

Consider half.

$$\text{Area} = \pi R t$$



$$x' = R \cos \theta \\ y' = R \sin \theta$$

$$I_{xx'} = \int_0^{\pi} t y'^2 R d\theta = \int_0^{\pi} t R^3 \sin^2 \theta d\theta = \underline{\underline{\pi R^3 t / 2}}.$$

$$I_{yy'} = \int_0^{\pi} t x'^2 R d\theta = \int_0^{\pi} t R^3 \cos^2 \theta d\theta = \pi R^3 t / 2 \quad (\text{again } \checkmark)$$

$$I_{xy'} = 0 \quad \text{by symmetry.}$$

Find centroid:

$$x'_c = 0 \\ (\pi R t) y'_c = \int_0^{\pi} t y' R d\theta = \int_0^{\pi} t \sin \theta R^2 d\theta \\ = t R^2 \left[-\cos \theta \right]_0^{\pi} = 2R^2 t \\ \rightarrow y'_c = \frac{2R^2 t}{\pi R t} = \underline{\underline{\frac{2}{\pi} R}}$$

3

Total cross-section:



$$I_{xx} = 2 I_{x'x'} = \underline{\underline{\pi R^3 t}}$$

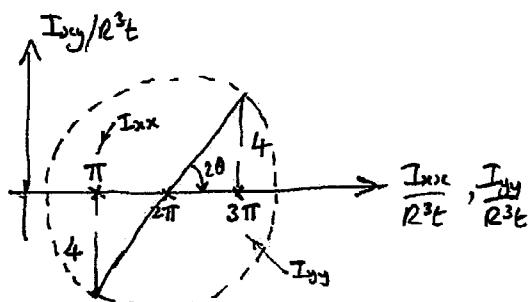
$$I_{yy} = 2 I_{y'y'} + 2 A x_c^2 \\ \downarrow R^2 = \pi R^3 t + 2(\pi R t) R^2 \\ = \underline{\underline{3\pi R^3 t}}$$

$$I_{xy} = 2 I_{xy'} + 2 A (x_c)(y_c)$$

$$= 0 + 2(\pi R t)(-R) \left(\frac{2}{\pi} R \right) = \underline{\underline{-4R^3 t}}$$

2

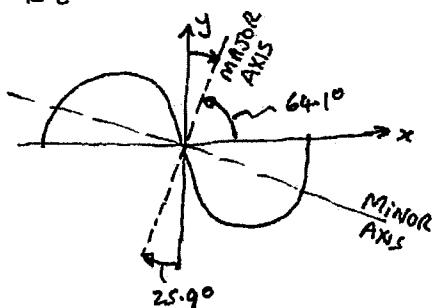
Mohr's circle



$$\tan 2\theta = \frac{4}{\pi}$$

$$2\theta = \tan^{-1} \left(\frac{4}{\pi} \right)$$

$$\rightarrow \underline{\underline{\theta = 25.9^\circ}}$$



Angle between Major Axis and x-axis

$$> 90 - 25.9 = 64.1^\circ$$

Q1 cont'd. a) Principal 2nd moments of area:

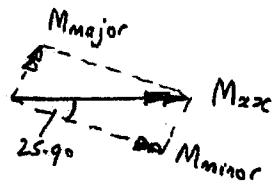
From Mohr's Circle Geometry $\lambda_{1,2} = \frac{2\pi}{R^3 t} \pm \sqrt{\frac{4^2 + \pi^2}{R^3 t}}$ = { 11.369 Major
1.197 Minor
centre of Mohr's Circ. radius of Mohr's Circ.

(Alternatively, find eigenvals of $I = R^3 t \begin{bmatrix} \pi & 4 \\ 4 & 3\pi \end{bmatrix}$)

Q1 b) Centroid is at $x=0, y=0$ by symmetry
and Shear centre is at $x=0, y=0$ by symmetry.

Q1 c) BM (M_{xx}) = $\int y dm = (\omega L) \frac{L}{2}$
 $= \rho g (2\pi R t) \frac{L^2}{2} = \underline{\underline{\pi \rho g R t L^2}}$

Resolve into Princ. Directions.

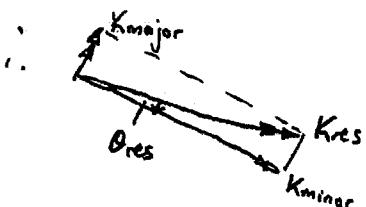


$$M_{\text{major}} = M_{xx} \sin 25.9^\circ = 0.4368 M_{xx}$$

$$M_{\text{minor}} = M_{xx} \cos 25.9^\circ = 0.8996 M_{xx}$$

$$K_{\text{major}} = \frac{M_{\text{major}}}{EI_{\text{major}}} = \frac{1}{E} \frac{M_{xx}}{R^3 t} \left(\frac{0.4368}{11.369} \right) = 0.0384 \frac{M_{xx}}{ER^3 t}$$

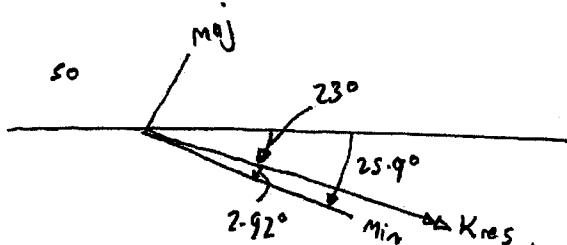
$$K_{\text{minor}} = \frac{M_{\text{minor}}}{EI_{\text{minor}}} = \frac{1}{E} \frac{M_{xx}}{R^3 t} \left(\frac{0.8996}{1.197} \right) = 0.7515 \frac{M_{xx}}{ER^3 t}$$



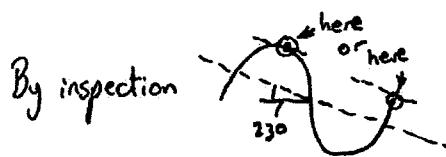
$$|K_{\text{res}}| = M_{xx} \sqrt{0.0384^2 + 0.7515^2} = 0.7525 \frac{M_{xx}}{ER^3 t}$$

$$\tan \theta_{\text{res}} = \frac{K_{\text{maj}}}{K_{\text{min}}} = \frac{0.0384}{0.7515} = 0.0511$$

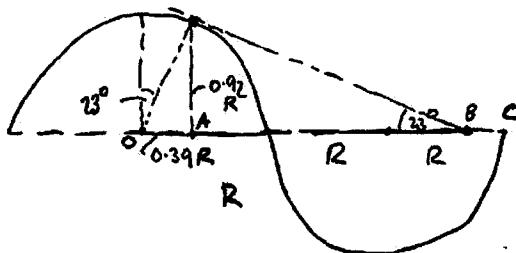
$$\Rightarrow \theta_{\text{res}} = 2.92^\circ$$



Q1 c) cont'd.

Need to find point on section furthest from axis of curvature K_{res} .

Geom:

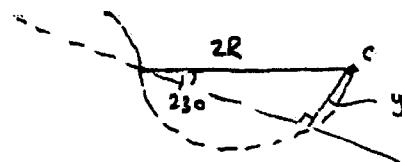


$$AB = \frac{0.92}{\tan 23^\circ} = 2.167$$

$$OB = 0.39 + 2.167 = 2.55R$$

but

$$OC = 3R$$

 $\therefore B$ to left of C $\therefore C$ is worst caseNeed distance of C from K_{res} axis:

$$\frac{y}{2R} = \sin 23^\circ$$

$$\rightarrow y = 2R \sin 23^\circ$$

$$= 0.7815R$$

$$\text{Now } K_{res} = 0.7525 \frac{M_{xx}}{ER^3t} = 0.7525 \left(\frac{\pi \rho g L^2 R t}{ER^3 t} \right)$$

$$\begin{aligned} \text{so } \frac{G}{atc} &= E K_{res} y = E \left(\frac{0.7525 \pi \rho g L^2 R t}{ER^3 t} \right) 0.7815 R \\ &= \pi (0.7525)(0.7815) \frac{\rho g L^2}{R} \end{aligned}$$

$$\begin{aligned} \text{so } \underline{\underline{\frac{G}{atc}}} &= 1.8474 \frac{\rho g L^2}{R} = 18.1 \frac{\rho L^2}{R} \end{aligned}$$

Q2.



Kinematical Indeterminacy = 1

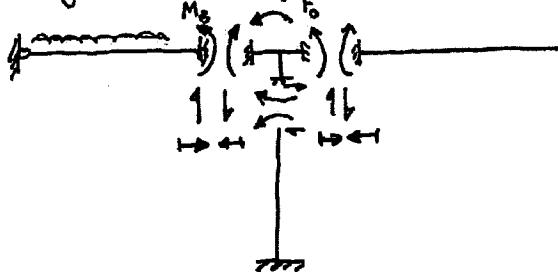


Statical Indeterminacy = 4



(4)

b) Kinematically determinate system



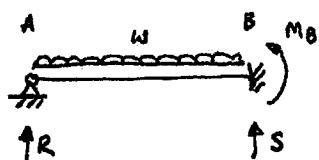
$EA = \infty$

$EI = 5 \times 10^4 \text{ kNm}^2$

$w = 10 \text{ kN/m}$

$M_B = ?$

Real system

Moments about A \rightarrow

$M_B + SL = (wL)\frac{L}{2}$

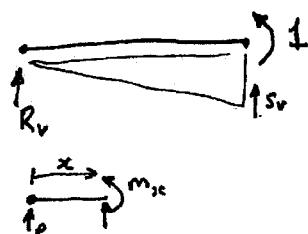
$S = \frac{wL}{2} - \frac{M_B}{L}$

Vertical force

$R + S = wL$

$\Rightarrow R = \frac{wL}{2} + \frac{M_B}{L}$

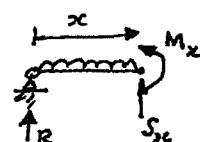
Virtual system



Top: Moments about B: $R_v L = \frac{1}{2} \Rightarrow R_v = \frac{1}{L}$

Bottom: Moments about RH $M_{xc} = R_v x_c$

$\therefore M_{xc} = \frac{x_c}{L}$

Moments about RH end: \rightarrow

$M_x + wx\left(\frac{x}{2}\right) = Rx$
 $= \left(\frac{wL}{2} + \frac{M_B}{L}\right)x$

$M_{xc} = \left(\frac{wL}{2} + \frac{M_B}{L}\right)x \Rightarrow -\frac{wx^2}{2}$

Real. Virt: $\frac{\text{EXT}}{\text{INT}} = \int_0^L \frac{M_x M_{xc}}{EI} dx$

Real

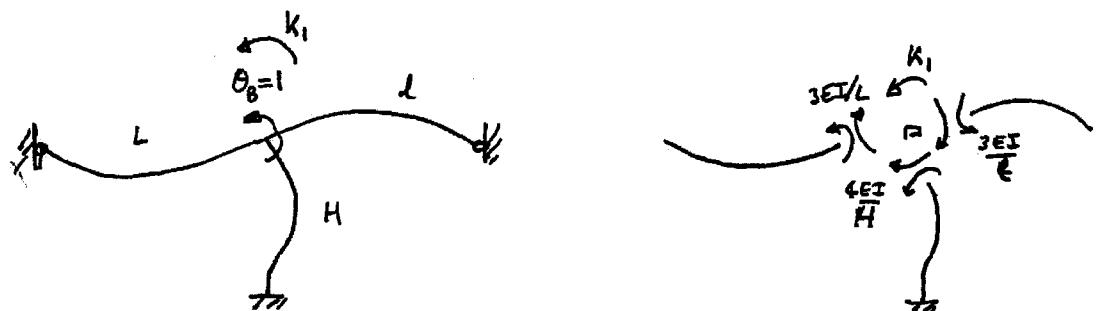
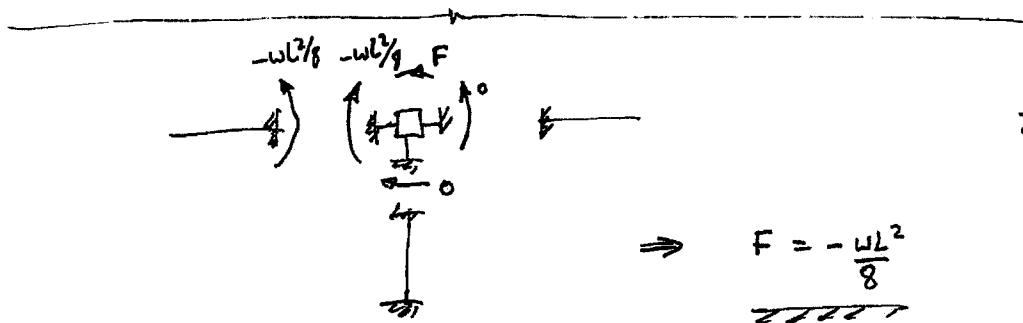
$1.0 = \int_0^L \left(\frac{wL}{2} + \frac{M_B}{L}\right) \left[\left(\frac{wL}{2} + \frac{M_B}{L}\right)x - \frac{wx^2}{2}\right] dx$

$\therefore 0 = \left(\frac{wL}{2} + \frac{M_B}{L}\right) \int_0^1 x^2 dx - \frac{w}{2L} \int_0^1 x^3 dx$

$= \left(\frac{wL}{2} + \frac{M_B}{L}\right) \frac{L^3}{3} - \frac{w}{2L} \frac{L^4}{4} \Rightarrow \frac{wL^3}{6} + \frac{M_B L}{3} = \frac{wL^3}{8}$

$$\Rightarrow M_B = \frac{3}{L} \left[\frac{wL^3}{8} - \frac{wL^3}{6} \right] = 3wL^2 \left[\frac{6-8}{48} \right] = -3wL^2 \frac{2}{48}$$

$$= -\frac{wL^2}{8}$$



$$K_1 = \frac{3EI}{L} + \frac{3EI}{\ell} + \frac{4EI}{H}$$

$$F + K_1 \theta_B = 0 \Rightarrow -\frac{wL^2}{8} + \left(\frac{3EI}{L} + \frac{3EI}{\ell} + \frac{4EI}{H} \right) \theta_B = 0$$

$$\Rightarrow \theta_B = \frac{\frac{wL^2}{8}}{\left(\frac{3EI}{L} + \frac{3EI}{\ell} + \frac{4EI}{H} \right)} = \frac{10(6)^2/8}{\left(\frac{3}{6} + \frac{3}{8} + \frac{4}{4} \right) EI} = \frac{45}{(1.875) EI}$$

$$= \frac{24}{EI} = \frac{24}{5 \times 10^4} = \underline{\underline{4.8 \times 10^{-4} \text{ rad}}}.$$

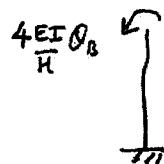
On BA:
$$= -\frac{wL^2}{8} + \frac{3EI}{L} \left(\frac{24}{EI} \right)$$

$$= -\frac{10(6)^2}{8} + \frac{72}{6} = -45 + 12 = \underline{\underline{-33 \text{ kNm}}}$$

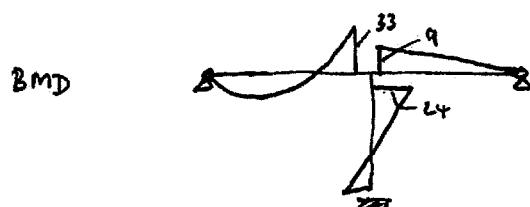
On BC :

$$\frac{3EI}{l} \theta_B = \frac{3EI}{8} \frac{24}{EI} = 9 \text{ kNm}$$

On BD:



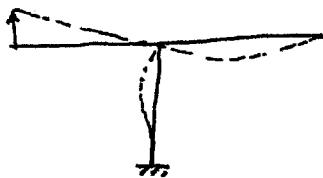
$$\frac{4EI}{H} \theta_B = \frac{4EI}{4} \frac{24}{EI} = 24 \text{ kNm}$$



(11)

Q2 c).

Vertical reaction at A



Vertical reaction at D.

(5)

Q3

(a) j th column = vector of nodal forces required to give unit displacement in j th component of nodal displacement vectors, and zero displacement along all other components.

2

(b) Marks awarded for clear explanation of eigenmodes/eigenvectors as static modes (e.g. force of one "shape" gives displacement of same "shape") and of eigenvalues as the generalized stiffnesses of these deflected shapes. (More general than concept of "buckling mode" which is only applicable at point of buckling). Useful in stability analysis as the eigenvalues (and the positivity thereof) measures how stable the structure is, and buckling corresponds to the first time an eigenvalue passes through zero.

4

c) i) By inspection, stiffness matrix

$$= \frac{EI}{L} \begin{bmatrix} s+7 & sc \\ sc & s+6 \end{bmatrix}$$

This ~~ggg~~ buckles when $\det = 0 \Rightarrow (s+6)(s+7) - s^2c^2 = 0$
 $s^2 + 13s + 42 - s^2c^2 = 0$
 $s^2(1-c^2) + 13s + 42 = 0.$

Interpolate:

P/P_E	s	c	$s^2(1-c^2)$	$13s+42$
2.0	0.1428	24.6841	-12.40	43.48
2.2	-0.5194	-7.5107	-14.9485	35.24
2.4	-1.3006	-3.3703	-17.5	25.092
2.6	-2.2490	-2.2312	-20.122	12.76

\leftarrow cross here.
 $\sim P/P_E = 2.5$

$$P/P_E \approx 2.5 \quad P_E = \frac{\pi^2 EI}{L^2} \quad \rightarrow \frac{L_E}{L} = \sqrt{\frac{P_E}{P}} = \sqrt{\frac{1}{2.5}} = 0.6325$$

6

Q3 cont'd.

$$\text{c ii) } \begin{bmatrix} M_B \\ M_E \end{bmatrix} = \frac{EI}{L} \begin{bmatrix} s+7 & sc \\ sc & s+6 \end{bmatrix} \begin{bmatrix} \theta_B \\ \theta_E \end{bmatrix}$$

$$\text{and } P = P_E, \text{ so } s = 2.4674, c = 1.0$$

$$\therefore \begin{bmatrix} M_B \\ M_E \end{bmatrix} = \frac{EI}{L} \begin{bmatrix} 9.4674 & 2.4674 \\ 2.4674 & 8.4674 \end{bmatrix} \begin{bmatrix} \theta_B \\ \theta_E \end{bmatrix}$$

$$\underline{M} = \begin{bmatrix} PL/100 \\ 0 \end{bmatrix} = \begin{bmatrix} \pi^2 EI L / 100 \\ 0 \end{bmatrix} = \frac{EI}{L} \begin{bmatrix} \pi^2 / 100 \\ 0 \end{bmatrix} \quad (\frac{EI}{L} \text{'s cancel})$$

$$\rightarrow \begin{bmatrix} \theta_B \\ \theta_E \end{bmatrix} = \begin{bmatrix} 9.4674 & 2.4674 \\ 2.4674 & 8.4674 \end{bmatrix}^{-1} \begin{bmatrix} \pi^2 / 100 \\ 0 \end{bmatrix}$$

$$= \frac{1}{(8.4674)(9.4674) - (2.4674)^2} \begin{bmatrix} 8.4674 & -2.4674 \\ -2.4674 & 9.4674 \end{bmatrix} \begin{bmatrix} \pi^2 / 100 \\ 0 \end{bmatrix}$$

denom = 74.076

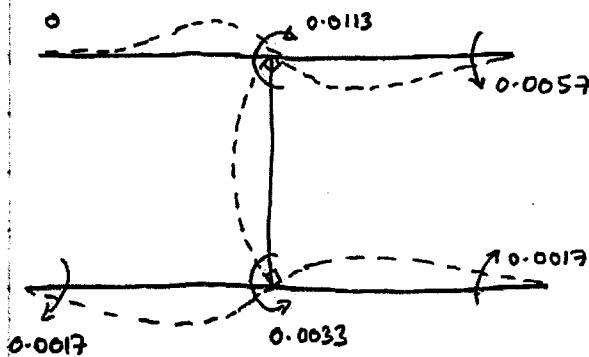
$$\rightarrow \theta_B = \frac{8.4674 \pi^2 / 100}{74.076} = \underline{0.0113 \text{ rads}}$$

$$\theta_E = \frac{-2.4674 \pi^2 / 100}{74.076} = \underline{-0.0033 \text{ rads}}$$

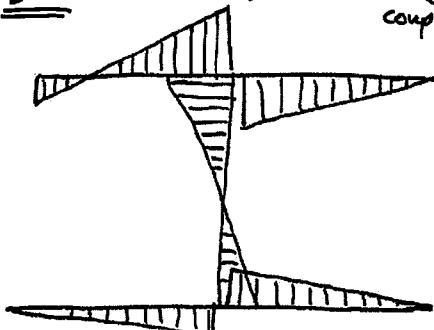
Also $|\theta_c| = |\theta_B|/2 = 0.0057 \text{ rads}$

$|\theta_D| = |\theta_F| = |\theta_E|/2 = 0.0017 \text{ rads.}$

Deflected shape:



BMD:



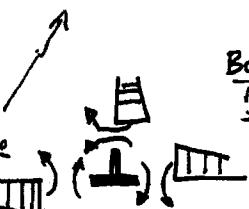
TOP NODE

All moments
on joint
anticlockwise,
resisting applied
couple

8

Moments on
joint self-equilibrate
(because no
applied couple there).

BOTTOM
NODE

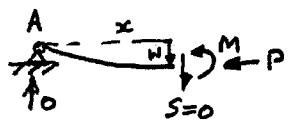


Q4. i)



$$M_A \Rightarrow S = 0$$

$$M_B \Rightarrow R = 0$$



$$\text{Equilib: } M_A - M + Pw = 0$$

$$\text{Const.t/Compat: } M = -EI \left(\frac{d^2w}{dx^2} - \frac{d^2w_0}{dx^2} \right)$$

$$\rightarrow EIw'' + Pw = EIw_0''$$

w_0 = limit. imperfection

$$\begin{aligned} \text{Let } w_0 = b \sin\left(\frac{\pi x}{L}\right) \\ \text{Initial half-sine wave bow} \\ \text{and } w = a \sin\left(\frac{\pi x}{L}\right) \\ \text{generally} \end{aligned} \quad \left. \begin{aligned} \left(-\frac{\pi^2 EI}{L^2} + P \right) a \sin\left(\frac{\pi x}{L}\right) &= \left(-\frac{\pi^2 EI}{L^2} \right) b \sin\left(\frac{\pi x}{L}\right) \\ \text{write } P_E = \pi^2 EI / L^2 & \\ \Rightarrow a &= b \frac{P_E}{P_E - P} = \frac{b}{1 - P/P_E} \end{aligned} \right.$$

$$\text{Moment at centre of beam } M_{\text{centre}} = Pa = Pb / (1 - P/P_E)$$

$$\text{Extreme fibre stress on concave side } \sigma_{\max} = \frac{P}{A} + \frac{M_{\max}}{I} = \frac{P}{A} + \frac{Pb}{(1 - P/P_E)} \frac{y_{\max}}{Ar^2}$$

$$\text{so } \sigma_{\max} = \sigma + \frac{\sigma}{(1 - \sigma/\sigma_E)} \left(\frac{b y_{\max}}{r^2} \right) \leftarrow \begin{matrix} \text{imperfection} \\ \text{parameter} \end{matrix}$$

$$\text{Setting } \sigma_{\max} = \sigma_y \text{ yield stress} \Rightarrow$$

$$\sigma_y = \sigma + \eta \frac{\sigma}{(1 - \sigma/\sigma_E)} = \sigma + \eta \frac{\sigma \sigma_E}{(\sigma_E - \sigma)}$$

$$\text{whence } (\sigma_y - \sigma)(\sigma_E - \sigma) = \eta \sigma \sigma_E$$

(8)

Q4

- b) i) Marks awarded for clear explanation of additional stiffness associated with warping restraint

ii) $406 \times 140 \times 46$ UB.

$$L = 6\text{m}, \quad I_{\text{minor}} = 538 \times 10^{-8} \text{ m}^4$$

$$J = 19 \times 10^{-8} \text{ m}^4$$

$$G = \frac{E}{2(1+\nu)} = \frac{E}{2.6} \quad \text{for } \nu = 0.3 \text{ (steel)} \text{ and } E = 210 \times 10^9 \text{ N/m}^2.$$

$$\Gamma = \frac{ID^2}{4}, \quad D = 403.2 - 11.2 = 392 \text{ mm} = 0.392 \text{ m}$$

$$\Gamma = \frac{538 \times 10^{-8} (0.392)^2}{4} = \underline{\underline{20.7 \times 10^{-8} \text{ m}^6}}.$$

$$\sqrt{1 + \frac{\pi^2 EI}{L^2 GS}} = \sqrt{1 + \frac{\pi^2 (2.6)}{6^2} \frac{20.7}{19}} = \sqrt{1.78} = \underline{\underline{1.33}}$$

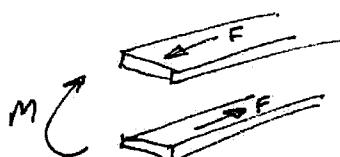
$$\frac{\pi}{L} \sqrt{GJEI} = \frac{\pi}{6} (210 \times 10^9) \sqrt{\frac{(19)(538)}{2.6}} \times 10^{-8}$$

$$= \underline{\underline{68.9 \text{ kNm}}}$$

$$\text{so } M_{cr} = \underline{\underline{1.33 (68.9)}} = \underline{\underline{91.7 \text{ kNm}}}$$

(2)

iii)



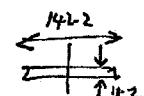
$$FD = M$$

D = dist between flange centroids
 $= 0.392 \text{ m}$

$\therefore FD$

$$\text{Also } f_{cr} = \frac{\pi^2 EI}{L^2}$$

$$I = \frac{bd^3}{12} \text{ for flange alone}$$



$$= \frac{(0.0112)(0.1422)^3}{12} = \underline{\underline{2.684 \times 10^{-6} \text{ m}^4}}$$

$$f_{cr} = \frac{\pi^2 EI}{L^2} = \frac{\pi^2}{6^2} (210 \times 10^9) (2.684 \times 10^{-6}) \text{ N} = 155 \text{ kN}$$

$$M_{cr} = FD = 155 \text{ kN} \times 0.392 \text{ m} = \underline{\underline{60.8 \text{ kNm}}}$$

(6)

Q4 a) Cont'd.

This should equal the second term (The warping restraint term) in the above eqn.

$$(\text{Check: } M_{\text{2nd}} = \frac{\pi}{L} \sqrt{GJ\epsilon} \left(\frac{\pi^2 EI}{L^2 G^2} \right)^{1/2}$$

$$= 68.9 \text{ kNm} \times \sqrt{0.78} = \underline{60.8 \text{ kNm}} \quad \checkmark$$

so discrepancy is omission of first term which arises due to St Venant torsional resistance, (due to shear stresses ).

(4)

3D4 Structural Analysis and Stability 2010-2011
Answers

1. a) $11.369R^3t$, $1.197R^3t$, 64.1°
c) $1.8474\rho g L^2 / R$
2. a) Kinematical = 1, Statical = 4 (Other answers possible if well-argued).
b) BA -33kNm, BC 9kNm, BD 24kNm
3. c) i) $0.6325L$
ii) $0.0113, -0.0033$
4. b) ii) 91.7kNm
iii) 60.8kNm