

3C7 Paper 2008

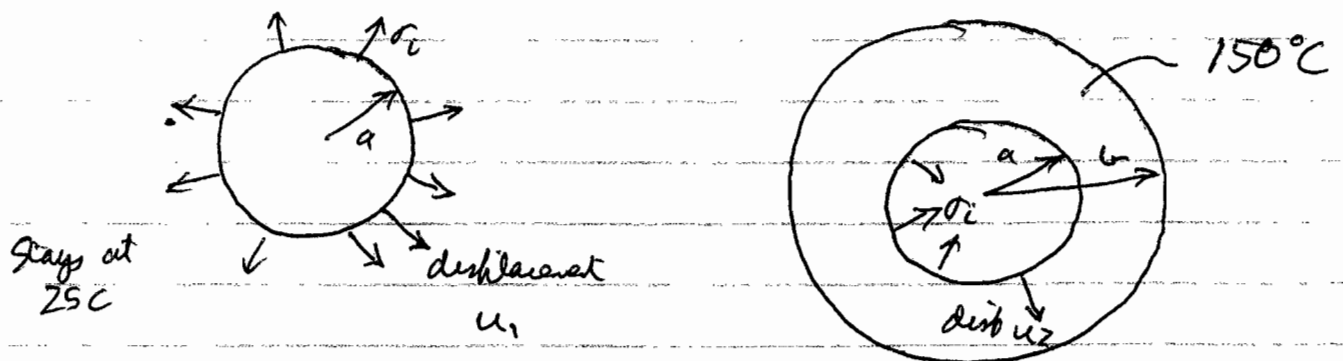
Solutions

1. (a) Plane stress occurs when there are no stresses on the surfaces of a thin sheet through the body (15%)
If x, y are in the plane then $\sigma_{zz} = 0$ τ_{xz} & $\tau_{yz} = 0$

(b) All dimensions increase with a strain of $\alpha (T - T_0)$ with no induced stresses (15%)

Surface displacement at the hole is $\alpha a (T - T_0)$

(c) Split into two:-



Basic method. Find displacements at interface in terms of σ_i and then use compatibility to find σ_i .

Inner disc $\sigma_{rr} = A - \frac{B}{r^2}$; $\sigma_{\theta\theta} = A + \frac{B}{r^2}$

$r \rightarrow 0$ stresses do not $\rightarrow \infty$ $\therefore B = 0$

$A = \sigma_i$
 $= \sigma_{rr} = \sigma_{\theta\theta}$

$$\epsilon_{\theta\theta} = \frac{\sigma_{\theta\theta} - \nu \sigma_{rr}}{E} = \frac{(1-\nu)\sigma_i}{E}$$

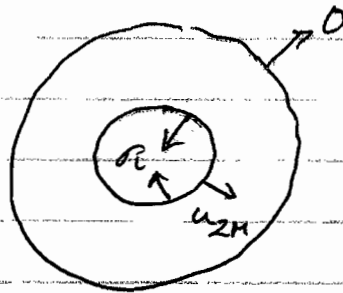
$$\epsilon_{\theta\theta} = \frac{u}{r} \quad \therefore \quad u_1 = \frac{a(1-\nu)\sigma_i}{E}$$

Outer annulus.

Consider effects of heating and interface stress separately, then add and check compatibility.

Thermal. $u_{2T} = a \alpha \Delta T$

Mechanical



$$\sigma_{rr} = A - \frac{B}{r^2}$$

$$\sigma_{\theta\theta} = A + \frac{B}{r^2}$$

$$\sigma_{rr} = 0 \text{ when } r = b$$

$$0 = A - \frac{B}{b^2}$$

$$-\sigma_i \quad r = a$$

$$\sigma_i = A - \frac{B}{a^2}$$

$$\therefore \sigma_i = \frac{B}{b^2} - \frac{B}{a^2} = \frac{(a^2 - b^2)B}{a^2 b^2}$$

$$\therefore B = \frac{a^2 b^2}{(a^2 - b^2)} \sigma_i$$

$$A = \frac{a^2 \sigma_i}{(a^2 - b^2)}$$

$$\frac{1}{E} \epsilon_{\theta\theta} = \sigma_{\theta\theta} - \nu \sigma_{rr} \quad \text{and} \quad \epsilon_{\theta\theta} = \frac{u}{r}$$

$$\therefore u_{2M} = \frac{a}{E} (\sigma_{\theta\theta} - \nu \sigma_{rr})$$

$$= \frac{a}{E} \sigma_i$$

$$(\sigma_{\theta\theta})_a = A + \frac{B}{a^2} = \frac{a^2 \sigma_i}{(a^2 - b^2)} + \frac{a^2 b^2}{a^2 (a^2 - b^2)} \sigma_i$$

$$= \frac{(a^2 + b^2)}{(a^2 - b^2)} \sigma_i$$

$$(\sigma_{rr})_a = A - \frac{B}{a^2} = \frac{(a^2 - b^2)}{(a^2 - b^2)} \sigma_i = \sigma_i \quad (\text{as expected})$$

$$\therefore u_{2M} = \frac{a}{E} \sigma_i \left(\frac{a^2 + b^2}{a^2 - b^2} - \nu \right)$$

$$\therefore u_1 = u_{2T} + u_{2M}$$

$$\frac{a}{E} (1 - \nu) \sigma_i = a \alpha \Delta T + \frac{a}{E} \sigma_i \left(\frac{a^2 + b^2}{a^2 - b^2} - \nu \right)$$

$$E \alpha \Delta T = \sigma_i \frac{(1 - \nu)(a^2 - b^2) - a^2 - b^2 + \nu a^2 + \nu b^2}{(a^2 - b^2)}$$

$$= \frac{\sigma_i}{(a^2 - b^2)} (\cancel{a^2} - \nu \cancel{a^2} + \nu \cancel{b^2} - b^2 - \cancel{a^2} - b^2 + \nu \cancel{a^2} + \nu \cancel{b^2})$$

$$= \frac{\sigma_i (-2b^2)}{(a^2 - b^2)} \Rightarrow \sigma_i = \frac{E \alpha \Delta T (b^2 - a^2)}{2b^2}$$

$$\text{or } \sigma_i = \frac{E \alpha \Delta T}{2} \left(1 - \frac{a^2}{b^2} \right)$$

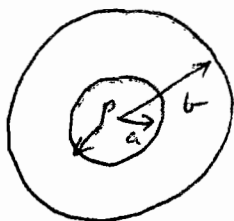
Put in numbers

$$\sigma_c = \frac{180 \cdot 10^3 \cdot 12 \cdot 10^{-6} \cdot 125}{2} \left(1 - \frac{(150)^2}{(250)^2} \right)$$
$$= \underline{\underline{86.4 \text{ MPa}}}$$

N.B. Poisson's ratio not needed but given in question since if any candidate makes an error they won't be able to get any answer without it.

2. Take external pressure as 0, internal pressure as p .

(a)



Elastic. Thick wall. Yield will occur on inner surface.

$$\sigma_{rr} = A - \frac{B}{r^2} \quad \sigma_{\theta\theta} = A + \frac{B}{r^2}$$

$$\sigma_{rr} = -p \text{ @ } r = a \quad ; \quad \sigma_{rr} = 0 \text{ @ } r = b \quad \Rightarrow A = \frac{B}{b^2}$$

$$-p = \frac{B}{b^2} + \frac{B}{a^2} \quad \Rightarrow \frac{B(a^2 + b^2)}{b^2 a^2} = -p$$

$$B = \frac{-p b^2 a^2}{(a^2 + b^2)} \quad A = \frac{-p a^2}{(a^2 + b^2)}$$

$$\Rightarrow \sigma_{rr} = -p \left[\frac{a^2}{(a^2 + b^2)} - \frac{b^2 a^2}{(a^2 + b^2) r^2} \right] = \frac{-p}{(1 - m^2)} \left[1 - \frac{b^2}{r^2} \right]$$

$$\sigma_{\theta\theta} = \frac{-p}{(1 - m^2)} \left(1 + \frac{b^2}{r^2} \right)$$

Yielding will occur first on inner surface and ~~that~~ it must involve σ_{rr} & $\sigma_{\theta\theta}$ since σ_{zz} will be the intermediate stress (bonus if they discuss why!)

$$|\sigma_{\theta\theta} - \sigma_{rr}| = 2k = \frac{p}{(m^2 - 1)} \cdot \frac{2b^2}{r^2}$$

$$\Rightarrow p = \frac{m^2 - 1}{2m^2} \cdot 2k$$

(b) Plastik everywhere.

$$\therefore |\sigma_{\theta\theta} - \sigma_{rr}| = 2K \quad \text{throughout.}$$

Equilibrium equation (data sheet) applies everywhere - not function of elastic/plastic.

$$\frac{\partial}{\partial r} (r \sigma_{rr}) + \frac{\partial \sigma_{r\theta}}{\partial \theta} - \sigma_{\theta\theta} = 0$$

\downarrow
0 everywhere

$$\Rightarrow r \cdot \frac{d\sigma_{rr}}{dr} + \sigma_{rr} - \sigma_{\theta\theta} = 0$$

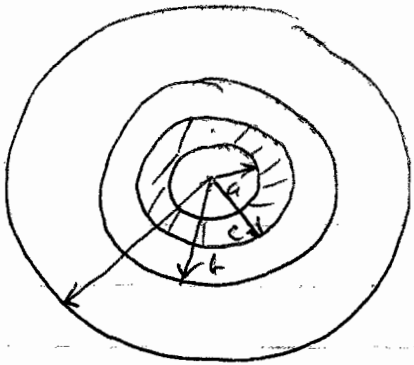
$$\therefore r \cdot \frac{d\sigma_{rr}}{dr} = 2K$$

$$\Rightarrow \int_{-b}^0 d\sigma_{rr} = \int_a^b \frac{2K}{r}$$

$$\Rightarrow [\sigma_{rr}]_{-b}^0 = 2K \ln \frac{b}{a}$$

$$\Rightarrow \frac{b}{2K} = \ln m$$

(c) Need to consider three tubes



Inner plastic from a to c

Second elastic from c to b

Outer elastic from b to R .

Intermediate pressures. p on inside

$$a = 100$$

$$b = 150$$

$$c = 135$$

R is unknown

p_1 at c

p_2 at b

0 at R

$$\text{For inner } \frac{p - p_1}{2k} = \ln\left(\frac{c}{a}\right) \quad \frac{240 - p_1}{420} = \ln\left(\frac{135}{100}\right)$$

$$\Rightarrow p_1 = 114 \text{ MPa}$$

Between c and b answer to (a) applies since yield just occurring at c

$$\therefore \frac{p_1 - p_2}{2k} = \frac{\left(\frac{b}{c}\right)^2 - 1}{2\left(\frac{b}{c}\right)^2} = \frac{(1.11)^2 - 1}{2(1.11)^2} \Rightarrow p_2 = 74 \text{ MPa}$$

From b out to D at point of yield at b .

$$\therefore \frac{p_2 - 0}{2k} = \frac{\left(\frac{R}{b}\right)^2 - 1}{2\left(\frac{R}{b}\right)^2} = \frac{74}{420} = 0.176$$

$$\Rightarrow \left(\frac{R}{b}\right)^2 = 1.54 \quad \Rightarrow \frac{R}{b} = 1.24 \quad \Rightarrow R = 186 \text{ mm}$$

$$\Rightarrow D = \underline{\underline{373 \text{ mm.}}}$$

$$3(a) \quad \phi = \frac{T_0}{4} \left\{ xy - \frac{xy^2}{h} - \frac{xy^3}{h^2} + \frac{Ly^2}{h} + \frac{Ly^3}{h^2} \right\}$$

$$\underline{\underline{\sigma_{yy} = \frac{\partial^2 \phi}{\partial x^2} = 0}} \quad \text{everywhere (by inspection - no } x^2 \text{ terms)}$$

$$\frac{\partial \phi}{\partial y} = \frac{T_0}{4} \left\{ x - \frac{2xy}{h} - \frac{3xy^2}{h^2} + \frac{2Ly}{h} + \frac{3Ly^2}{h^2} \right\}$$

$$\underline{\underline{\sigma_{xx} = \frac{\partial^2 \phi}{\partial y^2} = \frac{T_0}{4} \left\{ -\frac{2x}{h} - \frac{6xy}{h^2} + \frac{2L}{h} + \frac{6Ly}{h^2} \right\}}}$$

$$\underline{\underline{\sigma_{xy} = -\frac{\partial^2 \phi}{\partial x \partial y} = -\frac{T_0}{4} \left\{ 1 - \frac{2y}{h} - \frac{3y^2}{h^2} \right\}}}$$

$$(b) \text{ at } y = h \quad \sigma_{yy} = 0 \quad \checkmark$$

$$\sigma_{xy} = -\frac{T_0}{4} \left\{ 1 - 2 - 3 \right\} = +T_0 \quad \checkmark$$

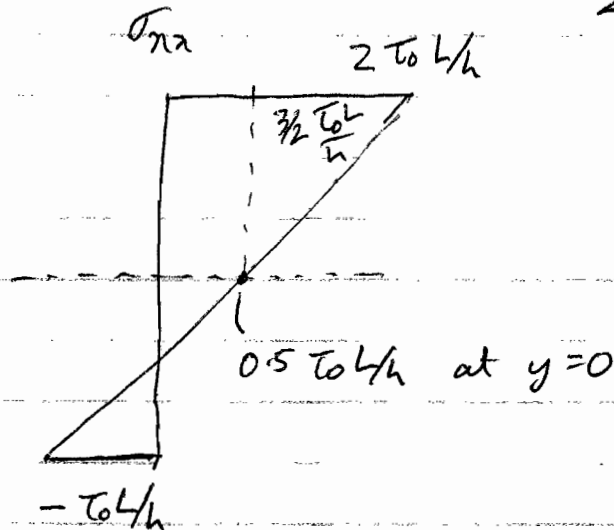
$$\text{at } y = -h \quad \sigma_{yy} = 0 \quad \checkmark$$

$$\sigma_{xy} = -\frac{T_0}{4} \left\{ 1 + 2 - 3 \right\} = 0 \quad \checkmark$$

(c) At $x=0$

$$\sigma_{xx} = \frac{T_0}{4} \left\{ \frac{2L}{h} + \frac{6Ly}{h^2} \right\} = \frac{T_0 L}{h} \left(\frac{1}{2} + \frac{3y}{2h} \right)$$

$$\sigma_{xy} = \frac{T_0}{4} \left\{ -1 + \frac{2y}{h} + \frac{3y^2}{h^2} \right\}$$



\therefore Horizontal Equilibrium should say

At root $b \left(0.5 \frac{T_0 L}{h} \right) \cancel{2L} = T_0 L b$ applied by T_0

which is satisfied

Moment equilibrium. Stress varies linearly

$$\therefore \text{Moment} = \frac{\sigma I}{y} = \frac{3 \cdot T_0 L}{2 h} \cdot b \cdot \frac{(2h)^3}{12 h}$$

$$= T_0 L b h$$

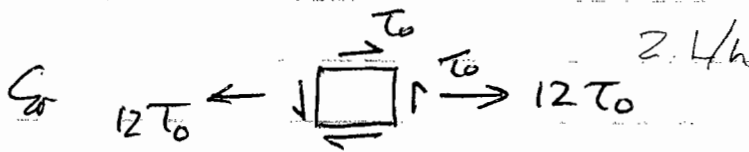
same.

Applied moment = $T_0 L b h$

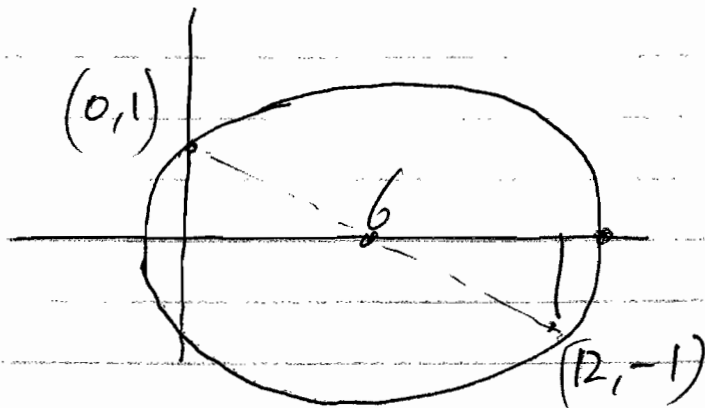
\therefore Moment equilibrium satisfied

(d) From (c) distribution of σ_{xx} is linear as this is the same as simple beam theory. ($\sigma_{xx} = 2\tau_0 L/h = 12\tau_0$ here)

But simple distribution of shear would give



Mohr's circle

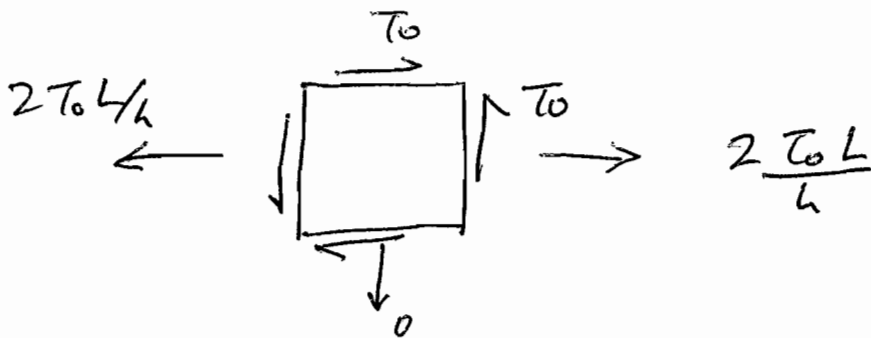


Radius = $\sqrt{37}$

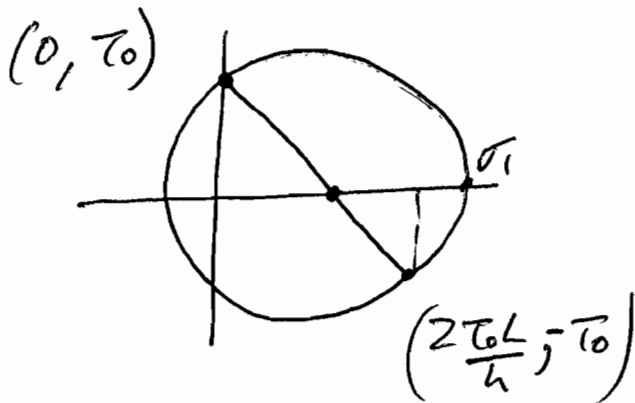
$\sigma_{max} = 6 + \sqrt{37} = 12.08 \tau_0$ } ($< 1\%$ difference)
 of $12 \tau_0$ }

(d)

The stress distribution at $x=0, y=h$ is



So Mohr's Circle



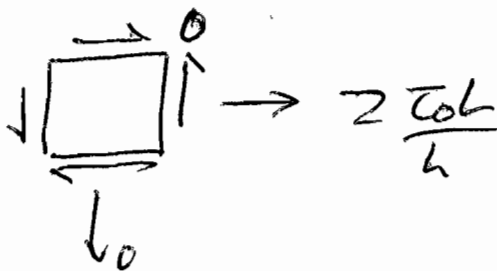
∴ Centre at $\frac{\tau_0 L}{h}$

$$\begin{aligned} \text{Radius} &= \sqrt{\tau_0^2 + \tau_0^2 \left(\frac{L}{h}\right)^2} \\ &= \tau_0 \sqrt{1 + \left(\frac{L}{h}\right)^2} \end{aligned}$$

∴ Principal tensile stress will be

$$\sigma_1 = \frac{\tau_0 L}{h} + \tau_0 \sqrt{1 + \left(\frac{L}{h}\right)^2} = \frac{\tau_0 L}{h} \left(1 + \left(1 + \left(\frac{L}{h}\right)^2\right)^{1/2}\right)$$

Simple beam theory gives

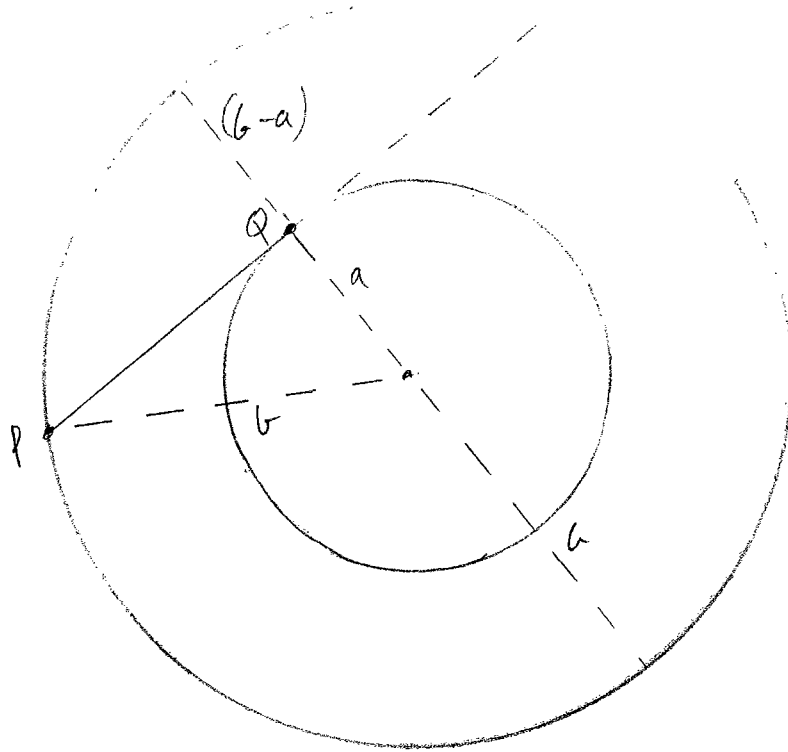


(No shear because no vertical load)

$$\sigma_{1s} = \frac{2\tau_0 L}{h}$$

Bonus if they note that discrepancy gets larger as $h/L \rightarrow 1$.

(a)



To find PQ

either

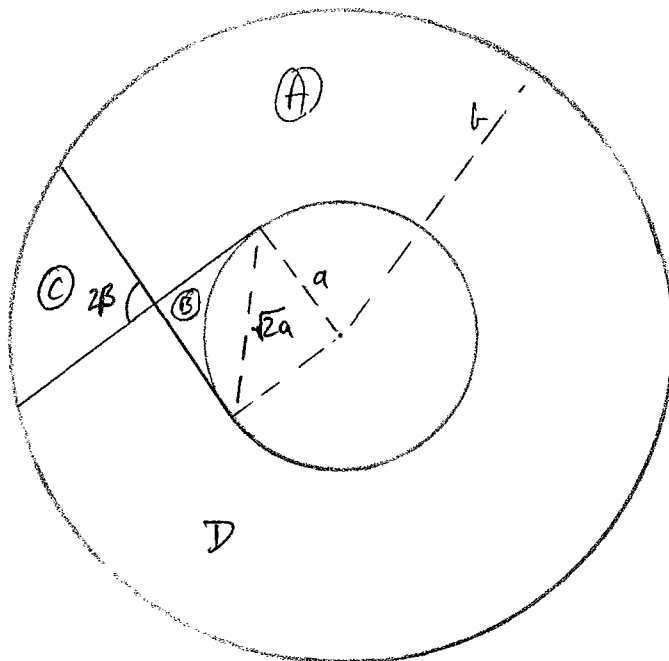
Bythagoras

$$(PQ)^2 + a^2 = b^2$$

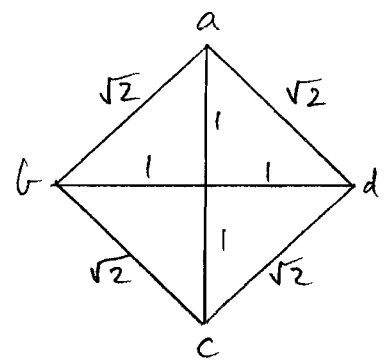
$$\Rightarrow PQ = \sqrt{b^2 - a^2} = \sqrt{(b-a)(b+a)}$$

or

Intersecting chords $PQ^2 = (b-a)(b+a)$



For $\beta = 45^\circ$



Upper bound

$$4(p \cdot \sqrt{2} a \cdot 1) = K (l_{ab} \cdot v_{ab} + l_{ac} \cdot v_{ac}) \times 2 \times 2$$

$$v_{ab} = v_{ac} = \sqrt{2} \quad (\text{from hodograph})$$

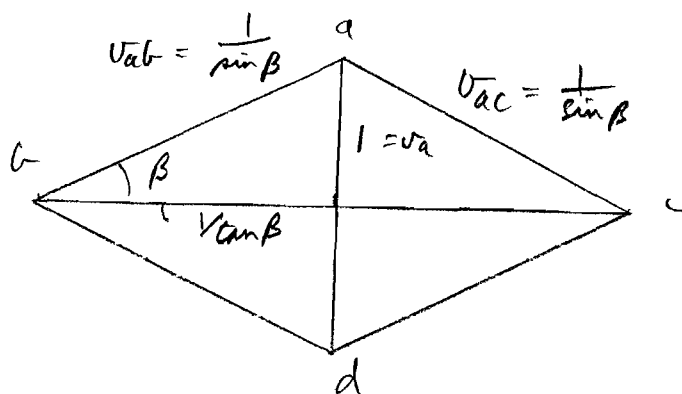
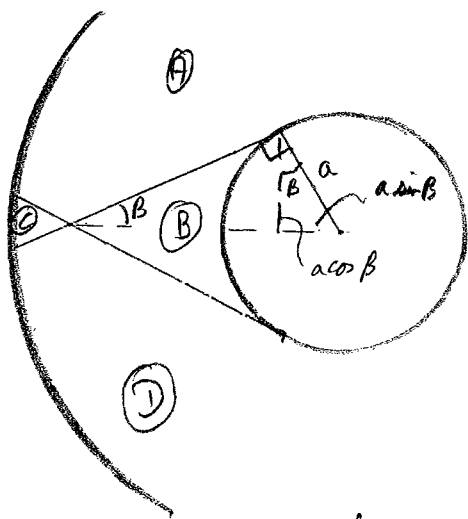
$$l_{ab} = a \quad l_{ac} + l_{ac} = \cancel{PQ} = \sqrt{b^2 - a^2}$$

$$\therefore \cancel{\sqrt{2}} p a = \cancel{K} \left(\cancel{a} \sqrt{2} + \sqrt{2} (\sqrt{b^2 - a^2} - \cancel{PQ}) \right)$$

$$p = K \left(\cancel{PQ} \frac{\sqrt{b^2 - a^2}}{a} \right) = \underline{\underline{K \left(\frac{b}{a} \right)^2 - 1}}$$

$$\text{If } b/a = 3$$

$$\underline{p = 2\sqrt{2} K.}$$

(6) If $\beta \neq 45^\circ$ 

$$l_{ab} = \frac{a}{\tan \beta}$$

Upper bound calculation

$$Z \left(p \cdot \cancel{Z} a \sin \beta \cdot 1 + \cancel{K} \cdot \cancel{Z} a \cos \beta \cdot \frac{1}{\tan \beta} \right)$$

$$= \cancel{K} \cdot K (l_{ab} \cdot v_{ab} + l_{ac} \cdot v_{ac})$$

But since $v_{ac} = v_{ab}$ then this = $\cancel{K} K v_{ab} (\underbrace{l_{ab} + l_{ac}}_{=PQ})$

$$\frac{\cos \beta}{\tan \beta} = \cancel{\cos \beta} \frac{\cos^2 \beta}{\sin \beta}$$

$$\therefore p a \left(\frac{\cancel{\sin^2 \beta} + \cos^2 \beta}{\cancel{\sin \beta}} \right) = K \frac{1}{\cancel{\sin \beta}} \sqrt{(b^2 - a^2)}$$

$$\underline{\underline{p = K \sqrt{\left(\frac{b}{a}\right)^2 - 1}}}$$

(which is the same as before - possibly slightly more concise).

