

## ENGINEERING TRIPoS PART IIB

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Wednesday 26 April 2006      2.30 to 4.00

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Module 4A3

### TURBOMACHINERY I

*Answer not more than two questions.*

*All questions carry the same number of marks.*

*The approximate percentage of marks allocated to each part of a question is indicated in the right margin.*

*Attachment:- Compressible Flow Data Book (38 pages).*

STATIONERY  
Single-sided script paper

SPECIAL REQUIREMENTS  
Engineering Data Book  
CUED approved calculator allowed

**You may not start to read the questions printed on the subsequent pages of this question paper until instructed that you may do so by the Invigilator**

- 1 (a) Derive the relationship

$$\psi = 2(1 - \Lambda - \phi \tan \alpha_1)$$

relating the stage loading coefficient  $\psi$ , the reaction  $\Lambda$ , the flow coefficient  $\phi$  and the interstage swirl  $\alpha_1$  of an axial flow turbine with repeating stages.

[30%]

(b) A 20% reaction high-pressure steam turbine stage has a loading coefficient of 1.85 at a flow coefficient of 0.35. Find the absolute and relative flow angles at inlet and outlet to the stator and rotor and sketch the velocity triangles at these locations.

[25%]

- (c) The total-to-total efficiency of the stage can be estimated from

$$\eta_{tt} = 1 - 0.04 \frac{V_2^2 + W_3^2}{\Delta h_0}.$$

Estimate the total-to-total and the total-to-static efficiency of this stage design.

[20%]

(d) Describe the features which make low reaction stages attractive to some steam turbine manufacturers.

[25%]

- 2 (a) Using a clearly labelled sketch, describe how the radial velocity profile in the rotor passage of a centrifugal impeller varies with increasing radius and what physical phenomena are responsible for these changes. In your answer, you should include discussion of the region close to the trailing edge. [20%]
- (b) A centrifugal compressor is designed with  $40^\circ$  backswept blades and a radial velocity to blade speed ratio at impeller exit of 0.3. Air enters the compressor axially at a rate of  $0.62 \text{ kg s}^{-1}$  at a stagnation temperature of 300 K and a stagnation pressure of 100 kPa. At design flow the stagnation pressure ratio across the impeller is 3, the slip factor is 0.9 and the total-to-total polytropic efficiency is 0.93. Determine the stagnation temperature at impeller exit. [15%]
- (c) Determine the blade speed at impeller exit and the absolute and relative flow velocities there. Sketch the velocity triangle at this location. [25%]
- (d) If the impeller tip radius is 40 mm at inlet and 75 mm at exit, find the axial width of the impeller at exit and the inlet hub-to-tip ratio necessary to keep the inlet Mach number at 0.4. [20%]
- (e) Explain why centrifugal compressors can generally operate with higher pressure coefficients than single stage axial machines. [20%]

(TURN OVER

- 3 (a) Show that the propulsive efficiency of an aircraft can be written as

$$\eta_p = \frac{(1+F)V_j V - V^2}{\frac{1}{2}(1+F)V_j^2 - \frac{1}{2}V^2}$$

where  $V$  is the velocity of the aircraft,  $V_j$  is the jet velocity and  $F$  is the ratio of fuel to air mass flow rates. Explain the physical meaning of the terms in the numerator and denominator. If the fuel mass flow rate can be considered small relative to the air mass flow rate, use the above equation to derive the Froude equation for propulsive efficiency.

[20%]

- (b) The fan of a turbofan engine is designed for an aircraft that cruises at Mach 0.85 at a height of 10000 m. The ambient pressure at this height is 26.5 kPa and the ambient temperature is 223.3 K. The fan pressure ratio is 1.5 and its total-to-total polytropic efficiency is 0.9. If the flow in the exit nozzle may be considered isentropic, calculate the velocity of the jet from the fan  $V_{jfan}$  and the propulsive efficiency of the aircraft  $\eta_p$  assuming that the velocity of the jet from the core is the same as that from the fan.

[35%]

- (c) The fan described in Part (b) is powered directly by the low-pressure turbine. The bypass ratio of the engine is 10 and the flow through the nozzle can be considered isentropic. Take  $\gamma = 1.3$  and  $c_p = 1250 \text{ J kg}^{-1}\text{K}^{-1}$  for the products of combustion.

- (i) Calculate the temperature drop across the low-pressure turbine.

[10%]

- (ii) If the total-to-total polytropic efficiency of the low-pressure turbine is 0.8 and the stagnation pressure and temperature at entry to the turbine are 290 kPa and 1050 K, calculate the ratio of the bypass jet velocity to the core jet velocity,  $V_{jfan} / V_{jcore}$ .

[20%]

- (iii) Describe how you would modify the equation for  $\eta_p$  to account for different core and fan jet velocities.

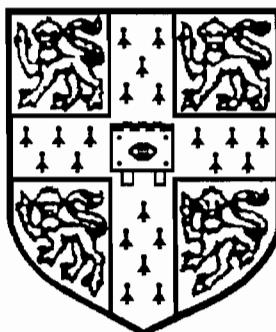
[15%]

**END OF PAPER**

# Compressible Flow Data Book

for Part II of the  
Engineering Tripos

**2004 Edition**



Cambridge University Engineering Department

# PERFECT GAS RELATIONS FOR COMPRESSIBLE FLOW

**Ratios of stagnation to static quantities**

$$\frac{T}{T_0} = \left(1 + \frac{\gamma-1}{2} M^2\right)^{-1}$$

$$\frac{P}{P_0} = \left(1 + \frac{\gamma-1}{2} M^2\right)^{-\frac{\gamma}{\gamma-1}}$$

$$\frac{\rho}{\rho_0} = \left(1 + \frac{\gamma-1}{2} M^2\right)^{-\frac{1}{\gamma-1}}$$

*Notes:*

- (1)  $T_0 = \text{const.}$  in adiabatic flow with no shaft work
- (2) If flow is isentropic,  $p_0 = \text{const.}$  and  $\rho_0 = \text{const.}$  when  $T_0 = \text{const.}$

**Mach number relations (see tables)**

$$\frac{V}{\sqrt{c_p T_0}} = \sqrt{\gamma-1} M \left(1 + \frac{\gamma-1}{2} M^2\right)^{-\frac{1}{2}}$$

$$\frac{\dot{m} \sqrt{c_p T_0}}{A p_0} = \frac{\gamma}{\sqrt{\gamma-1}} M \left(1 + \frac{\gamma-1}{2} M^2\right)^{-\frac{1}{2} \left(\frac{\gamma+1}{\gamma-1}\right)}$$

$$\frac{\dot{m} \sqrt{c_p T_0}}{A p} = \frac{\gamma}{\sqrt{\gamma-1}} M \left(1 + \frac{\gamma-1}{2} M^2\right)^{\frac{1}{2}}$$

$$\frac{F}{\dot{m} \sqrt{c_p T_0}} = \frac{\sqrt{\gamma-1}}{\gamma} \frac{1 + \gamma M^2}{M} \left(1 + \frac{\gamma-1}{2} M^2\right)^{-\frac{1}{2}} \quad \text{where} \quad F = (p + \rho V^2) A$$

$$\frac{\frac{1}{2} \rho V^2}{p_0} = \frac{1}{2} \gamma M^2 \left(1 + \frac{\gamma-1}{2} M^2\right)^{-\frac{\gamma}{\gamma-1}}$$

# ONE-DIMENSIONAL FLOW OF A PERFECT GAS

**Isentropic flow**

$$\frac{A}{A^*} = \frac{1}{M} \left\{ \frac{2}{\gamma+1} \left( 1 + \frac{\gamma-1}{2} M^2 \right) \right\}^{\frac{1}{2} \left( \frac{\gamma+1}{\gamma-1} \right)}$$

**Adiabatic constant area flow**

$$\frac{4c_f L_{\max}}{D} = \frac{1-M^2}{\gamma M^2} + \frac{\gamma+1}{2\gamma} \ln \left( \frac{(\gamma+1)M^2}{2 \left( 1 + \frac{\gamma-1}{2} M^2 \right)} \right)$$

**Normal shock waves in perfect gases**

$$VV_s = a^*{}^2$$

$$M_s = \left( \frac{1 + \frac{\gamma-1}{2} M^2}{\gamma M^2 - \frac{\gamma-1}{2}} \right)^{\frac{1}{2}}$$

$$\frac{p_{0s}}{p_0} = \left( \frac{\frac{\gamma+1}{2} M^2}{1 + \frac{\gamma-1}{2} M^2} \right)^{\frac{\gamma}{\gamma-1}} \left( \frac{2\gamma}{\gamma+1} M^2 - \frac{\gamma-1}{\gamma+1} \right)^{\frac{1}{1-\gamma}}$$

$$\frac{p_s}{p} = 1 + \frac{2\gamma}{\gamma+1} (M^2 - 1)$$

$$\frac{p_{0s}}{p} = \left( \frac{\gamma+1}{2} M^2 \right)^{\frac{\gamma}{\gamma-1}} \left( \frac{2\gamma}{\gamma+1} M^2 - \frac{\gamma-1}{\gamma+1} \right)^{\frac{1}{1-\gamma}}$$

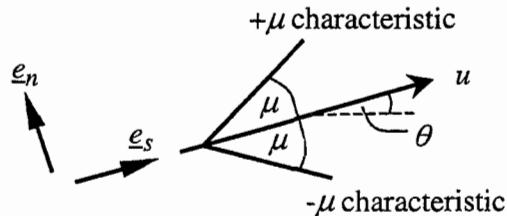
$$\frac{T_s}{T} = \frac{\gamma-1}{(\gamma+1)^2} \frac{2}{M^2} \left( 1 + \frac{\gamma-1}{2} M^2 \right) \left( \frac{2\gamma}{\gamma-1} M^2 - 1 \right)$$

$$\frac{\rho_s}{\rho} = \frac{(\gamma+1)M^2}{2 \left( 1 + \frac{\gamma-1}{2} M^2 \right)}$$

## TWO DIMENSIONAL SUPERSONIC FLOW

### Method of Characteristics for 2-D supersonic flow

Applicable to adiabatic ( $h_0 = \text{constant}$ ), isentropic flow



**Mach Number** 
$$M = u/c$$

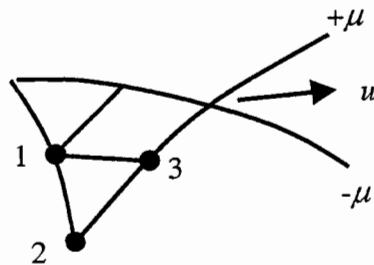
**Mach angle** 
$$\mu = \sin^{-1}\left(\frac{1}{M}\right)$$

**Prandtl-Meyer function** 
$$\nu = \int_1^M \sqrt{M^2 - 1} \frac{du}{u}$$

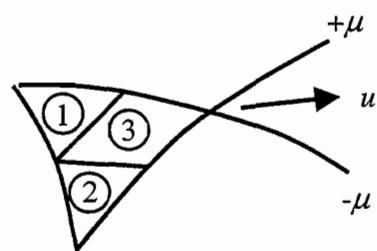
$$\nu = \sqrt{\frac{\gamma+1}{\gamma-1}} \tan^{-1} \sqrt{\frac{\gamma-1}{\gamma+1} (M^2 - 1)} - \tan^{-1} \sqrt{M^2 - 1} \quad \text{for a perfect gas}$$

### Calculations

#### Lattice Method



#### Field (or wave) method



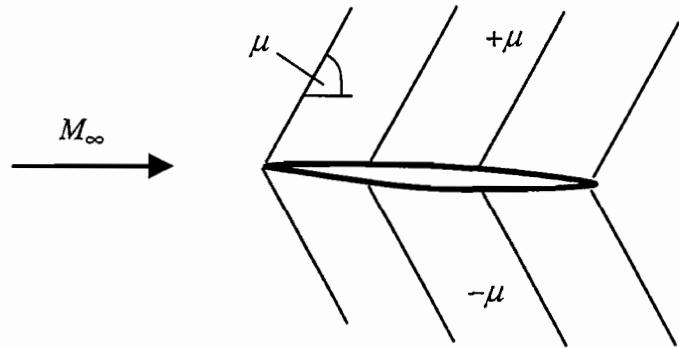
$$\nu_3 - \theta_3 = \nu_2 - \theta_2 \quad \text{along } +\mu$$

$$\nu_3 + \theta_3 = \nu_1 + \theta_1 \quad \text{along } -\mu$$

$$\nu_3 + \theta_3 = \nu_1 + \theta_1 \quad \text{across } +\mu$$

$$\nu_3 - \theta_3 = \nu_2 - \theta_2 \quad \text{across } -\mu$$

### Linearised Method of Characteristics (thin film theory)

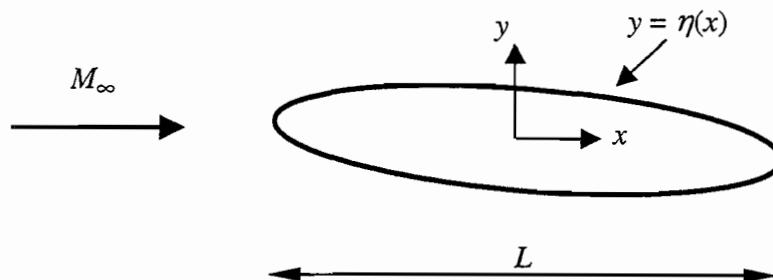


$$\mu \approx \sin^{-1}(1/M_\infty)$$

$$\Delta p \approx \pm \frac{\rho_\infty u_\infty^2 \Delta \theta}{\sqrt{M_\infty^2 - 1}} \quad \text{across } \pm \mu \text{ waves}$$

$$\text{Pressure coefficient } c_p = \frac{p - p_\infty}{\frac{1}{2} \rho_\infty u_\infty^2} = \pm \frac{2\theta}{\sqrt{M_\infty^2 - 1}} \quad \text{on upper/lower surface}$$

**Prandtl-Glauert rule for linearised potential flow past geometrically similar bodies**



$$\text{Pressure coefficient } c_p = \frac{p - p_\infty}{\frac{1}{2} \rho_\infty u_\infty^2}$$

For geometrically similar bodies with  $\frac{\eta}{L} = f\left(\frac{x}{L}\right)$  and  $c_p(M_\infty = 0) = c_{p0}$ ,

$$c_p = \frac{c_{p0}}{\sqrt{1 - M_\infty^2}} \quad \text{in subsonic flow}$$

$$c_p \propto \frac{1}{\sqrt{M_\infty^2 - 1}} \quad \text{in supersonic flow}$$

## Oblique Shock Relations (see tables)

$$\frac{p_2}{p_1} = 1 + \frac{2\gamma}{\gamma+1} (M_1^2 \sin^2 \beta - 1)$$

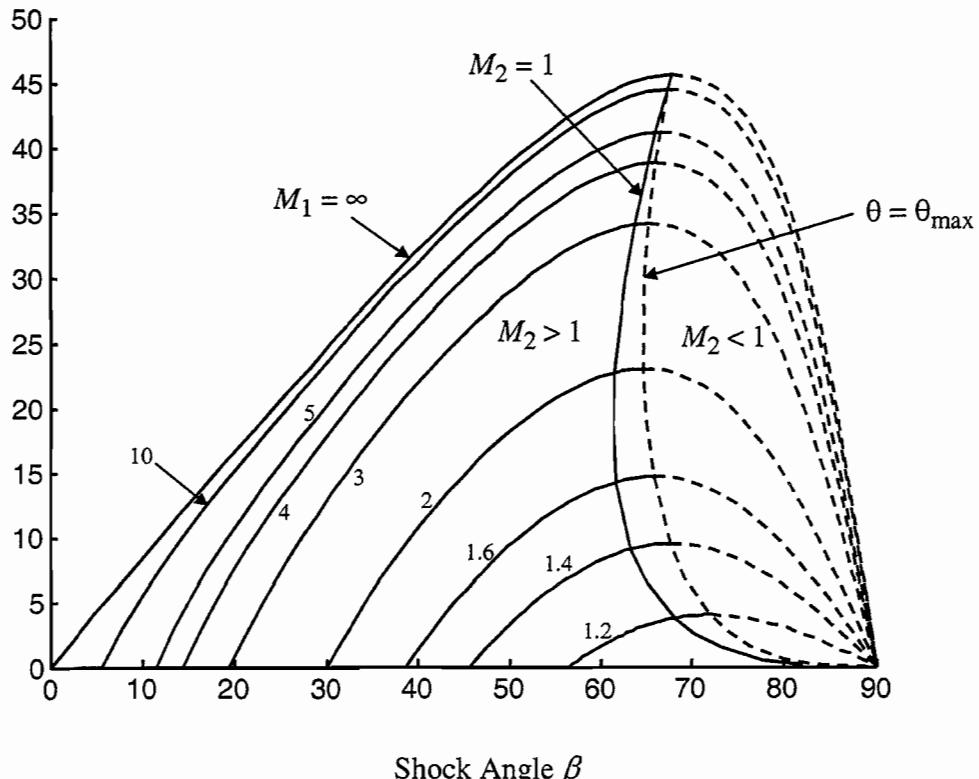
$$\frac{T_2}{T_1} = \frac{\gamma-1}{(\gamma+1)^2} \frac{2}{M_1^2 \sin^2 \beta} \left( 1 + \frac{\gamma-1}{2} M_1^2 \sin^2 \beta \right) \left( \frac{2\gamma}{\gamma-1} M_1^2 \sin^2 \beta - 1 \right)$$

$$\frac{\rho_2}{\rho_1} = \frac{(\gamma+1)M_1^2 \sin^2 \beta}{2 \left[ 1 + \frac{\gamma-1}{2} M_1^2 \sin^2 \beta \right]}$$

$$M_2 \sin(\beta - \theta) = \left[ \frac{1 + \frac{\gamma-1}{2} M_1^2 \sin^2 \beta}{\gamma M_1^2 \sin^2 \beta - \frac{\gamma-1}{2}} \right]^{\frac{1}{2}}$$

$$\frac{p_{02}}{p_{01}} = \left( \frac{\frac{\gamma+1}{2} M_1^2 \sin^2 \beta}{1 + \frac{\gamma-1}{2} M_1^2 \sin^2 \beta} \right)^{\frac{\gamma}{\gamma-1}} \left( \frac{2\gamma}{\gamma+1} M_1^2 \sin^2 \beta - \frac{\gamma-1}{\gamma+1} \right)^{\frac{1}{1-\gamma}}$$

$$\tan \theta = \frac{2 \cot \beta (M_1^2 \sin^2 \beta - 1)}{(\gamma+1)M_1^2 - 2(M_1^2 \sin^2 \beta - 1)}$$



# GAS FLOW TABLES ( $\gamma=1.400$ ): SUBSONIC FLOW

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{p_0}$
0.010	1.0000	0.9999	1.0000	0.0063	0.0221	0.0221	45.1813	7134.405	0.0001
0.020	0.9999	0.9997	0.9998	0.0126	0.0443	0.0443	22.5994	1778.450	0.0003
0.030	0.9998	0.9994	0.9996	0.0190	0.0664	0.0664	15.0761	787.0814	0.0006
0.040	0.9997	0.9989	0.9992	0.0253	0.0885	0.0886	11.3173	440.3522	0.0011
0.050	0.9995	0.9983	0.9988	0.0316	0.1105	0.1107	9.0644	280.0203	0.0017
0.060	0.9993	0.9975	0.9982	0.0379	0.1325	0.1329	7.5645	193.0311	0.0025
0.070	0.9990	0.9966	0.9976	0.0443	0.1545	0.1550	6.4947	140.6550	0.0034
0.080	0.9987	0.9955	0.9968	0.0506	0.1764	0.1772	5.6939	106.7182	0.0045
0.090	0.9984	0.9944	0.9960	0.0569	0.1983	0.1994	5.0723	83.4961	0.0056
0.100	0.9980	0.9930	0.9950	0.0632	0.2200	0.2216	4.5762	66.9216	0.0070
0.110	0.9976	0.9916	0.9940	0.0695	0.2417	0.2438	4.1714	54.6879	0.0084
0.120	0.9971	0.9900	0.9928	0.0758	0.2633	0.2660	3.8350	45.4080	0.0100
0.130	0.9966	0.9883	0.9916	0.0821	0.2849	0.2883	3.5513	38.2070	0.0117
0.140	0.9961	0.9864	0.9903	0.0884	0.3063	0.3105	3.3089	32.5113	0.0135
0.150	0.9955	0.9844	0.9888	0.0947	0.3276	0.3328	3.0996	27.9320	0.0155
0.160	0.9949	0.9823	0.9873	0.1009	0.3488	0.3551	2.9172	24.1978	0.0176
0.170	0.9943	0.9800	0.9857	0.1072	0.3699	0.3774	2.7569	21.1152	0.0198
0.180	0.9936	0.9776	0.9840	0.1135	0.3908	0.3997	2.6151	18.5427	0.0222
0.190	0.9928	0.9751	0.9822	0.1197	0.4116	0.4221	2.4889	16.3752	0.0246
0.200	0.9921	0.9725	0.9803	0.1260	0.4323	0.4445	2.3758	14.5333	0.0272
0.210	0.9913	0.9697	0.9783	0.1322	0.4528	0.4669	2.2740	12.9560	0.0299
0.220	0.9904	0.9668	0.9762	0.1385	0.4731	0.4893	2.1820	11.5961	0.0328
0.230	0.9895	0.9638	0.9740	0.1447	0.4933	0.5118	2.0985	10.4161	0.0357
0.240	0.9886	0.9607	0.9718	0.1509	0.5133	0.5343	2.0225	9.3865	0.0387
0.250	0.9877	0.9575	0.9694	0.1571	0.5332	0.5568	1.9530	8.4834	0.0419
0.260	0.9867	0.9541	0.9670	0.1633	0.5528	0.5794	1.8892	7.6876	0.0451
0.270	0.9856	0.9506	0.9645	0.1695	0.5723	0.6020	1.8306	6.9832	0.0485
0.280	0.9846	0.9470	0.9619	0.1757	0.5915	0.6246	1.7766	6.3572	0.0520
0.290	0.9835	0.9433	0.9592	0.1819	0.6106	0.6473	1.7267	5.7989	0.0555
0.300	0.9823	0.9395	0.9564	0.1881	0.6295	0.6700	1.6805	5.2993	0.0592
0.310	0.9811	0.9355	0.9535	0.1942	0.6481	0.6928	1.6377	4.8507	0.0629
0.320	0.9799	0.9315	0.9506	0.2003	0.6666	0.7156	1.5978	4.4467	0.0668
0.330	0.9787	0.9274	0.9476	0.2065	0.6848	0.7384	1.5608	4.0821	0.0707
0.340	0.9774	0.9231	0.9445	0.2126	0.7027	0.7613	1.5262	3.7520	0.0747
0.350	0.9761	0.9188	0.9413	0.2187	0.7205	0.7842	1.4939	3.4525	0.0788
0.360	0.9747	0.9143	0.9380	0.2248	0.7380	0.8072	1.4637	3.1801	0.0829
0.370	0.9733	0.9098	0.9347	0.2309	0.7553	0.8302	1.4354	2.9320	0.0872
0.380	0.9719	0.9052	0.9313	0.2369	0.7723	0.8532	1.4090	2.7054	0.0915
0.390	0.9705	0.9004	0.9278	0.2430	0.7891	0.8763	1.3841	2.4983	0.0959
0.400	0.9690	0.8956	0.9243	0.2490	0.8056	0.8995	1.3608	2.3085	0.1003
0.410	0.9675	0.8907	0.9207	0.2551	0.8219	0.9227	1.3388	2.1344	0.1048
0.420	0.9659	0.8857	0.9170	0.2611	0.8379	0.9460	1.3182	1.9744	0.1094
0.430	0.9643	0.8807	0.9132	0.2671	0.8536	0.9693	1.2988	1.8272	0.1140
0.440	0.9627	0.8755	0.9094	0.2730	0.8691	0.9927	1.2804	1.6915	0.1186
0.450	0.9611	0.8703	0.9055	0.2790	0.8843	1.0161	1.2632	1.5664	0.1234
0.460	0.9594	0.8650	0.9016	0.2850	0.8992	1.0396	1.2469	1.4509	0.1281
0.470	0.9577	0.8596	0.8976	0.2909	0.9138	1.0631	1.2315	1.3441	0.1329
0.480	0.9559	0.8541	0.8935	0.2968	0.9282	1.0867	1.2170	1.2453	0.1378
0.490	0.9542	0.8486	0.8894	0.3027	0.9423	1.1104	1.2033	1.1539	0.1426
0.500	0.9524	0.8430	0.8852	0.3086	0.9561	1.1341	1.1903	1.0691	0.1475

$\gamma=1.400$

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{p_0}$
0.510	0.9506	0.8374	0.8809	0.3145	0.9696	1.1579	1.1781	0.9904	0.1525
0.520	0.9487	0.8317	0.8766	0.3203	0.9828	1.1818	1.1665	0.9174	0.1574
0.530	0.9468	0.8259	0.8723	0.3262	0.9958	1.2057	1.1556	0.8496	0.1624
0.540	0.9449	0.8201	0.8679	0.3320	1.0084	1.2297	1.1452	0.7866	0.1674
0.550	0.9430	0.8142	0.8634	0.3378	1.0208	1.2538	1.1354	0.7281	0.1724
0.560	0.9410	0.8082	0.8589	0.3436	1.0328	1.2779	1.1261	0.6736	0.1774
0.570	0.9390	0.8022	0.8544	0.3493	1.0446	1.3021	1.1173	0.6229	0.1825
0.580	0.9370	0.7962	0.8498	0.3551	1.0561	1.3264	1.1090	0.5757	0.1875
0.590	0.9349	0.7901	0.8451	0.3608	1.0672	1.3507	1.1011	0.5317	0.1925
0.600	0.9328	0.7840	0.8405	0.3665	1.0781	1.3751	1.0937	0.4908	0.1976
0.610	0.9307	0.7778	0.8357	0.3722	1.0887	1.3996	1.0867	0.4527	0.2026
0.620	0.9286	0.7716	0.8310	0.3779	1.0990	1.4242	1.0800	0.4172	0.2076
0.630	0.9265	0.7654	0.8262	0.3835	1.1090	1.4489	1.0737	0.3841	0.2127
0.640	0.9243	0.7591	0.8213	0.3891	1.1186	1.4736	1.0678	0.3533	0.2177
0.650	0.9221	0.7528	0.8164	0.3948	1.1280	1.4984	1.0621	0.3246	0.2226
0.660	0.9199	0.7465	0.8115	0.4003	1.1371	1.5233	1.0568	0.2979	0.2276
0.670	0.9176	0.7401	0.8066	0.4059	1.1459	1.5483	1.0518	0.2730	0.2326
0.680	0.9153	0.7338	0.8016	0.4115	1.1544	1.5733	1.0471	0.2498	0.2375
0.690	0.9131	0.7274	0.7966	0.4170	1.1626	1.5984	1.0426	0.2282	0.2424
0.700	0.9107	0.7209	0.7916	0.4225	1.1705	1.6237	1.0384	0.2081	0.2473
0.710	0.9084	0.7145	0.7865	0.4280	1.1782	1.6490	1.0344	0.1895	0.2521
0.720	0.9061	0.7080	0.7814	0.4335	1.1855	1.6744	1.0307	0.1721	0.2569
0.730	0.9037	0.7016	0.7763	0.4389	1.1925	1.6999	1.0272	0.1561	0.2617
0.740	0.9013	0.6951	0.7712	0.4443	1.1993	1.7254	1.0239	0.1411	0.2664
0.750	0.8989	0.6886	0.7660	0.4497	1.2058	1.7511	1.0208	0.1273	0.2711
0.760	0.8964	0.6821	0.7609	0.4551	1.2119	1.7768	1.0179	0.1145	0.2758
0.770	0.8940	0.6756	0.7557	0.4605	1.2178	1.8027	1.0152	0.1026	0.2804
0.780	0.8915	0.6691	0.7505	0.4658	1.2234	1.8286	1.0126	0.0917	0.2849
0.790	0.8890	0.6625	0.7452	0.4711	1.2288	1.8547	1.0103	0.0816	0.2894
0.800	0.8865	0.6560	0.7400	0.4764	1.2338	1.8808	1.0081	0.0723	0.2939
0.810	0.8840	0.6495	0.7347	0.4817	1.2386	1.9070	1.0060	0.0638	0.2983
0.820	0.8815	0.6430	0.7295	0.4869	1.2431	1.9333	1.0041	0.0559	0.3026
0.830	0.8789	0.6365	0.7242	0.4921	1.2474	1.9598	1.0024	0.0488	0.3069
0.840	0.8763	0.6300	0.7189	0.4973	1.2514	1.9863	1.0008	0.0423	0.3112
0.850	0.8737	0.6235	0.7136	0.5025	1.2551	2.0129	0.9993	0.0363	0.3153
0.860	0.8711	0.6170	0.7083	0.5077	1.2585	2.0396	0.9979	0.0310	0.3195
0.870	0.8685	0.6106	0.7030	0.5128	1.2617	2.0665	0.9967	0.0261	0.3235
0.880	0.8659	0.6041	0.6977	0.5179	1.2646	2.0934	0.9956	0.0218	0.3275
0.890	0.8632	0.5977	0.6924	0.5230	1.2673	2.1204	0.9946	0.0179	0.3314
0.900	0.8606	0.5913	0.6870	0.5280	1.2698	2.1476	0.9937	0.0145	0.3352
0.910	0.8579	0.5849	0.6817	0.5331	1.2719	2.1748	0.9929	0.0115	0.3390
0.920	0.8552	0.5785	0.6764	0.5381	1.2739	2.2021	0.9922	0.0089	0.3427
0.930	0.8525	0.5721	0.6711	0.5431	1.2756	2.2296	0.9916	0.0067	0.3464
0.940	0.8498	0.5658	0.6658	0.5481	1.2770	2.2572	0.9911	0.0048	0.3499
0.950	0.8471	0.5595	0.6604	0.5530	1.2783	2.2848	0.9907	0.0033	0.3534
0.960	0.8444	0.5532	0.6551	0.5579	1.2793	2.3126	0.9903	0.0021	0.3569
0.970	0.8416	0.5469	0.6498	0.5628	1.2800	2.3405	0.9901	0.0011	0.3602
0.980	0.8389	0.5407	0.6445	0.5677	1.2806	2.3685	0.9899	0.0005	0.3635
0.990	0.8361	0.5345	0.6392	0.5725	1.2809	2.3966	0.9898	0.0001	0.3667
1.000	0.8333	0.5283	0.6339	0.5774	1.2810	2.4249	0.9897	0.0000	0.3698

# GAS FLOW TABLES ( $\gamma=1.400$ ): SUPERSONIC FLOW

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4 c_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{p_0}$	$M_s$	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	$\nu$	$M$
1.010	0.8306	0.5221	0.6287	0.5821	1.2809	2.4532	0.9898	0.0001	0.3728	0.9901	1.0000	1.0235	1.9152	1.0066	0.04	1.010
1.020	0.8278	0.5160	0.6234	0.5869	1.2806	2.4817	0.9899	0.0005	0.3758	0.9805	1.0000	1.0471	1.9379	1.0132	0.13	1.020
1.030	0.8250	0.5099	0.6181	0.5917	1.2801	2.5103	0.9900	0.0010	0.3787	0.9712	1.0000	1.0711	1.9610	1.0198	0.23	1.030
1.040	0.8222	0.5039	0.6129	0.5964	1.2793	2.5590	0.9903	0.0018	0.3815	0.9620	0.9999	1.0952	1.9844	1.0263	0.35	1.040
1.050	0.8193	0.4979	0.5977	0.6011	1.2784	2.5678	0.9905	0.0027	0.3842	0.9531	0.9999	1.1196	2.0083	1.0328	0.49	1.050
1.060	0.8165	0.4919	0.6024	0.6058	1.2773	2.5967	0.9909	0.0038	0.3869	0.9444	0.9998	1.1442	2.0325	1.0393	0.64	1.060
1.070	0.8137	0.4860	0.5972	0.6104	1.2760	2.6258	0.9913	0.0051	0.3895	0.9360	0.9996	1.1691	2.0570	1.0458	0.80	1.070
1.080	0.8108	0.4800	0.5920	0.6151	1.2745	2.6549	0.9917	0.0066	0.3919	0.9277	0.9994	1.1941	2.0819	1.0522	0.97	1.080
1.090	0.8080	0.4742	0.5869	0.6197	1.2728	2.6842	0.9922	0.0082	0.3944	0.9196	0.9992	1.2195	2.1072	1.0586	1.15	1.090
1.100	0.8052	0.4684	0.5817	0.6243	1.2709	2.7136	0.9928	0.0099	0.3967	0.9118	0.9989	1.2450	2.1328	1.0649	1.34	1.100
1.110	0.8023	0.4626	0.5766	0.6288	1.2689	2.7432	0.9934	0.0118	0.3990	0.9041	0.9986	1.2708	2.1598	1.0713	1.53	1.110
1.120	0.7994	0.4568	0.5714	0.6333	1.2667	2.7728	0.9940	0.0138	0.4011	0.8966	0.9982	1.2968	2.1851	1.0776	1.74	1.120
1.130	0.7966	0.4511	0.5663	0.6379	1.2643	2.8026	0.9947	0.0159	0.4032	0.8892	0.9978	1.3231	2.2118	1.0840	1.94	1.130
1.140	0.7937	0.4455	0.5612	0.6423	1.2618	2.8325	0.9954	0.0182	0.4052	0.8820	0.9973	1.3495	2.2388	1.0903	2.16	1.140
1.150	0.7908	0.4398	0.5562	0.6468	1.2590	2.8626	0.9961	0.0205	0.4072	0.8750	0.9967	1.3763	2.2661	1.0966	2.38	1.150
1.160	0.7879	0.4343	0.5511	0.6512	1.2562	2.8927	0.9969	0.0230	0.4090	0.8682	0.9961	1.4032	2.2937	1.1029	2.61	1.160
1.170	0.7851	0.4287	0.5461	0.6556	1.2531	2.9230	0.9978	0.0255	0.4108	0.8615	0.9953	1.4304	2.3217	1.1092	2.84	1.170
1.180	0.7822	0.4232	0.5411	0.6600	1.2500	2.9534	0.9986	0.0281	0.4125	0.8549	0.9946	1.4578	2.3590	1.1154	3.07	1.180
1.190	0.7793	0.4178	0.5361	0.6644	1.2466	2.9840	0.9995	0.0309	0.4141	0.8485	0.9937	1.4855	2.3786	1.1217	3.31	1.190
1.200	0.7764	0.4124	0.5311	0.6687	1.2432	3.0147	1.0004	0.0336	0.4157	0.8422	0.9928	1.5133	2.4075	1.1280	3.56	1.200
1.210	0.7735	0.4070	0.5262	0.6730	1.2396	3.0455	1.0014	0.0365	0.4171	0.8360	0.9918	1.5415	2.4367	1.1343	3.81	1.210
1.220	0.7706	0.4017	0.5213	0.6773	1.2358	3.0764	1.0024	0.0394	0.4185	0.8300	0.9907	1.5698	2.4663	1.1405	4.06	1.220
1.230	0.7677	0.3964	0.5164	0.6816	1.2319	3.1075	1.0034	0.0424	0.4198	0.8241	0.9896	1.5984	2.4961	1.1468	4.31	1.230
1.240	0.7648	0.3912	0.5115	0.6858	1.2279	3.1387	1.0045	0.0455	0.4211	0.8183	0.9884	1.6272	2.5263	1.1531	4.57	1.240
1.250	0.7619	0.3861	0.5067	0.6901	1.2238	3.1700	1.0055	0.0486	0.4223	0.8126	0.9871	1.6563	2.5568	1.1594	4.83	1.250

$\gamma=1.400$

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{4p_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2}\rho V^2}{p_0}$	$M_s$	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	$\nu$	$M$
1.260	0.7590	0.3809	0.5019	0.6943	1.2195	3.2015	1.0066	0.0517	0.4233	0.8071	0.9857	1.6855	2.5875	1.1657	5.09	1.260
1.270	0.7561	0.3759	0.4971	0.6984	1.2152	3.2331	1.0077	0.0549	0.4244	0.8016	0.9842	1.7151	2.6186	1.1720	5.36	1.270
1.280	0.7532	0.3708	0.4923	0.7026	1.2107	3.2648	1.0089	0.0582	0.4253	0.7963	0.9827	1.7448	2.6500	1.1783	5.63	1.280
1.290	0.7503	0.3693	0.4876	0.7067	1.2061	3.2967	1.0100	0.0615	0.4262	0.7911	0.9811	1.7748	2.6816	1.1846	5.90	1.290
1.300	0.7474	0.3609	0.4829	0.7108	1.2014	3.3287	1.0112	0.0648	0.4270	0.7860	0.9794	1.8050	2.7136	1.1909	6.17	1.300
1.310	0.7445	0.3560	0.4782	0.7149	1.1965	3.3608	1.0124	0.0682	0.4277	0.7809	0.9776	1.8355	2.7459	1.1972	6.44	1.310
1.320	0.7416	0.3512	0.4736	0.7189	1.1916	3.3931	1.0136	0.0716	0.4283	0.7760	0.9758	1.8661	2.7784	1.2035	6.72	1.320
1.330	0.7387	0.3464	0.4690	0.7229	1.1866	3.4255	1.0149	0.0750	0.4289	0.7712	0.9738	1.8971	2.8112	1.2099	7.00	1.330
1.340	0.7358	0.3417	0.4644	0.7270	1.1815	3.4581	1.0161	0.0785	0.4294	0.7664	0.9718	1.9282	2.8444	1.2162	7.28	1.340
1.350	0.7329	0.3370	0.4598	0.7309	1.1763	3.4907	1.0174	0.0820	0.4299	0.7618	0.9697	1.9596	2.8778	1.2226	7.56	1.350
1.360	0.7300	0.3323	0.4553	0.7349	1.1710	3.5236	1.0187	0.0855	0.4303	0.7572	0.9676	1.9912	2.9115	1.2290	7.84	1.360
1.370	0.7271	0.3277	0.4508	0.7388	1.1656	3.5566	1.0200	0.0890	0.4306	0.7527	0.9653	2.0231	2.9455	1.2354	8.13	1.370
1.380	0.7242	0.3232	0.4463	0.7427	1.1601	3.5897	1.0213	0.0926	0.4308	0.7483	0.9630	2.0551	2.9798	1.2418	8.41	1.380
1.390	0.7213	0.3187	0.4418	0.7466	1.1546	3.6229	1.0226	0.0962	0.4310	0.7440	0.9607	2.0875	3.0144	1.2482	8.70	1.390
1.400	0.7184	0.3142	0.4374	0.7505	1.1490	3.6563	1.0240	0.0997	0.4311	0.7397	0.9582	2.1200	3.0492	1.2547	8.99	1.400
1.410	0.7155	0.3098	0.4330	0.7543	1.1433	3.6899	1.0253	0.1033	0.4312	0.7355	0.9557	2.1528	3.0844	1.2612	9.28	1.410
1.420	0.7126	0.3055	0.4287	0.7581	1.1375	3.7236	1.0267	0.1069	0.4312	0.7314	0.9531	2.1858	3.1198	1.2676	9.57	1.420
1.430	0.7097	0.3012	0.4244	0.7619	1.1317	3.7574	1.0281	0.1106	0.4311	0.7274	0.9504	2.2191	3.1555	1.2741	9.86	1.430
1.440	0.7069	0.2969	0.4201	0.7657	1.1258	3.7914	1.0295	0.1142	0.4310	0.7235	0.9476	2.2525	3.1915	1.2807	10.15	1.440
1.450	0.7040	0.2927	0.4158	0.7694	1.1198	3.8255	1.0308	0.1178	0.4308	0.7196	0.9448	2.2863	3.2278	1.2872	10.44	1.450
1.460	0.7011	0.2886	0.4116	0.7732	1.1138	3.8598	1.0323	0.1215	0.4306	0.7157	0.9420	2.3202	3.2643	1.2938	10.73	1.460
1.470	0.6982	0.2845	0.4074	0.7769	1.1077	3.8942	1.0337	0.1251	0.4303	0.7120	0.9390	2.3544	3.3011	1.3003	11.02	1.470
1.480	0.6954	0.2804	0.4032	0.7805	1.1016	3.9287	1.0351	0.1288	0.4299	0.7083	0.9360	2.3888	3.3382	1.3069	11.32	1.480
1.490	0.6925	0.2764	0.3991	0.7842	1.0954	3.9634	1.0365	0.1324	0.4295	0.7047	0.9329	2.4235	3.3756	1.3136	11.61	1.490
1.500	0.6897	0.2724	0.3950	0.7878	1.0891	3.9983	1.0379	0.1361	0.4290	0.7011	0.9298	2.4583	3.4133	1.3202	11.91	1.500

$\gamma=1.400$

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4 c_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{p_0}$	$M_s$	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	$v$	$M$
1.510	0.6868	0.2685	0.3909	0.7914	1.0829	4.0333	1.0394	0.1397	0.4285	0.6976	0.9266	2.4935	3.4512	1.3269	12.20	1.510
1.520	0.6840	0.2646	0.3869	0.7950	1.0765	4.0684	1.0408	0.1433	0.4279	0.6941	0.9233	2.5288	3.4894	1.3336	12.49	1.520
1.530	0.6811	0.2608	0.3829	0.7986	1.0702	4.1037	1.0423	0.1470	0.4273	0.6907	0.9200	2.5644	3.5279	1.3403	12.79	1.530
1.540	0.6783	0.2570	0.3789	0.8021	1.0638	4.1392	1.0437	0.1506	0.4266	0.6874	0.9166	2.6002	3.5667	1.3470	13.09	1.540
1.550	0.6754	0.2533	0.3750	0.8057	1.0573	4.1748	1.0452	0.1543	0.4259	0.6841	0.9132	2.6363	3.6057	1.3538	13.38	1.550
1.560	0.6726	0.2496	0.3710	0.8092	1.0508	4.2105	1.0467	0.1579	0.4252	0.6809	0.9097	2.6725	3.6450	1.3606	13.68	1.560
1.570	0.6698	0.2459	0.3672	0.8126	1.0443	4.2464	1.0481	0.1615	0.4243	0.6777	0.9062	2.7091	3.6846	1.3674	13.97	1.570
1.580	0.6670	0.2423	0.3633	0.8161	1.0378	4.2825	1.0496	0.1651	0.4235	0.6746	0.9026	2.7458	3.7244	1.3742	14.27	1.580
1.590	0.6642	0.2388	0.3595	0.8195	1.0312	4.3187	1.0511	0.1688	0.4226	0.6715	0.8989	2.7828	3.7646	1.3811	14.56	1.590
1.600	0.6614	0.2353	0.3557	0.8230	1.0246	4.3551	1.0526	0.1724	0.4216	0.6684	0.8952	2.8200	3.8050	1.3880	14.86	1.600
1.610	0.6586	0.2318	0.3520	0.8263	1.0180	4.3916	1.0541	0.1760	0.4206	0.6655	0.8915	2.8575	3.8456	1.3949	15.16	1.610
1.620	0.6558	0.2284	0.3483	0.8297	1.0114	4.4282	1.0555	0.1795	0.4196	0.6625	0.8877	2.8951	3.8866	1.4018	15.45	1.620
1.630	0.6530	0.2250	0.3446	0.8331	1.0047	4.4651	1.0570	0.1831	0.4185	0.6596	0.8838	2.9331	3.9278	1.4088	15.75	1.630
1.640	0.6502	0.2217	0.3409	0.8364	0.9980	4.5020	1.0585	0.1867	0.4174	0.6568	0.8799	2.9712	3.9693	1.4158	16.04	1.640
1.650	0.6475	0.2184	0.3373	0.8397	0.9913	4.5392	1.0600	0.1902	0.4162	0.6540	0.8760	3.0096	4.0110	1.4228	16.34	1.650
1.660	0.6447	0.2151	0.3337	0.8430	0.9846	4.5765	1.0615	0.1938	0.4150	0.6512	0.8720	3.0482	4.0531	1.4299	16.63	1.660
1.670	0.6419	0.2119	0.3302	0.8462	0.9779	4.6139	1.0630	0.1973	0.4138	0.6485	0.8680	3.0871	4.0953	1.4369	16.93	1.670
1.680	0.6392	0.2088	0.3266	0.8495	0.9712	4.6515	1.0645	0.2008	0.4125	0.6458	0.8639	3.1261	4.1379	1.4440	17.22	1.680
1.690	0.6364	0.2057	0.3232	0.8527	0.9644	4.6892	1.0660	0.2043	0.4112	0.6431	0.8599	3.1655	4.1807	1.4512	17.52	1.690
1.700	0.6337	0.2026	0.3197	0.8559	0.9577	4.7272	1.0674	0.2078	0.4098	0.6405	0.8557	3.2050	4.2238	1.4583	17.81	1.700
1.710	0.6310	0.1996	0.3163	0.8591	0.9508	4.7652	1.0689	0.2113	0.4085	0.6380	0.8516	3.2448	4.2672	1.4655	18.10	1.710
1.720	0.6283	0.1966	0.3129	0.8622	0.9442	4.8035	1.0704	0.2147	0.4071	0.6355	0.8474	3.2848	4.3108	1.4727	18.40	1.720
1.730	0.6256	0.1936	0.3095	0.8654	0.9374	4.8418	1.0719	0.2182	0.4056	0.6330	0.8431	3.3251	4.3547	1.4800	18.69	1.730
1.740	0.6229	0.1907	0.3062	0.8685	0.9307	4.8804	1.0734	0.2216	0.4041	0.6305	0.8389	3.3655	4.3989	1.4873	18.98	1.740
1.750	0.6202	0.1878	0.3029	0.8716	0.9239	4.9191	1.0749	0.2250	0.4026	0.6281	0.8346	3.4063	4.4433	1.4946	19.27	1.750

$\gamma=1.400$

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{p_0}$	$M_s$	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	$v$	$M$
1.760	0.6175	0.1850	0.2996	0.8747	0.9172	4.9580	1.0764	0.2284	0.4011	0.6257	0.8302	3.4472	4.4880	1.5019	19.56	1.760
1.770	0.6148	0.1822	0.2964	0.8777	0.9104	4.9970	1.0779	0.2318	0.3996	0.6234	0.8259	3.4884	4.5330	1.5093	19.86	1.770
1.780	0.6121	0.1794	0.2931	0.8808	0.9037	5.0362	1.0793	0.2352	0.3980	0.6210	0.8215	3.5298	4.5782	1.5167	20.15	1.780
1.790	0.6095	0.1767	0.2900	0.8838	0.8970	5.0755	1.0808	0.2385	0.3964	0.6188	0.8171	3.5715	4.6237	1.5241	20.44	1.790
1.800	0.6068	0.1740	0.2868	0.8868	0.8902	5.1150	1.0823	0.2419	0.3947	0.6165	0.8127	3.6133	4.6695	1.5316	20.73	1.800
1.810	0.6041	0.1714	0.2837	0.8898	0.8835	5.1547	1.0838	0.2452	0.3931	0.6143	0.8082	3.6555	4.7155	1.5391	21.01	1.810
1.820	0.6015	0.1688	0.2806	0.8927	0.8768	5.1945	1.0852	0.2485	0.3914	0.6121	0.8038	3.6978	4.7618	1.5466	21.30	1.820
1.830	0.5989	0.1662	0.2776	0.8957	0.8701	5.2345	1.0867	0.2518	0.3897	0.6099	0.7993	3.7404	4.8084	1.5541	21.59	1.830
1.840	0.5963	0.1637	0.2745	0.8986	0.8634	5.2747	1.0882	0.2551	0.3879	0.6078	0.7948	3.7832	4.8552	1.5617	21.88	1.840
1.850	0.5936	0.1612	0.2715	0.9015	0.8568	5.3150	1.0896	0.2583	0.3862	0.6057	0.7902	3.8263	4.9023	1.5693	22.16	1.850
1.860	0.5910	0.1587	0.2686	0.9044	0.8501	5.3555	1.0911	0.2616	0.3844	0.6036	0.7857	3.8695	4.9497	1.5770	22.45	1.860
1.870	0.5884	0.1563	0.2656	0.9072	0.8435	5.3962	1.0926	0.2648	0.3826	0.6016	0.7811	3.9131	4.9973	1.5847	22.73	1.870
1.880	0.5859	0.1539	0.2627	0.9101	0.8368	5.4370	1.0940	0.2680	0.3808	0.5996	0.7765	3.9568	5.0452	1.5924	23.02	1.880
1.890	0.5833	0.1516	0.2598	0.9129	0.8302	5.4780	1.0955	0.2712	0.3790	0.5976	0.7720	4.0008	5.0934	1.6001	23.30	1.890
1.900	0.5807	0.1492	0.2570	0.9157	0.8237	5.5191	1.0969	0.2743	0.3771	0.5956	0.7674	4.0450	5.1418	1.6079	23.59	1.900
1.910	0.5782	0.1470	0.2542	0.9185	0.8171	5.5604	1.0984	0.2775	0.3753	0.5937	0.7627	4.0895	5.1905	1.6157	23.87	1.910
1.920	0.5756	0.1447	0.2514	0.9213	0.8106	5.6019	1.0998	0.2806	0.3754	0.5918	0.7581	4.1341	5.2394	1.6236	24.15	1.920
1.930	0.5731	0.1425	0.2486	0.9240	0.8041	5.6435	1.1012	0.2837	0.3775	0.5899	0.7535	4.1791	5.2886	1.6314	24.43	1.930
1.940	0.5705	0.1403	0.2459	0.9268	0.7976	5.6853	1.1027	0.2868	0.3696	0.5880	0.7488	4.2242	5.3381	1.6394	24.71	1.940
1.950	0.5680	0.1381	0.2432	0.9295	0.7911	5.7273	1.1041	0.2899	0.3677	0.5862	0.7442	4.2696	5.3878	1.6473	24.99	1.950
1.960	0.5655	0.1360	0.2405	0.9322	0.7846	5.7695	1.1055	0.2929	0.3657	0.5844	0.7395	4.3152	5.4378	1.6553	25.27	1.960
1.970	0.5630	0.1339	0.2378	0.9349	0.7782	5.8118	1.1069	0.2960	0.3638	0.5826	0.7349	4.3611	5.4881	1.6633	25.55	1.970
1.980	0.5605	0.1318	0.2352	0.9375	0.7718	5.8542	1.1084	0.2990	0.3618	0.5808	0.7302	4.4071	5.5386	1.6713	25.83	1.980
1.990	0.5580	0.1298	0.2326	0.9402	0.7655	5.8969	1.1098	0.3020	0.3598	0.5791	0.7255	4.4535	5.5894	1.6794	26.10	1.990
2.000	0.5556	0.1278	0.2300	0.9428	0.7591	5.9397	1.1112	0.3050	0.3579	0.5774	0.7209	4.5000	5.6404	1.6875	26.38	2.000

$\gamma=1.400$

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{p_0}$	$M_s$	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	$\nu$	$M$
2.010	0.5531	0.1258	0.2275	0.9454	0.7528	5.9827	1.1126	0.3080	0.3559	0.5757	0.7162	4.5468	5.6918	1.6956	26.66	2.010
2.020	0.5506	0.1239	0.2250	0.9480	0.7465	6.0258	1.1140	0.3109	0.3539	0.5740	0.7115	4.5938	5.7433	1.7038	26.93	2.020
2.030	0.5482	0.1220	0.2225	0.9506	0.7403	6.0692	1.1154	0.3138	0.3518	0.5723	0.7069	4.6411	5.7952	1.7120	27.20	2.030
2.040	0.5458	0.1201	0.2200	0.9531	0.7340	6.1126	1.1167	0.3168	0.3498	0.5707	0.7022	4.6885	5.8473	1.7203	27.48	2.040
2.050	0.5433	0.1182	0.2176	0.9557	0.7279	6.1563	1.1181	0.3197	0.3478	0.5691	0.6975	4.7363	5.8996	1.7285	27.75	2.050
2.060	0.5409	0.1164	0.2152	0.9582	0.7217	6.2001	1.1195	0.3225	0.3458	0.5675	0.6928	4.7842	5.9523	1.7369	28.02	2.060
2.070	0.5385	0.1146	0.2128	0.9607	0.7156	6.2441	1.1209	0.3254	0.3437	0.5659	0.6882	4.8324	6.0051	1.7452	28.29	2.070
2.080	0.5361	0.1128	0.2104	0.9632	0.7095	6.2883	1.1222	0.3282	0.3417	0.5643	0.6835	4.8808	6.0583	1.7536	28.56	2.080
2.090	0.5337	0.1111	0.2081	0.9657	0.7034	6.3326	1.1236	0.3310	0.3396	0.5628	0.6799	4.9295	6.1117	1.7620	28.83	2.090
2.100	0.5313	0.1094	0.2058	0.9681	0.6974	6.3772	1.1250	0.3339	0.3376	0.5613	0.6742	4.9783	6.1654	1.7705	29.10	2.100
2.110	0.5290	0.1077	0.2035	0.9706	0.6914	6.4218	1.1263	0.3366	0.3355	0.5598	0.6696	5.0275	6.2193	1.7789	29.36	2.110
2.120	0.5266	0.1060	0.2013	0.9730	0.6854	6.4667	1.1276	0.3394	0.3334	0.5583	0.6649	5.0768	6.2735	1.7875	29.63	2.120
2.130	0.5243	0.1043	0.1990	0.9754	0.6795	6.5117	1.1290	0.3422	0.3314	0.5568	0.6603	5.1264	6.3280	1.7960	29.90	2.130
2.140	0.5219	0.1027	0.1968	0.9778	0.6736	6.5569	1.1303	0.3449	0.3293	0.5554	0.6557	5.1762	6.3827	1.8046	30.16	2.140
2.150	0.5196	0.1011	0.1946	0.9802	0.6677	6.6023	1.1317	0.3476	0.3272	0.5540	0.6511	5.2263	6.4377	1.8132	30.43	2.150
2.160	0.5173	0.0996	0.1925	0.9825	0.6619	6.6478	1.1330	0.3503	0.3252	0.5525	0.6464	5.2765	6.4929	1.8219	30.69	2.160
2.170	0.5150	0.0980	0.1903	0.9849	0.6561	6.6936	1.1343	0.3530	0.3231	0.5511	0.6419	5.3271	6.5484	1.8306	30.95	2.170
2.180	0.5127	0.0965	0.1882	0.9872	0.6503	6.7395	1.1356	0.3556	0.3210	0.5498	0.6373	5.3778	6.6042	1.8393	31.21	2.180
2.190	0.5104	0.0950	0.1861	0.9895	0.6446	6.7855	1.1369	0.3583	0.3189	0.5484	0.6327	5.4288	6.6602	1.8481	31.47	2.190
2.200	0.5081	0.0935	0.1841	0.9918	0.6389	6.8318	1.1382	0.3609	0.3169	0.5471	0.6281	5.4800	6.7165	1.8569	31.73	2.200
2.210	0.5059	0.0921	0.1820	0.9941	0.6333	6.8782	1.1395	0.3635	0.3148	0.5457	0.6236	5.5315	6.7730	1.8657	31.99	2.210
2.220	0.5036	0.0906	0.1800	0.9964	0.6277	6.9248	1.1408	0.3661	0.3127	0.5444	0.6191	5.5831	6.8298	1.8746	32.25	2.220
2.230	0.5014	0.0892	0.1780	0.9986	0.6221	6.9715	1.1421	0.3687	0.3106	0.5431	0.6145	5.6351	6.8869	1.8835	32.51	2.230
2.240	0.4991	0.0878	0.1760	1.0009	0.6165	7.0185	1.1434	0.3712	0.3085	0.5418	0.6100	5.6872	6.9442	1.8924	32.76	2.240
2.250	0.4969	0.0865	0.1740	1.0031	0.6110	7.0656	1.1446	0.3738	0.3065	0.5406	0.6055	5.7396	7.0018	1.9014	33.02	2.250

$\gamma=1.400$

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4 c_f L_{\max}}{D} \frac{1}{2} \rho V^2$	$M_s$	$\frac{P_0 s}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	$\nu$	$M$	
2.260	0.4947	0.0851	0.1721	1.0053	0.6056	7.1129	1.1459	0.3763	0.3044	0.5393	0.6011	5.7922	7.0597	1.9104	33.27	2.260
2.270	0.4925	0.0838	0.1702	1.0075	0.6002	7.1603	1.1472	0.3788	0.3023	0.5381	0.5966	5.8451	7.1178	1.9194	33.53	2.270
2.280	0.4903	0.0825	0.1683	1.0097	0.5948	7.2080	1.1484	0.3813	0.3003	0.5368	0.5921	5.8981	7.1762	1.9285	33.78	2.280
2.290	0.4881	0.0812	0.1664	1.0118	0.5894	7.2558	1.1497	0.3838	0.2982	0.5356	0.5877	5.9515	7.2348	1.9376	34.03	2.290
2.300	0.4859	0.0800	0.1646	1.0140	0.5841	7.3038	1.1509	0.3862	0.2961	0.5344	0.5833	6.0050	7.2937	1.9468	34.28	2.300
2.310	0.4837	0.0787	0.1628	1.0161	0.5788	7.3520	1.1521	0.3887	0.2941	0.5332	0.5789	6.0588	7.3528	1.9560	34.53	2.310
2.320	0.4816	0.0775	0.1609	1.0182	0.5736	7.4003	1.1534	0.3911	0.2920	0.5321	0.5745	6.1128	7.4122	1.9652	34.78	2.320
2.330	0.4794	0.0763	0.1592	1.0204	0.5684	7.4488	1.1546	0.3935	0.2900	0.5309	0.5702	6.1671	7.4719	1.9745	35.03	2.330
2.340	0.4773	0.0751	0.1574	1.0224	0.5632	7.4975	1.1558	0.3959	0.2879	0.5297	0.5658	6.2215	7.5319	1.9838	35.28	2.340
2.350	0.4752	0.0740	0.1556	1.0245	0.5581	7.5464	1.1570	0.3983	0.2859	0.5286	0.5615	6.2763	7.5920	1.9931	35.53	2.350
2.360	0.4731	0.0728	0.1539	1.0266	0.5530	7.5955	1.1582	0.4006	0.2839	0.5275	0.5572	6.3312	7.6525	2.0025	35.77	2.360
2.370	0.4709	0.0717	0.1522	1.0286	0.5480	7.6447	1.1595	0.4030	0.2818	0.5264	0.5529	6.3864	7.7132	2.0119	36.02	2.370
2.380	0.4688	0.0706	0.1505	1.0307	0.5430	7.6941	1.1606	0.4053	0.2798	0.5253	0.5486	6.4418	7.7742	2.0213	36.26	2.380
2.390	0.4668	0.0695	0.1488	1.0327	0.5380	7.7437	1.1618	0.4076	0.2778	0.5242	0.5444	6.4975	7.8354	2.0308	36.50	2.390
2.400	0.4647	0.0684	0.1472	1.0347	0.5331	7.7935	1.1630	0.4099	0.2758	0.5231	0.5401	6.5533	7.8969	2.0403	36.75	2.400
2.410	0.4626	0.0673	0.1456	1.0367	0.5282	7.8434	1.1642	0.4122	0.2738	0.5221	0.5359	6.6095	7.9587	2.0499	36.99	2.410
2.420	0.4606	0.0663	0.1439	1.0387	0.5233	7.8935	1.1654	0.4144	0.2718	0.5210	0.5317	6.6658	8.0207	2.0595	37.23	2.420
2.430	0.4585	0.0653	0.1424	1.0407	0.5185	7.9438	1.1665	0.4167	0.2698	0.5200	0.5276	6.7224	8.0830	2.0691	37.47	2.430
2.440	0.4565	0.0643	0.1408	1.0426	0.5137	7.9943	1.1677	0.4189	0.2678	0.5189	0.5234	6.7792	8.1455	2.0788	37.71	2.440
2.450	0.4544	0.0633	0.1392	1.0446	0.5090	8.0450	1.1689	0.4211	0.2658	0.5179	0.5193	6.8363	8.2083	2.0885	37.95	2.450
2.460	0.4524	0.0623	0.1377	1.0465	0.5043	8.0958	1.1700	0.4233	0.2639	0.5169	0.5152	6.8935	8.2713	2.0982	38.18	2.460
2.470	0.4504	0.0613	0.1362	1.0484	0.4996	8.1468	1.1712	0.4255	0.2619	0.5159	0.5111	6.9511	8.3346	2.1080	38.42	2.470
2.480	0.4484	0.0604	0.1346	1.0503	0.4950	8.1980	1.1723	0.4277	0.2599	0.5149	0.5071	7.0088	8.3982	2.1178	38.66	2.480
2.490	0.4464	0.0594	0.1332	1.0522	0.4904	8.2494	1.1734	0.4298	0.2580	0.5140	0.5030	7.0668	8.4620	2.1276	38.89	2.490
2.500	0.4444	0.0585	0.1317	1.0541	0.4858	8.3010	1.1746	0.4320	0.2561	0.5130	0.4990	7.1250	8.5261	2.1375	39.12	2.500

$\gamma=1.400$

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2}\rho V^2}{p_0}$	$M_s$	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	$\nu$	$M$
2.510	0.4425	0.0576	0.1302	1.0560	0.4813	8.3527	1.1757	0.4341	0.5120	0.4950	7.1835	8.5905	2.1474	39.36	2.510	
2.520	0.4405	0.0567	0.1288	1.0578	0.4768	8.4046	1.1768	0.4362	0.5252	0.5111	0.4911	7.2421	8.6551	2.1574	39.59	2.520
2.530	0.4386	0.0559	0.1274	1.0597	0.4724	8.4567	1.1779	0.4383	0.5253	0.5102	0.4871	7.3011	8.7200	2.1674	39.82	2.530
2.540	0.4366	0.0550	0.1260	1.0615	0.4680	8.5090	1.1790	0.4404	0.2484	0.5092	0.4832	7.3602	8.7851	2.1774	40.05	2.540
2.550	0.4347	0.0542	0.1246	1.0633	0.4636	8.5615	1.1801	0.4425	0.2465	0.5083	0.4793	7.4196	8.8505	2.1875	40.28	2.550
2.560	0.4328	0.0533	0.1232	1.0651	0.4593	8.6141	1.1812	0.4445	0.2446	0.5074	0.4754	7.4792	8.9161	2.1976	40.51	2.560
2.570	0.4309	0.0525	0.1218	1.0669	0.4550	8.6670	1.1823	0.4466	0.2427	0.5065	0.4715	7.5391	8.9820	2.2077	40.74	2.570
2.580	0.4289	0.0517	0.1205	1.0687	0.4507	8.7200	1.1834	0.4486	0.2409	0.5056	0.4677	7.5991	9.0482	2.2179	40.96	2.580
2.590	0.4271	0.0509	0.1192	1.0705	0.4465	8.7732	1.1844	0.4506	0.2390	0.5047	0.4659	7.6595	9.1146	2.2281	41.19	2.590
2.600	0.4252	0.0501	0.1179	1.0722	0.4423	8.8265	1.1855	0.4526	0.2371	0.5039	0.4601	7.7200	9.1813	2.2383	41.41	2.600
2.610	0.4233	0.0493	0.1166	1.0740	0.4382	8.8801	1.1866	0.4546	0.2353	0.5030	0.4564	7.7808	9.2483	2.2486	41.64	2.610
2.620	0.4214	0.0486	0.1153	1.0757	0.4341	8.9338	1.1876	0.4565	0.2335	0.5022	0.4526	7.8418	9.3155	2.2590	41.86	2.620
2.630	0.4196	0.0478	0.1140	1.0774	0.4300	8.9877	1.1887	0.4585	0.2317	0.5013	0.4489	7.9031	9.3829	2.2693	42.09	2.630
2.640	0.4177	0.0471	0.1128	1.0791	0.4260	9.0418	1.1897	0.4604	0.2298	0.5005	0.4452	7.9645	9.4506	2.2797	42.31	2.640
2.650	0.4159	0.0464	0.1115	1.0808	0.4220	9.0961	1.1908	0.4624	0.2280	0.4996	0.4416	8.0263	9.5186	2.2902	42.53	2.650
2.660	0.4141	0.0457	0.1103	1.0825	0.4180	9.1506	1.1918	0.4643	0.2262	0.4988	0.4379	8.0882	9.5869	2.3006	42.75	2.660
2.670	0.4122	0.0450	0.1091	1.0842	0.4141	9.2052	1.1928	0.4662	0.2245	0.4980	0.4343	8.1504	9.6554	2.3111	42.97	2.670
2.680	0.4104	0.0443	0.1079	1.0859	0.4102	9.2601	1.1939	0.4681	0.2227	0.4972	0.4307	8.2128	9.7241	2.3217	43.19	2.680
2.690	0.4086	0.0436	0.1067	1.0875	0.4063	9.3151	1.1949	0.4700	0.2209	0.4964	0.4271	8.2755	9.7931	2.3323	43.40	2.690
2.700	0.4068	0.0430	0.1056	1.0892	0.4025	9.3703	1.1959	0.4718	0.2192	0.4956	0.4256	8.3383	9.8624	2.3429	43.62	2.700
2.710	0.4051	0.0423	0.1044	1.0908	0.3987	9.4257	1.1969	0.4737	0.2174	0.4949	0.4201	8.4015	9.9319	2.3536	43.84	2.710
2.720	0.4033	0.0417	0.1033	1.0924	0.3949	9.4812	1.1979	0.4755	0.2157	0.4941	0.4166	8.4648	10.0017	2.3642	44.05	2.720
2.730	0.4015	0.0410	0.1022	1.0941	0.3912	9.5370	1.1989	0.4773	0.2140	0.4933	0.4131	8.5284	10.0718	2.3750	44.27	2.730
2.740	0.3998	0.0404	0.1010	1.0957	0.3875	9.5929	1.1999	0.4791	0.2123	0.4926	0.4097	8.5922	10.1421	2.3858	44.48	2.740
2.750	0.3980	0.0398	0.0999	1.0973	0.3838	9.6490	1.2009	0.4809	0.2106	0.4918	0.4062	8.6563	10.2127	2.3966	44.69	2.750

$\gamma=1.400$

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2}\rho V^2}{p_0}$	$M_s$	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	$\nu$	$M$
2.760	0.3963	0.0392	0.0989	1.0988	0.3802	9.7053	1.2019	0.4827	0.2089	0.4911	0.4028	8.7205	10.2835	2.4074	44.91	2.760
2.770	0.3945	0.0386	0.0978	1.1004	0.3766	9.7618	1.2029	0.4845	0.2072	0.4903	0.3994	8.7851	10.3546	2.4183	45.12	2.770
2.780	0.3928	0.0380	0.0967	1.1020	0.3730	9.8185	1.2038	0.4863	0.2055	0.4896	0.3961	8.8498	10.4259	2.4292	45.33	2.780
2.790	0.3911	0.0374	0.0957	1.1035	0.3695	9.8753	1.2048	0.4880	0.2039	0.4889	0.3928	8.9148	10.4975	2.4402	45.54	2.790
2.800	0.3894	0.0368	0.0946	1.1051	0.3660	9.9324	1.2058	0.4898	0.2022	0.4882	0.3895	8.9800	10.5694	2.4512	45.75	2.800
2.810	0.3877	0.0363	0.0936	1.1066	0.3625	9.9896	1.2067	0.4915	0.2006	0.4875	0.3862	9.0455	10.6415	2.4622	45.95	2.810
2.820	0.3860	0.0357	0.0926	1.1081	0.3591	10.0470	1.2077	0.4932	0.1990	0.4868	0.3829	9.1111	10.7139	2.4733	46.16	2.820
2.830	0.3844	0.0352	0.0916	1.1096	0.3557	10.1046	1.2086	0.4949	0.1973	0.4861	0.3797	9.1771	10.7865	2.4844	46.37	2.830
2.840	0.3827	0.0347	0.0906	1.1111	0.3523	10.1624	1.2095	0.4966	0.1957	0.4854	0.3765	9.2432	10.8594	2.4955	46.57	2.840
2.850	0.3810	0.0341	0.0896	1.1126	0.3490	10.2204	1.2105	0.4983	0.1941	0.4847	0.3733	9.3096	10.9326	2.5067	46.78	2.850
2.860	0.3794	0.0336	0.0886	1.1141	0.3457	10.2785	1.2114	0.5000	0.1926	0.4840	0.3701	9.3762	11.0060	2.5179	46.98	2.860
2.870	0.3777	0.0331	0.0877	1.1156	0.3424	10.3368	1.2123	0.5016	0.1910	0.4833	0.3670	9.4431	11.0797	2.5292	47.19	2.870
2.880	0.3761	0.0326	0.0867	1.1171	0.3392	10.3954	1.2132	0.5033	0.1894	0.4827	0.3639	9.5101	11.1536	2.5405	47.39	2.880
2.890	0.3745	0.0321	0.0858	1.1185	0.3359	10.4541	1.2142	0.5049	0.1879	0.4820	0.3608	9.5775	11.2278	2.5518	47.59	2.890
2.900	0.3729	0.0317	0.0849	1.1199	0.3328	10.5130	1.2151	0.5065	0.1863	0.4814	0.3577	9.6450	11.3022	2.5632	47.79	2.900
2.910	0.3712	0.0312	0.0840	1.1214	0.3296	10.5720	1.2160	0.5081	0.1848	0.4807	0.3547	9.7128	11.3770	2.5746	47.99	2.910
2.920	0.3696	0.0307	0.0831	1.1228	0.3265	10.6313	1.2169	0.5097	0.1833	0.4801	0.3517	9.7808	11.4519	2.5861	48.19	2.920
2.930	0.3681	0.0302	0.0822	1.1242	0.3234	10.6908	1.2178	0.5113	0.1818	0.4795	0.3487	9.8491	11.5271	2.5976	48.39	2.930
2.940	0.3665	0.0298	0.0813	1.1256	0.3203	10.7504	1.2187	0.5129	0.1803	0.4788	0.3457	9.9175	11.6026	2.6091	48.59	2.940
2.950	0.3649	0.0293	0.0804	1.1270	0.3173	10.8102	1.2195	0.5145	0.1788	0.4782	0.3428	9.9863	11.6784	2.6206	48.78	2.950
2.960	0.3633	0.0289	0.0796	1.1284	0.3143	10.8702	1.2204	0.5160	0.1773	0.4776	0.3398	10.0552	11.7544	2.6322	48.98	2.960
2.970	0.3618	0.0285	0.0787	1.1298	0.3113	10.9304	1.2213	0.5176	0.1758	0.4770	0.3369	10.1244	11.8306	2.6439	49.18	2.970
2.980	0.3602	0.0281	0.0779	1.1312	0.3083	10.9908	1.2222	0.5191	0.1744	0.4764	0.3340	10.1938	11.9072	2.6555	49.37	2.980
2.990	0.3587	0.0276	0.0770	1.1325	0.3054	11.0514	1.2230	0.5206	0.1729	0.4758	0.3312	10.2635	11.9839	2.6673	49.56	2.990
3.000	0.3571	0.0272	0.0762	1.1339	0.3025	11.1122	1.2239	0.5222	0.1715	0.4752	0.3283	10.3333	12.0610	2.6790	49.76	3.000

# GAS FLOW TABLES ( $\gamma=1.333$ ): SUBSONIC FLOW

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{p_0}$
0.010	1.0000	0.9999	1.0000	0.0058	0.0231	0.0231	43.2958	7493.200	0.0001
0.020	0.9999	0.9997	0.9998	0.0115	0.0462	0.0462	21.6560	1868.007	0.0003
0.030	0.9999	0.9994	0.9996	0.0173	0.0693	0.0693	14.4464	826.7890	0.0006
0.040	0.9997	0.9989	0.9992	0.0231	0.0923	0.0924	10.8442	462.6179	0.0011
0.050	0.9996	0.9983	0.9988	0.0288	0.1153	0.1155	8.6851	294.2161	0.0017
0.060	0.9994	0.9976	0.9982	0.0346	0.1383	0.1386	7.2475	202.8455	0.0024
0.070	0.9992	0.9967	0.9976	0.0404	0.1612	0.1618	6.2222	147.8292	0.0033
0.080	0.9989	0.9957	0.9968	0.0461	0.1841	0.1849	5.4546	112.1800	0.0042
0.090	0.9987	0.9946	0.9960	0.0519	0.2069	0.2080	4.8587	87.7848	0.0054
0.100	0.9983	0.9934	0.9950	0.0577	0.2297	0.2312	4.3831	70.3719	0.0066
0.110	0.9980	0.9920	0.9940	0.0634	0.2523	0.2544	3.9949	57.5186	0.0080
0.120	0.9976	0.9905	0.9928	0.0692	0.2749	0.2775	3.6724	47.7680	0.0095
0.130	0.9972	0.9888	0.9916	0.0749	0.2974	0.3007	3.4003	40.2012	0.0111
0.140	0.9967	0.9870	0.9903	0.0807	0.3197	0.3239	3.1678	34.2155	0.0129
0.150	0.9963	0.9851	0.9888	0.0864	0.3420	0.3471	2.9670	29.4027	0.0148
0.160	0.9958	0.9831	0.9873	0.0921	0.3641	0.3704	2.7920	25.4777	0.0168
0.170	0.9952	0.9810	0.9857	0.0979	0.3861	0.3936	2.6383	22.2372	0.0189
0.180	0.9946	0.9787	0.9840	0.1036	0.4080	0.4169	2.5022	19.5326	0.0211
0.190	0.9940	0.9763	0.9822	0.1093	0.4298	0.4402	2.3809	17.2536	0.0235
0.200	0.9934	0.9738	0.9803	0.1150	0.4514	0.4635	2.2724	15.3166	0.0260
0.210	0.9927	0.9711	0.9783	0.1207	0.4728	0.4869	2.1747	13.6578	0.0285
0.220	0.9920	0.9684	0.9762	0.1264	0.4941	0.5102	2.0863	12.2273	0.0312
0.230	0.9913	0.9655	0.9740	0.1321	0.5152	0.5336	2.0061	10.9859	0.0340
0.240	0.9905	0.9625	0.9717	0.1378	0.5362	0.5570	1.9330	9.9026	0.0370
0.250	0.9897	0.9594	0.9694	0.1435	0.5569	0.5805	1.8662	8.9522	0.0400
0.260	0.9889	0.9562	0.9669	0.1492	0.5775	0.6040	1.8049	8.1146	0.0431
0.270	0.9880	0.9529	0.9644	0.1549	0.5979	0.6275	1.7486	7.3731	0.0463
0.280	0.9871	0.9494	0.9618	0.1605	0.6181	0.6510	1.6966	6.7140	0.0496
0.290	0.9862	0.9459	0.9591	0.1662	0.6380	0.6746	1.6486	6.1261	0.0530
0.300	0.9852	0.9422	0.9563	0.1718	0.6578	0.6982	1.6042	5.5998	0.0565
0.310	0.9843	0.9384	0.9534	0.1775	0.6774	0.7218	1.5629	5.1272	0.0601
0.320	0.9832	0.9346	0.9505	0.1831	0.6967	0.7455	1.5245	4.7016	0.0638
0.330	0.9822	0.9306	0.9475	0.1887	0.7158	0.7692	1.4888	4.3173	0.0675
0.340	0.9811	0.9265	0.9444	0.1943	0.7347	0.7929	1.4555	3.9693	0.0714
0.350	0.9800	0.9224	0.9412	0.1999	0.7533	0.8167	1.4244	3.6535	0.0753
0.360	0.9789	0.9181	0.9379	0.2055	0.7717	0.8405	1.3953	3.3663	0.0793
0.370	0.9777	0.9137	0.9346	0.2111	0.7898	0.8644	1.3680	3.1046	0.0834
0.380	0.9765	0.9093	0.9311	0.2167	0.8077	0.8883	1.3425	2.8655	0.0875
0.390	0.9753	0.9047	0.9276	0.2223	0.8253	0.9122	1.3185	2.6469	0.0917
0.400	0.9741	0.9001	0.9241	0.2278	0.8427	0.9362	1.2959	2.4466	0.0960
0.410	0.9728	0.8954	0.9204	0.2334	0.8598	0.9603	1.2747	2.2627	0.1003
0.420	0.9715	0.8906	0.9167	0.2389	0.8766	0.9843	1.2548	2.0937	0.1047
0.430	0.9701	0.8857	0.9130	0.2444	0.8932	1.0085	1.2360	1.9382	0.1091
0.440	0.9688	0.8807	0.9091	0.2499	0.9095	1.0326	1.2183	1.7949	0.1136
0.450	0.9674	0.8757	0.9052	0.2554	0.9255	1.0569	1.2016	1.6627	0.1182
0.460	0.9660	0.8706	0.9012	0.2609	0.9412	1.0811	1.1858	1.5405	0.1228
0.470	0.9645	0.8654	0.8972	0.2664	0.9567	1.1055	1.1710	1.4276	0.1274
0.480	0.9631	0.8601	0.8931	0.2718	0.9718	1.1299	1.1569	1.3231	0.1321
0.490	0.9616	0.8548	0.8890	0.2773	0.9867	1.1543	1.1436	1.2263	0.1368
0.500	0.9600	0.8494	0.8847	0.2827	1.0012	1.1788	1.1310	1.1365	0.1415

$\gamma=1.333$

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{p_0}$
0.510	0.9585	0.8439	0.8805	0.2881	1.0155	1.2033	1.1192	1.0532	0.1463
0.520	0.9569	0.8384	0.8761	0.2935	1.0295	1.2279	1.1079	0.9759	0.1511
0.530	0.9553	0.8328	0.8717	0.2989	1.0431	1.2526	1.0973	0.9041	0.1559
0.540	0.9537	0.8271	0.8673	0.3043	1.0565	1.2773	1.0872	0.8373	0.1608
0.550	0.9520	0.8214	0.8628	0.3097	1.0696	1.3021	1.0777	0.7752	0.1656
0.560	0.9504	0.8157	0.8583	0.3150	1.0823	1.3269	1.0687	0.7174	0.1705
0.570	0.9487	0.8099	0.8537	0.3204	1.0948	1.3518	1.0601	0.6636	0.1754
0.580	0.9470	0.8040	0.8490	0.3257	1.1069	1.3768	1.0520	0.6136	0.1803
0.590	0.9452	0.7981	0.8443	0.3310	1.1188	1.4018	1.0444	0.5669	0.1852
0.600	0.9434	0.7921	0.8396	0.3363	1.1303	1.4269	1.0371	0.5235	0.1901
0.610	0.9417	0.7861	0.8348	0.3416	1.1415	1.4521	1.0303	0.4830	0.1950
0.620	0.9398	0.7801	0.8300	0.3469	1.1524	1.4773	1.0238	0.4452	0.1999
0.630	0.9380	0.7740	0.8252	0.3521	1.1630	1.5026	1.0176	0.4101	0.2048
0.640	0.9362	0.7679	0.8203	0.3573	1.1733	1.5280	1.0118	0.3773	0.2096
0.650	0.9343	0.7618	0.8153	0.3626	1.1833	1.5534	1.0063	0.3467	0.2145
0.660	0.9324	0.7556	0.8104	0.3678	1.1930	1.5789	1.0011	0.3183	0.2194
0.670	0.9305	0.7494	0.8054	0.3729	1.2023	1.6045	0.9962	0.2918	0.2242
0.680	0.9285	0.7431	0.8003	0.3781	1.2114	1.6301	0.9916	0.2671	0.2290
0.690	0.9266	0.7368	0.7953	0.3833	1.2201	1.6559	0.9872	0.2441	0.2338
0.700	0.9246	0.7306	0.7902	0.3884	1.2285	1.6817	0.9831	0.2227	0.2386
0.710	0.9226	0.7242	0.7850	0.3935	1.2367	1.7075	0.9792	0.2028	0.2433
0.720	0.9205	0.7179	0.7799	0.3986	1.2445	1.7335	0.9755	0.1843	0.2480
0.730	0.9185	0.7116	0.7747	0.4037	1.2520	1.7595	0.9721	0.1671	0.2527
0.740	0.9164	0.7052	0.7695	0.4088	1.2592	1.7856	0.9688	0.1512	0.2574
0.750	0.9144	0.6988	0.7643	0.4139	1.2661	1.8118	0.9658	0.1364	0.2620
0.760	0.9123	0.6924	0.7590	0.4189	1.2727	1.8381	0.9629	0.1227	0.2666
0.770	0.9102	0.6860	0.7537	0.4239	1.2790	1.8644	0.9603	0.1100	0.2711
0.780	0.9080	0.6796	0.7484	0.4289	1.2850	1.8908	0.9578	0.0983	0.2756
0.790	0.9059	0.6732	0.7431	0.4339	1.2907	1.9174	0.9554	0.0875	0.2800
0.800	0.9037	0.6668	0.7378	0.4389	1.2961	1.9440	0.9533	0.0776	0.2844
0.810	0.9015	0.6603	0.7325	0.4438	1.3013	1.9706	0.9513	0.0685	0.2888
0.820	0.8993	0.6539	0.7271	0.4487	1.3061	1.9974	0.9494	0.0601	0.2930
0.830	0.8971	0.6475	0.7217	0.4536	1.3107	2.0243	0.9477	0.0524	0.2973
0.840	0.8949	0.6411	0.7164	0.4585	1.3149	2.0512	0.9461	0.0454	0.3015
0.850	0.8926	0.6346	0.7110	0.4634	1.3189	2.0782	0.9446	0.0391	0.3056
0.860	0.8904	0.6282	0.7056	0.4683	1.3226	2.1053	0.9433	0.0333	0.3097
0.870	0.8881	0.6218	0.7002	0.4731	1.3260	2.1326	0.9420	0.0281	0.3137
0.880	0.8858	0.6154	0.6948	0.4779	1.3292	2.1599	0.9409	0.0235	0.3176
0.890	0.8835	0.6090	0.6893	0.4827	1.3321	2.1873	0.9399	0.0193	0.3215
0.900	0.8812	0.6026	0.6839	0.4875	1.3347	2.2147	0.9390	0.0156	0.3253
0.910	0.8788	0.5963	0.6785	0.4923	1.3370	2.2423	0.9383	0.0124	0.3291
0.920	0.8765	0.5899	0.6731	0.4970	1.3391	2.2700	0.9376	0.0096	0.3328
0.930	0.8741	0.5836	0.6676	0.5018	1.3410	2.2978	0.9370	0.0072	0.3364
0.940	0.8717	0.5773	0.6622	0.5065	1.3425	2.3256	0.9365	0.0052	0.3400
0.950	0.8694	0.5710	0.6568	0.5111	1.3439	2.3536	0.9360	0.0035	0.3435
0.960	0.8670	0.5647	0.6514	0.5158	1.3449	2.3817	0.9357	0.0022	0.3469
0.970	0.8646	0.5585	0.6459	0.5205	1.3458	2.4098	0.9354	0.0012	0.3502
0.980	0.8621	0.5522	0.6405	0.5251	1.3464	2.4381	0.9353	0.0005	0.3535
0.990	0.8597	0.5460	0.6351	0.5297	1.3467	2.4664	0.9351	0.0001	0.3567
1.000	0.8573	0.5398	0.6297	0.5343	1.3468	2.4949	0.9351	0.0000	0.3598

# GAS FLOW TABLES ( $\gamma=1.333$ ): SUPERSONIC FLOW

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{p_0}$
1.010	0.8548	0.5337	0.6243	0.5389	1.3467	2.5234	0.9351	0.0001	0.3628
1.020	0.8524	0.5276	0.6189	0.5434	1.3464	2.5521	0.9352	0.0005	0.3658
1.030	0.8499	0.5215	0.6136	0.5479	1.3458	2.5809	0.9354	0.0011	0.3687
1.040	0.8474	0.5154	0.6082	0.5525	1.3450	2.6097	0.9356	0.0019	0.3715
1.050	0.8449	0.5093	0.6028	0.5569	1.3440	2.6387	0.9359	0.0029	0.3743
1.060	0.8424	0.5033	0.5975	0.5614	1.3428	2.6678	0.9363	0.0042	0.3769
1.070	0.8399	0.4974	0.5922	0.5659	1.3414	2.6970	0.9367	0.0056	0.3795
1.080	0.8374	0.4914	0.5869	0.5703	1.3397	2.7263	0.9371	0.0071	0.3820
1.090	0.8349	0.4855	0.5816	0.5747	1.3379	2.7557	0.9376	0.0089	0.3845
1.100	0.8323	0.4796	0.5763	0.5791	1.3359	2.7852	0.9381	0.0108	0.3868
1.110	0.8298	0.4738	0.5710	0.5835	1.3337	2.8148	0.9387	0.0128	0.3891
1.120	0.8272	0.4680	0.5658	0.5878	1.3313	2.8446	0.9394	0.0150	0.3913
1.130	0.8247	0.4622	0.5605	0.5922	1.3287	2.8744	0.9401	0.0173	0.3934
1.140	0.8221	0.4565	0.5553	0.5965	1.3259	2.9043	0.9408	0.0197	0.3954
1.150	0.8195	0.4508	0.5501	0.6008	1.3229	2.9344	0.9415	0.0223	0.3974
1.160	0.8170	0.4452	0.5449	0.6050	1.3198	2.9646	0.9424	0.0250	0.3993
1.170	0.8144	0.4396	0.5398	0.6093	1.3165	2.9949	0.9432	0.0277	0.4011
1.180	0.8118	0.4340	0.5347	0.6135	1.3131	3.0253	0.9441	0.0306	0.4028
1.190	0.8092	0.4285	0.5295	0.6177	1.3094	3.0558	0.9450	0.0335	0.4044
1.200	0.8066	0.4230	0.5245	0.6219	1.3057	3.0864	0.9459	0.0366	0.4060
1.210	0.8040	0.4176	0.5194	0.6261	1.3017	3.1172	0.9469	0.0397	0.4075
1.220	0.8014	0.4122	0.5143	0.6302	1.2976	3.1481	0.9479	0.0429	0.4089
1.230	0.7988	0.4068	0.5093	0.6344	1.2934	3.1791	0.9489	0.0462	0.4102
1.240	0.7962	0.4015	0.5043	0.6385	1.2890	3.2102	0.9500	0.0495	0.4115
1.250	0.7936	0.3963	0.4994	0.6426	1.2845	3.2414	0.9511	0.0529	0.4127
1.260	0.7909	0.3911	0.4944	0.6466	1.2798	3.2727	0.9522	0.0564	0.4138
1.270	0.7883	0.3859	0.4895	0.6507	1.2751	3.3042	0.9533	0.0599	0.4148
1.280	0.7857	0.3808	0.4846	0.6547	1.2701	3.3358	0.9545	0.0634	0.4158
1.290	0.7830	0.3757	0.4798	0.6587	1.2651	3.3675	0.9557	0.0670	0.4167
1.300	0.7804	0.3706	0.4749	0.6627	1.2599	3.3993	0.9569	0.0707	0.4175
1.310	0.7778	0.3657	0.4701	0.6667	1.2547	3.4313	0.9581	0.0744	0.4182
1.320	0.7751	0.3607	0.4654	0.6706	1.2493	3.4633	0.9594	0.0781	0.4189
1.330	0.7725	0.3558	0.4606	0.6746	1.2438	3.4955	0.9606	0.0819	0.4195
1.340	0.7698	0.3510	0.4559	0.6785	1.2382	3.5279	0.9619	0.0857	0.4200
1.350	0.7672	0.3462	0.4512	0.6824	1.2325	3.5603	0.9632	0.0895	0.4205
1.360	0.7646	0.3414	0.4465	0.6862	1.2266	3.5929	0.9645	0.0934	0.4209
1.370	0.7619	0.3367	0.4419	0.6901	1.2207	3.6256	0.9659	0.0973	0.4212
1.380	0.7593	0.3320	0.4373	0.6939	1.2147	3.6584	0.9672	0.1012	0.4215
1.390	0.7566	0.3274	0.4328	0.6977	1.2086	3.6914	0.9686	0.1051	0.4216
1.400	0.7540	0.3229	0.4282	0.7015	1.2025	3.7245	0.9700	0.1091	0.4218
1.410	0.7513	0.3183	0.4237	0.7053	1.1962	3.7577	0.9714	0.1130	0.4218
1.420	0.7487	0.3139	0.4192	0.7090	1.1899	3.7910	0.9728	0.1170	0.4218
1.430	0.7460	0.3094	0.4148	0.7127	1.1835	3.8245	0.9742	0.1210	0.4217
1.440	0.7434	0.3051	0.4104	0.7164	1.1770	3.8581	0.9756	0.1250	0.4216
1.450	0.7407	0.3007	0.4060	0.7201	1.1704	3.8918	0.9771	0.1290	0.4214
1.460	0.7381	0.2965	0.4017	0.7238	1.1638	3.9257	0.9785	0.1331	0.4212
1.470	0.7354	0.2922	0.3974	0.7275	1.1571	3.9597	0.9800	0.1371	0.4209
1.480	0.7328	0.2880	0.3931	0.7311	1.1504	3.9938	0.9815	0.1411	0.4205
1.490	0.7301	0.2839	0.3888	0.7347	1.1435	4.0281	0.9829	0.1452	0.4201
1.500	0.7275	0.2798	0.3846	0.7383	1.1367	4.0625	0.9844	0.1492	0.4196

$\gamma=1.333$

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2}\rho V^2}{p_0}$
1.510	0.7248	0.2758	0.3804	0.7419	1.1298	4.0970	0.9859	0.1532	0.4191
1.520	0.7222	0.2718	0.3763	0.7454	1.1228	4.1317	0.9874	0.1573	0.4185
1.530	0.7195	0.2678	0.3722	0.7489	1.1158	4.1665	0.9889	0.1613	0.4178
1.540	0.7169	0.2639	0.3681	0.7524	1.1087	4.2014	0.9905	0.1654	0.4171
1.550	0.7143	0.2600	0.3641	0.7559	1.1016	4.2365	0.9920	0.1694	0.4164
1.560	0.7116	0.2562	0.3600	0.7594	1.0945	4.2717	0.9935	0.1734	0.4156
1.570	0.7090	0.2524	0.3561	0.7629	1.0873	4.3070	0.9950	0.1775	0.4147
1.580	0.7064	0.2487	0.3521	0.7663	1.0801	4.3425	0.9966	0.1815	0.4138
1.590	0.7038	0.2450	0.3482	0.7697	1.0729	4.3782	0.9981	0.1855	0.4129
1.600	0.7011	0.2414	0.3443	0.7731	1.0656	4.4139	0.9997	0.1895	0.4119
1.610	0.6985	0.2378	0.3405	0.7765	1.0583	4.4498	1.0012	0.1935	0.4109
1.620	0.6959	0.2343	0.3367	0.7799	1.0510	4.4859	1.0028	0.1975	0.4098
1.630	0.6933	0.2308	0.3329	0.7832	1.0436	4.5220	1.0043	0.2015	0.4087
1.640	0.6907	0.2273	0.3291	0.7865	1.0363	4.5584	1.0059	0.2055	0.4075
1.650	0.6881	0.2239	0.3254	0.7898	1.0289	4.5948	1.0075	0.2094	0.4063
1.660	0.6855	0.2206	0.3217	0.7931	1.0215	4.6314	1.0090	0.2134	0.4051
1.670	0.6829	0.2172	0.3181	0.7964	1.0141	4.6682	1.0106	0.2173	0.4038
1.680	0.6803	0.2139	0.3145	0.7996	1.0066	4.7051	1.0122	0.2213	0.4025
1.690	0.6777	0.2107	0.3109	0.8028	0.9992	4.7421	1.0137	0.2252	0.4011
1.700	0.6751	0.2075	0.3074	0.8061	0.9918	4.7793	1.0153	0.2291	0.3997
1.710	0.6726	0.2044	0.3039	0.8093	0.9843	4.8166	1.0169	0.2330	0.3983
1.720	0.6700	0.2012	0.3004	0.8124	0.9769	4.8541	1.0184	0.2369	0.3968
1.730	0.6674	0.1982	0.2969	0.8156	0.9694	4.8917	1.0200	0.2407	0.3953
1.740	0.6649	0.1951	0.2935	0.8187	0.9620	4.9294	1.0216	0.2446	0.3938
1.750	0.6623	0.1922	0.2901	0.8218	0.9545	4.9673	1.0232	0.2484	0.3922
1.760	0.6597	0.1892	0.2868	0.8249	0.9471	5.0054	1.0247	0.2522	0.3906
1.770	0.6572	0.1863	0.2835	0.8280	0.9396	5.0435	1.0263	0.2560	0.3890
1.780	0.6546	0.1834	0.2802	0.8311	0.9322	5.0819	1.0279	0.2598	0.3874
1.790	0.6521	0.1806	0.2770	0.8341	0.9248	5.1204	1.0294	0.2636	0.3857
1.800	0.6496	0.1778	0.2737	0.8372	0.9173	5.1590	1.0310	0.2673	0.3840
1.810	0.6471	0.1751	0.2706	0.8402	0.9099	5.1978	1.0326	0.2711	0.3822
1.820	0.6445	0.1723	0.2674	0.8432	0.9025	5.2367	1.0341	0.2748	0.3805
1.830	0.6420	0.1697	0.2643	0.8461	0.8951	5.2758	1.0357	0.2785	0.3787
1.840	0.6395	0.1670	0.2612	0.8491	0.8878	5.3150	1.0373	0.2822	0.3769
1.850	0.6370	0.1644	0.2581	0.8521	0.8804	5.3544	1.0388	0.2858	0.3751
1.860	0.6345	0.1619	0.2551	0.8550	0.8731	5.3939	1.0404	0.2895	0.3732
1.870	0.6320	0.1593	0.2521	0.8579	0.8658	5.4336	1.0419	0.2931	0.3714
1.880	0.6295	0.1568	0.2491	0.8608	0.8585	5.4734	1.0435	0.2967	0.3695
1.890	0.6271	0.1544	0.2462	0.8636	0.8512	5.5134	1.0450	0.3003	0.3676
1.900	0.6246	0.1520	0.2433	0.8665	0.8439	5.5535	1.0466	0.3039	0.3656
1.910	0.6221	0.1496	0.2404	0.8693	0.8367	5.5938	1.0481	0.3074	0.3637
1.920	0.6197	0.1472	0.2376	0.8722	0.8295	5.6342	1.0497	0.3110	0.3617
1.930	0.6172	0.1449	0.2348	0.8750	0.8223	5.6748	1.0512	0.3145	0.3598
1.940	0.6148	0.1426	0.2320	0.8778	0.8152	5.7155	1.0527	0.3180	0.3578
1.950	0.6123	0.1404	0.2292	0.8805	0.8081	5.7564	1.0543	0.3215	0.3558
1.960	0.6099	0.1382	0.2265	0.8833	0.8010	5.7974	1.0558	0.3249	0.3537
1.970	0.6075	0.1360	0.2238	0.8860	0.7939	5.8386	1.0573	0.3284	0.3517
1.980	0.6051	0.1338	0.2212	0.8888	0.7869	5.8800	1.0588	0.3318	0.3497
1.990	0.6026	0.1317	0.2185	0.8915	0.7799	5.9215	1.0603	0.3352	0.3476
2.000	0.6002	0.1296	0.2159	0.8942	0.7729	5.9631	1.0619	0.3386	0.3455

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{\rho_2}{\rho_1}$	$M_2$	$\frac{T_2}{T_1}$	$\frac{p_{02}}{p_{01}}$	
1.05	0.558	79.937	1.0803	1.0567	1.0223	0.9845	0.99995	1.40	8.000	75.893	1.9842	1.6163	1.2276	0.8184	
									6.000	80.485	2.0575	1.6562	1.2423	0.7762	0.96806
									4.000	83.988	2.0949	1.6763	1.2497	0.7545	0.96286
									2.000	87.075	2.1140	1.6865	1.2535	0.7432	0.96009
1.10	1.515	76.297	1.1658	1.1157	1.0449	0.9711	0.99963								0.95865
1.15	2.000	67.003	1.1408	1.0986	1.0384	1.0434	0.99977	1.45	2.000	46.004	1.1028	1.0723	1.0284	1.3808	0.99990
	2.671	73.822	1.2565	1.1767	1.0678	0.9598	0.99879		4.000	48.679	1.2169	1.1503	1.0579	1.3091	0.99923
	2.000	81.173	1.3399	1.2316	1.0880	0.9007	0.99745		6.000	51.755	1.3463	1.2357	1.0895	1.2325	0.99733
1.20	2.000	61.050	1.1197	1.0841	1.0329	1.1113	0.99985		8.000	55.517	1.5000	1.3333	1.1250	1.1460	0.99325
	3.944	71.977	1.3525	1.2397	1.0910	0.9502	0.99720		10.000	61.046	1.7114	1.4613	1.1712	1.0317	0.98440
	2.000	83.861	1.4941	1.3297	1.1237	0.8551	0.99344		10.785	67.097	1.9147	1.5779	1.2135	0.9235	0.97269
1.25	2.000	56.844	1.1110	1.0780	1.0306	1.1696	0.99988		10.000	72.994	2.0764	1.6664	1.2461	0.8366	0.96147
	4.000	61.986	1.2541	1.1752	1.0672	1.0721	0.99882		8.000	78.197	2.1836	1.7232	1.2672	0.7777	0.95324
	5.286	70.540	1.4539	1.3045	1.1146	0.9423	0.99468		6.000	81.733	2.2355	1.7501	1.2774	0.7485	0.94905
	4.000	79.385	1.5944	1.3913	1.1459	0.8525	0.98975		4.000	84.702	2.2853	1.7654	1.2832	0.7316	0.94659
	2.000	85.211	1.6435	1.4210	1.1566	0.8209	0.98763		2.000	87.406	2.2812	1.7736	1.2862	0.7225	0.94526
1.30	2.000	53.474	1.1065	1.0749	1.0294	1.2244	0.99989		6.000	44.065	1.1030	1.0725	1.0284	1.4316	0.99990
	4.000	57.423	1.2334	1.1613	1.0621	1.1398	0.99906		8.000	46.543	1.2165	1.1500	1.0578	1.3615	0.99923
	6.000	63.459	1.4113	1.2775	1.1048	1.0274	0.99585		12.000	64.359	1.9668	1.6068	1.2241	0.9607	0.99739
	6.662	69.395	1.5608	1.3709	1.1386	0.9359	0.99108		12.113	66.589	2.0439	1.6489	1.2396	0.9213	0.96385
	6.000	75.372	1.6793	1.4423	1.1643	0.8636	0.98598		12.000	68.790	2.1147	1.6869	1.2527	0.8849	0.95849
	4.000	81.649	1.7634	1.4917	1.1822	0.8118	0.98169		10.000	75.995	2.3046	1.7855	1.2908	0.7854	0.94329
	2.000	86.058	1.7957	1.5103	1.1889	0.7918	0.97990		8.000	79.712	2.3746	1.8207	1.3042	0.7476	0.93725
	2.000	86.644							6.000	82.662	2.4155	1.8410	1.3121	0.7250	0.93363
	2.000	86.644							4.000	85.256	2.4404	1.8533	1.3168	0.7112	0.93141
	2.000	86.644							2.000	87.668	2.4540	1.8599	1.3194	0.7035	0.93018
1.35	2.000	50.634	1.1042	1.0733	1.0287	1.2774	0.99990								1.4821
	4.000	53.965	1.2238	1.1549	1.0596	1.1994	0.99916								0.99990
	6.000	58.232	1.3702	1.2512	1.0952	1.1089	0.99682								0.99923
	8.000	66.914	1.6327	1.4145	1.1543	0.9543	0.98812								0.99739
	8.048	68.470	1.6732	1.4387	1.1630	0.9307	0.98627								1.3414
	8.000	70.023	1.7114	1.4613	1.1712	0.9085	0.98440								1.2651
	6.000	78.660	1.8774	1.5569	1.2058	0.8111	0.97506								0.99375
	4.000	83.028	1.9283	1.5854	1.2163	0.7807	0.97182								1.1804
	2.000	86.644	1.9523	1.5988	1.2211	0.7662	0.97023								0.98738
	2.000	86.644													1.0758
	2.000	86.644													0.97615
	2.000	86.644													0.9198
	2.000	86.644													0.95362
	2.000	86.644													0.8014
	2.000	86.644													0.93367
	2.000	86.644													0.7515
	2.000	86.644													0.92496
	2.000	86.644													0.7229
	2.000	86.644													0.91995
	2.000	86.644													0.91673
	2.000	86.644													0.6928
	2.000	86.644													0.91470
	2.000	86.644													0.6862
	2.000	86.644													0.61356

Oblique Shock p1

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{M_2}{M_1}$	$\frac{p_{02}}{p_{01}}$	$\frac{\rho_2}{\rho_1}$	$\frac{p_2}{p_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{T_2}{T_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$	
													$T_2/T_1$	
1.60	2.000	40.724	1.1046	1.0289	1.5323	0.99990	1.70	6.000	84.848	3.1778	2.1865	1.4534	0.6547	0.85856
	4.000	42.931	1.2189	1.1516	1.0584	1.4638	0.99921	4.000	86.619	3.1933	2.1929	1.4562	0.6467	0.85695
	6.000	45.344	1.3446	1.2346	1.0891	1.3934	0.99736	2.000	88.325	3.2021	2.1965	1.4578	0.6421	0.85602
8.000	48.030	1.4843	1.3236	1.1215	1.3195	0.99376	1.75	2.000	36.689	1.1087	1.0765	1.0300	1.6816	0.99989
10.000	51.116	1.6430	1.4207	1.1565	1.2397	0.98766	0.97781	4.000	38.651	1.2271	1.1571	1.0605	1.6133	0.99913
12.000	60.537	2.0974	1.6777	1.2502	1.0232	0.95990	0.94204	6.000	40.756	1.3561	1.2421	1.09173	1.5441	
14.000	14.652	65.828	2.3192	1.7929	1.2936	0.9188	0.94204	8.000	43.034	1.4973	1.3317	1.1244	1.4733	0.99334
14.000	70.895	2.5000	1.8824	1.3281	0.8320	0.92598	0.91256	10.000	45.531	1.6529	1.4266	1.1586	1.3995	0.98721
12.000	75.900	2.6428	1.9504	1.3550	0.7611	0.91256	0.90574	12.000	48.319	1.8263	1.5279	1.1953	1.3210	0.97814
10.000	79.102	2.7132	1.9831	1.3682	0.7250	0.90574	0.90139	14.000	51.547	2.0245	1.6384	1.2357	1.2348	0.96524
8.000	81.691	2.7576	2.0035	1.3764	0.7018	0.89848	0.89848	16.000	55.589	2.2652	1.7654	1.2831	1.1329	0.94660
6.000	83.967	2.7870	2.0168	1.3819	0.6862	0.89660	0.8761	18.000	62.944	2.6670	1.9617	1.3595	0.91023	
4.000	86.061	2.8059	2.0254	1.3854	0.6761	0.89554	0.6703	18.121	65.134	2.7745	2.0112	1.3795	0.9189	0.89971
2.000	88.054	2.8166	2.0302	1.3873	0.6703	0.89554		18.000	67.269	2.8728	2.0554	1.3977	0.8766	0.88991
1.65	2.000	39.267	1.1058	1.0744	1.0292	1.5823	0.99990	14.000	76.988	3.2251	2.1651	1.4441	0.7635	0.86389
	4.000	41.377	1.2212	1.1531	1.0590	1.5140	0.99919	12.000	79.465	3.2868	2.2312	1.4731	0.6878	0.84714
	6.000	43.665	1.3475	1.2365	1.0898	1.4444	0.99730	10.000	81.570	3.3295	2.2484	1.4808	0.6669	0.84266
	8.000	46.181	1.4869	1.3252	1.1221	1.3720	0.99367	8.000	83.451	3.3598	2.2606	1.4862	0.6518	0.83947
	10.000	49.007	1.6429	1.4206	1.1665	1.2952	0.98766	6.000	85.190	3.3811	2.2691	1.4901	0.6409	0.83722
	12.000	52.312	1.8224	1.5257	1.1945	1.2104	0.97837	4.000	86.838	3.3954	2.2748	1.4926	0.6337	0.83571
	14.000	56.541	2.0441	1.6490	1.2396	1.1090	0.96384	2.000	88.432	3.4036	2.2780	1.4941	0.6295	0.83485
	15.855	65.547	2.4653	1.8655	1.3215	0.9184	0.92915							
	14.000	73.864	2.7642	2.0065	1.3776	0.7782	0.90073	1.80	2.000	35.538	1.1104	1.0776	1.7312	0.99988
	12.000	77.411	2.8587	2.0491	1.3951	0.7317	0.89132	1.80	2.000	37.444	1.2306	1.1594	1.0613	1.6624
	10.000	80.102	2.9157	2.0744	1.4056	0.7029	0.88557	4.000	39.481	1.3615	1.2455	1.0931	1.5932	0.99701
	8.000	82.389	2.9539	2.0911	1.4126	0.6833	0.88169	6.000	41.673	1.5044	1.3360	1.1260	1.5225	0.99310
	6.000	84.446	2.9798	2.1024	1.4174	0.6697	0.87904	8.000	44.057	1.6611	1.4315	1.1604	1.4494	0.98683
	4.000	86.364	2.9968	2.1097	1.4205	0.6607	0.87730	10.000	46.686	1.8345	1.5326	1.1970	1.3725	0.97766
	2.000	88.200	3.0065	2.1139	1.4222	0.6556	0.87631	12.000	49.661	2.0295	1.6411	1.2367	1.2896	0.96489
1.70	2.000	37.927	1.1072	1.0295	1.6320	0.99989	18.000	57.995	2.5516	1.9072	1.3379	1.0766	1.3725	0.97766
	4.000	39.957	1.2239	1.1550	1.5638	0.99916	19.183	64.987	2.9376	2.0839	1.4096	0.9195	0.88335	
	6.000	42.145	1.3514	1.2390	1.0907	1.4946	0.98722	18.000	71.424	3.2297	2.2079	1.4628	0.7956	0.85313
	8.000	44.528	1.4914	1.3280	1.1231	1.4232	0.99353	16.000	75.324	3.3707	2.2650	1.4882	0.7327	0.83832
	10.000	47.167	1.6466	1.4228	1.1573	1.3482	0.98750	14.000	78.020	3.4505	2.2965	1.5025	0.6958	0.82990
	12.000	50.168	1.8216	1.1943	1.2674	0.97841	12.000	80.214	3.5041	2.3174	1.5121	0.6703	0.82423	
	14.000	53.771	2.0273	1.2362	1.1757	0.96504	10.000	82.128	3.5424	2.3322	1.5189	0.6518	0.82018	
	16.000	58.794	2.2999	1.7831	1.2898	1.0569	0.94369	8.000	83.865	3.5702	2.3428	1.5239	0.6381	0.81725
	17.012	65.319	2.6171	1.9383	1.3502	0.9185	0.91502	6.000	85.485	3.5899	2.3503	1.5274	0.6283	0.81516
	16.000	71.426	2.8629	2.0510	1.3959	0.8077	0.89090	4.000	87.028	3.6032	2.3554	1.5298	0.6216	0.81376
	14.000	75.670	2.9984	2.1104	1.4208	0.7439	0.87713	2.000	88.525	3.6108	2.3583	1.5311	0.6178	0.81295
	12.000	78.555	3.0722	2.1421	1.4342	0.7080	0.86953							
	10.000	80.906	3.1208	2.1626	1.4431	0.6838	0.86450							
	8.000	82.965	3.1544	2.1767	1.4492	0.6667	0.86100							

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{\rho_2}{\rho_1}$	$M_2$	$\frac{T_2}{T_1}$	$\frac{p_{02}}{p_{01}}$	
1.85	2.000	34.466	1.1121	1.0788	1.0309	0.99988	1.7805	0.99988	1.95	2.000	32.528	1.1160	1.0319	1.8790	
	4.000	36.323	1.2343	1.1619	1.0623	0.99905	1.7114	0.99905	4.000	4.000	34.304	1.2424	1.1674	1.8085	
	6.000	38.302	1.3672	1.2492	1.0945	0.99689	1.6418	0.99689	6.000	6.000	36.191	1.3801	1.2575	1.99221	
	8.000	40.424	1.5123	1.3409	1.1278	0.99284	1.5711	0.99284	8.000	8.000	38.204	1.5302	1.3521	1.1318	
	10.000	42.717	1.6709	1.4373	1.1625	0.98638	1.4983	0.98638	10.000	10.000	40.360	1.6938	1.4509	1.1674	
	12.000	45.223	1.8453	1.5388	1.1922	0.97701	1.4224	0.97701	12.000	12.000	42.688	1.8726	1.5542	1.2049	
	14.000	48.014	2.0395	1.6465	1.2387	0.96417	1.3415	0.96417	14.000	14.000	45.230	2.0693	1.6625	1.2446	
	16.000	51.232	2.2697	1.7631	1.2822	0.94697	1.2524	0.94697	16.000	16.000	48.059	2.2879	1.7770	1.2875	
	18.000	55.227	2.5275	1.8956	1.3333	0.92345	1.1476	0.92345	18.000	18.000	51.320	2.5368	1.9001	1.3351	
	20.000	62.099	2.9519	2.0902	1.4123	0.9818	0.9818	0.9818	20.000	20.000	55.381	2.8378	2.0397	1.3913	
	20.198	64.872	3.1062	2.1565	1.4404	0.9205	0.86601	0.86601	22.000	22.000	62.860	3.3464	2.2553	1.89342	
	20.000	67.544	3.2437	2.2136	1.4653	0.8648	0.85167	0.85167	22.092	22.092	64.716	3.4603	2.3003	0.84087	
	18.000	73.440	3.5019	2.3165	1.5117	0.7560	0.82446	0.82446	22.000	22.000	66.523	3.5655	2.3410	0.94470	
	16.000	76.511	3.6090	2.3576	1.5308	0.7085	0.81314	0.81314	20.000	20.000	72.926	3.8872	2.4601	1.5953	
	14.000	78.861	3.6772	2.3833	1.5429	0.6773	0.80593	0.80593	18.000	18.000	75.964	4.0086	2.5030	1.7045	
	12.000	80.844	3.7252	2.4011	1.5514	0.6548	0.80088	0.80088	16.000	16.000	78.253	4.0857	2.5297	1.77114	
	10.000	82.606	3.7601	2.4140	1.5576	0.6381	0.79719	0.79719	14.000	14.000	80.165	4.1401	2.5484	1.6710	
	8.000	84.222	3.7858	2.4234	1.5622	0.6257	0.79449	0.79449	12.000	12.000	81.849	4.1804	2.5620	1.6317	
	6.000	85.740	3.8042	2.4301	1.5655	0.6166	0.79255	0.79255	10.000	10.000	83.381	4.2106	2.5722	1.6142	
	4.000	87.193	3.8167	2.4346	1.5677	0.6105	0.79124	0.79124	8.000	8.000	84.808	4.2333	2.5798	1.6036	
	2.000	88.606	3.8239	2.4373	1.5689	0.6069	0.79048	0.79048	6.000	6.000	86.163	4.2497	2.5853	1.5957	
									4.000	4.000	87.467	4.2609	2.5890	1.6458	
									2.000	2.000	88.741	4.2674	2.5912	1.6469	
												1.6469	0.5872	0.74441	
1.90	2.000	33.466	1.1140	1.0801	1.0314	0.99987	1.8298	0.99987	2.00	2.000	31.647	1.1180	1.0829	1.9280	
	4.000	35.279	1.2382	1.0633	1.0633	0.99901	1.7600	0.99901	4.000	4.000	33.390	1.2468	1.1702	1.0324	
	6.000	37.209	1.3735	1.2533	1.0959	0.99675	1.6901	0.99675	6.000	6.000	35.241	1.3871	1.2620	0.9986	
	8.000	39.272	1.5209	1.3463	1.1297	0.99254	1.6191	0.99254	8.000	8.000	37.210	1.5400	1.3581	0.99644	
	10.000	41.490	1.6818	1.4438	1.1649	1.5464	0.98586	1.4709	0.97624	10.000	10.000	39.314	1.7066	1.4584	0.99186
	12.000	43.898	1.8582	1.5460	1.2019	1.4720	0.97624	1.3913	0.96319	12.000	12.000	41.575	1.8884	1.5631	1.1702
	14.000	46.550	2.0530	1.6538	1.2414	1.3913	0.96319	1.2844	0.94605	14.000	14.000	44.029	2.0876	1.6724	1.2081
	16.000	49.544	2.2718	1.7688	1.3052	0.94605	1.3052	0.94605	16.000	16.000	46.731	2.3076	1.7870	1.2483	
	18.000	53.095	2.5263	1.8951	1.3231	1.2077	0.92356	1.2077	16.000	16.000	49.785	2.5546	1.9086	1.2913	
	20.000	57.900	2.8557	2.0477	1.3946	1.0835	0.89162	1.4720	1.0216	18.000	18.000	53.423	2.8429	2.0420	1.4034
	21.167	64.783	3.2805	2.2886	1.4720	0.9216	0.84781	1.4720	0.9216	20.000	20.000	58.457	3.2228	2.2051	1.4304
	20.000	71.057	3.6012	2.3546	1.5294	0.7935	0.81397	1.2844	0.92356	22.000	22.000	58.457	3.2228	2.2051	1.4304
	18.000	74.861	3.7578	2.4131	1.5572	0.7274	0.79744	1.2844	0.92356	22.974	22.974	64.669	3.6458	2.3715	1.5373
	16.000	77.463	3.8466	2.4455	1.5729	0.6884	0.78810	0.78810	22.000	22.000	70.332	3.9714	2.4899	1.5950	
	14.000	79.565	3.9068	2.4671	1.5836	0.6611	0.78178	0.78178	20.000	20.000	74.270	4.1570	2.5541	1.6276	
	12.000	81.383	3.9504	2.4826	1.5913	0.6409	0.77721	0.77721	18.000	18.000	76.862	4.2589	2.5883	1.6454	
	10.000	83.020	3.9828	2.4940	1.5970	0.6257	0.77383	0.77383	16.000	16.000	78.921	4.3277	2.6110	1.6574	
	8.000	84.534	4.0068	2.5024	1.6012	0.6142	0.771133	0.771133	14.000	14.000	80.684	4.3777	2.6274	1.6662	
	6.000	85.965	4.0241	2.5084	1.6042	0.6058	0.76953	0.76953	12.000	12.000	82.257	4.4153	2.6396	1.6727	
	4.000	87.338	4.0359	2.5125	1.6063	0.6001	0.76830	0.76830	10.000	10.000	83.700	4.4438	2.6487	1.6777	
	2.000	88.677	4.0428	2.5149	1.6075	0.5967	0.76759	0.76759	8.000	8.000	85.052	4.4653	2.6556	1.6815	
											6.000	86.339	4.4810	2.6606	1.6842
											4.000	87.582	4.4917	2.6640	1.6861
											2.000	88.798	4.4977	2.6660	1.6873

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{M_2}{M_1}$	$\frac{p_{02}}{p_{01}}$	$\frac{T_2}{T_1}$	$\frac{\rho_2}{\rho_1}$	$M_2$	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	$\frac{\rho_2}{\rho_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$
2.05	2.000	4.000	30.816	1.1200	1.0843	1.030	1.9771	0.99985	2.10	4.000	87.778	4.9706	2.8097
			32.532	1.2512	1.1732	1.065	1.9050	0.99885	1.8330	0.99627	88.894	4.9764	2.8113
			6.000	34.350	1.3943	1.2666	1.1008	0.99985	1.1362	1.7605	0.99148	2.15	2.000
			8.000	36.281	1.5502	1.3644	1.4664	1.1730	1.6868	0.98396	4.000	29.293	1.1243
			10.000	38.341	1.7201	1.4664	1.5726	1.2116	1.6111	0.97330	6.000	30.960	1.2606
			12.000	40.547	1.9053	1.6831	1.2076	1.2522	1.5326	0.95914	8.000	32.725	1.4094
			14.000	42.928	2.3300	1.7983	1.2956	1.4500	1.94112	0.94112	10.000	34.596	1.5719
			16.000	45.528	2.5774	1.9195	1.3427	1.3614	0.91878	1.3614	12.000	36.584	1.7490
			18.000	48.428	4.4215	2.6700	2.0497	1.3953	1.2630	0.89120	14.000	38.702	1.9417
			20.000	51.785	2.8600	2.1980	3.2057	1.4585	1.1444	0.85565	16.000	40.971	2.1518
			22.000	56.032	2.4419	3.8367	23.814	1.5712	0.9257	0.78913	18.000	43.422	2.3813
			22.000	72.193	4.2777	2.5946	2.6416	1.6487	0.7626	0.74336	24.000	46.104	2.6337
			20.000	75.324	2.6736	1.6738	1.6738	0.7056	0.72876	20.000	49.106	2.9150	
			18.000	77.614	4.5107	2.6894	2.6894	0.6694	0.6688	0.71981	22.000	52.618	2.3284
			16.000	79.498	4.5734	2.6898	2.7043	1.7003	0.64422	0.71356	24.000	57.217	3.6452
			14.000	81.138	4.6199	2.7043	2.7084	0.6219	0.70894	0.70894	25.376	64.616	4.2352
			12.000	82.617	4.6553	2.7152	2.7152	1.7145	0.60622	0.70545	24.000	71.164	4.6641
			10.000	83.983	4.6824	2.7236	2.7236	1.7192	0.5939	0.70278	22.000	74.564	4.8442
			8.000	85.269	4.7029	2.7299	2.7299	1.7228	0.5846	0.70077	20.000	76.920	4.9500
			6.000	86.497	4.7179	2.7344	2.7344	1.7254	0.5776	0.69930	18.000	78.817	5.0234
			4.000	87.685	4.7283	2.7376	2.7376	1.7272	0.5728	0.69829	16.000	80.444	5.0776
			2.000	88.849	4.7343	2.7394	2.7394	1.7282	0.5700	0.69770	14.000	81.896	5.1191
2.10	2.000	4.000	30.033	41.912	1.1222	1.035	1.035	1.035	0.99984	0.99980	8.000	85.639	5.1951
			6.000	33.513	1.2558	1.1763	1.0676	1.9530	0.99880	0.99880	6.000	86.767	5.2091
			8.000	35.412	1.4017	1.2714	1.1025	1.8801	0.99609	0.99609	4.000	87.862	5.2187
			10.000	37.433	1.7342	1.4746	1.4746	1.1386	1.8069	0.99108	2.000	88.936	5.2244
			12.000	39.592	1.9230	1.5825	1.2152	1.6564	0.97216	1.5777	2.0260	28.592	1.1266
			14.000	41.912	2.1290	1.6944	1.2565	1.3004	1.4954	0.93899	4.000	30.238	1.2654
			16.000	44.430	2.3547	1.8107	1.3478	1.4078	1.4078	0.91626	6.000	31.981	1.4173
			18.000	47.210	2.6041	1.9322	2.0607	1.3999	1.3122	0.88870	8.000	33.827	1.5832
			20.000	50.365	2.8848	2.2019	1.4602	1.2019	0.85466	10.000	35.785	1.7641	
			22.000	54.169	3.2152	2.3820	1.5424	1.0493	0.80628	12.000	37.869	1.9611	
			24.000	59.767	3.6739	2.5116	1.6058	0.9273	0.76858	14.000	40.095	2.1756	
			24.614	64.621	4.0332	2.6098	1.6568	0.8245	0.73867	16.000	42.489	2.4095	
			24.000	69.104	4.3238	2.6870	1.6987	0.7345	0.71445	18.000	45.092	2.6658	
			22.000	73.521	4.5644	2.7244	1.7197	0.6870	0.70251	20.000	47.975	2.9494	
			20.000	76.189	4.6852	2.7488	1.7336	0.6543	0.69468	22.000	51.277	3.2704	
			18.000	78.257	4.7652	2.7662	1.7436	0.6299	0.68906	24.000	55.356	3.6552	
			16.000	80.001	4.8232	2.8669	1.7512	0.6111	0.68484	26.000	62.695	4.2918	
			14.000	81.539	4.8669	2.7792	0.9806	0.5964	0.68162	26.103	64.620	4.4426	
			12.000	82.938	4.9006	2.7892	1.7570	0.5849	0.67914	26.000	66.480	4.5807	
			10.000	84.237	4.9264	2.7968	1.7615	0.5849	0.67726	24.000	72.560	4.9728	
			8.000	85.463	4.9461	2.8025	1.7649	0.5760	0.67588	22.000	75.420	5.1222	

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{\rho_2}{\rho_1}$	$M_2$	$\frac{T_2}{T_1}$	$\frac{p_{02}}{p_{01}}$
2.20	20.000	77.549	5.2175	2.8799	1.8117	0.6568	0.65185	2.30	16.000	40.816	2.4701	1.8678	1.3224	1.6676
	18.000	79.308	5.2856	2.8987	1.8234	0.6296	0.64562		18.000	43.299	2.7360	1.9936	1.3724	1.5804
	16.000	80.839	5.3369	2.9127	1.8323	0.6086	0.64096		20.000	46.007	3.0276	2.1230	1.4261	1.4885
14.000	82.216	5.3764	2.9235	1.8391	0.5921	0.65739	0.5921		22.000	49.026	3.3514	2.2573	1.4847	1.3894
12.000	83.483	5.4073	2.9318	1.8444	0.5789	0.63462	0.63462		24.000	52.536	3.7216	2.3998	1.5508	1.2788
10.000	84.670	5.4313	2.9382	1.8485	0.5686	0.63247	0.63247		26.000	57.077	4.1819	2.5625	1.6319	1.1425
8.000	85.798	5.4497	2.9431	1.8517	0.5605	0.63083	0.63083		27.454	64.653	4.8739	2.7813	1.7524	0.9338
6.000	86.883	5.4633	2.9468	1.8540	0.5545	0.62962	0.62962		26.000	71.264	5.3682	2.9212	1.8377	0.7743
4.000	87.938	5.4727	2.9493	1.8556	0.5503	0.62879	0.62879		24.000	74.512	5.5649	2.9736	1.8714	0.7060
2.000	88.973	5.4782	2.9507	1.8565	0.5479	0.62830	0.62830		22.000	76.770	5.6817	3.0039	1.8915	0.6635
2.25	2.000	27.926	1.1288	1.0903	1.0353	2.1725	0.99982		18.000	80.133	5.8238	3.0399	1.9158	0.6092
	4.000	29.555	1.2703	1.1859	1.0712	2.0962	0.98861		16.000	81.509	5.8705	3.0515	1.9238	0.5945
	6.000	31.277	1.4254	1.2864	1.1080	2.0203	0.99548		14.000	82.764	5.9071	3.0606	1.9301	0.5757
	8.000	33.102	1.5949	1.3916	1.1461	1.9443	0.98973		12.000	83.928	5.9360	3.0677	1.9350	0.5638
	10.000	35.034	1.7798	1.5011	1.1856	1.8674	0.98079		10.000	85.026	5.9586	3.0732	1.9389	0.5543
	12.000	37.088	1.9812	1.6147	1.2270	1.7891	0.98827		8.000	86.074	5.9761	3.0775	1.9419	0.5469
	14.000	39.277	2.2004	1.7319	1.2705	1.7088	0.95189		6.000	87.085	5.9890	3.0807	1.9441	0.5413
	16.000	41.623	2.4392	1.8527	1.3166	1.6257	0.93152		4.000	88.070	5.9980	3.0828	1.9456	0.5374
	18.000	44.161	2.7000	1.9770	1.3657	1.5388	0.90703		2.000	89.039	6.0033	3.0841	1.9465	0.5352
	20.000	46.948	2.9871	2.1055	1.4187	1.4466	0.87829							
	22.000	50.091	3.3085	2.2400	1.4770	1.4770	0.84486	2.35	2.000	26.692	1.1334	1.0935	1.0365	2.2698
	24.000	53.837	3.6830	2.3854	1.5440	1.2318	0.80532		4.000	28.289	1.2804			
	26.000	59.122	4.1839	2.5632	1.6323	1.0792	0.75298		6.000	29.979	1.4420			
	26.795	64.633	4.6556	2.7513	1.7145	0.9321	0.70542		8.000	31.765	1.6189			
	26.000	69.627	5.0238	2.8250	1.7783	0.8115	0.66991		10.000	33.657	1.8124			
	24.000	73.634	5.2707	2.8946	1.8209	0.7254	0.64698		12.000	35.662	2.0232			
	22.000	76.145	5.4009	2.9301	1.8433	0.6775	0.63519		14.000	37.790	2.2526			
	20.000	78.098	5.4884	2.9534	1.8583	0.6441	0.62739		16.000	40.060	2.5021			
	18.000	79.744	5.5523	2.9703	1.8693	0.6189	0.62175		18.000	42.497	2.7736			
	16.000	81.192	5.6011	2.9830	1.8776	0.5993	0.61749		20.000	45.140	3.0108			
	14.000	82.504	5.6391	2.9929	1.8842	0.5836	0.61418		22.000	48.059	3.2759			
	12.000	83.716	5.6688	3.0006	1.8893	0.5711	0.61161		24.000	51.393	3.7677			
	10.000	84.856	5.6921	3.0065	1.8932	0.5612	0.60960		26.000	55.500	4.2092			
	8.000	85.942	5.7100	3.0111	1.8963	0.5535	0.60806		28.000	62.973	4.9459			
	6.000	86.988	5.7233	3.0145	1.8986	0.5477	0.60692		28.082	64.679	5.0977			
	4.000	88.007	5.7324	3.0168	1.9002	0.5437	0.60614		28.000	66.328	5.2377			
	2.000	89.008	5.7378	3.0182	1.9011	0.5413	0.60568		26.000	72.454	5.6907			
2.30	2.000	27.294	1.1311	1.0919	1.0359	2.2212	0.99981		22.000	77.317	5.9657	3.0750	1.9401	0.6510
	4.000	28.906	1.2753	1.1892	1.0724	2.1437	0.99854		20.000	79.014	6.0423			
	6.000	30.611	1.4336	1.2916	1.1099	2.0667	0.99526		18.000	80.483	6.1001			
	8.000	32.415	1.6068	1.3988	1.1487	1.9896	0.98923		16.000	81.798	6.1451			
	10.000	34.326	1.7959	1.5104	1.1890	1.9117	0.97989		14.000	83.001	6.1806			
	12.000	36.354	2.0019	1.6260	1.2311	1.8325	0.96684		12.000	84.122	6.2087			
	14.000	38.510	2.2261	1.7452	1.2755	1.7514	0.94982		8.000	85.182	6.2308			

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$	
2.35	6.000	87.174	6.2606	3.1453	1.9904	0.5353	0.562772	2.45	26.000	53.045	4.3053	2.6037	1.6535	0.74055
	4.000	88.129	6.2694	3.1474	1.9919	0.5315	0.56203		28.000	57.780	4.8455	2.7729	1.7475	1.1385
	2.000	89.068	6.2745	3.1486	1.9928	0.5293	0.56162		29.253	64.744	5.5614	2.9727	1.8708	0.9386
2.40	2.000	26.120	1.1358	1.0951	1.0371	2.3184	0.99979	24.000	70.828	6.0810	3.1029	1.9598	0.7837	0.57709
	4.000	27.702	1.2856	1.1960	1.0749	2.2383	0.98839	22.000	74.185	6.3161	3.1582	1.9999	0.7082	0.55836
	6.000	29.377	1.4505	1.3023	1.1138	2.1589	0.98478	20.000	79.752	6.6146	3.2101	2.0230	0.6623	0.54787
	8.000	31.149	1.6314	1.4137	1.1540	2.0794	0.98818	18.000	81.089	6.6682	3.2372	2.0599	0.5842	0.53157
	10.000	33.023	1.8292	1.5295	1.1959	1.9994	0.97797	16.000	82.299	6.7105	3.2464	2.0671	0.5681	0.52845
	12.000	35.007	2.0450	1.6495	1.2398	1.9181	0.96377	14.000	83.416	6.7442	3.2536	2.0728	0.5550	0.52599
	14.000	37.112	2.2798	1.7729	1.2860	1.8350	0.94538	12.000	84.462	6.7710	3.2594	2.0774	0.5444	0.52403
	16.000	39.351	2.5351	1.8993	1.3348	1.7497	0.92274	10.000	85.455	6.7923	3.2640	2.0810	0.5359	0.52249
	18.000	41.748	2.8128	2.0285	1.3866	1.6613	0.85952	8.000	86.408	6.8088	3.2675	2.0838	0.5292	0.52129
	20.000	44.336	3.1155	2.1604	1.4421	1.5689	0.86505	6.000	87.331	6.8211	3.2701	2.0859	0.5242	0.52041
	22.000	47.174	3.4480	2.2955	1.5021	1.4709	0.83015	4.000	88.232	6.8296	3.2719	2.0873	0.5207	0.51979
	24.000	50.371	3.8196	2.4357	1.5682	1.3644	0.79093	2.000	89.119	6.8346	3.2730	2.0882	0.5186	0.51943
	26.000	54.184	4.2521	2.5861	1.6442	1.2426	0.74598							
	28.000	59.656	4.8382	2.7707	1.7462	1.0779	0.68761	2.50	2.000	25.050	1.1405	1.0984	1.0384	2.4155
	28.681	64.710	5.3269	2.9100	1.8305	0.9370	0.64187		4.000	26.609	1.2961	1.2029	1.0775	2.3326
	28.000	69.291	5.7130	3.0119	1.8968	0.8201	0.60781		6.000	28.259	1.4679	1.3133	1.1177	2.2505
	26.000	73.400	6.0048	3.0845	1.9468	0.7260	0.56331		8.000	30.005	1.6568	1.4289	1.1595	2.1685
	24.000	75.889	6.1539	3.1203	1.9722	0.6751	0.57121		10.000	31.851	1.8639	1.5493	1.2031	2.0859
	22.000	77.803	6.2534	3.1436	1.9892	0.6397	0.56329		12.000	33.802	2.0900	1.6737	1.2488	2.0022
	20.000	79.402	6.3260	3.1605	2.0016	0.6129	0.55758		14.000	35.866	2.3364	1.8015	1.2969	0.94057
	18.000	80.800	6.3816	3.1732	2.0111	0.5919	0.55326		16.000	38.057	2.6042	1.9322	1.3478	1.8295
	16.000	82.059	6.4251	3.1831	2.0185	0.5751	0.54990		18.000	40.389	2.8949	2.0652	1.4018	1.7394
	14.000	83.217	6.4596	3.1909	2.0244	0.5615	0.54726		20.000	42.890	3.2109	2.2002	1.4594	1.6458
	12.000	84.299	6.4870	3.1971	2.0290	0.5505	0.54517		22.000	45.602	3.5558	2.3373	1.5213	1.5475
	10.000	85.324	6.5087	3.2019	2.0327	0.5416	0.54352		24.000	48.600	3.9361	2.4775	1.5887	1.4426
	8.000	86.306	6.5254	3.2057	2.0356	0.5348	0.54225		26.000	52.036	4.3657	2.6235	1.6641	1.3268
	6.000	87.255	6.5379	3.2085	2.0377	0.5296	0.54131		28.000	56.335	4.8844	2.7844	1.7542	1.1888
	4.000	88.182	6.5466	3.2104	2.0392	0.5260	0.54065		29.797	64.782	5.8014	3.0342	1.9120	0.9402
	2.000	89.094	6.5517	3.2115	2.0400	0.5238	0.54027		28.000	71.949	6.4249	3.1831	2.0185	0.7573
2.45	2.000	25.572	1.1381	1.0968	1.0377	2.3670	0.98978		24.000	74.856	6.6273	3.2282	2.0529	0.53460
	4.000	27.143	1.2908	1.1994	1.0762	2.2855	0.99831		22.000	76.939	6.7526	3.2555	2.0742	0.6509
	6.000	28.805	1.4591	1.3078	1.1157	2.2048	0.99453		20.000	80.070	6.9082	3.2744	2.0893	0.51894
	8.000	30.563	1.6440	1.4212	1.1567	2.1241	0.98761		18.000	81.353	6.9602	3.2885	2.1007	0.5962
	10.000	32.422	1.8463	1.5393	1.194	2.0428	0.97695		16.000	82.518	7.0014	3.3080	2.1095	0.51048
	12.000	34.388	2.0672	1.6615	1.2442	1.9603	0.96215		14.000	83.598	7.0343	3.3148	2.1221	0.5489
	14.000	36.472	2.3078	1.7871	1.2914	1.8762	0.94302		12.000	84.612	7.0607	3.3202	2.1266	0.5387
	16.000	38.685	2.5692	1.9156	1.3412	1.7898	0.91955		10.000	85.576	7.0816	3.3245	2.1301	0.5304
	18.000	41.047	2.8532	2.0466	1.3941	1.7006	0.89187		8.000	86.502	7.0979	3.3278	2.1329	0.5240
	20.000	43.588	3.1623	2.1800	1.4506	1.6077	0.86018		6.000	87.400	7.1100	3.3303	2.1350	0.5191
	22.000	46.358	3.5007	2.3160	1.5115	1.5097	0.82459		4.000	88.277	7.1184	3.3320	2.1364	0.49947
	24.000	49.445	3.8759	2.4560	1.5781	1.4042	0.786502		2.000	89.142	7.1234	3.3330	2.1372	0.49913

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{T_2}{T_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{\rho_2}{\rho_1}$	$M_2$	$\frac{T_2}{T_1}$	$\frac{p_{02}}{p_{01}}$	
2.55	2.000	24.550	1.1429	1.1001	1.0390	2.4639	0.99976	2.60	30.814	64.866	6.2972	3.1538	0.55984	
	4.000	26.099	1.3015	1.2065	1.0788	2.3796	0.99814		30.000	69.778	6.7777	3.2609	0.8111	
	6.000	27.739	1.4768	1.3189	1.1198	2.2961	0.99399		28.000	73.590	7.0906	3.3263	0.52354	
	8.000	29.474	1.6699	1.4367	1.1623	2.2128	0.98642		26.000	75.955	7.2555	3.3596	0.50138	
	10.000	31.307	1.8817	1.5593	1.2067	2.1288	0.97479		24.000	77.778	7.3665	2.1785	0.49015	
	12.000	33.244	2.1133	1.6861	1.2534	2.0438	0.95871		22.000	79.299	7.4481	2.3974	0.48276	
	14.000	35.293	2.3656	1.8162	1.3025	1.9573	0.93803		20.000	80.626	7.5108	3.4095	0.47742	
	16.000	37.463	2.6399	1.9490	1.3545	1.8687	0.91283		18.000	81.815	7.5602	3.4189	0.47336	
	18.000	39.770	2.9378	2.0840	1.4097	1.7776	0.89333		16.000	82.906	7.5997	3.4264	0.46768	
	20.000	42.236	3.2611	2.2207	1.4685	1.6832	0.84985		14.000	83.922	7.6316	3.4324	0.46566	
	22.000	44.899	3.6130	2.3591	1.5315	1.5845	0.81272		12.000	84.879	7.6572	3.4372	0.46405	
	24.000	47.822	3.9995	2.4998	1.5999	1.4797	0.772709		10.000	85.792	7.6775	3.4411	0.46277	
	26.000	51.130	4.4319	2.6449	1.6756	1.3655	0.727272		8.000	86.671	7.6934	3.4440	0.46178	
	28.000	55.131	4.9401	2.8007	1.7638	1.2334	0.67784		6.000	87.524	7.7053	3.4462	0.46104	
	30.000	61.449	5.6866	3.0051	1.8923	1.0385	0.61007		4.000	88.359	7.7135	3.4478	0.46053	
	30.317	64.823	6.0466	3.0946	1.9539	0.9418	0.57989		2.000	89.183	7.7184	3.4487	0.46022	
	30.000	67.966	6.3519	3.1664	2.0060	0.8568	0.55557						0.5045	
	28.000	72.844	6.7595	3.2569	2.0754	0.7364	0.52487							
	26.000	75.440	6.9402	3.2952	2.1061	0.6793	0.51190							
	24.000	77.380	7.0575	3.3195	2.1260	0.6405	0.50368							
	22.000	78.978	7.1423	3.3368	2.1404	0.6115	0.49783							
	20.000	80.360	7.2068	3.3499	2.1514	0.5887	0.49343							
	18.000	81.594	7.2575	3.3600	2.1600	0.5703	0.49002							
	16.000	82.720	7.2978	3.3680	2.1668	0.5554	0.48732							
	14.000	83.766	7.3301	3.3744	2.1723	0.5432	0.48517							
	12.000	84.750	7.3561	3.3795	2.1767	0.5333	0.48345							
	10.000	85.688	7.3767	3.3835	2.1802	0.5253	0.48209							
	8.000	86.590	7.3927	3.3866	2.1829	0.5190	0.48104							
	6.000	87.464	7.4047	3.3890	2.1849	0.5142	0.48025							
	4.000	88.320	7.4131	3.3906	2.1864	0.5109	0.47971							
	2.000	89.163	7.4180	3.3916	2.1872	0.5090	0.47939							
	24.071	25.611	1.3070	1.2100	1.0801	2.4265	0.99805	2.5123	0.99975	30.000	57.877	5.7097	3.0110	0.8963
	27.241	1.4858	1.3245	1.1218	1.23416	2.2568	0.98579	31.288	64.910	6.5531	3.2118	2.0403	0.9447	0.54016
	6.000	28.966	1.6831	1.4445	1.1651	2.1715	0.97365	30.000	70.983	7.1564	3.3397	2.1428	0.7814	0.49687
	8.000	30.789	1.8998	1.5695	1.2105	2.0852	0.96690	28.000	74.230	7.4211	3.3922	2.1877	0.7039	0.47918
	10.000	32.714	1.6986	1.2369	1.2980	1.9973	0.93541	26.000	76.415	7.5742	3.4216	2.2137	0.6565	0.46930
	12.000	34.749	2.3955	1.8311	1.3082	1.9075	0.90930	24.000	78.138	7.6801	3.4415	2.2316	0.6224	0.46262
	14.000	36.901	2.6767	1.9662	1.3613	1.9075	0.90930	22.000	79.592	7.7589	3.4562	2.2449	0.5962	0.46677
	16.000	39.185	2.9817	2.1032	1.4177	1.8152	0.87884	20.000	80.870	7.8200	3.4674	2.2553	0.5752	0.45396
	20.000	41.621	3.3126	2.2417	1.4778	1.7199	0.84443	18.000	82.020	7.8684	3.4763	2.2634	0.5582	0.45101
	22.000	44.242	3.6723	2.3814	1.5421	1.6205	0.80645	16.000	83.079	7.9073	3.4833	2.2700	0.5442	0.44866
	24.000	47.102	4.0658	2.5229	1.6116	1.5157	0.76520	14.000	84.066	7.9387	3.4890	2.2753	0.5327	0.44677
	26.000	50.305	4.5028	2.6675	1.6880	1.4025	0.72060	12.000	84.998	7.9640	3.4935	2.2796	0.5234	0.44526
	28.000	54.088	5.0067	2.8201	1.7754	1.2744	0.67151	8.000	86.746	7.9999	3.4972	2.2830	0.5158	0.44406
	30.000	59.352	5.6706	3.0010	1.8896	1.1062	0.61145	6.000	87.579	8.0116	3.5021	2.2877	0.5053	0.44242

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{\rho_2}{\rho_1}$	$M_2$	$\frac{T_2}{T_1}$	$\frac{p_{02}}{p_{01}}$
2.65	4.000	88.396	8.0198	3.5036	2.2891	0.5021	0.44194	2.75	24.000	45.225	4.2794	2.5951	1.6490	1.6181
	2.000	89.200	8.0247	3.5044	2.2899	0.5003	0.44165		26.000	48.206	4.7375	2.7404	1.7288	1.5056
2.70	2.000	23.173	1.1503	1.1051	1.0409	2.6090	0.99972	30.000	55.674	5.2490	5.8507	3.0466	1.9204	1.2416
	4.000	24.696	1.3179	1.2172	1.0827	2.5201	0.99786	32.000	62.549	6.7812	3.2616	2.0791	1.0209	0.52329
2.70	6.000	26.311	1.5042	1.3360	1.1260	2.4321	0.99311	32.173	65.002	7.0807	3.3243	2.1300	0.9476	0.50207
	8.000	28.019	1.7102	1.4605	1.1709	2.3444	0.98446	30.000	72.678	7.3448	3.3773	2.1748	0.8812	0.48420
2.70	10.000	29.824	1.9369	1.5902	1.2180	2.2561	0.97125	28.000	75.285	8.0870	3.4773	2.2644	0.7401	0.45066
	12.000	31.728	2.1855	1.7241	1.2676	2.1669	0.95309	26.000	77.202	8.2233	3.5393	2.3235	0.6378	0.43010
2.70	14.000	33.739	2.4569	1.8614	1.3199	2.0763	0.92991	24.000	78.766	8.3214	3.5561	2.3400	0.6071	0.42454
	16.000	35.862	2.7523	2.0010	1.3754	1.9838	0.90191	22.000	80.110	8.3960	3.5688	2.3526	0.5829	0.42037
2.70	18.000	38.109	3.0727	2.1423	1.4343	1.8890	0.86948	20.000	81.303	8.4545	3.5786	2.3625	0.5634	0.41714
	20.000	40.496	3.4200	2.2845	1.4970	1.7915	0.83311	18.000	82.386	8.5014	3.5864	2.3704	0.6789	0.43799
2.70	22.000	43.049	3.7964	2.4273	1.5641	1.8905	0.79337	16.000	83.387	8.5392	3.5927	2.3768	0.6378	0.43010
	24.000	45.809	4.2059	2.5706	1.6362	1.5848	0.75072	14.000	84.324	8.5699	3.5978	2.3820	0.6071	0.42454
2.70	26.000	48.852	4.6560	2.7155	1.7146	1.4723	0.70538	12.000	85.212	8.5948	3.6019	2.3862	0.5145	0.40951
	28.000	52.334	5.1626	2.8645	1.8022	1.3488	0.65692	10.000	86.062	8.6146	3.6051	2.3895	0.5072	0.40845
2.70	30.000	56.687	5.7730	3.0271	1.9071	1.2018	0.60268	8.000	86.882	8.6301	3.6077	2.3922	0.5015	0.40762
	31.741	64.956	6.8143	3.2687	2.0847	0.9462	0.52090	6.000	87.680	8.6418	3.6096	2.3941	0.4972	0.40700
2.70	30.000	71.913	7.5186	3.4110	2.2042	0.7587	0.47286	4.000	88.462	8.6499	3.6109	2.3955	0.4942	0.40656
	28.000	74.790	7.7529	3.4551	2.2439	0.6907	0.45808	2.000	89.234	8.6547	3.6117	2.3963	0.4924	0.40631
2.70	26.000	76.828	7.8967	3.4814	2.2682	0.6468	0.44930							
	24.000	78.466	7.9983	3.4997	2.2854	0.6145	0.44321	2.80	2.000	22.344	23.854	1.3292	1.0854	1.0422
2.70	22.000	79.862	8.0748	3.5133	2.2984	0.5893	0.43870							
	20.000	81.095	8.1345	3.5238	2.3085	0.5691	0.43522	4.000		1.1553	1.3476	1.3292	1.0854	1.0422
2.70	18.000	82.210	8.1821	3.5321	2.3165	0.5527	0.43247	6.000		1.1553	1.3476	1.3292	1.0854	1.0422
	16.000	83.238	8.2204	3.5388	2.3230	0.5391	0.43027	8.000		1.1553	1.3476	1.3292	1.0854	1.0422
2.70	14.000	84.199	8.2515	3.5441	2.3282	0.5279	0.42850	10.000		1.1553	1.3476	1.3292	1.0854	1.0422
	12.000	85.109	8.2765	3.5484	2.3324	0.5188	0.422708	12.000		1.1553	1.3476	1.3292	1.0854	1.0422
2.70	10.000	85.978	8.2965	3.5518	2.3358	0.5114	0.42595	14.000		1.1553	1.3476	1.3292	1.0854	1.0422
	8.000	86.816	8.3121	3.5545	2.3385	0.5056	0.42506	16.000		1.1553	1.3476	1.3292	1.0854	1.0422
2.70	6.000	87.631	8.3238	3.5565	2.3404	0.5012	0.42441	18.000		1.1553	1.3476	1.3292	1.0854	1.0422
	4.000	88.430	8.3319	3.5579	2.3418	0.4981	0.42395	20.000		1.1553	1.3476	1.3292	1.0854	1.0422
2.70	2.000	89.218	8.3367	3.5587	2.3426	0.4962	0.42368	22.000		1.1553	1.3476	1.3292	1.0854	1.0422
	22.750	22.000	8.000	27.575	1.7239	1.1738	0.98377	32.587	65.050	7.3524	3.3788	2.1761	0.9490	0.48369
2.75	4.000	24.267	1.3236	1.5135	1.0841	2.5667	0.98776	30.000	54.786	5.9387	3.0683	2.9135	1.8328	1.4163
	6.000	25.873	1.3417	1.1280	2.4772	0.99279	32.000	60.433	6.7529	3.2555	2.0743	1.0909	0.52535	0.52535
2.75	2.000	1.1528	1.1068	1.2209	1.4686	1.1738	2.3879	32.587	65.050	7.3524	3.3788	2.1761	0.9490	0.48369
	4.000	10.000	29.372	1.9558	1.6007	1.2219	2.2982	32.000	69.211	7.8278	3.4689	2.2566	0.8307	0.45348
2.75	12.000	31.269	2.2104	1.7371	1.2724	2.2074	0.95109	30.000	73.328	8.2272	3.5399	2.3241	0.7243	0.42988
	14.000	33.269	2.4885	1.8768	1.3259	2.1153	0.92704	28.000	75.728	8.4241	3.5735	2.3574	0.6684	0.41882
2.75	16.000	35.381	2.7912	2.0188	1.3826	2.0213	0.89806	26.000	77.543	8.5544	3.5952	2.3794	0.6296	0.41169
	18.000	37.612	3.1197	2.1622	1.4429	1.9253	0.86461	24.000	79.042	8.6495	3.6108	2.3954	0.6002	0.40659
2.75	20.000	39.980	3.4757	2.3063	1.5070	1.8265	0.82724	22.000	80.339	8.7224	3.6227	2.4077	0.5769	0.40273
	22.000	42.504	3.8610	2.4506	1.5755	1.7245	0.78659	20.000	81.496	8.7800	3.6319	2.4177	0.5580	0.39771

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$M_2$	$\frac{T_2}{T_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{p_{02}}{p_{01}}$	
2.80	18.000	82.550	8.8262	3.6393	2.4252	0.5425	0.39731	2.90	6.000	24.666	1.5421	1.3594	1.1344	0.99178	
	16.000	83.525	8.8637	3.6453	2.4316	0.5297	0.39538		8.000	26.350	1.7663	1.4933	1.1828	2.5175	0.98153
	14.000	84.440	8.8942	3.6501	2.4367	0.5191	0.39382		10.000	28.129	2.0143	1.6328	1.2336	2.4229	0.96597
	12.000	85.308	8.9188	3.6540	2.4409	0.5103	0.39256		12.000	30.007	2.2873	1.7767	1.2874	2.3273	0.94475
	10.000	86.140	8.9385	3.6571	2.4442	0.5033	0.39156		14.000	31.985	2.5863	1.9238	1.3444	2.2304	0.97794
	8.000	86.943	8.9540	3.6595	2.4468	0.4977	0.39078		16.000	34.069	2.9123	2.0729	1.4050	2.1318	0.88591
	6.000	87.725	8.9656	3.6613	2.4487	0.4935	0.39019		18.000	36.264	3.2663	2.2229	1.4694	2.0313	0.84930
	4.000	88.492	8.9737	3.6626	2.4501	0.4905	0.38978		20.000	38.584	3.6496	2.3729	1.5380	1.9285	0.80886
	2.000	89.248	8.9784	3.6633	2.4509	0.4887	0.38954		22.000	41.044	4.0638	2.5222	1.6112	1.8229	0.76540
2.85	2.000	21.954	1.1579	1.1103	1.0429	2.7537	0.98968	28.000	49.655	5.5328	2.9652	1.8659	1.4788	0.62347	
	4.000	23.457	1.3249	1.2283	1.0868	2.6598	0.99755		30.000	53.274	6.1364	3.1161	1.9692	1.3453	0.57262
	6.000	25.052	1.5325	1.3535	1.1323	2.5670	0.98213		32.000	57.931	6.8791	3.2824	2.0957	1.1827	0.51624
	8.000	26.742	1.7520	1.4850	1.1798	2.4744	0.98230		33.363	65.145	7.9116	3.4841	2.2708	0.9516	0.44840
	10.000	28.526	1.9946	1.6220	1.2297	2.3815	0.96735		32.000	71.287	8.6350	3.6085	2.3930	0.7771	0.40736
	12.000	30.410	2.2613	1.7634	1.2824	2.2876	0.94692		30.000	74.392	8.9347	3.6565	2.4435	0.6985	0.39175
	14.000	32.394	2.5532	1.9080	1.3382	2.1923	0.92105		28.000	76.490	9.1095	3.6836	2.4730	0.6500	0.38301
	16.000	34.486	2.8712	2.0547	1.3974	2.0953	0.89006		26.000	78.142	9.2307	3.7020	2.4934	0.6149	0.37709
	18.000	36.692	3.2165	2.2025	1.4604	1.9964	0.85451		24.000	79.533	9.3212	3.7156	2.5087	0.5878	0.37275
	20.000	39.025	3.5904	2.3505	1.5275	1.8950	0.81511		22.000	80.750	9.3915	3.7260	2.5205	0.5660	0.36942
	22.000	41.505	3.9948	2.4982	1.5991	1.7906	0.77258		20.000	81.843	9.4475	3.7343	2.5300	0.5482	0.36680
	24.000	44.160	4.4325	2.6451	1.6757	1.6825	0.72766		18.000	82.845	9.4928	3.7409	2.5376	0.5335	0.36469
	26.000	47.042	4.9089	2.7916	1.7585	1.5692	0.68081		16.000	83.775	9.5296	3.7462	2.5438	0.5212	0.36299
	28.000	50.247	5.4345	2.9391	1.8490	1.4481	0.63219		14.000	84.651	9.5597	3.7506	2.5489	0.5111	0.36161
	30.000	53.992	6.0344	3.0917	1.9518	1.3127	0.58089		12.000	85.484	9.5842	3.7541	2.5027	0.5057	0.36049
	32.000	59.037	6.8013	2.2659	2.0825	1.1407	0.52183		10.000	86.283	9.6038	3.7570	2.5563	0.4959	0.35960
	32.984	65.097	7.6294	3.4320	2.2230	0.9503	0.46580		8.000	87.055	9.6191	3.7592	2.5588	0.4906	0.35890
	32.000	70.389	8.2421	3.5425	2.3266	0.8001	0.42903		6.000	87.808	9.6306	3.7608	2.5608	0.4865	0.35838
	30.000	73.893	8.5802	3.5995	2.3837	0.7107	0.41030		4.000	88.546	9.6387	3.7620	2.5621	0.4836	0.35802
	28.000	76.127	8.7648	3.6295	2.4149	0.6588	0.40050		2.000	89.275	9.6434	3.7626	2.5629	0.4819	0.35780
	26.000	77.855	8.8902	3.6495	2.4360	0.6220	0.39402								
	24.000	79.297	8.9827	3.6640	2.4516	0.5938	0.38933								
	22.000	80.552	9.0543	3.6751	2.4637	0.5713	0.38874								
	20.000	81.676	9.1110	3.6838	2.4733	0.5530	0.38294								
	18.000	82.702	9.1567	3.6908	2.4810	0.53779	0.38069								
	16.000	83.655	9.1938	3.6964	2.4872	0.5253	0.37888								
	14.000	84.549	9.2241	3.7010	2.4923	0.5150	0.37741								
	12.000	85.399	9.2486	3.7047	2.4964	0.5064	0.37623								
	10.000	86.213	9.2683	3.7077	2.4998	0.4995	0.37528								
	8.000	87.001	9.2836	3.7100	2.5023	0.4940	0.37454								
	6.000	87.768	9.2952	3.7117	2.5043	0.4899	0.37399								
	4.000	88.520	9.3033	3.7129	2.5057	0.4870	0.37360								
	2.000	89.262	9.3080	3.7136	2.5065	0.4853	0.37338								
2.90	2.000	21.578	1.1604	1.1120	1.0435	2.8019	0.99966								
	4.000	23.076	1.3406	1.2320	1.0882	2.7062	0.99744								

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{\rho_2}{\rho_1}$	$\frac{p_{02}}{p_{01}}$	$M_2$	$\frac{T_2}{T_1}$	$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{\rho_2}{\rho_1}$	$M_2$	$\frac{T_2}{T_1}$	$\frac{p_{02}}{p_{01}}$
2.95	32.000	6.9741	3.3023	2.1119	1.2199	0.50950	0.43150	0.9528	0.3194	3.00	14.000	84.837	10.2483	3.8459
	33.726	65.193	3.5350	2.3194	1.4204	0.58752	0.58752	0.7585	2.4577	12.000	85.638	10.2726	3.8491	2.6688
	32.000	72.020	3.6696	2.3194	1.4204	0.58752	0.58752	0.7585	2.4577	10.000	86.408	10.2921	3.8517	2.6721
30.000	74.838	9.2917	3.7112	2.5037	0.6877	0.37416	0.37416	0.6877	2.5037	8.000	87.154	10.3074	3.8537	2.6747
28.000	76.821	9.4585	3.7359	2.5318	0.6420	0.36628	0.36628	0.6420	2.5318	6.000	87.881	10.3190	3.8553	2.6766
26.000	78.407	9.5762	3.7530	2.5516	0.6084	0.36086	0.36086	0.6084	2.5516	4.000	88.594	10.3270	3.8563	2.6779
24.000	79.752	9.6649	3.7657	2.5666	0.5821	0.35684	0.35684	0.5821	2.5666	2.000	89.299	10.3318	3.8569	2.6787
22.000	80.935	9.7342	3.7742	2.5782	0.5610	0.35374	0.35374	0.5610	2.5782	1.000	89.935	10.3374	3.8575	2.6797
20.000	82.000	9.7896	3.7834	2.5875	0.5437	0.35128	0.35128	0.5437	2.5875	0.000	90.581	10.3421	3.8581	2.6806
18.000	82.978	9.8345	3.7896	2.5951	0.5293	0.34931	0.34931	0.5293	2.5951	4.000	91.224	10.3471	3.8589	2.6815
16.000	83.889	9.8712	3.7947	2.6013	0.5173	0.34771	0.34771	0.5173	2.6013	12.000	91.867	10.3521	3.8597	2.6824
14.000	84.747	9.9012	3.7989	2.6063	0.5074	0.34641	0.34641	0.5074	2.6063	6.000	92.509	10.3571	3.8605	2.6833
12.000	85.563	9.9255	3.8023	2.6104	0.4992	0.34536	0.34536	0.4992	2.6104	8.000	93.151	10.3621	3.8613	2.6842
10.000	86.348	9.9450	3.8050	2.6137	0.4925	0.34452	0.34452	0.4925	2.6137	10.000	93.794	10.3671	3.8621	2.6851
8.000	87.106	9.9604	3.8071	2.6163	0.4872	0.34386	0.34386	0.4872	2.6163	12.000	94.436	10.3721	3.8629	2.6860
6.000	87.845	9.9719	3.8087	2.6182	0.4832	0.34336	0.34336	0.4832	2.6182	14.000	95.078	10.3771	3.8637	2.6869
4.000	88.571	9.9799	3.8098	2.6196	0.4804	0.34302	0.34302	0.4804	2.6196	16.000	95.720	10.3821	3.8645	2.6878
2.000	89.288	9.9847	3.8104	2.6204	0.4788	0.34282	0.34282	0.4788	2.6204	18.000	96.362	10.3861	3.8653	2.6887
3.00	2.000	20.867	1.1656	1.0449	2.9881	0.99963	0.99963	2.9881	1.0449	24.000	42.361	42.361	2.4284	1.4972
	4.000	22.355	1.2395	1.0909	2.7988	0.99721	0.99721	2.7988	1.0909	26.000	45.110	52.806	2.8973	1.8226
	6.000	23.936	1.5616	1.3714	1.1387	2.7008	0.99105	0.99105	1.3714	28.000	48.102	54.002	3.0455	1.9196
	8.000	25.611	1.7953	1.5101	1.1888	2.6031	0.97993	0.97993	1.5101	30.000	51.455	6.4722	3.1938	2.0265
	10.000	27.383	2.0545	1.6546	1.2417	2.5050	0.96308	0.96308	1.6546	32.000	55.456	7.1967	3.3478	2.1497
	12.000	29.251	2.3404	1.8036	1.2977	2.4060	0.94022	0.94022	1.8036	34.000	61.505	8.2161	3.5380	2.3222
	14.000	31.218	2.6540	1.9556	1.3571	2.3056	0.91148	0.91148	1.9556	34.407	65.288	8.7895	3.6335	2.4197
	16.000	33.288	2.9964	2.1095	1.4204	2.2037	0.87734	0.87734	2.1095	34.000	68.742	9.2596	3.7064	2.4983
	18.000	35.467	3.3685	2.2641	1.4878	2.1000	0.83855	0.83855	2.2641	32.000	73.184	9.7779	3.7817	2.5856
	20.000	37.764	3.7713	2.4181	1.5596	1.9941	0.79602	0.79602	2.4181	30.000	75.604	10.0154	3.8146	2.6255
	22.000	40.192	4.2064	2.5708	1.6362	1.8858	0.75068	0.75068	2.5708	28.000	77.406	10.1703	3.8355	2.6516
	24.000	42.775	4.6761	2.7216	1.7181	1.7744	0.70340	0.70340	2.7216	26.000	78.880	10.2825	3.8505	2.6705
	26.000	45.552	5.1844	2.8706	1.8060	1.6589	0.65491	0.65491	2.8706	24.000	80.145	10.3683	3.8617	2.6849
	28.000	48.586	5.7388	3.0184	1.9012	1.5374	0.60560	0.60560	3.0184	22.000	81.267	10.4361	3.8705	2.6963
	30.000	52.014	6.3559	3.1673	2.0067	1.4059	0.55526	0.55526	3.1673	20.000	82.284	10.4906	3.8776	2.7055
	32.000	56.182	7.0810	3.3244	2.1300	1.2541	0.50205	0.50205	3.3244	18.000	83.221	10.5350	3.8833	2.7129
	34.000	63.673	8.2682	3.5470	2.3310	1.0029	0.42755	0.42755	3.5470	16.000	84.095	10.5714	3.8879	2.7190
	34.073	65.241	8.4917	3.5848	2.3688	0.9540	0.41510	0.41510	3.5848	14.000	84.921	10.6012	3.8917	2.7240
	34.000	66.749	8.6971	3.6186	2.4035	0.9083	0.40406	0.40406	3.6186	12.000	85.709	10.6255	3.8948	2.7281
	36.000	72.642	9.3988	3.7271	2.5217	0.7428	0.36908	0.36908	2.5217	10.000	86.466	10.6450	3.8973	2.7314
	38.000	75.239	9.6517	3.7638	2.5643	0.6779	0.35743	0.35743	2.5643	8.000	87.199	10.6603	3.8992	2.7340
	38.000	77.126	9.8121	3.7865	2.5913	0.6345	0.35029	0.35029	2.5913	6.000	87.914	10.6719	3.9007	2.7359
	38.000	78.652	9.9268	3.8024	2.6106	0.6022	0.34530	0.34530	2.6106	4.000	88.617	10.6799	3.9017	2.7373
	38.000	79.956	10.0139	3.8144	2.6253	0.5768	0.34157	0.34157	2.6253	2.000	89.310	10.6847	3.9023	2.7381
	38.000	81.106	10.0824	3.8237	2.6368	0.5563	0.33868	0.33868	2.6368	0.000	89.310	10.6847	3.9023	2.7381
	38.000	82.147	10.1373	3.8311	2.6460	0.5394	0.33638	0.33638	2.6460	4.000	89.310	10.6847	3.9023	2.7381
	38.000	83.103	10.1819	3.8371	2.6536	0.5253	0.33453	0.33453	2.6536	0.000	89.310	10.6847	3.9023	2.7381
	38.000	83.996	10.2184	3.8420	2.6597	0.5136	0.33302	0.33302	2.6597	4.000	89.310	10.6847	3.9023	2.7381

$\frac{p_{02}}{p_{01}} = \frac{1.0449}{1.0909} = 0.9556$

Oblique Shock p10

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$	
3.10	6.000	23.258	1.5815	1.1431	2.7894	0.99027	3.15	28.000	47.216	6.0688	3.1000	1.9577	1.6194	0.57808	
	8.000	24.927	1.8249	1.5271	2.6881	0.97822		30.000	50.449	6.7158	3.2475	2.0680	1.4886	0.52806	
	10.000	26.692	2.0956	1.6767	2.5864	0.96004		32.000	54.201	7.4487	3.3975	2.1924	1.3441	0.47738	
	12.000	28.554	2.3949	1.8308	1.3081	2.4837		34.000	59.196	8.3736	3.5650	2.3489	1.1632	0.42162	
	14.000	30.513	2.7286	1.9879	1.3701	2.3798	0.90473	35.033	65.382	9.4008	3.7274	2.5221	0.9575	0.36898	
	16.000	32.574	3.0831	2.1467	1.4362	2.2743	0.86841	34.000	70.719	10.1474	3.8325	2.6478	0.7974	0.33596	
	18.000	34.739	3.4740	2.3057	1.5067	2.1672	0.82741	32.000	74.089	10.5396	3.8839	2.7137	0.7064	0.32016	
	20.000	37.017	3.8973	2.4637	1.5819	2.0581	0.78278	30.000	76.244	10.7550	3.9111	2.7499	0.6531	0.31190	
	22.000	39.421	4.3543	2.6198	1.6621	1.9468	0.73556	28.000	77.906	10.9014	3.9292	2.7745	0.6152	0.30644	
	24.000	41.968	4.8470	2.7733	1.7477	1.8329	0.68676	26.000	79.289	11.0097	3.9424	2.7927	0.5860	0.30248	
	26.000	44.692	5.3788	2.9241	1.8395	1.7154	0.63718	24.000	80.490	11.0936	3.9524	2.8068	0.5627	0.29947	
	28.000	47.646	5.9563	3.0727	1.9385	1.5928	0.58731	22.000	81.560	11.1602	3.9604	2.8180	0.5436	0.29710	
	30.000	50.935	6.5922	3.2205	2.0470	1.4620	0.53722	20.000	82.535	11.2142	3.9668	2.8270	0.5278	0.29520	
	32.000	54.800	7.3197	3.3723	2.1705	1.3157	0.48586	18.000	83.436	11.2583	3.9720	2.8334	0.5145	0.29366	
	34.000	60.205	8.2768	3.5485	2.3325	1.1241	0.42706	16.000	84.279	11.2945	3.9762	2.8405	0.5035	0.29240	
	34.726	65.335	9.0925	3.6810	2.4701	0.9564	0.38385	14.000	85.076	11.3243	3.9797	2.8455	0.4942	0.29138	
	34.000	69.872	9.7174	3.7732	2.5754	0.8203	0.35449	12.000	85.838	11.3486	3.9825	2.8496	0.4865	0.29054	
	32.000	73.661	10.1577	3.8339	2.6495	0.7171	0.33553	10.000	86.571	11.3682	3.9848	2.8529	0.4803	0.28987	
	30.000	75.938	10.3831	3.8636	2.6874	0.6607	0.32634	8.000	87.281	11.3835	3.9866	2.8555	0.4754	0.28935	
	28.000	77.666	10.5334	3.8831	2.7126	0.6212	0.32040	6.000	87.976	11.3951	3.9879	2.8574	0.4716	0.28895	
	26.000	79.091	10.6435	3.8971	2.7311	0.5911	0.31614	4.000	88.657	11.4032	3.9889	2.8588	0.4690	0.28867	
	24.000	80.324	10.7282	3.9077	2.7454	0.5671	0.31291	2.000	89.330	11.4080	3.9894	2.8596	0.4674	0.28851	
	22.000	81.419	10.7954	3.9161	2.7567	0.5476	0.31038								
	20.000	82.413	10.8496	3.9228	2.7658	0.5314	0.30836								
	18.000	83.381	10.8938	3.9282	2.7732	0.5179	0.30672								
	16.000	84.189	10.9301	3.9327	2.7793	0.5067	0.30539								
	14.000	85.001	10.9599	3.9363	2.7843	0.4973	0.30430								
	12.000	85.775	10.9842	3.9393	2.7884	0.4895	0.30341								
	10.000	86.520	11.0037	3.9416	2.7917	0.4832	0.30270								
	8.000	87.242	11.0190	3.9435	2.7942	0.4781	0.30215								
	6.000	87.945	11.0306	3.9449	2.7962	0.4743	0.30173								
	4.000	88.637	11.0387	3.9458	2.7975	0.4716	0.30144								
	2.000	89.321	11.0434	3.9464	2.7983	0.4701	0.30127								
3.15	2.000	19.891	1.1734	1.0428	0.99958	1.0421	0.99683	26.000	43.920	5.5816	2.9750	1.8743	0.61919	0.35463	
	4.000	21.366	1.2699	1.0951	2.9371	0.99683	2.8336	0.98986	28.000	46.811	6.1840	3.1274	1.9774	0.56880	
	6.000	22.937	1.5915	1.1453	2.8336	0.98986	1.981	0.97734	30.000	49.994	6.8427	3.2747	2.0895	1.5144	
	8.000	24.603	1.8399	1.5357	2.7304	0.98986	2.2093	0.82172	32.000	53.651	7.5832	3.4223	2.2152	1.3711	
	10.000	26.366	2.1166	1.6878	2.1250	0.95846	0.77603	0.77603	34.000	58.350	8.4906	2.8252	1.7784	1.8893	
	12.000	28.225	2.4226	1.8445	2.1314	2.5222	0.93300	34.000	65.327	9.7141	3.7727	2.5748	1.7695		
	14.000	30.181	2.7592	2.0042	2.1367	2.4165	0.90123	34.000	71.408	10.5657	3.8872	2.7181	0.7791	0.31914	
	16.000	32.238	3.1273	2.1654	2.4443	2.3092	0.86382	34.000	74.475	10.9242	3.9320	2.8783	0.6967	0.30560	
	18.000	34.398	3.5279	2.2366	1.5163	2.2627	0.82172	32.000	76.526	11.1314	3.9570	2.8131	0.6461	0.29812	
	20.000	36.668	3.9617	2.4866	1.5933	2.0895	0.77603	28.000	78.130	11.2746	3.9792	2.8851	0.6096	0.29310	
	22.000	39.061	4.4302	2.6444	1.6753	1.9767	0.72789	26.000	79.475	11.3814	3.9864	2.8551	0.5812	0.28942	
	24.000	41.594	4.9349	2.7982	1.7629	1.8613	0.67833	24.000	80.646	11.4644	3.9959	2.8690	0.5555	0.28660	

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{\rho_2}{\rho_1}$	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$	
3.20	22.000	81.694	11.5307	4.0035	2.8802	0.5398	0.28438	3.30	2.000	19.009	1.1812	1.1262	3.1858	0.99953	
	20.000	82.649	11.5844	4.0096	2.8892	0.5243	0.28260		4.000	20.475	1.3880	1.2626	3.0748	0.99642	
	18.000	83.533	11.6285	4.0146	2.8966	0.5113	0.28115		6.000	22.039	1.6222	1.4082	1.1520	2.9653	0.98858
	16.000	84.363	11.6647	4.0187	2.9026	0.5004	0.27996		8.000	23.699	1.8859	1.5617	1.2076	2.8563	0.97453
	14.000	85.147	11.6945	4.0220	2.9076	0.4913	0.27899		10.000	25.457	2.1807	1.7216	1.2666	2.7468	0.95347
	12.000	85.897	11.7188	4.0247	2.9117	0.4837	0.27820		12.000	27.310	2.5078	1.8861	1.3296	2.6364	0.92526
	10.000	86.619	11.7385	4.0269	2.9150	0.4776	0.27757		14.000	29.261	2.8688	2.0536	1.3970	2.5248	0.89031
	8.000	87.320	11.7539	4.0286	2.9176	0.4727	0.27707		16.000	31.308	3.2640	2.2219	1.4690	2.4118	0.84954
	6.000	88.003	11.7655	4.0299	2.9196	0.4690	0.27669		18.000	33.456	3.6947	2.3898	1.5460	2.2974	0.80409
	4.000	88.675	11.7736	4.0308	2.9209	0.4664	0.27643		20.000	35.710	4.1617	2.5557	1.6284	2.1813	0.75527
	2.000	89.340	11.7784	4.0313	2.9217	0.4649	0.27628		22.000	38.077	4.6655	2.7184	1.7163	2.0636	0.70444
3.25	2.000	19.293	1.1786	1.1244	1.0482	3.1380	0.99955	3.0290	0.99656	28.000	46.062	6.4212	3.0118	0.60108	
	4.000	20.762	1.3818	1.2586	1.0979	3.0290	0.99656	3.0290	0.99656	30.000	49.163	7.1057	3.1342	0.6955	
	6.000	22.328	1.6119	1.4019	1.1498	2.9215	0.98902	3.0290	0.98902	32.000	52.667	7.8658	3.4758	0.50034	
	8.000	23.990	1.8704	1.5530	1.2044	2.8145	0.97549	34.000	56.963	8.7622	3.6291	2.4144	1.4218	0.45116	
	10.000	25.749	2.1590	1.7103	1.2624	2.7070	0.95518	35.882	65.518	10.3564	3.8602	2.6829	1.8215	0.32741	
	12.000	27.604	2.4791	1.8722	1.3242	2.5986	0.92789	34.000	72.501	11.3896	3.9873	2.8565	2.0178	0.28914	
	14.000	29.556	2.8318	2.0370	1.3901	2.4889	0.89402	32.000	75.148	11.7036	4.0230	2.9092	2.1342	0.27869	
	16.000	31.606	3.2179	2.2030	1.4607	2.3779	0.85437	30.000	77.029	11.8983	4.0445	2.9418	2.2630	0.27247	
	18.000	33.757	3.6384	2.3687	1.5360	2.2653	0.81004	28.000	78.535	12.0364	4.0595	2.9650	2.6336	0.26817	
	20.000	36.016	4.0940	2.5326	1.6165	2.1511	0.76227	26.000	79.812	12.1408	4.0706	2.9825	2.7525	0.26497	
	22.000	38.390	4.5858	2.6937	1.7024	2.0350	0.71232	24.000	80.932	12.2227	4.0793	2.9963	2.8507	0.26251	
	24.000	40.898	5.1156	2.8513	1.7941	1.9168	0.66129	22.000	81.938	12.2884	4.0862	3.0073	0.5328	0.26055	
	26.000	43.563	5.6858	3.0049	1.8922	1.7958	0.61015	20.000	82.859	12.3420	4.0918	3.0163	0.5178	0.25896	
	28.000	46.426	6.3015	3.1548	1.9974	1.7607	0.55950	18.000	83.714	12.3860	4.0964	3.0236	0.5052	0.25767	
	30.000	49.566	6.9727	3.4490	2.1116	1.5394	0.50960	16.000	84.517	12.4423	4.1001	3.0297	0.4946	0.25662	
	32.000	53.141	7.7223	3.4494	2.2387	1.3970	0.45998	14.000	85.278	12.4523	4.1032	3.0348	0.4858	0.25578	
	34.000	57.616	8.6213	3.6062	2.3907	1.2287	0.40809	12.000	86.007	12.4767	4.1057	3.0389	0.4785	0.25504	
	35.610	65.473	10.0327	3.8170	2.6285	0.9596	0.34078	10.000	86.708	12.4964	4.1077	3.0422	0.4725	0.25448	
	34.000	71.993	10.9786	3.9386	2.7875	0.7636	0.30361	8.000	87.390	12.5120	4.1093	3.0448	0.4677	0.25403	
	32.000	74.827	11.3120	3.9783	2.8434	0.6878	0.29180	6.000	88.056	12.5237	4.1105	3.0467	0.4641	0.25369	
	30.000	76.787	11.5124	4.0014	2.8771	0.6396	0.28499	4.000	88.710	12.5319	4.1114	3.0481	0.4616	0.25346	
	28.000	78.339	11.6529	4.0173	2.9007	0.60443	0.28035	2.000	89.357	12.5367	4.1119	3.0489	0.4601	0.25332	
	26.000	79.649	11.7584	4.0291	2.9184	0.5767	0.27692								
	24.000	80.793	11.8408	4.0382	2.9322	0.5545	0.27429								
	22.000	81.819	11.9067	4.0454	2.9433	0.5362	0.27220								
	20.000	82.757	11.9604	4.0513	2.9523	0.5210	0.27052								
	18.000	83.626	12.0044	4.0560	2.9596	0.5082	0.26916								
	16.000	84.442	12.0407	4.0599	2.9657	0.4974	0.26804								
	14.000	85.214	12.0705	4.0631	2.9707	0.4885	0.26712								
	12.000	85.953	12.0949	4.0658	2.9748	0.4810	0.26637								
	10.000	86.665	12.1145	4.0679	2.9781	0.4750	0.26577								
	8.000	87.356	12.1300	4.0695	2.9807	0.4702	0.26530								
	6.000	88.030	12.1417	4.0707	2.9827	0.4665	0.26495								
	4.000	88.693	12.1498	4.0716	2.9840	0.4639	0.26470								
	2.000	89.348	12.1547	4.0721	2.9848	0.4624	0.26455								

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{T_2}{T_1}$	$\frac{p_2}{p_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{\rho_2}{\rho_1}$	$M_2$	$\frac{T_2}{T_1}$	$\frac{p_{02}}{p_{01}}$
3.35	24,000	40,264	2.9033	1.8263	1.9704	0.64409	3.40	34,000	73,352	12,2131	4,0783	2,9946	0.7279	0.26279
	26,000	42,898	3.0588	1.9288	1.8468	0.59200		32,000	75,717	12,4992	4,1080	3,0426	0.6653	0.25440
	28,000	45,716	6,5433	3,2097	2,0386	0.54090		30,000	77,467	12,6849	4,1268	3,0738	0.6225	0.24914
3.40	2,000	18,467	5,3024	2.9416	2.2416	0.49109	3.40	34,000	73,352	12,2131	4,0783	2,9946	0.7279	0.26279
	4,000	19,928	5,8998	3.0588	2.0386	0.54090		32,000	75,717	12,4992	4,1080	3,0426	0.6653	0.25440
	6,000	21,488	6,5433	3,2097	2,0386	0.54090		30,000	77,467	12,6849	4,1268	3,0738	0.6225	0.24914
3.45	2,000	18,467	5,3024	2.9416	2.2416	0.49109	3.45	2,000	78,891	12,8193	4,1402	3,0963	0.5902	0.24542
	4,000	19,928	5,8998	3.0588	2.0386	0.54090		26,000	80,110	12,9221	4,1503	3,1135	0.5646	0.24263
	6,000	21,488	6,5433	3,2097	2,0386	0.54090		24,000	81,185	13,0033	4,1582	3,1271	0.5437	0.24046
3.50	2,000	18,467	5,3024	2.9416	2.2416	0.49109	3.50	2,000	78,891	12,8193	4,1402	3,0963	0.5902	0.24542
	4,000	19,928	5,8998	3.0588	2.0386	0.54090		26,000	80,110	12,9221	4,1503	3,1135	0.5646	0.24263
	6,000	21,488	6,5433	3,2097	2,0386	0.54090		24,000	81,185	13,0033	4,1582	3,1271	0.5437	0.24046
3.60	2,000	18,467	5,3024	2.9416	2.2416	0.49109	3.60	2,000	82,156	13,0688	4,1645	3,1381	0.5264	0.23872
	4,000	19,928	5,8998	3.0588	2.0386	0.54090		22,000	83,047	13,1224	4,1697	3,1471	0.5119	0.23732
	6,000	21,488	6,5433	3,2097	2,0386	0.54090		20,000	83,876	13,1665	4,1739	3,1545	0.4997	0.23617
3.70	2,000	18,467	5,3024	2.9416	2.2416	0.49109	3.70	2,000	84,656	13,2030	4,1774	3,1606	0.4894	0.23522
	4,000	19,928	5,8998	3.0588	2.0386	0.54090		24,000	85,396	13,2331	4,1802	3,1657	0.4808	0.23445
	6,000	21,488	6,5433	3,2097	2,0386	0.54090		22,000	86,105	13,2578	4,1826	3,1698	0.4736	0.23381
3.80	2,000	18,467	5,3024	2.9416	2.2416	0.49109	3.80	2,000	86,789	13,2777	4,1844	3,1731	0.4678	0.23330
	4,000	19,928	5,8998	3.0588	2.0386	0.54090		24,000	87,453	13,2934	4,1859	3,1757	0.4632	0.23290
	6,000	21,488	6,5433	3,2097	2,0386	0.54090		22,000	88,103	13,3052	4,1870	3,1777	0.4596	0.23260
3.90	2,000	18,467	5,3024	2.9416	2.2416	0.49109	3.90	2,000	88,741	13,3135	4,1878	3,1791	0.4572	0.23239
	4,000	19,928	5,8998	3.0588	2.0386	0.54090		20,000	89,372	13,3184	4,1883	3,1799	0.4557	0.23227
4.00	2,000	18,467	5,3024	2.9416	2.2416	0.49109	4.00	2,000	18,209	19,668	1,4063	1,1316	1,0509	0.99947
	4,000	19,928	5,8998	3.0588	2.0386	0.54090		24,000	21,226	1,6536	1,2743	1,1036	3,2118	0.99597
	6,000	21,488	6,5433	3,2097	2,0386	0.54090		22,000	22,884	1,9331	1,5881	1,2172	2,9809	0.97149
4.10	2,000	18,467	5,3024	2.9416	2.2416	0.49109	4.10	2,000	24,639	2,2468	1,7559	1,2796	2,8653	0.94812
	4,000	19,928	5,8998	3.0588	2.0386	0.54090		20,000	26,491	2,5962	1,9284	1,3463	2,7486	0.91701
	6,000	21,488	6,5433	3,2097	2,0386	0.54090		18,000	28,438	2,9823	2,1035	1,4178	2,6309	0.87878
4.20	2,000	18,467	5,3024	2.9416	2.2416	0.49109	4.20	2,000	30,481	3,4063	2,2791	1,4946	2,5118	0.83456
	4,000	19,928	5,8998	3.0588	2.0386	0.54090		24,000	32,621	3,8688	2,4535	1,5769	2,3915	0.78577
	6,000	21,488	6,5433	3,2097	2,0386	0.54090		22,000	34,863	4,3706	2,6251	1,6649	2,2698	0.73391
4.30	2,000	18,467	5,3024	2.9416	2.2416	0.49109	4.30	2,000	37,213	4,9123	2,7926	1,7590	2,1468	0.68049
	4,000	19,928	5,8998	3.0588	2.0386	0.54090		32,000	39,683	5,4951	2,9552	2,0224	0.62680	
	6,000	21,488	6,5433	3,2097	2,0386	0.54090		26,000	42,292	6,1211	3,1125	1,9666	1,8960	0.57385
4.40	2,000	18,467	5,3024	2.9416	2.2416	0.49109	4.40	2,000	45,073	6,7941	3,2644	2,0813	1,7667	0.52235
	4,000	19,928	5,8998	3.0588	2.0386	0.54090		30,000	48,080	7,5215	3,4115	2,2047	1,6329	0.47267
	6,000	21,488	6,5433	3,2097	2,0386	0.54090		24,000	51,420	8,3194	3,5558	2,3397	1,4914	0.42466
4.50	2,000	18,467	5,3024	2.9416	2.2416	0.49109	4.50	2,000	55,344	9,2294	3,7018	2,4932	1,3339	0.37715
	4,000	19,928	5,8998	3.0588	2.0386	0.54090		36,000	60,903	10,4358	3,8705	2,6962	1,1265	0.32424
	6,000	21,488	6,5433	3,2097	2,0386	0.54090		30,000	65,647	11,3584	3,9837	2,8512	0,9634	0.29020
4.60	2,000	18,467	5,3024	2.9416	2.2416	0.49109	4.60	2,000	69,850	12,0718	4,0633	2,9709	2,8032	0.26708
	4,000	19,928	5,8998	3.0588	2.0386	0.54090		32,000	73,716	12,6278	4,1211	3,0642	0,7184	0.25074
	6,000	21,488	6,5433	3,2097	2,0386	0.54090		26,000	75,970	12,9035	4,1485	3,1104	0,6589	0.24313
4.70	2,000	18,467	5,3024	2.9416	2.2416	0.49109	4.70	2,000	77,665	13,0858	4,1662	3,1410	0,6175	0.23828
	4,000	19,928	5,8998	3.0588	2.0386	0.54090		30,000	79,054	13,2189	4,1789	3,1633	0,5860	0.23481
	6,000	21,488	6,5433	3,2097	2,0386	0.54090		24,000	80,246	13,3210	4,1885	3,1804	0,5609	0.23220
4.80	2,000	18,467	5,3024	2.9416	2.2416	0.49109	4.80	2,000	81,302	13,4020	4,1961	3,1939	0,5404	0.23016
	4,000	19,928	5,8998	3.0588	2.0386	0.54090		22,000	82,256	13,4675	4,2021	3,2049	0,5234	0.22852
	6,000	21,488	6,5433	3,2097	2,0386	0.54090		20,000	83,134	13,5211	4,2071	3,2139	0,5091	0.22719
4.90	2,000	18,467	5,3024	2.9416	2.2416	0.49109	4.90	2,000	83,951	13,5654	4,2111	3,2213	0,4971	0.22611

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$M_2$	$\frac{T_2}{T_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{p_{02}}{p_{01}}$
3.45	16,000	84,720	13,6020	4,2145	3,2275	0.4869	0.22521	3.55	2,000	17,715	1,1947	1,1353	1,0523	3,4246	0.99943
	14,000	85,451	13,6322	4,2172	3,2325	0.4784	0.22448	4,000	19,170	1,4187	1,2822	1,1065	3,3029	0.99566	
	12,000	86,151	13,6570	4,2195	3,2367	0.4714	0.22388	6,000	20,726	1,6748	1,4396	1,1634	3,1829	0.98619	
	10,000	86,826	13,66770	4,2213	3,2400	0.4656	0.22340	8,000	22,383	1,98653	1,6059	1,2238	3,0633	0.96935	
	8,000	87,482	13,6928	4,2227	3,2427	0.4610	0.22302	10,000	24,138	2,2920	1,7791	1,2883	2,9433	0.94435	
	6,000	88,125	13,7047	4,2238	3,2447	0.4575	0.22273	12,000	25,989	2,6566	1,95669	1,3576	2,8224	0.91123	
	4,000	88,756	13,7130	4,2245	3,2461	0.4551	0.22253	14,000	27,936	3,0603	2,1370	1,4321	2,7003	0.87077	
	2,000	89,379	13,7180	4,2250	3,2469	0.4536	0.22241	16,000	29,977	3,5040	2,3174	1,5121	2,5771	0.82424	
3.50	2,000	17,958	1,1920	1,1335	1,0516	3,3769	0.9945	20,000	34,352	4,5148	2,6714	1,6901	2,3271	0.71939	
	4,000	19,415	1,4125	1,4333	1,1050	3,2574	0.99582	24,000	36,692	5,0827	2,8419	1,7885	2,2005	0.66437	
	6,000	20,972	1,6642	1,7783	1,1611	3,1396	0.98669	26,000	41,738	6,3495	3,1659	2,0069	1,8935	0.60946	
	8,000	22,629	1,9491	1,5970	1,2205	3,0222	0.97044	28,000	44,488	7,0535	3,3187	2,1254	1,8117	0.50395	
	10,000	24,384	2,2693	1,7675	1,2839	2,9044	0.94626	30,000	47,447	7,8120	3,4660	2,2539	1,6762	0.45445	
	12,000	26,236	2,6262	1,9426	1,3519	2,7856	0.91415	32,000	50,705	8,6392	3,6092	2,3937	1,5342	0.40714	
	14,000	28,182	3,0211	2,1202	1,4249	2,6657	0.87481	34,000	54,463	9,5691	3,7520	2,5504	1,3790	0.36118	
	16,000	30,225	3,4549	2,2982	1,5033	2,5445	0.82942	36,000	59,399	10,7262	3,9075	2,7450	1,1885	0.31299	
	18,000	32,363	3,9283	2,4747	1,5874	2,4222	0.77952	37,091	65,729	12,0520	4,0612	2,9676	0,9651	0.26768	
	20,000	34,602	4,4421	2,6482	1,6774	2,9286	0.72668	36,000	71,121	12,9969	4,1576	3,1261	0,7943	0.24063	
	22,000	36,947	4,9969	2,8173	1,7737	2,1739	0,67245	34,000	74,353	13,4667	4,2021	3,2048	0,7018	0,22854	
	24,000	39,410	5,5936	2,9811	1,8764	2,0478	0,61813	32,000	76,427	13,7265	4,2257	3,2483	0,6473	0,22221	
	26,000	42,009	6,2345	3,1392	1,9860	1,9199	0,56478	30,000	78,025	13,9083	4,2415	3,2779	0,6083	0,21803	
	28,000	44,774	6,9227	3,2916	2,1032	1,7894	0,51313	28,000	79,351	14,0342	4,2530	3,2998	0,5782	0,21501	
	30,000	47,755	7,6654	3,4388	2,2291	1,6549	0,46353	26,000	80,497	14,1355	4,2618	3,3168	0,5541	0,21271	
	32,000	51,053	8,4777	3,5825	2,3664	1,5131	0,41586	24,000	81,517	14,2163	4,2687	3,3343	0,5343	0,21090	
	34,000	54,888	9,3968	3,7268	2,5214	1,57570	0,36917	22,000	82,442	14,2819	4,2743	3,3413	0,5178	0,20944	
	36,000	60,090	10,5715	3,8879	2,7191	1,5194	0,31891	20,000	83,294	14,3358	4,2789	3,3503	0,5039	0,20826	
	38,000	65,689	11,7027	4,0229	2,9090	0,9643	0,27872	18,000	84,090	14,3804	4,2827	3,3578	0,4922	0,20729	
	40,000	70,545	12,5396	4,1121	3,0494	0,8105	0,25324	16,000	84,839	14,4173	4,2858	3,3640	0,4823	0,20649	
	42,000	74,048	13,0455	4,1623	3,1342	0,7098	0,23934	14,000	85,552	14,4478	4,2883	3,3691	0,4740	0,20583	
	44,000	76,207	13,3126	4,1787	3,1791	0,6529	0,23241	12,000	86,235	14,4729	4,2904	3,4733	0,4671	0,20529	
	46,000	77,851	13,4920	4,2044	3,2090	0,6128	0,22791	10,000	86,895	14,4981	4,2921	3,3767	0,4615	0,20485	
	48,000	79,207	13,6238	4,2165	3,2311	0,5820	0,22468	8,000	87,537	14,5091	4,2934	3,3794	0,4570	0,20451	
	50,000	80,375	13,7255	4,2256	3,2481	0,5574	0,22223	6,000	88,165	14,5212	4,2944	3,3814	0,4535	0,20425	
	52,000	81,413	13,8064	4,2329	3,2617	0,5373	0,22031	4,000	88,782	14,5296	4,2951	3,3828	0,4511	0,20407	
	54,000	82,352	13,8719	4,2387	3,2727	0,5205	0,21877	2,000	89,392	14,5346	4,2956	3,3836	0,4497	0,20397	
	56,000	83,216	13,9256	4,2435	3,2817	0,5065	0,21751								
	58,000	84,022	13,9700	4,2474	3,2891	0,4946	0,21649								
	60,000	84,781	14,0067	4,2506	3,2952	0,4846	0,21564								
	62,000	85,503	14,0371	4,2532	3,3003	0,4762	0,21494								
	64,000	86,194	14,0620	4,2554	3,3045	0,4692	0,21438								
	66,000	86,862	14,0822	4,2572	3,3079	0,4635	0,21392								
	68,000	87,510	14,0980	4,2585	3,3105	0,4590	0,21356								
	70,000	88,145	14,1100	4,2596	3,3125	0,4555	0,21329								
	72,000	88,769	14,1184	4,2603	3,3139	0,4531	0,21310								
	74,000	89,386	14,1234	4,2607	3,3148	0,4516	0,21298								

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$
3.60	20.000	34.110	4.5883	2.6945	1.7029	2.3552	0.71207	3.65	37.513	65.808	12.7662	4.1349	3.0874	0.9668
	22.000	36.448	5.1699	2.8666	1.8035	2.2267	0.65625		36.000	72.054	13.9006	4.2413	3.2775	0.7684
	24.000	38.898	5.7953	3.0327	1.9109	2.0973	0.60079		34.000	74.894	14.3206	4.2776	3.3478	0.6877
	26.000	41.478	6.4663	3.1924	2.0255	1.9664	0.54674		32.000	76.827	14.5690	4.2984	3.3894	0.6371
	28.000	44.215	7.1862	3.3457	2.1479	1.8335	0.49483		30.000	78.345	14.7420	4.3126	3.4183	0.6000
	30.000	47.153	7.9610	3.4930	2.2791	1.6971	0.44543		28.000	79.617	14.8713	4.3231	3.4400	0.1962
	32.000	50.376	8.8038	3.6357	2.4215	1.5547	0.39847		26.000	80.723	14.9723	4.3311	3.4569	0.5712
	34.000	54.066	9.7460	3.7772	2.5802	1.4002	0.35321		24.000	81.712	15.0533	4.3376	3.4705	0.19697
	36.000	58.793	10.8943	3.9283	2.7733	1.2149	0.30670		22.000	82.610	15.1191	4.3427	3.4815	0.19202
	37.306	65.769	12.4065	4.0985	3.0271	0.9660	0.25708		20.000	83.440	15.1734	4.3470	3.4906	0.19096
	36.000	71.617	13.4496	4.2005	3.2019	0.7805	0.22897		18.000	84.215	15.2184	4.3505	3.4981	0.19009
	34.000	74.634	13.8916	4.2405	3.2760	0.6945	0.21831		16.000	84.947	15.2557	4.3534	3.5043	0.18937
	32.000	76.633	14.1452	4.2626	3.3184	0.6420	0.21249		14.000	85.644	15.2866	4.3558	3.5095	0.4699
	30.000	78.190	14.3199	4.2776	3.3477	0.6041	0.20861		12.000	86.313	15.3120	4.3577	3.5137	0.4632
	28.000	79.487	14.4500	4.2885	3.3695	0.5746	0.20578		10.000	86.959	15.3325	4.3593	3.5172	0.4576
	26.000	80.614	14.5512	4.2969	3.3864	0.5510	0.20362		8.000	87.587	15.3487	4.3606	3.5199	0.4532
	24.000	81.617	14.6320	4.3036	3.3999	0.5315	0.20191		6.000	88.201	15.3609	4.3615	3.5219	0.4499
	22.000	82.528	14.6976	4.3090	3.4109	0.5152	0.20054		4.000	88.807	15.3695	4.3622	3.5234	0.4475
	20.000	83.369	14.7517	4.3134	3.4200	0.5015	0.19942		2.000	89.405	15.3746	4.3625	3.5242	0.4461
	18.000	84.154	14.7965	4.3170	3.4275	0.4899	0.19849							0.18710
	16.000	84.894	14.8336	4.3200	3.4337	0.4801	0.19774							0.18720
	14.000	85.599	14.8643	4.3225	3.4388	0.4719	0.19711	3.70	2.000	17.027	1.2029	1.0544	0.9936	
	12.000	86.275	14.8895	4.3245	3.4430	0.4651	0.19660		4.000	18.478	1.4377	1.2942	1.1108	0.9515
	10.000	86.928	14.9099	4.3262	3.4465	0.4595	0.19619		6.000	20.032	1.7073	1.4589	1.1703	0.98461
	8.000	87.562	14.9260	4.3274	3.4491	0.4551	0.19586		8.000	21.688	2.0146	1.6330	1.2337	0.96594
	6.000	88.184	14.9381	4.3284	3.4512	0.4517	0.19562		10.000	23.444	2.3615	1.8141	1.3017	0.93840
	4.000	88.794	14.9466	4.3291	3.4526	0.4493	0.19545		12.000	24.297	2.7498	1.9998	1.3749	0.9018
	2.000	89.398	14.9517	4.3295	3.4534	0.4479	0.19535		14.000	27.246	3.1808	2.1877	1.4539	0.85825
									16.000	29.287	3.6554	2.3751	1.5391	0.6728
									18.000	31.423	4.1745	2.5600	1.6306	0.5420
									20.000	33.653	4.7382	2.7406	1.7289	0.4105
														0.69731
														0.42765
														0.27783
														0.64001
														0.58349
														0.33742
														0.52883
														0.47677
														0.23710
														0.42765
														0.20791
														0.38140
														0.19442
														0.6324
														0.19104
														0.5680
														0.18855
														0.5451
														0.18664
														0.5261
														0.18512
														0.18389

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{\rho_2}{\rho_1}$	$M_2$	$\frac{T_2}{T_1}$	$\frac{p_{02}}{p_{01}}$	
3.70	20.000	83.507	15.6008	4.3797	3.5621	0.4969	0.18289	3.75	2.000	89.416	16.2379	4.4261	3.6687	0.4428	
	18.000	84.274	15.6460	4.3831	3.5696	0.4856	0.18206								0.17169
	16.000	84.998	15.6836	4.3859	3.5759	0.4760	0.18138								
	14.000	85.687	15.7147	4.3882	3.5811	0.4680	0.18082	3.80	2.000	89.416	16.2379	4.4261	3.6687	0.4428	
	12.000	86.348	15.7402	4.3901	3.5854	0.4613	0.18035								
	10.000	86.988	15.7609	4.3916	3.5889	0.4558	0.17998								
	8.000	87.610	15.7772	4.3928	3.5916	0.4515	0.17969								
	6.000	88.219	15.7896	4.3937	3.5937	0.4481	0.17947								
	4.000	88.817	15.7982	4.3944	3.5951	0.4458	0.17932								
	2.000	89.411	15.8033	4.3947	3.5960	0.4444	0.17922								
3.75	2.000	16.810	1.2055	1.1426	1.0551	3.6149	0.98933	20.000	22.000	35.556	5.5299	1.1445	1.0558	3.6624	
	4.000	18.260	1.4440	1.2982	1.1123	3.4840	0.99497								0.99931
	6.000	19.814	1.7184	1.4654	1.1727	3.3550	0.98405								0.99479
	8.000	21.470	2.0312	1.6420	1.2370	3.2264	0.96476								0.98349
	10.000	23.227	2.3849	1.8258	1.3062	3.0974	0.95634								0.96355
	12.000	25.081	2.7813	2.0142	1.3808	2.9674	0.89905	30.000	46.105	8.5816	3.5997	2.3840	1.7556	2.4644	
	14.000	27.030	3.2217	2.2046	1.4614	2.8363	0.85397	32.000	49.218	9.4923	3.7408	2.5375	1.6313	2.3283	
	16.000	29.072	3.7069	2.3943	1.5482	2.7042	0.80280	34.000	52.702	10.4940	3.8780	2.1080	1.21919	0.62373	
	18.000	31.207	4.2379	2.5813	1.6417	2.5712	0.74744	36.000	56.894	11.6543	4.0175	2.0548	0.51113	0.56627	
	20.000	33.438	4.8148	2.7637	1.7422	2.4376	0.68987	38.000	64.192	13.4871	4.2039	2.2414	1.9166	0.51113	
	22.000	35.767	5.4382	2.9401	1.8497	2.3034	0.63185	38.092	65.921	13.8756	4.2390	2.2414	1.9166	0.45902	
	24.000	38.204	6.1086	3.1095	1.9645	2.1688	0.57486	38.000	67.568	14.2269	4.2696	2.32194	1.4778	0.32194	
	26.000	40.762	6.8272	3.2714	2.0869	2.0333	0.51996	36.000	73.114	15.2586	4.3536	2.32194	1.4778	0.32194	
	28.000	43.464	7.5969	3.4259	2.2175	1.8964	0.46786	36.000	75.572	15.6341	4.3822	2.32194	1.4778	0.32194	
	30.000	46.350	8.4228	3.5733	2.3572	1.7570	0.41888	32.000	77.342	15.8710	4.3997	2.3840	1.7761	0.41022	
	32.000	49.486	9.3159	3.7148	2.5078	1.6129	0.37300	30.000	78.762	16.0402	4.4120	2.4212	1.6313	0.36471	
	34.000	53.014	10.3013	3.8529	2.6736	1.4594	0.32964	28.000	79.967	16.1687	4.4212	2.4212	1.6313	0.36471	
	36.000	57.310	11.4538	3.9947	2.8672	1.2839	0.28696	26.000	81.022	16.2697	4.4284	2.4284	1.6313	0.36471	
	37.906	65.884	13.5007	4.2052	3.2105	0.9683	0.22770	24.000	81.969	16.3512	4.4341	2.4341	1.6313	0.36471	
	36.000	72.794	14.8041	3.4287	3.4287	0.7481	0.19834	22.000	82.833	16.4178	4.4387	2.4387	1.6313	0.36471	
	34.000	75.361	15.1917	3.4384	3.4936	0.6755	0.19061	20.000	83.634	16.4729	4.4426	2.4426	1.6313	0.36471	
	32.000	77.180	15.4318	3.5369	3.5338	0.6280	0.18602	18.000	84.383	16.5186	4.4457	2.4457	1.6313	0.36471	
	30.000	78.631	15.6021	3.4798	3.5623	0.5926	0.18286	16.000	85.092	16.5567	4.4484	2.4484	1.6313	0.36471	
	28.000	79.856	15.7307	3.4894	3.5838	0.5649	0.18053	14.000	85.767	16.5882	4.4505	2.4505	1.6313	0.36471	
	26.000	80.927	15.8316	3.4968	3.6007	0.5423	0.17872	12.000	86.415	16.6141	4.4523	2.4523	1.6313	0.36471	
	24.000	81.887	15.9128	3.4028	3.6143	0.5237	0.17728	10.000	87.043	16.6352	4.4537	2.4537	1.6313	0.36471	
	22.000	82.762	15.9792	4.4076	3.6254	0.5080	0.17612	8.000	87.653	16.6518	4.4549	2.4549	1.6313	0.36471	
	20.000	83.572	16.0339	4.4115	3.6345	0.4948	0.17517	6.000	88.251	16.6643	4.4557	2.4557	1.6313	0.36471	
	18.000	84.330	16.0794	4.4148	3.6422	0.4836	0.17439	4.000	88.839	16.6731	4.4563	2.4563	1.6313	0.36471	
	16.000	85.045	16.1172	4.4175	3.6485	0.4741	0.17374	2.000	89.421	16.6783	4.4567	2.4567	1.6313	0.36471	
	14.000	85.727	16.1485	4.4198	3.6537	0.4662	0.17321								
	12.000	86.382	16.1743	4.4216	3.6580	0.4595	0.17277								
	10.000	87.016	16.1951	4.4231	3.6615	0.4541	0.17242	3.85	2.000	89.421	16.6783	4.4567	2.4567	1.6313	
	8.000	87.632	16.2116	4.4242	3.6643	0.4498	0.17214								
	6.000	88.235	16.2240	4.4251	3.6663	0.4465	0.17193								
	4.000	88.829	16.2327	4.4257	3.6678	0.4441	0.17178								

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{\rho_2}{\rho_1}$	$M_2$	$\frac{T_2}{T_1}$	$\frac{p_{02}}{p_{01}}$	
3.85	10.000	22.812	2.4328	1.8495	1.3153	0.93209	3.90	24.000	37.584	6.4345	3.1853	2.2371	2.0201	0.54918	
	12.000	24.668	2.8456	2.0432	1.3927	0.9386	0.89264	26.000	40.126	7.2035	3.3492	2.1508	2.0968	0.49366	
	14.000	26.619	3.3050	2.2386	1.4764	2.9028	0.84523	28.000	42.802	8.0258	3.5046	2.2901	1.9558	0.44158	
	16.000	28.664	3.8121	2.4330	1.5688	2.7661	0.79172	30.000	45.646	8.9059	3.6519	2.4387	1.8131	0.39322	
	18.000	30.799	4.3670	2.6239	1.6643	2.6287	0.73428	32.000	48.716	9.8536	3.7923	2.5983	1.6668	0.34848	
	20.000	33.028	4.9706	2.8097	1.7691	2.4909	0.67493	34.000	52.126	10.8901	3.9278	2.7726	1.5130	0.30686	
	22.000	35.353	5.6230	2.9887	1.8814	2.3529	0.61558	36.000	56.149	12.0723	4.0633	2.9710	1.3425	0.26706	
	24.000	37.783	6.3245	3.1601	2.0013	2.2146	0.55770	38.000	62.087	13.6897	4.2224	3.2421	1.1106	0.22309	
	26.000	40.330	7.0764	3.3234	2.1293	2.0760	0.50236	38.445	65.991	14.6407	4.3043	3.4014	0.9704	0.20173	
	28.000	43.014	7.8808	3.4785	2.2656	1.9364	0.45026	38.000	69.501	15.4023	4.3647	3.5289	0.8527	0.18658	
	30.000	45.871	8.7425	3.6259	2.4111	1.7948	0.40167	36.000	73.678	16.1768	4.4218	3.6584	0.7240	0.17273	
	32.000	48.961	9.6715	3.7666	2.5677	1.6493	0.35654	34.000	75.956	16.5334	4.4468	3.7181	0.6600	0.16682	
	34.000	52.407	10.6904	3.9030	2.7390	1.4957	0.31434	32.000	77.640	16.7653	4.4626	3.7569	0.6160	0.16313	
	36.000	56.508	11.8605	4.0404	2.9355	1.3239	0.27366	30.000	79.006	16.9380	4.4738	3.7843	0.5828	0.16052	
	38.000	62.939	13.5472	4.2095	3.2183	1.0767	0.22655	28.000	80.172	17.0613	4.4823	3.8064	0.5563	0.15857	
	38.272	65.956	14.2556	4.2721	3.3369	0.9697	0.21003	26.000	81.199	17.1629	4.4890	3.8234	0.5347	0.15705	
	38.000	68.733	14.8512	4.3214	3.4366	0.8764	0.19738	24.000	82.121	17.2449	4.4943	3.8371	0.5168	0.15583	
	36.000	73.407	15.7160	4.3883	3.5814	0.7314	0.18079	22.000	82.966	17.3122	4.4986	3.8483	0.5016	0.15485	
	34.000	75.770	16.0813	4.4150	3.6425	0.6449	0.17436	20.000	83.749	17.3680	4.5022	3.8576	0.4888	0.15404	
	32.000	77.495	16.3155	4.4316	3.6816	0.6198	0.17039	18.000	84.483	17.4143	4.5052	3.8654	0.4780	0.15337	
	30.000	78.888	16.4889	4.4433	3.7098	0.5859	0.16762	16.000	85.177	17.4529	4.5076	3.8718	0.4688	0.15281	
	28.000	80.072	16.6122	4.4522	3.7313	0.5591	0.16555	14.000	85.840	17.4850	4.5097	3.8772	0.4610	0.15235	
	26.000	81.112	16.7135	4.4591	3.7482	0.5372	0.16394	12.000	86.477	17.5113	4.5114	3.8816	0.4545	0.15198	
	24.000	82.047	16.7952	4.4646	3.7619	0.5190	0.16266	10.000	87.093	17.5327	4.5127	3.8852	0.4492	0.15167	
	22.000	82.901	16.8622	4.4691	3.7731	0.5037	0.16162	8.000	87.693	17.5496	4.5138	3.8880	0.4450	0.15143	
	20.000	83.692	16.9175	4.4728	3.7823	0.4907	0.16076	6.000	88.280	17.5623	4.5146	3.8901	0.4418	0.15125	
	18.000	84.434	16.9636	4.4758	3.7900	0.4798	0.16006	4.000	88.858	17.5713	4.5151	3.8916	0.4395	0.15113	
	16.000	85.136	17.0019	4.4784	3.7964	0.4705	0.15947	2.000	89.430	17.5766	4.5155	3.8925	0.4382	0.15105	
	14.000	85.804	17.0337	4.4805	3.8017	0.4627	0.15899								
	12.000	86.447	17.0598	4.4822	3.8061	0.4561	0.15859								
	10.000	87.068	17.0810	4.4836	3.8097	0.4508	0.15827	3.95	2.000	16.001	17.447	1.4697	1.3144	0.9923	
	8.000	87.674	17.0978	4.4847	3.8125	0.4465	0.15802			14.000	26.234	3.3902	2.2727	1.4917	
	6.000	88.266	17.1104	4.4855	3.8146	0.4433	0.15783			16.000	28.281	3.9194	2.4716	1.5858	
	4.000	88.849	17.1193	4.4861	3.8161	0.4410	0.15770			18.000	30.417	4.4992	2.6664	1.6874	
	2.000	89.426	17.1245	4.4865	3.8169	0.4397	0.15762			20.000	32.646	5.1304	2.8554	1.7967	
	3.90	2.000	16.196	17.642	1.4633	1.3104	1.1167	1.4849	1.7517	1.0571	3.7573	0.99926	0.99441	0.99141	0.9923
	6.000	19.196	1.4849	1.4849	1.7517	1.3200	1.2472	1.6694	2.0821	1.0571	3.7573	0.99926	0.99441	0.99141	0.9923
	8.000	20.854	2.0821	1.6694	2.0821	1.1167	1.1167	1.1167	1.1167	1.0571	3.7573	0.99926	0.99441	0.99141	0.9923
	10.000	22.614	2.4570	1.8614	2.4570	1.3200	1.1797	3.4830	0.92990	1.0571	3.7573	0.99926	0.99441	0.99141	0.9923
	12.000	24.472	2.8783	2.0578	2.8783	1.3200	1.2472	1.2472	0.96105	1.0571	3.7573	0.99926	0.99441	0.99141	0.9923
	14.000	26.424	3.3474	2.2557	1.4840	1.4840	1.4840	1.4840	0.88935	1.0571	3.7573	0.99926	0.99441	0.99141	0.9923
	16.000	28.469	3.8655	2.4523	1.5763	1.5763	1.5763	1.5763	0.78611	1.0571	3.7573	0.99926	0.99441	0.99141	0.9923
	18.000	30.605	4.4329	2.6452	1.6758	1.6758	1.6758	1.6758	0.72761	1.0571	3.7573	0.99926	0.99441	0.99141	0.9923
	20.000	32.834	5.0501	2.8326	1.7828	1.7828	1.7828	1.7828	0.66743	1.0571	3.7573	0.99926	0.99441	0.99141	0.9923
	22.000	35.157	5.7171	3.0129	1.8975	1.8975	1.8975	1.8975	0.60746	1.0571	3.7573	0.99926	0.99441	0.99141	0.9923

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{\rho_2}{\rho_1}$	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$	
3.95	38.000	61.406	13.8667	4.2383	3.2718	1.1389	0.21889	4.00	28.000	80.359	17.9765	4.5402	3.9594	0.5513	0.14555	
	38.612	66.026	15.0309	4.3358	3.4667	0.9711	0.19376		26.000	81.359	18.0787	4.5464	3.9765	0.5302	0.14419	
	38.000	70.101	15.9275	4.4038	3.6167	0.8345	0.17703		24.000	82.261	18.1615	4.5514	3.9903	0.5126	0.14310	
	36.000	73.928	16.6412	4.4541	3.7361	0.7172	0.16509		22.000	83.087	18.2296	4.5555	4.00117	0.4978	0.14221	
	34.000	76.131	16.9904	4.4776	3.7945	0.6554	0.15965		20.000	83.854	18.2861	4.5588	4.0111	0.4852	0.14148	
	32.000	77.777	17.2203	4.4927	3.8330	0.6125	0.15620		18.000	84.574	18.3331	4.5616	4.0190	0.4746	0.14087	
	30.000	79.120	17.3877	4.5035	3.8609	0.5798	0.15375		16.000	85.256	18.3723	4.5639	4.0255	0.4655	0.14037	
	28.000	80.268	17.5161	4.5117	3.8824	0.5537	0.15191		14.000	85.907	18.4049	4.5659	4.0310	0.4579	0.13996	
	26.000	81.281	17.6179	4.5181	3.8994	0.5324	0.15047		12.000	86.533	18.4317	4.5674	4.0355	0.4515	0.13962	
	24.000	82.192	17.7003	4.5232	3.9132	0.5147	0.14932		10.000	87.139	18.4535	4.5687	4.0391	0.4463	0.13934	
	22.000	83.028	17.7680	4.5274	3.9245	0.4997	0.14838		8.000	87.730	18.4707	4.5697	4.0420	0.4421	0.13912	
	20.000	83.803	17.8241	4.5309	3.9339	0.4870	0.14761		6.000	88.307	18.4837	4.5705	4.0442	0.4390	0.13896	
	18.000	84.529	17.8708	4.5338	3.9417	0.4762	0.14698		4.000	88.876	18.4928	4.5710	4.0457	0.4367	0.13885	
	16.000	85.218	17.9097	4.5362	3.9482	0.4671	0.14645		2.000	89.439	18.4982	4.5713	4.0466	0.4354	0.13878	
	14.000	85.874	17.9420	4.5381	3.9536	0.4594	0.14601									
	12.000	86.505	17.9686	4.5398	3.9581	0.4530	0.14566									
	10.000	87.116	17.9902	4.5411	3.9617	0.4477	0.14537									
	8.000	87.711	18.0072	4.5421	3.9645	0.4435	0.14514									
	6.000	88.294	18.0201	4.5429	3.9667	0.4404	0.14497									
	4.000	88.868	18.0291	4.5434	3.9682	0.4381	0.14485									
	2.000	89.435	18.0345	4.5438	3.9691	0.4368	0.14478									
4.00	2.000	15.813	1.2194	1.1519	1.0586	1.1519	0.9920	3.8521	0.9920	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	
	4.000	17.258	1.4763	1.3185	1.1196	1.1196	3.7089	3.7089	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401
	6.000	18.812	1.7743	1.4980	1.1844	1.1844	3.5679	3.5679	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401
	8.000	20.471	2.1166	1.6879	1.2540	1.2540	3.4273	3.4273	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401
	10.000	22.234	2.5061	1.8853	1.3293	1.3293	3.2860	3.2860	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401
	12.000	24.095	2.9445	2.0870	1.4109	1.4109	3.1439	3.1439	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401
	14.000	26.050	3.4334	2.2898	1.4994	1.4994	3.0009	3.0009	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401
	16.000	28.098	3.9741	2.4909	1.5954	1.5954	2.8570	2.8570	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401
	18.000	30.236	4.5667	2.6877	1.6991	1.6991	2.7128	2.7128	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401
	20.000	32.464	5.2116	2.8782	1.8107	1.8107	2.5686	2.5686	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401
	22.000	34.786	5.9090	3.0611	1.9304	1.9304	2.4246	2.4246	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401
	24.000	37.208	6.6592	3.2352	2.0583	2.0583	2.2809	2.2809	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401
	26.000	39.740	7.4625	3.4002	2.1947	2.1947	2.1374	2.1374	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401
	28.000	42.402	8.3215	3.5561	2.3401	2.3401	1.9935	1.9935	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401
	30.000	45.224	9.2397	3.7034	2.4949	2.4949	1.8485	1.8485	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401
	32.000	48.258	10.2259	3.8430	2.6609	2.6609	1.7006	1.7006	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401
	34.000	51.605	11.2995	3.9768	2.8413	2.8413	1.5463	1.5463	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401
	36.000	55.495	12.5100	4.1091	3.0444	3.0444	1.3776	1.3776	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401
	38.000	60.827	14.0647	4.2556	3.3049	3.3049	1.1637	1.1637	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401
	38.774	66.059	15.4261	4.3665	3.5329	3.5329	0.9717	0.9717	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401
	38.000	70.601	16.4407	4.4403	3.7026	3.7026	0.8196	0.8196	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401
	36.000	74.161	17.1095	4.4855	3.8144	3.8144	0.7109	0.7109	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401
	34.000	76.297	17.4525	4.5076	3.8718	3.8718	0.6511	0.6511	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401
	32.000	77.908	17.6808	4.5220	3.9099	3.9099	0.6090	0.6090	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401
	30.000	79.227	17.8479	4.5324	3.9379	3.9379	0.5769	0.5769	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401	0.99401