

## Part IIa Paper 3C6, 2010

(a) If spring is infinitely stiff, the base is fixed and each beam is constrained to have zero displacement and zero slope at its attachment to A-B. So follow the usual method for a cantilever.

$$EI \frac{d^4 u}{dx^4} = m\omega^2 u$$

$$\text{where } w(x,t) = u(x)e^{i\omega t}$$

$$\text{so } u = K_1 \cos \alpha x + K_2 \sin \alpha x + K_3 \cosh \alpha x + K_4 \sinh \alpha x$$

$$\text{with } \alpha^4 = m\omega^2/EI$$

Beam is clamped at  $x=0$ , free at  $x=L$

$$\text{so } \begin{cases} u(0) = 0 \\ u'(0) = 0 \end{cases} \rightarrow \begin{aligned} K_1 + K_3 &= 0 \\ K_2 + K_4 &= 0 \end{aligned}$$

$$u''(L) = 0 \rightarrow K_1(-\cos \alpha L - \cosh \alpha L) + K_2(\sin \alpha L - \sinh \alpha L) = 0$$

$$u'''(L) = 0 \rightarrow K_1(\sin \alpha L - \sinh \alpha L) + K_2(-\cos \alpha L - \cosh \alpha L) = 0$$

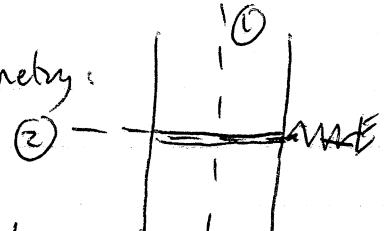
$$\text{So determinant } |\dots| = 0$$

$$\rightarrow (\cos \alpha L + \cosh \alpha L)^2 = -(\sin \alpha L - \sinh \alpha L)(\sin \alpha L + \sinh \alpha L)$$

$$\therefore \cos^2 \alpha L + 2 \cos \alpha L \cosh \alpha L + \cosh^2 \alpha L = \sinh^2 \alpha L - \sin^2 \alpha L$$

$$\therefore \cos \alpha L \cosh \alpha L = -1$$

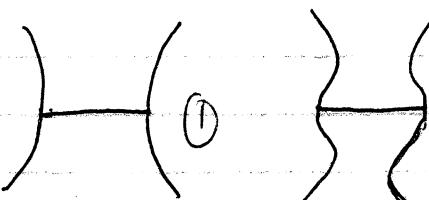
(b) The system has two planes of symmetry:



Each vibration mode must be either symmetric or antisymmetric with respect to each plane, so there are four groups of modes sym/sym, sym/anti, anti/sym, anti/anti. Sketches of the first few in each group:

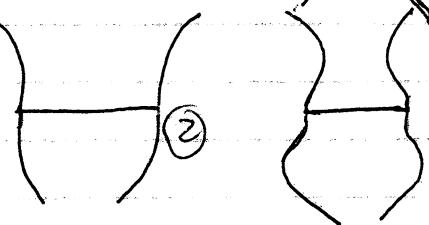
1b(Cont)

ss:



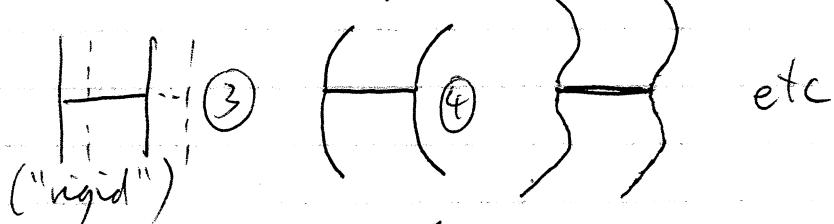
etc

sa:



etc

as:



etc

ad:



etc

1(c) With the spring removed, mode ③ has zero frequency. Modes ①, ② and ④ are all unchanged because they exert no net force on AB so it stays still. Mode ⑤ is the only one of this set which changes, together with the corresponding as mode of every higher group of 4.

To find this frequency, consider a single beam which is still free at  $x=L$ , but which has  $\begin{cases} u' = 0 & \text{(no rotation)} \\ u''' = 0 & \text{(no shear force)} \end{cases}$  at  $x=0$ .

So repeat calculation from (b):

$$u'(0) = 0 \rightarrow K_2 + K_4 = 0$$

$$u'''(0) = 0 \rightarrow -K_2 + K_4 = 0$$

$$\therefore K_2 = K_4 = 0$$

$$u''(L) = 0 \rightarrow -K_1 \cos \alpha L + K_3 \cosh \alpha L = 0$$

$$u''(L) = 0 \rightarrow K_1 \sin \alpha L + K_3 \sinh \alpha L = 0$$

$$\therefore \cos \alpha L \sinh \alpha L + \sin \alpha L \cosh \alpha L = 0$$

$$\therefore \tan \alpha L = -\tanh \alpha L$$

$$2 \text{ (a) Governing equation from data sheet } \rho \frac{\partial^2 w}{\partial t^2} - E \frac{\partial^2 w}{\partial x^2} = 0$$

Boundary conditions:  $\begin{cases} w = 0 \text{ at } x = 0 \\ \frac{\partial w}{\partial x} = 0 \text{ at } x = L \text{ (no strain)} \end{cases}$

So let  $w = u(x) e^{i\omega t}$  for modes:

$$\text{Then } u'' = -\frac{\omega^2}{c^2} u, \quad c^2 = \frac{E}{\rho}$$

$$\text{General solution } u = K_1 \sin \frac{\omega x}{c} + K_2 \cos \frac{\omega x}{c}$$

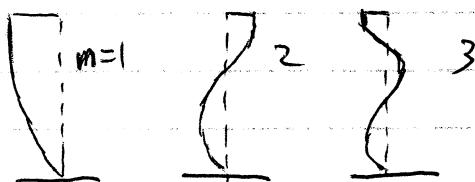
$$u(0) = 0 \rightarrow K_2 = 0$$

$$u'(L) = 0 \rightarrow K_1 \frac{\omega}{c} \cos \frac{\omega L}{c} = 0$$

$$\text{So natural frequencies where } \cos \frac{\omega L}{c} = 0, \text{ so } \frac{\omega L}{c} = (n-1/2)\pi \quad n=1, 2, 3, \dots$$

$$\text{Corresponding mode shape } u_n(x) = K_1 \sin \frac{(n-1/2)\pi x}{L}$$

Sketches:



(b) Static equilibrium under end force must have  $w = \lambda x$  for some value  $\lambda$ .

$$\text{Condition at } x=L \text{ is } F = EA \frac{\partial w}{\partial x} = \lambda EA, \therefore \lambda = \frac{F}{EA}$$

Now at  $t=0$  force at  $x=L$  jumps down from  $F$  to 0.

This is a negative step by  $-F$ , superposed on the static solution at  $t=0$ .

$$\therefore \text{Transient response } w(x, t) = \frac{Fx}{EA} - Fx \times (\text{step response})$$

Use Data Sheet result P3 #10, but first need to normalise modes.

2 Cont

$$\text{We require } \int_0^L K_1^2 \sin^2 \frac{w_n x}{c} \cdot p A dx = 1$$

$$\text{But } \int_0^L \sin^2 \frac{(n-k_2)\pi x}{L} dx = \frac{1}{2}, \text{ so need } K_1^2 = \frac{2}{pAL}$$

$$\begin{aligned} \text{So transient response is } w(x, t) &= \frac{F_0}{EA} - F \sum_{n=1}^{\infty} \frac{2}{pAL} \frac{u_n(x)u_n(L)}{w_n^2} (1 - \cos w_n t) \\ &= \frac{F_0}{EA} - F \sum_{n=1}^{\infty} \frac{\sin \frac{(n-k_2)\pi x}{L} \sin \frac{(n-k_2)\pi x}{L}}{c^2 ((n-k_2)\pi)^2} \left[ 1 - \cos \frac{(n-k_2)\pi ct}{L} \right] \\ &= \frac{F_0}{EA} \left\{ x - \frac{2L}{\pi^2} \sum_{n=1}^{\infty} \frac{(-1)^{n+1}}{(n-k_2)^2} \sin \frac{(n-k_2)\pi x}{L} \left[ 1 - \cos \frac{(n-k_2)\pi ct}{L} \right] \right\} \end{aligned} \quad (1)$$

But using the given Fourier series, the first two terms cancel, leaving

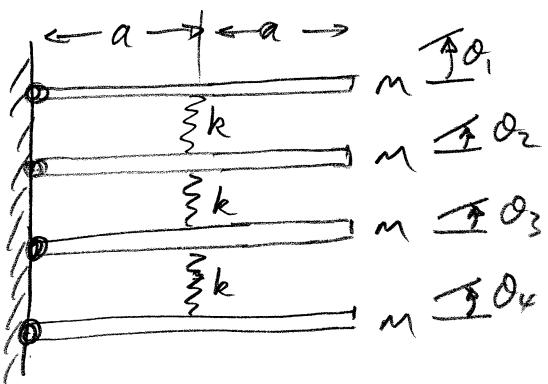
$$w(x, t) = \frac{2LF}{EA\pi^2} \sum_{n=1}^{\infty} \frac{\sin \frac{(n-k_2)\pi x}{L}}{L} \cos \frac{(n-k_2)\pi ct}{L} \frac{(-1)^{n+1}}{(n-\frac{1}{2})^2}$$

(c) This cancellation had to happen in order that when small damping is added we get  $w \rightarrow 0$  as  $t \rightarrow \infty$ .

Data Sheet formula says that the factor  $[1 - \cos w_n t]$  in (1) is replaced by  $[1 - \cos w_n t e^{-w_n S_n t}]$

The exponential factor  $\rightarrow 0$  as  $t \rightarrow \infty$ , leaving the two terms in (1) which cancelled.

3.



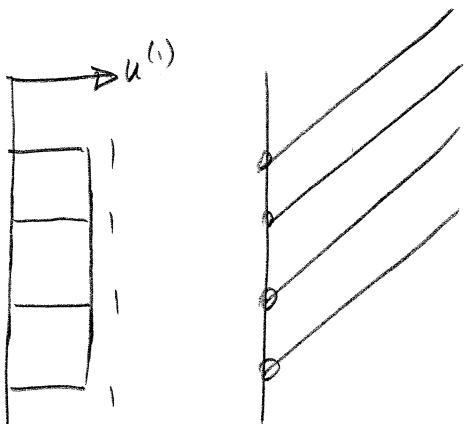
$$T_1 = \frac{1}{2} I_G \dot{\theta}_1^2 + \frac{1}{2} m(a\dot{\theta})^2$$

$$= \frac{1}{2} [ \frac{4}{3} ma^2 \dot{\theta}_1^2 ]$$

$$\Rightarrow T = \frac{1}{2} [ \frac{4}{3} ma^2 (\dot{\theta}_1^2 + \dot{\theta}_2^2 + \dot{\theta}_3^2 + \dot{\theta}_4^2) ]$$

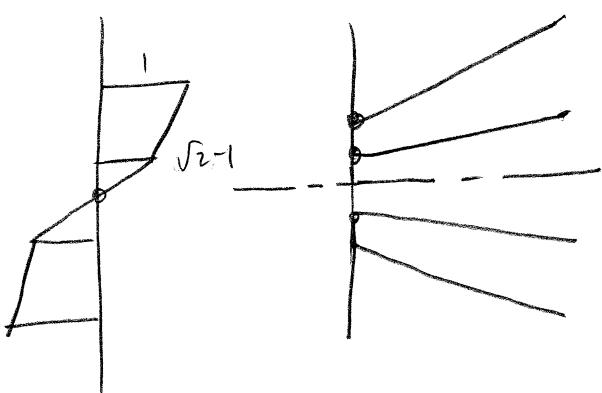
$$V = \frac{1}{2} k [ a\theta_1 - a\theta_2 ]^2 + \frac{1}{2} k [ a\theta_2 - a\theta_3 ]^2 + \frac{1}{2} k [ a\theta_3 - a\theta_4 ]^2$$

$$\therefore V = \frac{1}{2} ka^2 [ \theta_1^2 + 2\theta_2^2 + 2\theta_3^2 + \theta_4^2 - 2\theta_1\theta_2 - 2\theta_2\theta_3 - 2\theta_3\theta_4 ]$$



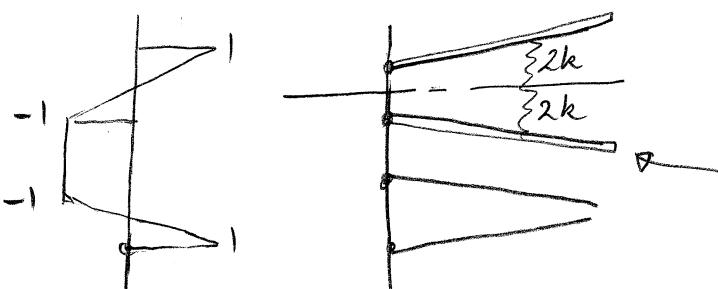
Rigid body mode  $\omega_1 = 0$ .

$$u^{(1)} = [1 \ 1 \ 1 \ 1]^T$$



$$u^{(2)} = [1 \ \alpha \ -\alpha \ -1]$$

$$\omega_2$$

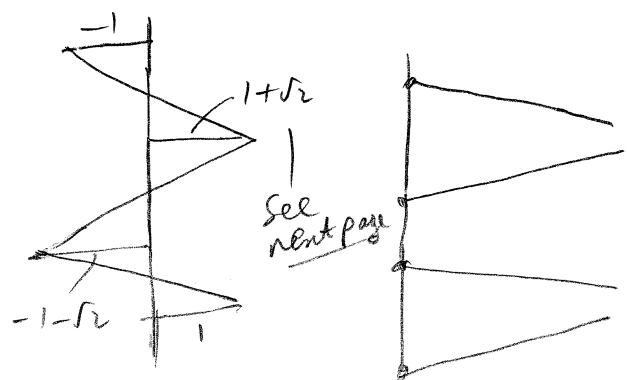


$$u^{(3)} = [1 \ -1 \ -1 \ 1]$$

$$\frac{4}{3} ma^2 \ddot{\theta} + 2k(a\theta) = 0$$

$$\ddot{\theta} + \frac{3}{2} \frac{k}{m} \theta = 0$$

$$\omega_3 = \sqrt{\frac{3k}{2m}}$$



$$u^{(4)} = [-1 \ \alpha \ -\alpha \ 1]$$

Same form as  $u^{(2)}$

3 (Cont)

$$\text{Rayleigh} \quad \omega^2 = \frac{V_{\max}}{T^2} \quad \text{with } u = [1 \ \alpha \ -\alpha \ -1]$$

$$\omega^2 = \frac{1}{2} k a^2 [(1-\alpha)^2 + (\alpha+\alpha)^2 + (-\alpha+1)^2]$$

$$= \frac{1}{2} k a^2 [2-4\alpha + 6\alpha^2]$$

$$T = \frac{1}{2} \frac{4}{3} m a^2 [1 + \alpha^2 + \alpha^2 + 1] = \frac{1}{2} \frac{4}{3} m a^2 [2 + 2\alpha^2]$$

$$\therefore \omega^2 = \frac{\cancel{k} k a^2 [2-4\alpha+6\alpha^2]}{\cancel{k} \frac{4}{3} m a^2 [2+2\alpha^2]} = \frac{\frac{3}{4} k}{m} \frac{[2-4\alpha+6\alpha^2]}{[2+2\alpha^2]}$$

To find  $\alpha$ :

$$\frac{d\omega^2}{d\alpha} = 0 : \frac{3k}{4m} \left[ \frac{(2+2\alpha^2)(-4+12\alpha) - (2-4\alpha+6\alpha^2)(4\alpha)}{(2+2\alpha^2)^2} \right] = 0$$

$$\Rightarrow -8 + 24\alpha - 8\alpha^2 + 24\alpha^3 - 8\alpha + 16\alpha^2 - 24\alpha^3 = 0$$

$$\Rightarrow 8\alpha^2 + 16\alpha - 8 = 0$$

$$\Rightarrow \alpha^2 + 2\alpha - 1 = 0 \Rightarrow \alpha = -1 \pm \sqrt{2}$$

$$\text{for } \alpha = -1 - \sqrt{2}, \quad \omega^2 = \frac{3k}{4m} \left[ \frac{2-4(-1-\sqrt{2})+6(-1-\sqrt{2})^2}{2+2(-1-\sqrt{2})^2} \right]$$

$$= -2.414$$

$$= \frac{3k}{4m} \left[ \frac{6+4\sqrt{2}}{2+\sqrt{2}} \right] = 2.561 \frac{k}{m}$$

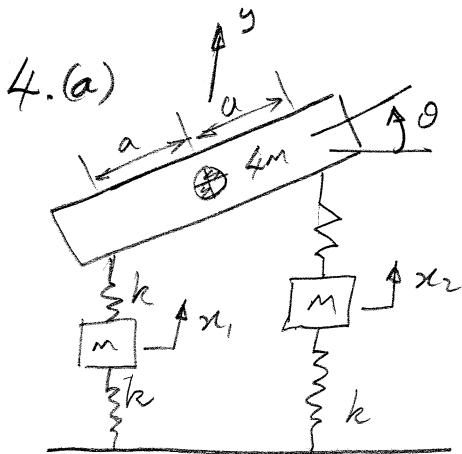
$$\Rightarrow \underline{\omega_4 = 1.60 \sqrt{\frac{k}{m}}}$$

$$\text{for } \alpha = -1 + \sqrt{2}, \quad \omega^2 = \frac{3k}{4m} \left[ \frac{2-4(-1+\sqrt{2})+6(-1+\sqrt{2})^2}{2+2(-1+\sqrt{2})^2} \right]$$

$$= 0.414$$

$$= \frac{3k}{4m} \left[ \frac{6-4\sqrt{2}}{2-\sqrt{2}} \right] = 0.4393 \frac{k}{m}$$

$$\underline{\omega_2 = 0.663 \sqrt{\frac{k}{m}}}$$



$$T = \frac{1}{2} m \dot{x}_1^2 + \frac{1}{2} m \dot{x}_2^2 + \frac{1}{2} 4m \dot{y}^2 + \frac{1}{2} m a^2 \dot{\theta}^2$$

$$[M] = m \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 4 & 0 \\ 0 & 0 & 0 & a^2 \end{bmatrix}$$

$$\begin{aligned} V &= \frac{1}{2} k x_1^2 + \frac{1}{2} k x_2^2 + \frac{1}{2} k (y - a\theta - x_1)^2 + \frac{1}{2} k (y + a\theta - x_2)^2 \\ &= \frac{1}{2} k (2x_1^2 + 2x_2^2 + 2y^2 + 2a^2\theta^2 - 2x_1 y - 2x_2 y + 2ax_1\theta - 2ax_2\theta) \end{aligned}$$

Hence by inspection of the quadratic form :

$$V = \frac{1}{2} [x_1 \ x_2 \ y \ \theta] k \begin{bmatrix} 2 & 0 & -1 & a \\ 0 & 2 & -1 & -a \\ -1 & -1 & 2 & 0 \\ a & -a & 0 & 2a^2 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ y \\ \theta \end{bmatrix}$$

(or do this by Lagrange)  $[k]$

(b) For the natural modes,  $([k] - \omega^2 [m]) u = 0$

For  $u = [1 \ 1 \ \alpha \ 0]^T$  this is

$$\begin{bmatrix} 2k - \omega^2 m & 0 & -k & ak \\ 0 & 2k - \omega^2 m & -k & -ak \\ -k & -k & 2k - 4\omega^2 m & 0 \\ ak & -ak & 0 & a^2(2k - \omega^2 m) \end{bmatrix} \begin{Bmatrix} 1 \\ 1 \\ \alpha \\ 0 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \\ 0 \\ 0 \end{Bmatrix} \quad (1)$$

First row of (1) is:  $2k - \omega^2 m - k\alpha = 0 \Rightarrow (2-\alpha)k = \omega^2 m \quad (2)$

Third row of (1) is:  $-2k + (2k - 4\omega^2 m)\alpha = 0 \Rightarrow (\alpha - 1)k = 2\alpha\omega^2 m \quad (3)$

$$\frac{(2-\alpha)}{(3)} \Rightarrow \frac{2-\alpha}{\alpha-1} = \frac{1}{2\alpha} \Rightarrow 2\alpha^2 - 3\alpha - 1 = 0$$

$$\alpha = \frac{3}{4} \pm \frac{\sqrt{17}}{4} = \frac{1.7808}{-0.2803} \quad \left. \right\} \quad (2) \Rightarrow \omega^2 = 0.219 \text{ k/m} \quad \underline{\underline{\text{& } \omega^2 = 2.281 \text{ k/m}}}$$

4(b) Similarly for  $\underline{u} = [1 \ -1 \ 0 \ \beta]^T$

1st row of (1):  $2k - \omega^2 m + ak\beta = 0 \Rightarrow (2 + a\beta)k = \omega^2 m \quad (4)$

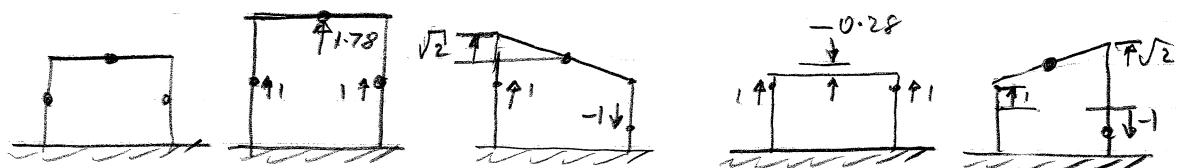
4th row of (1):  $2ak + a^2(2k - \omega^2 m)\beta = 0 \Rightarrow 2k(1 + a\beta) = \omega^2 m a\beta \quad (5)$

$$\frac{(4)}{(5)} \Rightarrow \frac{2 + a\beta}{2(1 + a\beta)} = \frac{1}{a\beta} \Rightarrow 2\beta + (a\beta)^2 = 2 + 2a\beta \Rightarrow a\beta = \pm \sqrt{2}$$

So (4) gives  $\omega^2 = (2 \pm \sqrt{2}) \frac{k}{m} = \frac{0.586 \text{ k/m}}{3.414 \text{ k/m}}$

Mode shapes:

(independent  
of the  
value of 'a')



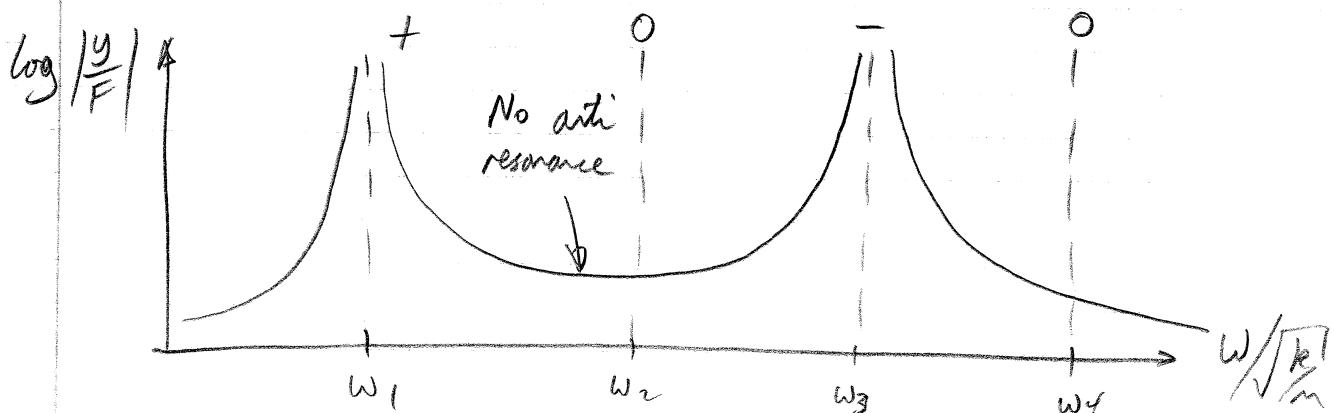
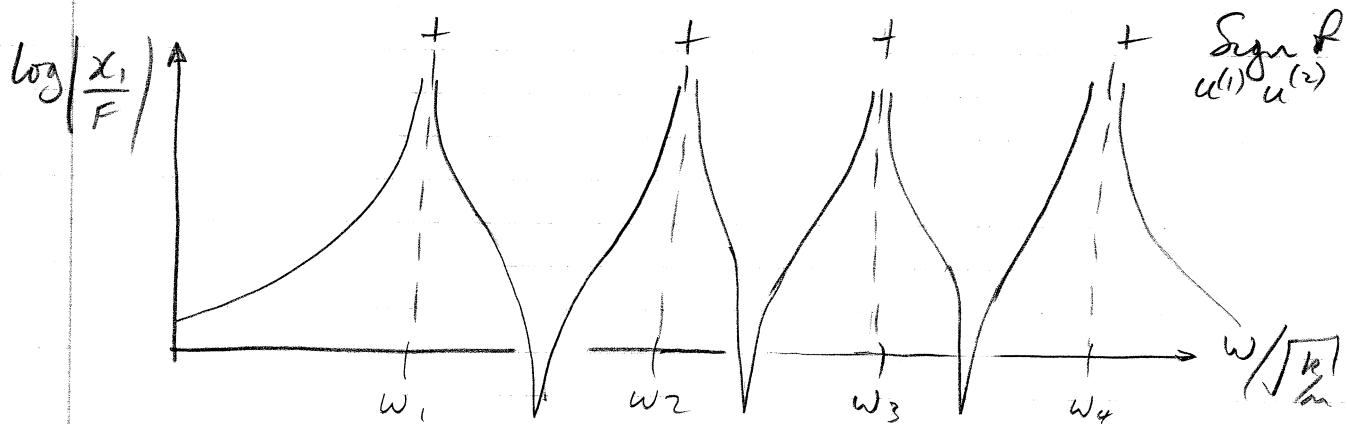
$$\alpha = 1.7808 \\ \omega^2 = 0.219 \frac{\text{k}}{\text{m}}$$

$$\alpha\beta = -\sqrt{2} \\ \omega^2 = 0.586 \frac{\text{k}}{\text{m}}$$

$$\alpha = -0.280 \\ \omega^2 = 2.281 \frac{\text{k}}{\text{m}}$$

$$\alpha\beta = \sqrt{2} \\ \omega^2 = 3.414 \frac{\text{k}}{\text{m}}$$

(c)  $\frac{x_j}{F_k} = \sum_{n=1}^N \frac{u_j^{(n)} u_k^{(n)}}{\omega_n^2 - \omega_n^2}$



## ENGINEERING TRIPPOS PART IIA

Module 3C6 Examination, 2010

## Answers

2 (a)  $u(0,t)=0, u'(L,t)=0$

(b)  $u(x,t) = \frac{2LF}{EA\pi^2} \sum_{n=1}^{\infty} \frac{(-1)^{n+1}}{(n-1/2)^2} \sin\left(\frac{(n-1/2)\pi x}{L}\right) \cos\left(\frac{(n-1/2)\pi ct}{L}\right).$

(c)  $u(x,t) = \frac{2LF}{EA\pi^2} \sum_{n=1}^{\infty} \frac{(-1)^{n+1}}{(n-1/2)^2} \sin\left(\frac{(n-1/2)\pi x}{L}\right) \cos\left(\frac{(n-1/2)\pi ct}{L}\right) e^{-\zeta_n \omega_n t}$

3 (a)  $T = \frac{2}{3}ma^2(\dot{\theta}_1^2 + \dot{\theta}_2^2 + \dot{\theta}_3^2 + \dot{\theta}_4^2),$

$V = \frac{1}{2}ka^2[(\theta_1 - \theta_2)^2 + (\theta_2 - \theta_3)^2 + (\theta_3 - \theta_4)^2]$

(c)  $[1 \ 1 \ 1 \ 1]^T, \quad \omega_1 = 0, \quad \omega_3 = \sqrt{\frac{3k}{2m}}$

(d)  $\alpha = -1 \pm \sqrt{2}, \quad \omega_2 = 0.663\sqrt{\frac{k}{m}}, \quad \omega_4 = 1.60\sqrt{\frac{k}{m}}, \quad \text{all frequencies are exact.}$

4 (a)  $T = \frac{1}{2}m\dot{x}_1^2 + \frac{1}{2}m\dot{x}_2^2 + 2m\dot{y}^2 + \frac{1}{2}ma^2\dot{\theta}^2, \quad [M] = m \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 4 & 0 \\ 0 & 0 & 0 & a^2 \end{bmatrix}$

$V = \frac{1}{2}k[x_1^2 + x_2^2 + (y - a\theta - x_1)^2 + (y + a\theta - x_2)^2]$

(b)  $\alpha = \frac{3}{4} \pm \frac{\sqrt{17}}{4}, \quad \omega^2 = 0.219\frac{k}{m}, 2.281\frac{k}{m}; \quad \beta = \pm\frac{\sqrt{2}}{a}, \quad \omega^2 = (2 \pm \sqrt{2})\frac{k}{m}$

Final Version