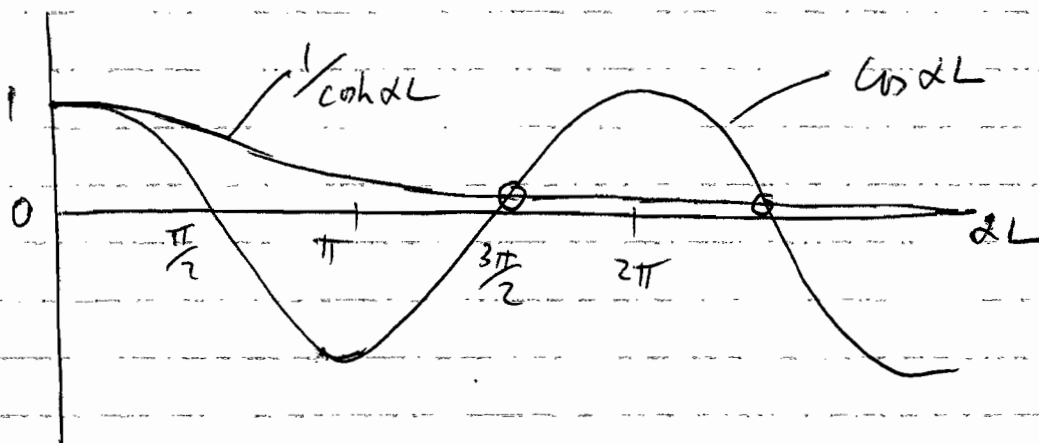
For non-trivial solution the determinant of this set must vanish. This gives the eigenvalue equation.

Modes:



(b) Given  $\cos \alpha L \cosh \alpha L = 1$

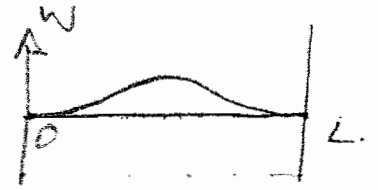
$$\rightarrow \cos \alpha L = \frac{1}{\cosh \alpha L}$$



So  $\alpha L \approx (n + \frac{1}{2})\pi, n = 1, 2, 3, \dots$

$$\text{So } \omega_n \approx \sqrt{\frac{EI}{m}} \frac{(n + \frac{1}{2})^2 \pi^2}{L^2}$$

2.(c) Put  $w = x^2(L-x)^2$  into (1)  
 $= x^2 - 2x^3L + xL^4$



$\rightarrow f = EI \cdot 24 = \text{constant}$   
 So this is the solution to static response to a uniform load, e.g. self-weight.

Expression satisfies  $w=0, w'=0$  at  $x=0, L$   
 so it fits the clamped-clamped beam.  
 If "looks plausible" to approximate the lowest mode,  
 so use Rayleigh.

From data sheet

$$V = \frac{1}{2} EI \int_0^L w''^2 dx = \frac{1}{2} EI \int_0^L (12x^2 - 12xL + 2L^2)^2 dx$$

$$= 2EI \int_0^L (36x^4 + 36x^2L^2 + L^4 - 72x^3L + 12x^2L^2 - 12xL^3) dx$$

$$= 2EI \left[ \frac{36L^5}{5} + \frac{36L^5}{3} + L^5 - \frac{72L^5}{4} + \frac{12L^5}{3} - \frac{12L^5}{2} \right]$$

$$= 2EIL^5 \left( \frac{36+60+5-90+20-30}{5} \right) = \frac{2}{5} EIL^5$$

$$\bar{T} = \frac{1}{2} m \int_0^L w^2 dx = \frac{1}{2} m \frac{L^9}{630} \quad \text{as given}$$

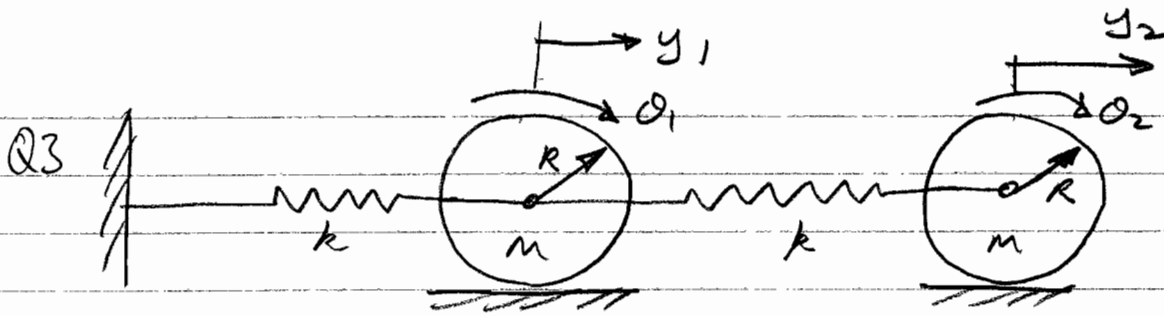
$$\text{So } \omega^2 \approx \frac{V}{\bar{T}} = \frac{EI}{mL^4} \cdot \frac{2}{5} \left/ \left( \frac{1}{2} \times \frac{1}{630} \right) \right. = 504 \frac{EI}{mL^4}$$

Approximate answer from (b) with  $n=1$  is

$$\omega^2 \approx \frac{EI}{mL^4} \left( \frac{3\pi}{2} \right)^4 \approx 493 \frac{EI}{mL^4}$$

The two are very close, and we know from the graph in (b) that  $493 \frac{EI}{mL^4}$  is a bit too low. On the

other hand the Rayleigh estimate is too high, so true result must lie between the two.



$$(a) T = \frac{1}{2} m \dot{y}_1^2 + \frac{1}{2} J \dot{\theta}_1^2 + \frac{1}{2} m \dot{y}_2^2 + \frac{1}{2} J \dot{\theta}_2^2$$

$$J = MR^2/2 \quad \& \quad \dot{y}_1 = R\dot{\theta}_1, \quad \dot{y}_2 = R\dot{\theta}_2$$

$$\text{So } T = \frac{1}{2} m \dot{y}_1^2 + \frac{1}{2} \frac{MR^2}{2} \left(\frac{\dot{y}_1}{R}\right)^2 + \frac{1}{2} m \dot{y}_2^2 + \frac{1}{2} \frac{MR^2}{2} \left(\frac{\dot{y}_2}{R}\right)^2$$

$$T = \frac{1}{2} \left( \frac{3}{2} m \dot{y}_1^2 + \frac{3}{2} m \dot{y}_2^2 \right) \Rightarrow [M] = \frac{3m}{2} \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix}$$

$$V = \frac{1}{2} k y_1^2 + \frac{1}{2} k (y_2 - y_1)^2$$

$$= \frac{1}{2} k (y_1^2 + y_2^2 - 2y_2 y_1 + y_1^2)$$

$$= \frac{1}{2} k (2y_1^2 - 2y_1 y_2 + y_2^2) \quad [K] = k \begin{bmatrix} 2 & -1 \\ -1 & 1 \end{bmatrix}$$

(b) Eigenvalue problem is  $([K] - \omega^2 [M]) \begin{Bmatrix} y_1 \\ y_2 \end{Bmatrix} = 0$

$$\begin{bmatrix} 2k - \omega^2 \frac{3m}{2} & -k \\ -k & k - \omega^2 \frac{3m}{2} \end{bmatrix} \begin{Bmatrix} y_1 \\ y_2 \end{Bmatrix} = 0$$

$$\text{C.E.} \Rightarrow (2k - \omega^2 \frac{3m}{2})(k - \omega^2 \frac{3m}{2}) - k^2 = 0$$

$$k^2 - \omega^2 \frac{3m}{2} (2k + k) + \omega^4 \frac{9m^2}{4} = 0$$

$$\text{or } \frac{9m^2}{4} \omega^4 - \frac{9k}{2} m \omega^2 + k^2 = 0$$

$$\omega^2 = \frac{\frac{3}{2} k m \pm \sqrt{\frac{81}{4} k^2 m^2 - 9m^2 k^2}}{\frac{3}{2} \frac{9m^2}{4}}$$

$$\frac{3}{2} \frac{9m^2}{4}$$

$$\omega^2 = \frac{\frac{3km}{2} \pm \frac{km\sqrt{5}}{2}}{\frac{3m^2}{2}} = \frac{k}{m} \pm \frac{\sqrt{5}k}{3m} = 0.255 \frac{k}{m}, 1.745 \frac{k}{m}$$

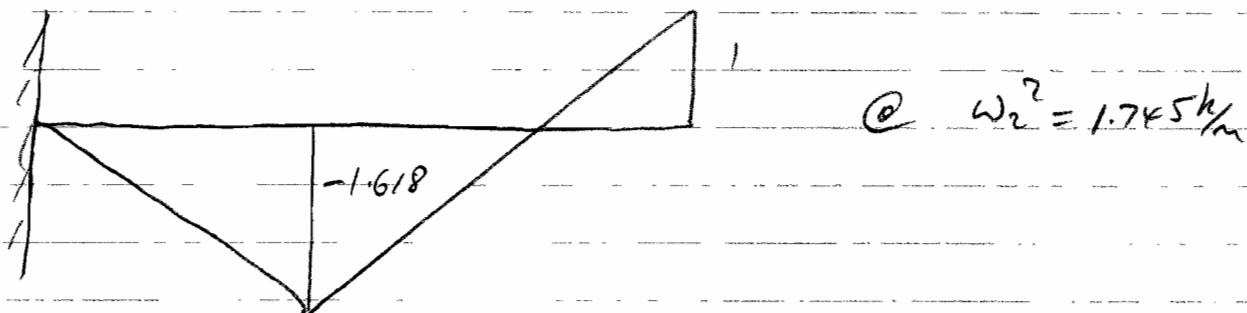
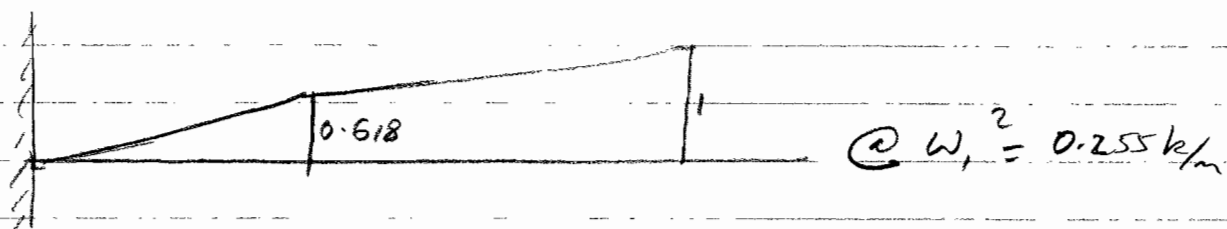
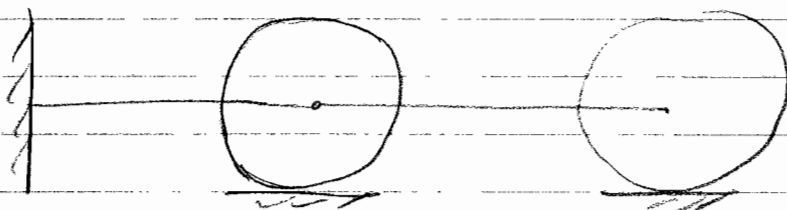
### 3. (Cont) Mode shapes

$$(2k - \omega^2 \frac{3m}{2}) y_1 - k y_2 = 0$$

$$\Rightarrow \frac{y_1}{y_2} = \frac{k}{2k - \omega^2 \frac{3m}{2}}$$

$$\textcircled{a} \omega_1^2 = 0.255 \text{ k/m}, \quad y_1/y_2^{(1)} = \frac{1}{2 - \frac{3}{2}(0.255)} = 0.618$$

$$\textcircled{c} \omega_2^2 = 1.745 \text{ k/m}, \quad y_1/y_2^{(2)} = \frac{1}{2 - \frac{3}{2}(1.745)} = -1.618$$



### (c) Transient response

$$\begin{Bmatrix} y_1 \\ y_2 \end{Bmatrix} = \begin{Bmatrix} 0.618 \\ 1 \end{Bmatrix} (A \cos \omega_1 t + B \sin \omega_1 t) + \begin{Bmatrix} -1.618 \\ 1 \end{Bmatrix} (C \cos \omega_2 t + D \sin \omega_2 t)$$

Initial Conds  $\begin{Bmatrix} y_1 \\ y_2 \end{Bmatrix} = \begin{Bmatrix} R \pi/4 \\ 0 \end{Bmatrix} \quad @ t=0.$

Initial displacement  $\Rightarrow$  sin terms are zero  $B=D=0.$

3. (cont)

$$\begin{Bmatrix} R\pi/4 \\ 0 \end{Bmatrix} = \begin{Bmatrix} 0.618 \\ 1 \end{Bmatrix} A + \begin{Bmatrix} -1.618 \\ 1 \end{Bmatrix} C$$

$$\Rightarrow R\pi/4 = 0.618A - 1.618C$$

$$0 = A + C \Rightarrow A = -C$$

$$\text{So } R\pi/4 = 0.618A + 1.618A \Rightarrow A = 0.351R$$

$$C = -0.351R$$

$$\text{So } y_1 = 0.217R \cos \omega_1 t + 0.568R \cos \omega_2 t$$

$$\& y_2 = 0.351R \cos \omega_1 t - 0.351R \cos \omega_2 t$$

check  $t=0$ ,  $y_1 = 0.785R = \pi R/4 \checkmark$   
 $y_2 = 0 \checkmark$

$$\text{At } t = \sqrt{\frac{m}{k}}, \quad y_1 = 0.217R \cos \sqrt{0.255} + 0.568R \cos \sqrt{1.745}$$

$$\& \theta_1 = y_1/R = \underline{0.33} \text{ radians } (18.9^\circ)$$

(d)  $k_2$  is increased by 20%

Calculate new  $\omega$ 's using Rayleigh's quotient with old mode shapes.

$$\text{Rayleigh: } \omega^2 = \frac{V_{\max}}{T^*} = \frac{\frac{1}{2} k y_1^2 + \frac{1}{2} (1.2k)(y_2 - y_1)^2}{\frac{1}{2} 3/2 m (y_1^2 + y_2^2)}$$

$$= \frac{2}{3} k/m \left[ \frac{y_1^2 + 1.2(y_2 - y_1)^2}{y_1^2 + y_2^2} \right]$$

$$\begin{pmatrix} y_1^{(0)} \\ y_2^{(0)} \end{pmatrix} = \begin{pmatrix} 0.618 \\ 1 \end{pmatrix} \Rightarrow \omega_1^2 \approx \frac{2}{3} k/m \left( \frac{0.618^2 + 1.2(0.382)^2}{0.618^2 + 1^2} \right) = 0.269 \frac{k}{m}$$

(higher as expected)

$$3. \text{ Cont } \begin{cases} y_1(t) \\ y_2(t) \end{cases} = \begin{cases} -1.618 \\ 1 \end{cases} \Rightarrow \omega_2^2 \approx \frac{2}{3} \frac{k}{m} \left( \frac{1.618^2 + 1.2 (2.618)^2}{1.618^2 + 1^2} \right)$$

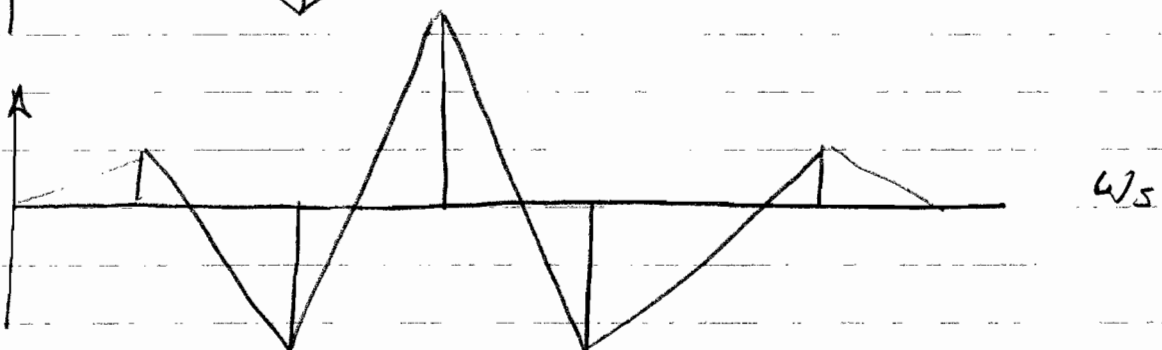
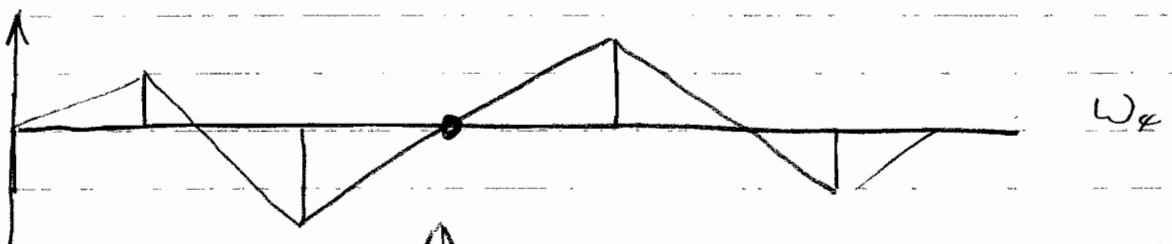
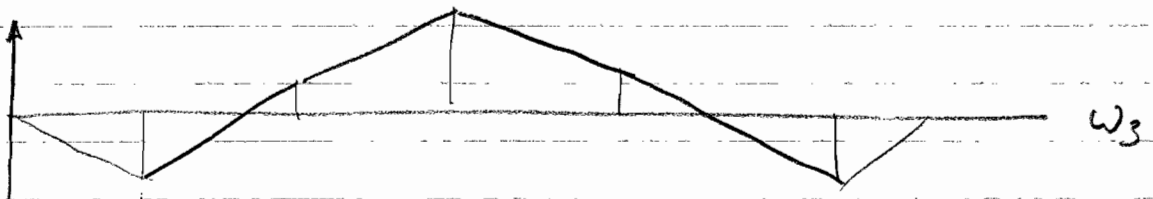
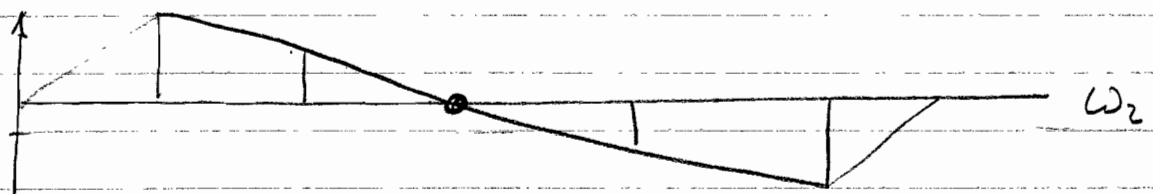
$$\omega_2^2 = \underline{\underline{2.0 \text{ k/m}}}$$

$$\text{At } t = \sqrt{\frac{m}{k}}, \quad \theta_1 \approx 0.217 \cos \sqrt{0.269} + 0.568 \cos \sqrt{2.0}$$
$$= \underline{\underline{0.277 \text{ rads}}}$$



$$(a) \quad T = \frac{1}{2} J (\dot{\theta}_1^2 + \dot{\theta}_2^2 + \dot{\theta}_3^2 + \dot{\theta}_4^2 + \dot{\theta}_5^2)$$

$$V = \frac{1}{2} k [(\theta_2 - \theta_1)^2 + (\theta_3 - \theta_2)^2 + (\theta_4 - \theta_3)^2 + (\theta_5 - \theta_4)^2]$$



Modes are either symmetric or antisymmetric

(c) Assume  $\underline{q} = [1 \quad 0.5 \quad 0 \quad -0.5 \quad -1]$

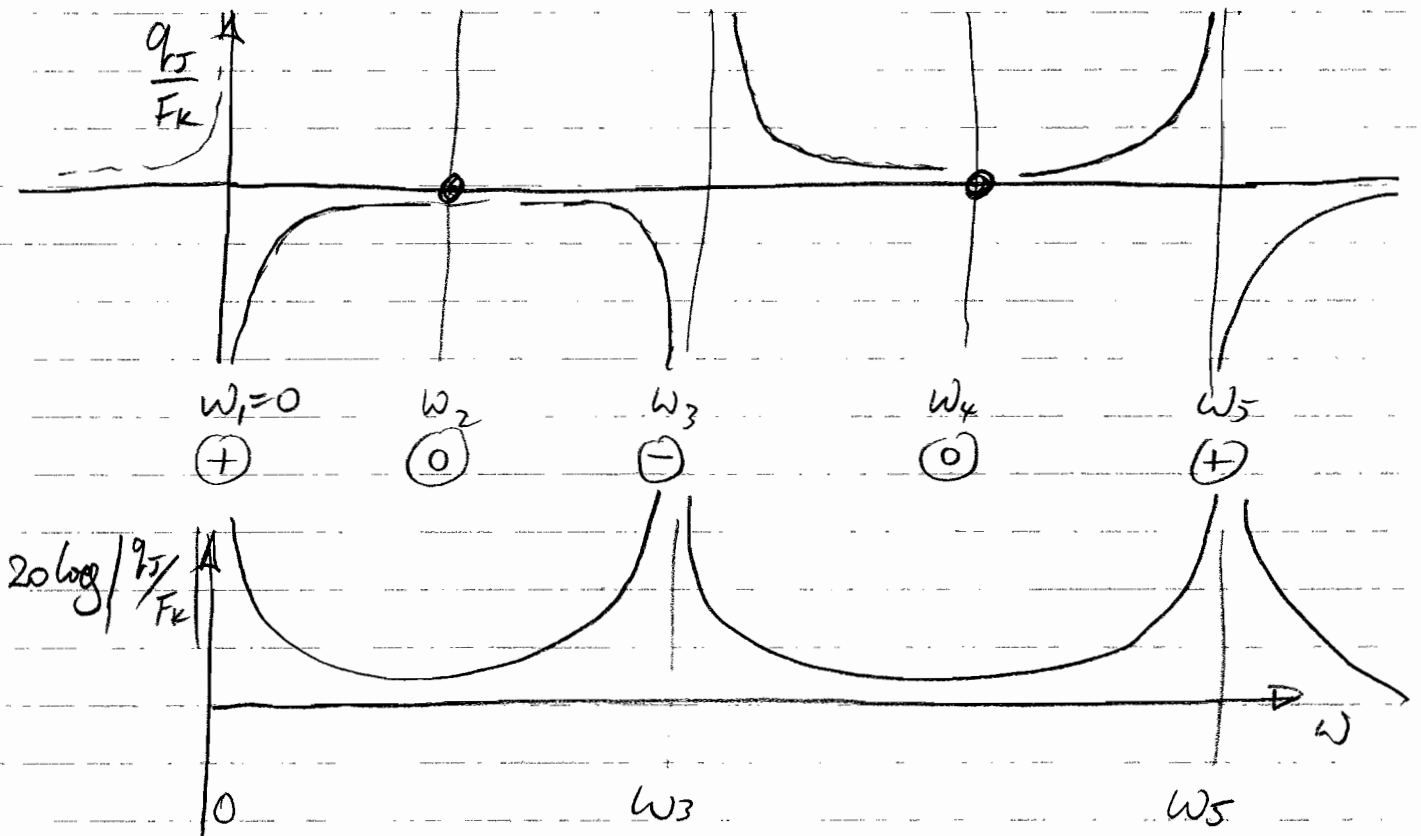
Rayleigh:  $\omega^2 \approx \frac{V_{max}}{T^*}$

$$\bar{\omega}^2 \approx \frac{\frac{1}{2} k [(0.5-1)^2 + (0-0.5)^2 + (-0.5-0)^2 + (-1+0.5)^2]}{\frac{1}{2} J [1^2 + 0.5^2 + 0^2 + (0.5)^2 + (-1)^2]}$$

$$= \frac{k}{J} \left( \frac{4(0.5)^2}{2(0.5^2)+2} \right) = \underline{\underline{0.4 k/J}} \quad \left( \begin{array}{l} 0.382 k/J \\ \text{exact} \end{array} \right)$$

(d)  $\frac{q_j}{F_k} = \sum_N \frac{u_j^{(n)} u_k^{(n)}}{\omega_n^2 - \omega^2}$  with  $J=3, k=1$

Signs of  $u_j^{(n)} u_k^{(n)}$  are:  $[+, 0, -, 0, +]$



(d) Flywheel has different moment of inertia, no damping, shaft is not massless, etc

## ENGINEERING TRIPOS PART IIB

Module 3C6 Examination, 2006

## Answers

$$1 \quad (\text{b}) \quad x=0, \theta=0; \quad x=L, d\theta/dx=0. \quad \omega_n = \sqrt{\frac{G}{\rho}} \left( n + \frac{1}{2} \right) \frac{\pi}{L}; \quad u_n(x) = \sin \left( n + \frac{1}{2} \right) \frac{\pi x}{L}$$

$$(\text{c}) \quad K \frac{\partial^2 \theta}{\partial t^2} = -GJ \frac{\partial \theta}{\partial x} \quad \text{at } x=L; \quad \tan kL = \frac{J\rho}{Kk}, \quad \text{with } k^2 = \frac{\rho\omega^2}{G}$$

$$2 \quad (\text{b}) \quad \omega_n^2 \approx \left( n + \frac{1}{2} \right)^4 \pi^4 \frac{EI}{mL^4}; \quad (\text{c}) \quad \omega^2 = 504 \frac{EI}{mL^4}$$

$$3 \quad (\text{a}) \quad T = \frac{1}{2} \left( \frac{3}{2} m \dot{y}_1^2 + \frac{3}{2} m \dot{y}_2^2 \right); \quad V = \frac{1}{2} k y_1^2 + \frac{1}{2} k (y_2 - y_1)^2; \quad \frac{3m}{2} \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix}; \quad k \begin{bmatrix} 2 & -1 \\ -1 & 1 \end{bmatrix}$$

$$(\text{b}) \quad \omega^2 = \frac{k}{m} \left( 1 \pm \frac{\sqrt{5}}{3} \right); \quad [0.618 \quad 1]^T; \quad [-1.618 \quad 1]^T$$

$$(\text{c}) \quad 0.33 \text{ rads}; \quad (\text{d}) \quad 0.28 \text{ rads}$$

$$4 \quad (\text{a}) \quad T = \frac{1}{2} J (\dot{\theta}_1^2 + \dot{\theta}_2^2 + \dot{\theta}_3^2 + \dot{\theta}_4^2 + \dot{\theta}_5^2)$$

$$V = \frac{1}{2} k [(\theta_2 - \theta_1)^2 + (\theta_3 - \theta_2)^2 + (\theta_4 - \theta_3)^2 + (\theta_5 - \theta_4)^2]$$

$$(\text{c}) \quad \omega_1^2 \approx 0.4 \frac{k}{J}$$