

ENGINEERING TRIPOS PART IIB

Friday 27 April 2007 9 to 10.30

Module 4A11

TURBOMACHINERY II

Answer not more than two questions.

All questions carry the same number of marks.

The approximate percentage of marks allocated to each part of a question is indicated in the right margin.

Attachment: Compressible Flow Data Book (38 pages).

STATIONERY

Single-sided script paper

SPECIAL REQUIREMENTS

Engineering Data Book

CUED approved calculator allowed

You may not start to read the questions
printed on the subsequent pages of this
question paper until instructed that you
may do so by the Invigilator

- 1 (a) Derive the Simple Radial Equilibrium equation (SRE)

$$\frac{dh_0}{dr} - T \frac{ds}{dr} - V_x \frac{dV_x}{dr} = \frac{V_\theta}{r} \frac{d(rV_\theta)}{dr}$$

from the momentum equation in the radial direction: $\rho V_\theta^2 / r = dp/dr$. State all the assumptions involved in your derivation. [20%]

- (b) Give a physical explanation for each term in the Simple Radial Equilibrium equation derived in (a). What additional assumptions are required to further reduce the equation to a relationship between the axial velocity distribution in the radial direction and the vortex distribution:

$$-V_x \frac{dV_x}{dr} = \frac{V_\theta}{r} \frac{d(rV_\theta)}{dr} \quad [20\%]$$

- (c) For a given vortex distribution at a certain axial location, the simplified SRE in part (b) above can be integrated to give the V_x distribution at this location. Describe how the integral constant is determined. [10%]

- (d) For a low hub-to-tip ratio free vortex design turbine stage with cylindrical endwalls, the flow through the stage can be assumed to have parallel streamlines. At the inlet of the stage the flow properties are uniform. Downstream of the stator the radial distributions of static pressure and density follow the relation: $p(r)/p_h = (\rho(r)/\rho_h)^\gamma$, where γ is the ratio of specific heat capacities and the subscript h denotes the location on the hub. Discuss the applicability of the simplified SRE derived in part (b) at the exit of the stator row and derive an expression for the radial distribution of velocity. [25%]

- (e) It is found that in reality the velocity distribution behind the stator does not follow the analytical prediction and that the stage loss is high. Discuss possible causes for this and suggest what change(s) to the stator could be made to improve the design. [25%]

2 (a) A two dimensional section of a transonic air compressor rotor with upstream relative Mach number 1.25 produces a static pressure ratio $p_2 / p_1 = 1.8$, of which 75% is estimated to be due to shock compression ($(p_2 / p_1)_{shock} = 1.6$) and the remaining 25% due to diffusion downstream of the shock. ($\gamma = 1.4$ for air.)

(i) Assuming that there is negligible flow turning from the upstream to the passage shock, sketch a possible blade profile and shockwave pattern for the section. Discuss quantitatively whether the main passage shock is normal or oblique with the help of the Engineering Data Book provided. [15%]

(ii) Estimate the specific entropy production, Δs , due to the shockwave. [15%]

(iii) Assume the specific entropy produced in this blade section due to the viscous drag force equals that due to the shockwave estimated in part (ii), i.e., $\Delta s_{total} = 2\Delta s_{shock}$. Estimate the isentropic efficiency of this blade section. Comment on your result. [20%]

(iv) Estimate the loss of stagnation pressure ratio of the blade section due to the production of entropy. Comment on the validity of your estimation in relation to the total entropy production and compare with the estimate of isentropic efficiency in part (iii). [20%]

(b) As exit Mach number increases beyond sonic condition for a transonic turbine blade, the actual loading can still increase but the flow turning reduces.

(i) With the help of a sketch, explain why this is so. [15%]

(ii) Describe the limit loading condition for turbine blades. A transonic turbine blade section has exit metal angle of 62° and it reaches the limit loading condition with an exit Mach number of 1.75. Calculate the deviation angle at this condition. [15%]

- 3 (a) (i) Explain the causes of secondary flow and suggest reasons why secondary flow might be expected to increase blade row losses. [30%]
- (ii) Sketch loss coefficient contours at blade row exit for an axial flow turbine and for an axial compressor, both with significant secondary flow. Explain any differences between the two sketches. [20%]
- (iii) Suggest how a low aspect ratio turbine stator could be redesigned so as to reduce secondary flow. [20%]
- (b) (i) With reference to the short-comings of other methods, explain the dominance of Euler / Navier-Stokes methods for turbomachinery blade design. Why is the finite-volume approach preferred to the finite-difference technique for this application? [20%]
- (ii) Comment on the most significant source(s) of errors / inaccuracies that Euler / Navier-Stokes methods for turbomachinery blade design may have. Suggest how they might be improved. [10%]

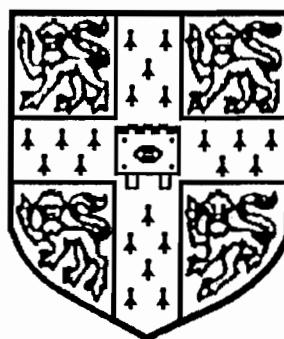
END OF PAPER

Compressible Flow

Data Book

for Part II of the
Engineering Tripos

2004 Edition



Cambridge University Engineering Department

PERFECT GAS RELATIONS FOR COMPRESSIBLE FLOW

Ratios of stagnation to static quantities

$$\frac{T}{T_0} = \left(1 + \frac{\gamma-1}{2} M^2\right)^{-1}$$

$$\frac{p}{p_0} = \left(1 + \frac{\gamma-1}{2} M^2\right)^{-\frac{\gamma}{\gamma-1}}$$

$$\frac{\rho}{\rho_0} = \left(1 + \frac{\gamma-1}{2} M^2\right)^{-\frac{1}{\gamma-1}}$$

Notes:

- (1) $T_0 = \text{const.}$ in adiabatic flow with no shaft work
- (2) If flow is isentropic, $p_0 = \text{const.}$ and $\rho_0 = \text{const.}$ when $T_0 = \text{const.}$

Mach number relations (see tables)

$$\frac{V}{\sqrt{c_p T_0}} = \sqrt{\gamma-1} M \left(1 + \frac{\gamma-1}{2} M^2\right)^{-\frac{1}{2}}$$

$$\frac{\dot{m} \sqrt{c_p T_0}}{A p_0} = \frac{\gamma}{\sqrt{\gamma-1}} M \left(1 + \frac{\gamma-1}{2} M^2\right)^{-\frac{1}{2} \left(\frac{\gamma+1}{\gamma-1}\right)}$$

$$\frac{\dot{m} \sqrt{c_p T_0}}{A p} = \frac{\gamma}{\sqrt{\gamma-1}} M \left(1 + \frac{\gamma-1}{2} M^2\right)^{\frac{1}{2}}$$

$$\frac{F}{\dot{m} \sqrt{c_p T_0}} = \frac{\sqrt{\gamma-1}}{\gamma} \frac{1+\gamma M^2}{M} \left(1 + \frac{\gamma-1}{2} M^2\right)^{-\frac{1}{2}} \quad \text{where} \quad F = (p + \rho V^2) A$$

$$\frac{\frac{1}{2} \rho V^2}{p_0} = \frac{1}{2} \gamma M^2 \left(1 + \frac{\gamma-1}{2} M^2\right)^{-\frac{\gamma}{\gamma-1}}$$

ONE-DIMENSIONAL FLOW OF A PERFECT GAS

Isentropic flow

$$\frac{A}{A^*} = \frac{1}{M} \left\{ \frac{2}{\gamma+1} \left(1 + \frac{\gamma-1}{2} M^2 \right) \right\}^{\frac{1}{2(\gamma-1)}}$$

Adiabatic constant area flow

$$\frac{4c_f L_{\max}}{D} = \frac{1-M^2}{\gamma M^2} + \frac{\gamma+1}{2\gamma} \ln \left(\frac{(\gamma+1)M^2}{2 \left(1 + \frac{\gamma-1}{2} M^2 \right)} \right)$$

Normal shock waves in perfect gases

$$VV_s = a^{*2}$$

$$M_s = \left(\frac{1 + \frac{\gamma-1}{2} M^2}{\gamma M^2 - \frac{\gamma-1}{2}} \right)^{\frac{1}{2}}$$

$$\frac{p_{0s}}{p_0} = \left(\frac{\frac{\gamma+1}{2} M^2}{1 + \frac{\gamma-1}{2} M^2} \right)^{\frac{\gamma}{\gamma-1}} \left(\frac{2\gamma}{\gamma+1} M^2 - \frac{\gamma-1}{\gamma+1} \right)^{\frac{1}{1-\gamma}}$$

$$\frac{p_s}{p} = 1 + \frac{2\gamma}{\gamma+1} (M^2 - 1)$$

$$\frac{p_{0s}}{p} = \left(\frac{\gamma+1}{2} M^2 \right)^{\frac{\gamma}{\gamma-1}} \left(\frac{2\gamma}{\gamma+1} M^2 - \frac{\gamma-1}{\gamma+1} \right)^{\frac{1}{1-\gamma}}$$

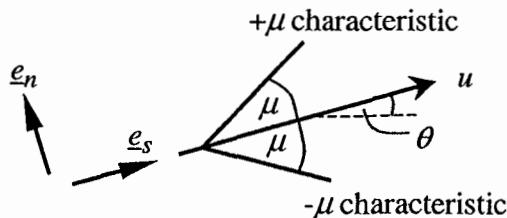
$$\frac{T_s}{T} = \frac{\gamma-1}{(\gamma+1)^2} \frac{2}{M^2} \left(1 + \frac{\gamma-1}{2} M^2 \right) \left(\frac{2\gamma}{\gamma-1} M^2 - 1 \right)$$

$$\frac{\rho_s}{\rho} = \frac{(\gamma+1)M^2}{2 \left(1 + \frac{\gamma-1}{2} M^2 \right)}$$

TWO DIMENSIONAL SUPERSONIC FLOW

Method of Characteristics for 2-D supersonic flow

Applicable to adiabatic ($h_0 = \text{constant}$), isentropic flow



Mach Number

$$M = u/c$$

Mach angle

$$\mu = \sin^{-1}\left(\frac{1}{M}\right)$$

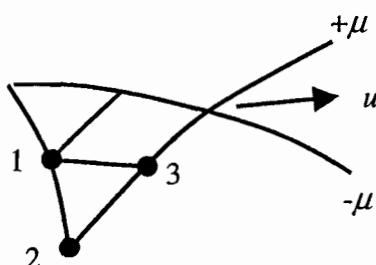
Prandtl-Meyer function

$$\nu = \int_1^M \sqrt{M^2 - 1} \frac{du}{u}$$

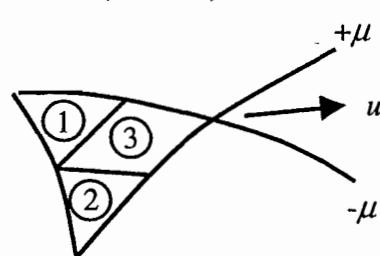
$$\nu = \sqrt{\frac{\gamma+1}{\gamma-1}} \tan^{-1} \sqrt{\frac{\gamma-1}{\gamma+1} (M^2 - 1)} - \tan^{-1} \sqrt{M^2 - 1} \quad \text{for a perfect gas}$$

Calculations

Lattice Method



Field (or wave) method



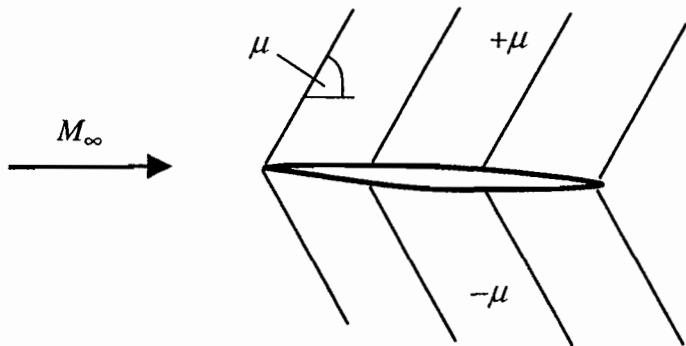
$$\nu_3 - \theta_3 = \nu_2 - \theta_2 \quad \text{along } +\mu$$

$$\nu_3 + \theta_3 = \nu_1 + \theta_1 \quad \text{along } -\mu$$

$$\nu_3 + \theta_3 = \nu_1 + \theta_1 \quad \text{across } +\mu$$

$$\nu_3 - \theta_3 = \nu_2 - \theta_2 \quad \text{across } -\mu$$

Linearised Method of Characteristics (thin film theory)

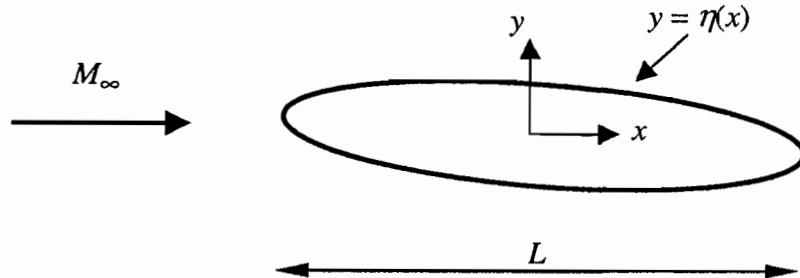


$$\mu \approx \sin^{-1}(1/M_\infty)$$

$$\Delta p \approx \pm \frac{\rho_\infty u_\infty^2 \Delta \theta}{\sqrt{M_\infty^2 - 1}} \quad \text{across } \pm \mu \text{ waves}$$

$$\text{Pressure coefficient } c_p = \frac{p - p_\infty}{\frac{1}{2} \rho_\infty u_\infty^2} = \pm \frac{2\theta}{\sqrt{M_\infty^2 - 1}} \quad \text{on upper/lower surface}$$

Prandtl-Glauert rule for linearised potential flow past geometrically similar bodies



$$\text{Pressure coefficient } c_p = \frac{p - p_\infty}{\frac{1}{2} \rho_\infty u_\infty^2}$$

For geometrically similar bodies with $\frac{\eta}{L} = f\left(\frac{x}{L}\right)$ and $c_p(M_\infty = 0) = c_{p0}$,

$$c_p = \frac{c_{p0}}{\sqrt{1 - M_\infty^2}} \quad \text{in subsonic flow}$$

$$c_p \propto \frac{1}{\sqrt{M_\infty^2 - 1}} \quad \text{in supersonic flow}$$

$\gamma=1.400$

M	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m}\sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m}\sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m}\sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2}\rho V^2}{p_0}$
0.510	0.9506	0.8374	0.8809	0.3145	0.9696	1.1579	1.1781	0.9904	0.1525
0.520	0.9487	0.8317	0.8766	0.3203	0.9828	1.1818	1.1665	0.9174	0.1574
0.530	0.9468	0.8259	0.8723	0.3262	0.9958	1.2057	1.1556	0.8496	0.1624
0.540	0.9449	0.8201	0.8679	0.3320	1.0084	1.2297	1.1452	0.7866	0.1674
0.550	0.9430	0.8142	0.8634	0.3378	1.0208	1.2538	1.1354	0.7281	0.1724
0.560	0.9410	0.8082	0.8589	0.3436	1.0328	1.2779	1.1261	0.6736	0.1774
0.570	0.9390	0.8022	0.8544	0.3493	1.0446	1.3021	1.1173	0.6229	0.1825
0.580	0.9370	0.7962	0.8498	0.3551	1.0561	1.3264	1.1090	0.5757	0.1875
0.590	0.9349	0.7901	0.8451	0.3608	1.0672	1.3507	1.1011	0.5317	0.1925
0.600	0.9328	0.7840	0.8405	0.3665	1.0781	1.3751	1.0937	0.4908	0.1976
0.610	0.9307	0.7778	0.8357	0.3722	1.0887	1.3996	1.0867	0.4527	0.2026
0.620	0.9286	0.7716	0.8310	0.3779	1.0990	1.4242	1.0800	0.4172	0.2076
0.630	0.9265	0.7654	0.8262	0.3835	1.1090	1.4489	1.0737	0.3841	0.2127
0.640	0.9243	0.7591	0.8213	0.3891	1.1186	1.4736	1.0678	0.3533	0.2177
0.650	0.9221	0.7528	0.8164	0.3948	1.1280	1.4984	1.0621	0.3246	0.2226
0.660	0.9199	0.7465	0.8115	0.4003	1.1371	1.5233	1.0568	0.2979	0.2276
0.670	0.9176	0.7401	0.8066	0.4059	1.1459	1.5483	1.0518	0.2730	0.2326
0.680	0.9153	0.7338	0.8016	0.4115	1.1544	1.5733	1.0471	0.2498	0.2375
0.690	0.9131	0.7274	0.7966	0.4170	1.1626	1.5984	1.0426	0.2282	0.2424
0.700	0.9107	0.7209	0.7916	0.4225	1.1705	1.6237	1.0384	0.2081	0.2473
0.710	0.9084	0.7145	0.7865	0.4280	1.1782	1.6490	1.0344	0.1895	0.2521
0.720	0.9061	0.7080	0.7814	0.4335	1.1855	1.6744	1.0307	0.1721	0.2569
0.730	0.9037	0.7016	0.7763	0.4389	1.1925	1.6999	1.0272	0.1561	0.2617
0.740	0.9013	0.6951	0.7712	0.4443	1.1993	1.7254	1.0239	0.1411	0.2664
0.750	0.8989	0.6886	0.7660	0.4497	1.2058	1.7511	1.0208	0.1273	0.2711
0.760	0.8964	0.6821	0.7609	0.4551	1.2119	1.7768	1.0179	0.1145	0.2758
0.770	0.8940	0.6756	0.7557	0.4605	1.2178	1.8027	1.0152	0.1026	0.2804
0.780	0.8915	0.6691	0.7505	0.4658	1.2234	1.8286	1.0126	0.0917	0.2849
0.790	0.8890	0.6625	0.7452	0.4711	1.2288	1.8547	1.0103	0.0816	0.2894
0.800	0.8865	0.6560	0.7400	0.4764	1.2338	1.8808	1.0081	0.0723	0.2939
0.810	0.8840	0.6495	0.7347	0.4817	1.2386	1.9070	1.0060	0.0638	0.2983
0.820	0.8815	0.6430	0.7295	0.4869	1.2431	1.9333	1.0041	0.0559	0.3026
0.830	0.8789	0.6365	0.7242	0.4921	1.2474	1.9598	1.0024	0.0488	0.3069
0.840	0.8763	0.6300	0.7189	0.4973	1.2514	1.9863	1.0008	0.0423	0.3112
0.850	0.8737	0.6235	0.7136	0.5025	1.2551	2.0129	0.9993	0.0363	0.3153
0.860	0.8711	0.6170	0.7083	0.5077	1.2585	2.0396	0.9979	0.0310	0.3195
0.870	0.8685	0.6106	0.7030	0.5128	1.2617	2.0665	0.9967	0.0261	0.3235
0.880	0.8659	0.6041	0.6977	0.5179	1.2646	2.0934	0.9956	0.0218	0.3275
0.890	0.8632	0.5977	0.6924	0.5230	1.2673	2.1204	0.9946	0.0179	0.3314
0.900	0.8606	0.5913	0.6870	0.5280	1.2698	2.1476	0.9937	0.0145	0.3352
0.910	0.8579	0.5849	0.6817	0.5331	1.2719	2.1748	0.9929	0.0115	0.3390
0.920	0.8552	0.5785	0.6764	0.5381	1.2739	2.2021	0.9922	0.0089	0.3427
0.930	0.8525	0.5721	0.6711	0.5431	1.2756	2.2296	0.9916	0.0067	0.3464
0.940	0.8498	0.5658	0.6658	0.5481	1.2770	2.2572	0.9911	0.0048	0.3499
0.950	0.8471	0.5595	0.6604	0.5530	1.2783	2.2848	0.9907	0.0033	0.3534
0.960	0.8444	0.5532	0.6551	0.5579	1.2793	2.3126	0.9903	0.0021	0.3569
0.970	0.8416	0.5469	0.6498	0.5628	1.2800	2.3405	0.9901	0.0011	0.3602
0.980	0.8389	0.5407	0.6445	0.5677	1.2806	2.3685	0.9899	0.0005	0.3635
0.990	0.8361	0.5345	0.6392	0.5725	1.2809	2.3966	0.9898	0.0001	0.3667
1.000	0.8333	0.5283	0.6339	0.5774	1.2810	2.4249	0.9897	0.0000	0.3698

GAS FLOW TABLES ($\gamma=1.400$): SUPERSONIC FLOW

M	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{1}{2} \rho V^2$	M_s	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	ν	M
1.010	0.8306	0.5221	0.6287	0.5821	1.2809	2.4532	0.9898	0.0001	0.3728	0.9901	1.0000	1.0235	1.9152	1.0066	0.04	1.010
1.020	0.8278	0.5160	0.6234	0.5869	1.2806	2.4817	0.9899	0.0005	0.3758	0.9805	1.0000	1.0471	1.9379	1.0132	0.13	1.020
1.030	0.8250	0.5099	0.6181	0.5917	1.2801	2.5103	0.9900	0.0010	0.3787	0.9712	1.0000	1.0711	1.9610	1.0198	0.23	1.030
1.040	0.8222	0.5039	0.6129	0.5964	1.2793	2.5390	0.9903	0.0018	0.3815	0.9620	0.9999	1.0952	1.9844	1.0263	0.35	1.040
1.050	0.8193	0.4979	0.6077	0.6011	1.2784	2.5678	0.9905	0.0027	0.3842	0.9531	0.9999	1.1196	2.0083	1.0328	0.49	1.050
1.060	0.8165	0.4919	0.6024	0.6058	1.2773	2.5967	0.9909	0.0038	0.3869	0.9444	0.9998	1.1442	2.0325	1.0393	0.64	1.060
1.070	0.8137	0.4860	0.5972	0.6104	1.2760	2.6258	0.9913	0.0051	0.3895	0.9360	0.9996	1.1691	2.0570	1.0458	0.80	1.070
1.080	0.8108	0.4800	0.5920	0.6151	1.2745	2.6549	0.9917	0.0066	0.3919	0.9277	0.9994	1.1941	2.0819	1.0522	0.97	1.080
1.090	0.8080	0.4742	0.5869	0.6197	1.2728	2.6842	0.9922	0.0082	0.3944	0.9196	0.9992	1.2195	2.1072	1.0586	1.15	1.090
1.100	0.8052	0.4684	0.5817	0.6243	1.2709	2.7136	0.9928	0.0099	0.3967	0.9118	0.9989	1.2450	2.1328	1.0649	1.34	1.100
1.110	0.8023	0.4626	0.5766	0.6288	1.2689	2.7432	0.9934	0.0118	0.3990	0.9041	0.9986	1.2708	2.1588	1.0713	1.53	1.110
1.120	0.7994	0.4568	0.5714	0.6333	1.2667	2.7728	0.9940	0.0138	0.4011	0.8966	0.9982	1.2968	2.1851	1.0776	1.74	1.120
1.130	0.7966	0.4511	0.5663	0.6379	1.2643	2.8026	0.9947	0.0159	0.4032	0.8892	0.9978	1.3231	2.2118	1.0840	1.94	1.130
1.140	0.7937	0.4455	0.5612	0.6423	1.2618	2.8325	0.9954	0.0182	0.4052	0.8820	0.9973	1.3495	2.2388	1.0903	2.16	1.140
1.150	0.7908	0.4398	0.5562	0.6468	1.2590	2.8626	0.9961	0.0205	0.4072	0.8750	0.9967	1.3763	2.2661	1.0966	2.38	1.150
1.160	0.7879	0.4343	0.5511	0.6512	1.2562	2.8927	0.9969	0.0230	0.4090	0.8682	0.9961	1.4032	2.2937	1.1029	2.61	1.160
1.170	0.7851	0.4287	0.5461	0.6556	1.2531	2.9230	0.9978	0.0255	0.4108	0.8615	0.9953	1.4304	2.3217	1.1092	2.84	1.170
1.180	0.7822	0.4232	0.5411	0.6600	1.2500	2.9534	0.9986	0.0281	0.4125	0.8549	0.9946	1.4578	2.3500	1.1154	3.07	1.180
1.190	0.7793	0.4178	0.5361	0.6644	1.2466	2.9840	0.9995	0.0309	0.4141	0.8485	0.9937	1.4855	2.3786	1.1217	3.31	1.190
1.200	0.7764	0.4124	0.5311	0.6687	1.2432	3.0147	1.0004	0.0336	0.4157	0.8422	0.9928	1.5133	2.4075	1.1280	3.56	1.200
1.210	0.7735	0.4070	0.5262	0.6730	1.2396	3.0455	1.0014	0.0365	0.4171	0.8360	0.9918	1.5415	2.4367	1.1343	3.81	1.210
1.220	0.7706	0.4017	0.5213	0.6773	1.2358	3.0764	1.0024	0.0394	0.4185	0.8300	0.9907	1.5698	2.4663	1.1405	4.06	1.220
1.230	0.7677	0.3964	0.5164	0.6816	1.2319	3.1075	1.0034	0.0424	0.4198	0.8241	0.9896	1.5984	2.4961	1.1468	4.31	1.230
1.240	0.7648	0.3912	0.5115	0.6858	1.2279	3.1387	1.0045	0.0455	0.4211	0.8183	0.9884	1.6272	2.5263	1.1531	4.57	1.240
1.250	0.7619	0.3861	0.5067	0.6901	1.2238	3.1700	1.0055	0.0486	0.4223	0.8126	0.9871	1.6563	2.5568	1.1594	4.83	1.250

$\gamma=1.400$

M	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap}$	$\frac{F}{m \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{p_0}$	M_s	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	ν	M
1.260	0.7590	0.3809	0.5019	0.6943	1.2195	3.2015	0.0066	0.0517	0.4223	0.8071	0.9857	1.6855	2.5875	1.1657	5.09	1.260
1.270	0.7561	0.3759	0.4971	0.6984	1.2152	3.2331	0.0077	0.0549	0.4244	0.8016	0.9842	1.7151	2.6186	1.1720	5.36	1.270
1.280	0.7532	0.3708	0.4923	0.7026	1.2107	3.2648	0.0089	0.0582	0.4253	0.7963	0.9827	1.7448	2.6500	1.1783	5.63	1.280
1.290	0.7503	0.3658	0.4876	0.7067	1.2061	3.2967	0.0100	0.0615	0.4262	0.7911	0.9811	1.7748	2.6816	1.1846	5.90	1.290
1.300	0.7474	0.3609	0.4829	0.7108	1.2014	3.3287	0.0112	0.0648	0.4270	0.7860	0.9794	1.8050	2.7136	1.1909	6.17	1.300
1.310	0.7445	0.3560	0.4782	0.7149	1.1965	3.3608	0.0124	0.0682	0.4277	0.7809	0.9776	1.8355	2.7459	1.1972	6.44	1.310
1.320	0.7416	0.3512	0.4736	0.7189	1.1916	3.3931	0.0136	0.0716	0.4283	0.7760	0.9758	1.8661	2.7784	1.2035	6.72	1.320
1.330	0.7387	0.3464	0.4690	0.7229	1.1866	3.4255	0.0149	0.0750	0.4289	0.7712	0.9738	1.8971	2.8112	1.2099	7.00	1.330
1.340	0.7358	0.3417	0.4644	0.7270	1.1815	3.4581	0.0161	0.0785	0.4294	0.7664	0.9718	1.9282	2.8444	1.2162	7.28	1.340
1.350	0.7329	0.3370	0.4598	0.7309	1.1763	3.4907	0.0174	0.0820	0.4299	0.7618	0.9697	1.9596	2.8778	1.2226	7.56	1.350
1.360	0.7300	0.3323	0.4553	0.7349	1.1710	3.5236	0.0187	0.0855	0.4303	0.7572	0.9676	1.9912	2.9115	1.2290	7.84	1.360
1.370	0.7271	0.3277	0.4508	0.7388	1.1656	3.5566	0.0200	0.0890	0.4306	0.7527	0.9653	2.0231	2.9455	1.2354	8.13	1.370
1.380	0.7242	0.3232	0.4463	0.7427	1.1601	3.5897	0.0213	0.0926	0.4308	0.7483	0.9630	2.0551	2.9798	1.2418	8.41	1.380
1.390	0.7213	0.3187	0.4418	0.7466	1.1546	3.6229	0.0226	0.0962	0.4310	0.7440	0.9607	2.0875	3.0144	1.2482	8.70	1.390
1.400	0.7184	0.3142	0.4374	0.7505	1.1490	3.6563	0.0240	0.0997	0.4311	0.7397	0.9582	2.1200	3.0492	1.2547	8.99	1.400
1.410	0.7155	0.3098	0.4330	0.7543	1.1433	3.6899	0.0253	0.1033	0.4312	0.7355	0.9557	2.1528	3.0844	1.2612	9.28	1.410
1.420	0.7126	0.3055	0.4287	0.7581	1.1375	3.7236	0.0267	0.1069	0.4312	0.7314	0.9531	2.1858	3.1198	1.2676	9.57	1.420
1.430	0.7097	0.3012	0.4244	0.7619	1.1317	3.7574	0.0281	0.1106	0.4311	0.7274	0.9504	2.2191	3.1555	1.2741	9.86	1.430
1.440	0.7069	0.2969	0.4201	0.7657	1.1258	3.7914	0.0295	0.1142	0.4310	0.7235	0.9476	2.2525	3.1915	1.2807	10.15	1.440
1.450	0.7040	0.2927	0.4158	0.7694	1.1198	3.8255	0.0308	0.1178	0.4308	0.7196	0.9448	2.2863	3.2278	1.2872	10.44	1.450
1.460	0.7011	0.2886	0.4116	0.7732	1.1138	3.8598	0.0323	0.1215	0.4306	0.7157	0.9420	2.3202	3.2643	1.2938	10.73	1.460
1.470	0.6982	0.2845	0.4074	0.7769	1.1077	3.8942	0.0337	0.1251	0.4303	0.7120	0.9390	2.3544	3.3011	1.3003	11.02	1.470
1.480	0.6954	0.2804	0.4032	0.7805	1.1016	3.9287	0.0351	0.1288	0.4299	0.7083	0.9360	2.3888	3.3382	1.3069	11.32	1.480
1.490	0.6925	0.2764	0.3991	0.7842	1.0954	3.9634	0.0365	0.1324	0.4295	0.7047	0.9329	2.4235	3.3756	1.3136	11.61	1.490
1.500	0.6897	0.2724	0.3950	0.7878	1.0891	3.9983	0.0379	0.1361	0.4290	0.7011	0.9298	2.4583	3.4133	1.3202	11.91	1.500

$\gamma=1.400$

M	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4 c_f L_{\max}}{D}$	$\frac{1}{2} \rho V^2}{p_0}$	M_s	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	v	M
1.510	0.68668	0.2685	0.3909	0.7914	1.0829	4.0333	0.1397	0.4285	0.6976	0.9266	2.4935	3.4512	1.3269	12.20	1.510	
1.520	0.68440	0.2646	0.3869	0.7950	1.0765	4.0684	1.0408	0.1433	0.4279	0.6941	0.9233	2.5288	3.4894	1.3336	12.49	1.520
1.530	0.68111	0.2608	0.3829	0.7986	1.0702	4.1037	1.0423	0.1470	0.4273	0.6907	0.9200	2.5644	3.5279	1.3403	12.79	1.530
1.540	0.67883	0.2570	0.3789	0.8021	1.0638	4.1392	1.0437	0.1506	0.4266	0.6874	0.9166	2.6002	3.5667	1.3470	13.09	1.540
1.550	0.6754	0.2533	0.3750	0.8057	1.0573	4.1748	1.0452	0.1543	0.4259	0.6841	0.9132	2.6363	3.6057	1.3538	13.38	1.550
1.560	0.6726	0.2496	0.3710	0.8092	1.0508	4.2105	1.0467	0.1579	0.4252	0.6809	0.9097	2.6725	3.6450	1.3606	13.68	1.560
1.570	0.6698	0.2459	0.3672	0.8126	1.0443	4.2464	1.0481	0.1615	0.4243	0.6777	0.9062	2.7091	3.6846	1.3674	13.97	1.570
1.580	0.6670	0.2423	0.3633	0.8161	1.0378	4.2825	1.0496	0.1651	0.4235	0.6746	0.9026	2.7458	3.7244	1.3742	14.27	1.580
1.590	0.6642	0.2388	0.3595	0.8195	1.0312	4.3187	1.0511	0.1688	0.4226	0.6715	0.8989	2.7828	3.7646	1.3811	14.56	1.590
1.600	0.6614	0.2353	0.3557	0.8230	1.0246	4.3551	1.0526	0.1724	0.4216	0.6684	0.8952	2.8200	3.8050	1.3880	14.86	1.600
1.610	0.6586	0.2318	0.3520	0.8263	1.0180	4.3916	1.0541	0.1760	0.4206	0.6655	0.8915	2.8575	3.8456	1.3949	15.16	1.610
1.620	0.6558	0.2284	0.3483	0.8297	1.0114	4.4282	1.0555	0.1795	0.4196	0.6625	0.8877	2.8951	3.8866	1.4018	15.45	1.620
1.630	0.6530	0.2250	0.3446	0.8331	1.0047	4.4651	1.0570	0.1831	0.4185	0.6596	0.8838	2.9331	3.9278	1.4088	15.75	1.630
1.640	0.6502	0.2217	0.3409	0.8364	0.9980	4.5020	1.0585	0.1867	0.4174	0.6568	0.8799	2.9712	3.9693	1.4158	16.04	1.640
1.650	0.6475	0.2184	0.3373	0.8397	0.9913	4.5392	1.0600	0.1902	0.4162	0.6540	0.8760	3.0096	4.0110	1.4228	16.34	1.650
1.660	0.6447	0.2151	0.3337	0.8430	0.9846	4.5765	1.0615	0.1938	0.4150	0.6512	0.8720	3.0482	4.0531	1.4299	16.63	1.660
1.670	0.6419	0.2119	0.3302	0.8462	0.9779	4.6139	1.0630	0.1973	0.4138	0.6485	0.8680	3.0871	4.0953	1.4369	16.93	1.670
1.680	0.6392	0.2088	0.3266	0.8495	0.9712	4.6515	1.0645	0.2008	0.4125	0.6458	0.8639	3.1261	4.1379	1.4440	17.22	1.680
1.690	0.6364	0.2057	0.3222	0.8527	0.9644	4.6892	1.0660	0.2043	0.4112	0.6431	0.8599	3.1655	4.1807	1.4512	17.52	1.690
1.700	0.6337	0.2026	0.3197	0.8559	0.9577	4.7272	1.0674	0.2078	0.4098	0.6405	0.8557	3.2050	4.2238	1.4583	17.81	1.700
1.710	0.6310	0.1996	0.3163	0.8591	0.9509	4.7652	1.0689	0.2113	0.4085	0.6380	0.8516	3.2448	4.2672	1.4655	18.10	1.710
1.720	0.6283	0.1966	0.3129	0.8622	0.9442	4.8035	1.0704	0.2147	0.4071	0.6355	0.8474	3.2848	4.3108	1.4727	18.40	1.720
1.730	0.6256	0.1936	0.3095	0.8654	0.9374	4.8418	1.0719	0.2182	0.4056	0.6330	0.8431	3.3251	4.3547	1.4800	18.69	1.730
1.740	0.6229	0.1907	0.3062	0.8685	0.9307	4.8804	1.0734	0.2216	0.4041	0.6305	0.8389	3.3655	4.3989	1.4873	18.98	1.740
1.750	0.6202	0.1878	0.3029	0.8716	0.9239	4.9191	1.0749	0.2250	0.4026	0.6281	0.8346	3.4063	4.4433	1.4946	19.27	1.750

$\gamma=1.400$

M	$\frac{T}{T_0}$	$\frac{P}{P_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4 c_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{p_0}$	M_s	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	v	M
1.760	0.6175	0.1850	0.2996	0.8747	0.9172	4.9580	1.0764	0.2284	0.4011	0.6257	0.8302	3.4472	4.4880	1.5019	19.56	1.760
1.770	0.6148	0.1822	0.2964	0.8777	0.9104	4.9970	1.0779	0.2318	0.3996	0.6234	0.8259	3.4884	4.5330	1.5093	19.86	1.770
1.780	0.6121	0.1794	0.2931	0.8808	0.9037	5.0362	1.0793	0.2352	0.3990	0.6210	0.8215	3.5298	4.5782	1.5167	20.15	1.780
1.790	0.6095	0.1767	0.2900	0.8838	0.8970	5.0755	1.0808	0.2385	0.3964	0.6188	0.8171	3.5715	4.6237	1.5241	20.44	1.790
1.800	0.6068	0.1740	0.2868	0.8868	0.8902	5.1150	1.0823	0.2419	0.3947	0.6165	0.8127	3.6133	4.6695	1.5316	20.73	1.800
1.810	0.6041	0.1714	0.2837	0.8898	0.8835	5.1547	1.0838	0.2452	0.3931	0.6143	0.8082	3.6555	4.7155	1.5391	21.01	1.810
1.820	0.6015	0.1688	0.2806	0.8927	0.8768	5.1945	1.0852	0.2485	0.3914	0.6121	0.8038	3.6978	4.7618	1.5466	21.30	1.820
1.830	0.5989	0.1662	0.2776	0.8957	0.8701	5.2345	1.0867	0.2518	0.3897	0.6099	0.7993	3.7404	4.8084	1.5541	21.59	1.830
1.840	0.5963	0.1637	0.2745	0.8986	0.8634	5.2747	1.0882	0.2551	0.3879	0.6078	0.7948	3.7832	4.8552	1.5617	21.88	1.840
1.850	0.5936	0.1612	0.2715	0.9015	0.8568	5.3150	1.0896	0.2583	0.3862	0.6057	0.7902	3.8263	4.9023	1.5693	22.16	1.850
1.860	0.5910	0.1587	0.2686	0.9044	0.8501	5.3555	1.0911	0.2616	0.3844	0.6036	0.7857	3.8695	4.9497	1.5770	22.45	1.860
1.870	0.5884	0.1563	0.2656	0.9072	0.8435	5.3962	1.0926	0.2648	0.3826	0.6016	0.7811	3.9131	4.9973	1.5847	22.73	1.870
1.880	0.5859	0.1539	0.2627	0.9101	0.8368	5.4370	1.0940	0.2680	0.3808	0.5996	0.7765	3.9568	5.0452	1.5924	23.02	1.880
1.890	0.5833	0.1516	0.2598	0.9129	0.8302	5.4780	1.0955	0.2712	0.3790	0.5976	0.7720	4.0008	5.0934	1.6001	23.30	1.890
1.900	0.5807	0.1492	0.2570	0.9157	0.8237	5.5191	1.0969	0.2743	0.3771	0.5956	0.7674	4.0450	5.1418	1.6079	23.59	1.900
1.910	0.5782	0.1470	0.2542	0.9185	0.8171	5.5604	1.0984	0.2775	0.3753	0.5937	0.7627	4.0895	5.1905	1.6157	23.87	1.910
1.920	0.5756	0.1447	0.2514	0.9213	0.8106	5.6019	1.0998	0.2806	0.3734	0.5918	0.7581	4.1341	5.2394	1.6236	24.15	1.920
1.930	0.5731	0.1425	0.2486	0.9240	0.8041	5.6435	1.1012	0.2837	0.3715	0.5899	0.7555	4.1791	5.2886	1.6314	24.43	1.930
1.940	0.5705	0.1403	0.2459	0.9268	0.7976	5.6853	1.1027	0.2868	0.3696	0.5880	0.7488	4.2242	5.3381	1.6394	24.71	1.940
1.950	0.5680	0.1381	0.2432	0.9295	0.7911	5.7273	1.1041	0.2899	0.3677	0.5862	0.7442	4.2696	5.3878	1.6473	24.99	1.950
1.960	0.5655	0.1360	0.2405	0.9322	0.7846	5.7695	1.1055	0.2929	0.3657	0.5844	0.7395	4.3152	5.4378	1.6553	25.27	1.960
1.970	0.5630	0.1339	0.2378	0.9349	0.7782	5.8118	1.1069	0.2960	0.3638	0.5826	0.7349	4.3611	5.4881	1.6633	25.55	1.970
1.980	0.5605	0.1318	0.2352	0.9375	0.7718	5.8542	1.1084	0.2990	0.3618	0.5808	0.7302	4.4071	5.5386	1.6713	25.83	1.980
1.990	0.5580	0.1298	0.2326	0.9402	0.7655	5.8969	1.1098	0.3020	0.3598	0.5791	0.7255	4.4535	5.5894	1.6794	26.10	1.990
2.000	0.5556	0.1278	0.2300	0.9428	0.7591	5.9397	1.1112	0.3050	0.3579	0.5774	0.7209	4.5000	5.6404	1.6875	26.38	2.000

$\gamma=1.400$

M	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{1}{2} \rho V^2}{p_0}$	M_s	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	v	M
2.010	0.5531	0.1258	0.2275	0.9454	0.7528	5.9827	1.1126	0.3080	0.3559	0.5757	0.7162	4.5468	5.6918	1.6956	26.66	2.010
2.020	0.5506	0.1239	0.2250	0.9480	0.7485	6.0258	1.1140	0.3109	0.3539	0.5740	0.7115	4.5938	5.7433	1.7038	26.93	2.020
2.030	0.5482	0.1220	0.2225	0.9506	0.7403	6.0692	1.1154	0.3138	0.3518	0.5723	0.7069	4.6411	5.7952	1.7120	27.20	2.030
2.040	0.5458	0.1201	0.2200	0.9531	0.7340	6.1126	1.1167	0.3168	0.3498	0.5707	0.7022	4.6885	5.8473	1.7203	27.48	2.040
2.050	0.5433	0.1182	0.2176	0.9557	0.7279	6.1563	1.1181	0.3197	0.3478	0.5691	0.6975	4.7363	5.8996	1.7285	27.75	2.050
2.060	0.5409	0.1164	0.2152	0.9582	0.7217	6.2001	1.1195	0.3225	0.3458	0.5675	0.6928	4.7842	5.9523	1.7369	28.02	2.060
2.070	0.5385	0.1146	0.2128	0.9607	0.7156	6.2441	1.1209	0.3254	0.3437	0.5659	0.6882	4.8324	6.0051	1.7452	28.29	2.070
2.080	0.5361	0.1128	0.2104	0.9632	0.7095	6.2883	1.1222	0.3282	0.3417	0.5643	0.6835	4.8808	6.0583	1.7536	28.56	2.080
2.090	0.5337	0.1111	0.2081	0.9657	0.7034	6.3326	1.1236	0.3310	0.3396	0.5628	0.6789	4.9295	6.1117	1.7620	28.83	2.090
2.100	0.5313	0.1094	0.2058	0.9681	0.6974	6.3772	1.1250	0.3339	0.3376	0.5613	0.6742	4.9783	6.1654	1.7705	29.10	2.100
2.110	0.5290	0.1077	0.2035	0.9706	0.6914	6.4218	1.1263	0.3366	0.3355	0.5598	0.6696	5.0275	6.2193	1.7789	29.36	2.110
2.120	0.5266	0.1060	0.2013	0.9730	0.6854	6.4667	1.1276	0.3394	0.3334	0.5583	0.6649	5.0768	6.2735	1.7875	29.63	2.120
2.130	0.5243	0.1043	0.1990	0.9754	0.6795	6.5117	1.1290	0.3422	0.3314	0.5568	0.6603	5.1264	6.3280	1.7960	29.90	2.130
2.140	0.5219	0.1027	0.1968	0.9778	0.6736	6.5569	1.1303	0.3449	0.3293	0.5554	0.6557	5.1762	6.3827	1.8046	30.16	2.140
2.150	0.5196	0.1011	0.1946	0.9802	0.6677	6.6023	1.1317	0.3476	0.3272	0.5540	0.6511	5.2263	6.4377	1.8132	30.43	2.150
2.160	0.5173	0.0996	0.1925	0.9825	0.6619	6.6478	1.1330	0.3503	0.3252	0.5525	0.6464	5.2765	6.4929	1.8219	30.69	2.160
2.170	0.5150	0.0980	0.1903	0.9849	0.6561	6.6936	1.1343	0.3530	0.3231	0.5511	0.6419	5.3271	6.5484	1.8306	30.95	2.170
2.180	0.5127	0.0965	0.1882	0.9872	0.6503	6.7395	1.1356	0.3556	0.3210	0.5498	0.6373	5.3778	6.6042	1.8393	31.21	2.180
2.190	0.5104	0.0950	0.1861	0.9895	0.6446	6.7855	1.1369	0.3583	0.3189	0.5484	0.6327	5.4288	6.6602	1.8481	31.47	2.190
2.200	0.5081	0.0935	0.1841	0.9918	0.6389	6.8318	1.1382	0.3609	0.3169	0.5471	0.6281	5.4800	6.7165	1.8569	31.73	2.200
2.210	0.5059	0.0921	0.1820	0.9941	0.6333	6.8782	1.1395	0.3635	0.3148	0.5457	0.6236	5.5315	6.7730	1.8657	31.99	2.210
2.220	0.5036	0.0906	0.1800	0.9964	0.6277	6.9248	1.1408	0.3661	0.3127	0.5444	0.6191	5.5831	6.8298	1.8746	32.25	2.220
2.230	0.5014	0.0892	0.1780	0.9986	0.6221	6.9715	1.1421	0.3687	0.3106	0.5431	0.6145	5.6351	6.8869	1.8835	32.51	2.230
2.240	0.4991	0.0878	0.1760	1.0009	0.6165	7.0185	1.1434	0.3712	0.3085	0.5418	0.6100	5.6872	6.9442	1.8924	32.76	2.240
2.250	0.4969	0.0865	0.1740	1.0031	0.6110	7.0656	1.1446	0.3738	0.3065	0.5406	0.6055	5.7396	7.0018	1.9014	33.02	2.250

$\gamma=1.400$

M	$\frac{T}{T_0}$	$\frac{P}{P_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{p_0}$	M_s	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	ν	M
2.010	0.5531	0.1258	0.2275	0.9454	0.7528	5.9827	1.1126	0.3080	0.3559	0.5757	0.7162	4.5468	5.6918	1.6956	26.66	2.010
2.020	0.5506	0.1239	0.2250	0.9480	0.7465	6.0258	1.1140	0.3109	0.3539	0.5740	0.7115	4.5938	5.7433	1.7038	26.93	2.020
2.030	0.5482	0.1220	0.2225	0.9506	0.7403	6.0692	1.1154	0.3138	0.3518	0.5723	0.7069	4.6411	5.7952	1.7120	27.20	2.030
2.040	0.5458	0.1201	0.2200	0.9531	0.7340	6.1126	1.1167	0.3168	0.3498	0.5707	0.7022	4.6885	5.8473	1.7203	27.48	2.040
2.050	0.5433	0.1182	0.2176	0.9557	0.7279	6.1563	1.1181	0.3197	0.3478	0.5691	0.6975	4.7363	5.8996	1.7285	27.75	2.050
2.060	0.5409	0.1164	0.2152	0.9582	0.7217	6.2001	1.1195	0.3225	0.3458	0.5675	0.6928	4.7842	5.9523	1.7369	28.02	2.060
2.070	0.5385	0.1146	0.2128	0.9607	0.7156	6.2441	1.1209	0.3254	0.3437	0.5659	0.6882	4.8324	6.0051	1.7452	28.29	2.070
2.080	0.5361	0.1128	0.2104	0.9632	0.7095	6.2883	1.1222	0.3282	0.3417	0.5643	0.6835	4.8808	6.0583	1.7536	28.56	2.080
2.090	0.5337	0.1111	0.2081	0.9657	0.7034	6.3326	1.1236	0.3310	0.3396	0.5628	0.6789	4.9295	6.1117	1.7620	28.83	2.090
2.100	0.5313	0.1094	0.2058	0.9681	0.6974	6.3772	1.1250	0.3339	0.3376	0.5613	0.6742	4.9783	6.1654	1.7705	29.10	2.100
2.110	0.5290	0.1077	0.2035	0.9706	0.6914	6.4218	1.1263	0.3366	0.3355	0.5598	0.6696	5.0275	6.2193	1.7789	29.36	2.110
2.120	0.5266	0.1060	0.2013	0.9730	0.6854	6.4667	1.1276	0.3394	0.3334	0.5583	0.6649	5.0768	6.2735	1.7875	29.63	2.120
2.130	0.5243	0.1043	0.1990	0.9754	0.6795	6.5117	1.1290	0.3422	0.3314	0.5568	0.6603	5.1264	6.3280	1.7960	29.90	2.130
2.140	0.5219	0.1027	0.1968	0.9778	0.6736	6.5569	1.1303	0.3449	0.3293	0.5554	0.6557	5.1762	6.3827	1.8046	30.16	2.140
2.150	0.5196	0.1011	0.1946	0.9802	0.6677	6.6023	1.1317	0.3476	0.3272	0.5540	0.6511	5.2263	6.4377	1.8132	30.43	2.150
2.160	0.5173	0.0996	0.1925	0.9825	0.6619	6.6478	1.1330	0.3503	0.3252	0.5525	0.6464	5.2765	6.4929	1.8219	30.69	2.160
2.170	0.5150	0.0980	0.1903	0.9849	0.6561	6.6936	1.1343	0.3530	0.3231	0.5511	0.6419	5.3271	6.5484	1.8306	30.95	2.170
2.180	0.5127	0.0965	0.1882	0.9872	0.6503	6.7395	1.1356	0.3556	0.3210	0.5498	0.6373	5.3778	6.6042	1.8393	31.21	2.180
2.190	0.5104	0.0950	0.1861	0.9895	0.6446	6.7855	1.1369	0.3583	0.3189	0.5484	0.6327	5.4288	6.6602	1.8481	31.47	2.190
2.200	0.5081	0.0935	0.1841	0.9918	0.6389	6.8318	1.1382	0.3609	0.3169	0.5471	0.6281	5.4800	6.7165	1.8569	31.73	2.200
2.210	0.5059	0.0921	0.1820	0.9941	0.6333	6.8782	1.1395	0.3635	0.3148	0.5457	0.6236	5.5315	6.7730	1.8657	31.99	2.210
2.220	0.5036	0.0906	0.1800	0.9964	0.6277	6.9248	1.1408	0.3661	0.3127	0.5444	0.6191	5.5831	6.8298	1.8746	32.25	2.220
2.230	0.5014	0.0892	0.1780	0.9986	0.6221	6.9715	1.1421	0.3687	0.3106	0.5431	0.6145	5.6351	6.8869	1.8835	32.51	2.230
2.240	0.4991	0.0878	0.1760	1.0009	0.6165	7.0185	1.1434	0.3712	0.3085	0.5418	0.6100	5.6872	6.9442	1.8924	32.76	2.240
2.250	0.4969	0.0865	0.1740	1.0031	0.6110	7.0656	1.1446	0.3738	0.3065	0.5406	0.6055	5.7396	7.0018	1.9014	33.02	2.250

$\gamma=1.400$

M	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap}$	$\frac{F}{\sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2}\rho V^2}{p_0}$	M_s	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	ν	M
1.760	0.6175	0.1850	0.2996	0.8747	0.9172	4.9580	0.2284	0.4011	0.6257	0.8302	3.4472	4.4880	1.5019	19.56	1.760	
1.770	0.6148	0.1822	0.2964	0.8777	0.9104	4.9970	0.1079	0.2318	0.3996	0.6234	0.8229	3.4884	4.5330	1.5093	19.86	1.770
1.780	0.6121	0.1794	0.2931	0.8808	0.9037	5.0362	1.0793	0.2352	0.3990	0.6210	0.8215	3.5298	4.5782	1.5167	20.15	1.780
1.790	0.6095	0.1767	0.2900	0.8838	0.8970	5.0755	1.0808	0.2385	0.3964	0.6188	0.8171	3.5715	4.6237	1.5241	20.44	1.790
1.800	0.6068	0.1740	0.2868	0.8868	0.8902	5.1150	1.0823	0.2419	0.3947	0.6165	0.8127	3.6133	4.6695	1.5316	20.73	1.800
1.810	0.6041	0.1714	0.2837	0.8898	0.8835	5.1547	1.0838	0.2452	0.3931	0.6143	0.8082	3.6555	4.7155	1.5391	21.01	1.810
1.820	0.6015	0.1688	0.2806	0.8927	0.8768	5.1945	1.0852	0.2485	0.3914	0.6121	0.8038	3.6978	4.7618	1.5466	21.30	1.820
1.830	0.5989	0.1662	0.2776	0.8957	0.8701	5.2345	1.0867	0.2518	0.3897	0.6099	0.7993	3.7404	4.8084	1.5541	21.59	1.830
1.840	0.5963	0.1637	0.2745	0.8986	0.8634	5.2747	1.0882	0.2551	0.3879	0.6078	0.7948	3.7832	4.8552	1.5617	21.88	1.840
1.850	0.5936	0.1612	0.2715	0.9015	0.8568	5.3150	1.0896	0.2583	0.3862	0.6057	0.7902	3.8263	4.9023	1.5693	22.16	1.850
1.860	0.5910	0.1587	0.2686	0.9044	0.8501	5.3555	1.0911	0.2616	0.3844	0.6036	0.7857	3.8695	4.9497	1.5770	22.45	1.860
1.870	0.5884	0.1563	0.2656	0.9072	0.8435	5.3962	1.0926	0.2648	0.3826	0.6016	0.7811	3.9131	4.9973	1.5847	22.73	1.870
1.880	0.5859	0.1539	0.2627	0.9101	0.8368	5.4370	1.0940	0.2680	0.3808	0.5996	0.7765	3.9568	5.0452	1.5924	23.02	1.880
1.890	0.5833	0.1516	0.2598	0.9129	0.8302	5.4780	1.0955	0.2712	0.3790	0.5976	0.7720	4.0008	5.0934	1.6001	23.30	1.890
1.900	0.5807	0.1492	0.2570	0.9157	0.8237	5.5191	1.0969	0.2743	0.3771	0.5956	0.7674	4.0450	5.1418	1.6079	23.59	1.900
1.910	0.5782	0.1470	0.2542	0.9185	0.8171	5.5604	1.0984	0.2775	0.3753	0.5937	0.7627	4.0895	5.1905	1.6157	23.87	1.910
1.920	0.5756	0.1447	0.2514	0.9213	0.8106	5.6019	1.0998	0.2806	0.3734	0.5918	0.7581	4.1341	5.2394	1.6236	24.15	1.920
1.930	0.5731	0.1425	0.2486	0.9240	0.8041	5.6435	1.1012	0.2837	0.3715	0.5899	0.7535	4.1791	5.2886	1.6314	24.43	1.930
1.940	0.5705	0.1403	0.2459	0.9268	0.7976	5.6853	1.1027	0.2868	0.3696	0.5880	0.7488	4.2242	5.3381	1.6394	24.71	1.940
1.950	0.5680	0.1381	0.2432	0.9295	0.7911	5.7273	1.1041	0.2899	0.3677	0.5862	0.7442	4.2696	5.3878	1.6473	24.99	1.950
1.960	0.5655	0.1360	0.2405	0.9322	0.7846	5.7695	1.1055	0.2929	0.3657	0.5844	0.7395	4.3152	5.4378	1.6553	25.27	1.960
1.970	0.5630	0.1339	0.2378	0.9349	0.7782	5.8118	1.1069	0.2960	0.3638	0.5826	0.7349	4.3611	5.4881	1.6633	25.55	1.970
1.980	0.5605	0.1318	0.2352	0.9375	0.7718	5.8542	1.1084	0.2990	0.3618	0.5808	0.7302	4.4071	5.5386	1.6713	25.83	1.980
1.990	0.5580	0.1298	0.2326	0.9402	0.7655	5.8969	1.1098	0.3020	0.3598	0.5791	0.7255	4.4535	5.5894	1.6794	26.10	1.990
2.000	0.5556	0.1278	0.2300	0.9428	0.7591	5.9397	1.1112	0.3050	0.3579	0.5774	0.7209	4.5000	5.6404	1.6875	26.38	2.000

$\gamma=1.400$

M	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{p_0}$	M_s	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	v	M
2.010	0.5531	0.1258	0.2275	0.9454	0.7528	5.9827	1.1126	0.3080	0.3559	0.5757	0.7162	4.5468	5.6918	1.6956	26.66	2.010
2.020	0.5506	0.1239	0.2250	0.9480	0.7465	6.0258	1.1140	0.3109	0.3539	0.5740	0.7115	4.5938	5.7433	1.7038	26.93	2.020
2.030	0.5482	0.1220	0.2225	0.9506	0.7403	6.0692	1.1154	0.3138	0.3518	0.5723	0.7069	4.6411	5.7952	1.7120	27.20	2.030
2.040	0.5458	0.1201	0.2200	0.9531	0.7340	6.1126	1.1167	0.3168	0.3498	0.5707	0.7022	4.6885	5.8473	1.7203	27.48	2.040
2.050	0.5433	0.1182	0.2176	0.9557	0.7279	6.1563	1.1181	0.3197	0.3478	0.5691	0.6975	4.7363	5.8996	1.7285	27.75	2.050
2.060	0.5409	0.1164	0.2152	0.9582	0.7217	6.2001	1.1195	0.3225	0.3458	0.5675	0.6928	4.7842	5.9523	1.7369	28.02	2.060
2.070	0.5385	0.1146	0.2128	0.9607	0.7156	6.2441	1.1209	0.3254	0.3437	0.5659	0.6882	4.8324	6.0051	1.7452	28.29	2.070
2.080	0.5361	0.1128	0.2104	0.9632	0.7095	6.2883	1.1222	0.3282	0.3417	0.5643	0.6835	4.8808	6.0583	1.7536	28.56	2.080
2.090	0.5337	0.1111	0.2081	0.9657	0.7034	6.3326	1.1236	0.3310	0.3396	0.5628	0.6789	4.9295	6.1117	1.7620	28.83	2.090
2.100	0.5313	0.1094	0.2058	0.9681	0.6974	6.3772	1.1250	0.3339	0.3376	0.5613	0.6742	4.9783	6.1654	1.7705	29.10	2.100
2.110	0.5290	0.1077	0.2035	0.9706	0.6914	6.4218	1.1263	0.3366	0.3355	0.5598	0.6696	5.0275	6.2193	1.7789	29.36	2.110
2.120	0.5266	0.1060	0.2013	0.9730	0.6854	6.4667	1.1276	0.3394	0.3334	0.5583	0.6649	5.0768	6.2785	1.7875	29.63	2.120
2.130	0.5243	0.1043	0.1990	0.9754	0.6795	6.5117	1.1290	0.3422	0.3314	0.5568	0.6603	5.1264	6.3280	1.7960	29.90	2.130
2.140	0.5219	0.1027	0.1968	0.9778	0.6736	6.5569	1.1303	0.3449	0.3293	0.5554	0.6557	5.1762	6.3827	1.8046	30.16	2.140
2.150	0.5196	0.1011	0.1946	0.9802	0.6677	6.6023	1.1317	0.3476	0.3272	0.5540	0.6511	5.2263	6.4377	1.8132	30.43	2.150
2.160	0.5173	0.0996	0.1925	0.9825	0.6619	6.6478	1.1330	0.3503	0.3252	0.5525	0.6464	5.2765	6.4929	1.8219	30.69	2.160
2.170	0.5150	0.0980	0.1903	0.9849	0.6561	6.6936	1.1343	0.3530	0.3231	0.5511	0.6419	5.3271	6.5484	1.8306	30.95	2.170
2.180	0.5127	0.0965	0.1882	0.9872	0.6503	6.7395	1.1356	0.3556	0.3210	0.5498	0.6373	5.3778	6.6042	1.8393	31.21	2.180
2.190	0.5104	0.0950	0.1861	0.9895	0.6446	6.7855	1.1369	0.3583	0.3189	0.5484	0.6327	5.4288	6.6602	1.8481	31.47	2.190
2.200	0.5081	0.0935	0.1841	0.9918	0.6389	6.8318	1.1382	0.3609	0.3169	0.5471	0.6281	5.4800	6.7165	1.8569	31.73	2.200
2.210	0.5059	0.0921	0.1820	0.9941	0.6333	6.8782	1.1395	0.3635	0.3148	0.5457	0.6236	5.5315	6.7730	1.8657	31.99	2.210
2.220	0.5036	0.0906	0.1800	0.9964	0.6277	6.9248	1.1408	0.3661	0.3127	0.5444	0.6191	5.5831	6.8298	1.8746	32.25	2.220
2.230	0.5014	0.0892	0.1780	0.9986	0.6221	6.9715	1.1421	0.3687	0.3106	0.5431	0.6145	5.6351	6.8869	1.8835	32.51	2.230
2.240	0.4991	0.0878	0.1760	1.0009	0.6165	7.0185	1.1434	0.3712	0.3085	0.5418	0.6100	5.6872	6.9442	1.8924	32.76	2.240
2.250	0.4969	0.0865	0.1740	1.0031	0.6110	7.0656	1.1446	0.3738	0.3065	0.5406	0.6055	5.7396	7.0018	1.9014	33.02	2.250

$\gamma=1.400$

M	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4 c_f L_{\max}}{D}$	$\frac{1}{2} \rho V^2}{p_0}$	M_s	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{T_s}{T}$	$\frac{P_{0s}}{P}$	$\frac{P_s}{P}$	$\frac{T_s}{T}$	ν	M
2.260	0.4947	0.0851	0.1721	1.0053	0.6056	7.1129	1.1459	0.3763	0.3044	0.5393	0.6011	5.7922	7.0597	1.9104	33.27	2.260		
2.270	0.4925	0.0838	0.1702	1.0075	0.6002	7.1603	1.1472	0.3788	0.3023	0.5381	0.5966	5.8451	7.1178	1.9194	33.53	2.270		
2.280	0.4903	0.0825	0.1683	1.0097	0.5948	7.2080	1.1484	0.3813	0.3003	0.5368	0.5921	5.8981	7.1762	1.9285	33.78	2.280		
2.290	0.4881	0.0812	0.1664	1.0118	0.5894	7.2558	1.1497	0.3838	0.2982	0.5356	0.5877	5.9515	7.2348	1.9376	34.03	2.290		
2.300	0.4859	0.0800	0.1646	1.0140	0.5841	7.3038	1.1509	0.3862	0.2961	0.5344	0.5833	6.0050	7.2937	1.9468	34.28	2.300		
2.310	0.4837	0.0787	0.1628	1.0161	0.5788	7.3520	1.1521	0.3887	0.2941	0.5332	0.5789	6.0598	7.3528	1.9560	34.53	2.310		
2.320	0.4816	0.0775	0.1609	1.0182	0.5736	7.4003	1.1534	0.3911	0.2920	0.5321	0.5745	6.1128	7.4122	1.9652	34.78	2.320		
2.330	0.4794	0.0763	0.1592	1.0204	0.5684	7.4488	1.1546	0.3935	0.2900	0.5309	0.5702	6.1671	7.4719	1.9745	35.03	2.330		
2.340	0.4773	0.0751	0.1574	1.0224	0.5632	7.4975	1.1558	0.3959	0.2879	0.5297	0.5658	6.2215	7.5319	1.9838	35.28	2.340		
2.350	0.4752	0.0740	0.1556	1.0245	0.5581	7.5464	1.1570	0.3983	0.2859	0.5286	0.5615	6.2763	7.5920	1.9931	35.53	2.350		
2.360	0.4731	0.0728	0.1539	1.0266	0.5530	7.5955	1.1582	0.4006	0.2839	0.5275	0.5572	6.3312	7.6525	2.0025	35.77	2.360		
2.370	0.4709	0.0717	0.1522	1.0286	0.5480	7.6447	1.1595	0.4030	0.2818	0.5264	0.5529	6.3864	7.7132	2.0119	36.02	2.370		
2.380	0.4688	0.0706	0.1505	1.0307	0.5430	7.6941	1.1606	0.4053	0.2798	0.5253	0.5486	6.4418	7.7742	2.0213	36.26	2.380		
2.390	0.4668	0.0695	0.1488	1.0327	0.5380	7.7437	1.1618	0.4076	0.2778	0.5242	0.5444	6.4975	7.8354	2.0308	36.50	2.390		
2.400	0.4647	0.0684	0.1472	1.0347	0.5331	7.7935	1.1630	0.4099	0.2758	0.5231	0.5401	6.5533	7.8969	2.0403	36.75	2.400		
2.410	0.4626	0.0673	0.1456	1.0367	0.5282	7.8434	1.1642	0.4122	0.2738	0.5221	0.5359	6.6095	7.9587	2.0499	36.99	2.410		
2.420	0.4606	0.0663	0.1439	1.0387	0.5233	7.8935	1.1654	0.4144	0.2718	0.5210	0.5317	6.6658	8.0207	2.0595	37.23	2.420		
2.430	0.4585	0.0653	0.1424	1.0407	0.5185	7.9438	1.1665	0.4167	0.2698	0.5200	0.5276	6.7224	8.0830	2.0691	37.47	2.430		
2.440	0.4565	0.0643	0.1408	1.0426	0.5137	7.9943	1.1677	0.4189	0.2678	0.5189	0.5234	6.7792	8.1455	2.0788	37.71	2.440		
2.450	0.4544	0.0633	0.1392	1.0446	0.5090	8.0450	1.1689	0.4211	0.2658	0.5179	0.5193	6.8363	8.2083	2.0885	37.95	2.450		
2.460	0.4524	0.0623	0.1377	1.0465	0.5043	8.0958	1.1700	0.4233	0.2639	0.5169	0.5152	6.8835	8.2713	2.0982	38.18	2.460		
2.470	0.4504	0.0613	0.1362	1.0484	0.4996	8.1468	1.1712	0.4255	0.2619	0.5159	0.5111	6.9511	8.3346	2.1080	38.42	2.470		
2.480	0.4484	0.0604	0.1346	1.0503	0.4950	8.1980	1.1723	0.4277	0.2599	0.5149	0.5071	7.0088	8.3982	2.1178	38.66	2.480		
2.490	0.4464	0.0594	0.1332	1.0522	0.4904	8.2494	1.1734	0.4298	0.2580	0.5140	0.5030	7.0668	8.4620	2.1276	38.89	2.490		
2.500	0.4444	0.0585	0.1317	1.0541	0.4858	8.3010	1.1746	0.4320	0.2561	0.5130	0.4990	7.1250	8.5261	2.1375	39.12	2.500		

$\gamma=1.400$

M	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4 c_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{p_0}$	M_s	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	ν	M
2.510	0.4425	0.0576	0.1302	1.0560	0.4813	8.3527	1.1757	0.4341	0.2541	0.5120	0.4950	7.1835	8.5905	2.1474	39.36	2.510
2.520	0.4405	0.0567	0.1288	1.0578	0.4768	8.4046	1.1768	0.4362	0.2522	0.5111	0.4911	7.2421	8.6551	2.1574	39.59	2.520
2.530	0.4386	0.0559	0.1274	1.0597	0.4724	8.4567	1.1779	0.4383	0.2503	0.5102	0.4871	7.3011	8.7200	2.1674	39.82	2.530
2.540	0.4366	0.0550	0.1260	1.0615	0.4680	8.5090	1.1790	0.4404	0.2484	0.5092	0.4832	7.3602	8.7851	2.1774	40.05	2.540
2.550	0.4347	0.0542	0.1246	1.0633	0.4636	8.5615	1.1801	0.4425	0.2465	0.5083	0.4793	7.4196	8.8505	2.1875	40.28	2.550
2.560	0.4328	0.0533	0.1232	1.0651	0.4593	8.6141	1.1812	0.4445	0.2446	0.5074	0.4754	7.4792	8.9161	2.1976	40.51	2.560
2.570	0.4309	0.0525	0.1218	1.0669	0.4550	8.6670	1.1823	0.4466	0.2427	0.5065	0.4715	7.5391	8.9820	2.2077	40.74	2.570
2.580	0.4289	0.0517	0.1205	1.0687	0.4507	8.7200	1.1834	0.4486	0.2409	0.5056	0.4677	7.5991	9.0482	2.2179	40.96	2.580
2.590	0.4271	0.0509	0.1192	1.0705	0.4465	8.7732	1.1844	0.4506	0.2390	0.5047	0.4639	7.6595	9.1146	2.2281	41.19	2.590
2.600	0.4252	0.0501	0.1179	1.0722	0.4423	8.8265	1.1855	0.4526	0.2371	0.5039	0.4601	7.7200	9.1813	2.2383	41.41	2.600
2.610	0.4233	0.0493	0.1166	1.0740	0.4382	8.8801	1.1866	0.4546	0.2353	0.5030	0.4564	7.7808	9.2483	2.2486	41.64	2.610
2.620	0.4214	0.0486	0.1153	1.0757	0.4341	8.9338	1.1876	0.4565	0.2335	0.5022	0.4526	7.8418	9.3155	2.2580	41.86	2.620
2.630	0.4196	0.0478	0.1140	1.0774	0.4300	8.9877	1.1887	0.4585	0.2317	0.5013	0.4489	7.9031	9.3829	2.2683	42.09	2.630
2.640	0.4177	0.0471	0.1128	1.0791	0.4260	9.0418	1.1897	0.4604	0.2298	0.5005	0.4452	7.9645	9.4506	2.2797	42.31	2.640
2.650	0.4159	0.0464	0.1115	1.0808	0.4220	9.0961	1.1908	0.4624	0.2280	0.4996	0.4416	8.0263	9.5186	2.2902	42.53	2.650
2.660	0.4141	0.0457	0.1103	1.0825	0.4180	9.1506	1.1918	0.4643	0.2262	0.4988	0.4379	8.0882	9.5869	2.3006	42.75	2.660
2.670	0.4122	0.0450	0.1091	1.0842	0.4141	9.2052	1.1928	0.4662	0.2245	0.4980	0.4343	8.1504	9.6554	2.3111	42.97	2.670
2.680	0.4104	0.0443	0.1079	1.0859	0.4102	9.2601	1.1939	0.4681	0.2227	0.4972	0.4307	8.2128	9.7241	2.3217	43.19	2.680
2.690	0.4086	0.0436	0.1067	1.0875	0.4063	9.3151	1.1949	0.4700	0.2209	0.4964	0.4271	8.2755	9.7931	2.3323	43.40	2.690
2.700	0.4068	0.0430	0.1056	1.0892	0.4025	9.3703	1.1959	0.4718	0.2192	0.4956	0.4236	8.3383	9.8624	2.3429	43.62	2.700
2.710	0.4051	0.0423	0.1044	1.0908	0.3987	9.4257	1.1969	0.4737	0.2174	0.4949	0.4201	8.4015	9.9319	2.3536	43.84	2.710
2.720	0.4033	0.0417	0.1033	1.0924	0.3949	9.4812	1.1979	0.4755	0.2157	0.4941	0.4166	8.4648	10.0017	2.3642	44.05	2.720
2.730	0.4015	0.0410	0.1022	1.0941	0.3912	9.5370	1.1989	0.4773	0.2140	0.4933	0.4131	8.5284	10.0718	2.3750	44.27	2.730
2.740	0.3998	0.0404	0.1010	1.0957	0.3875	9.5929	1.1999	0.4791	0.2123	0.4926	0.4097	8.5922	10.1421	2.3858	44.48	2.740
2.750	0.3980	0.0398	0.0999	1.0973	0.3858	9.6490	1.2009	0.4809	0.2106	0.4918	0.4062	8.6563	10.2127	2.3966	44.69	2.750

$\gamma=1.400$

M	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2}\rho V^2}{p_0}$	M_s	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	v	M
2.760	0.3963	0.0392	0.0989	1.0988	0.3802	9.7053	1.2019	0.4827	0.2089	0.4911	0.4028	8.7205	10.2835	2.4074	44.91	2.760
2.770	0.3945	0.0386	0.0978	1.1004	0.3766	9.7618	1.2029	0.4845	0.2072	0.4903	0.3994	8.7851	10.3546	2.4183	45.12	2.770
2.780	0.3928	0.0380	0.0967	1.1020	0.3730	9.8185	1.2038	0.4863	0.2055	0.4896	0.3961	8.8498	10.4259	2.4292	45.33	2.780
2.790	0.3911	0.0374	0.0957	1.1035	0.3695	9.8753	1.2048	0.4880	0.2039	0.4889	0.3928	8.9148	10.4975	2.4402	45.54	2.790
2.800	0.3894	0.0368	0.0946	1.1051	0.3660	9.9324	1.2058	0.4898	0.2022	0.4882	0.3895	8.9800	10.5694	2.4512	45.75	2.800
2.810	0.3877	0.0363	0.0936	1.1066	0.3625	9.9896	1.2067	0.4915	0.2006	0.4875	0.3862	9.0455	10.6415	2.4622	45.95	2.810
2.820	0.3860	0.0357	0.0926	1.1081	0.3591	10.0470	1.2077	0.4932	0.1990	0.4868	0.3829	9.1111	10.7139	2.4733	46.16	2.820
2.830	0.3844	0.0352	0.0916	1.1096	0.3557	10.1046	1.2086	0.4949	0.1973	0.4861	0.3797	9.1771	10.7865	2.4844	46.37	2.830
2.840	0.3827	0.0347	0.0906	1.1111	0.3523	10.1624	1.2095	0.4966	0.1957	0.4854	0.3765	9.2432	10.8594	2.4955	46.57	2.840
2.850	0.3810	0.0341	0.0896	1.1126	0.3490	10.2204	1.2105	0.4983	0.1941	0.4847	0.3733	9.3096	10.9326	2.5067	46.78	2.850
2.860	0.3794	0.0336	0.0886	1.1141	0.3457	10.2785	1.2114	0.5000	0.1926	0.4840	0.3701	9.3762	11.0060	2.5179	46.98	2.860
2.870	0.3777	0.0331	0.0877	1.1156	0.3424	10.3368	1.2123	0.5016	0.1910	0.4833	0.3670	9.4431	11.0797	2.5292	47.19	2.870
2.880	0.3761	0.0326	0.0867	1.1171	0.3392	10.3954	1.2132	0.5033	0.1894	0.4827	0.3639	9.5101	11.1536	2.5405	47.39	2.880
2.890	0.3745	0.0321	0.0858	1.1185	0.3359	10.4541	1.2142	0.5049	0.1879	0.4820	0.3608	9.5775	11.2278	2.5518	47.59	2.890
2.900	0.3729	0.0317	0.0849	1.1199	0.3328	10.5130	1.2151	0.5065	0.1863	0.4814	0.3577	9.6450	11.3022	2.5632	47.79	2.900
2.910	0.3712	0.0312	0.0840	1.1214	0.3296	10.5720	1.2160	0.5081	0.1848	0.4807	0.3547	9.7128	11.3770	2.5746	47.99	2.910
2.920	0.3696	0.0307	0.0831	1.1228	0.3265	10.6313	1.2169	0.5097	0.1833	0.4801	0.3517	9.7808	11.4519	2.5861	48.19	2.920
2.930	0.3681	0.0302	0.0822	1.1242	0.3234	10.6908	1.2178	0.5113	0.1818	0.4795	0.3487	9.8491	11.5271	2.5976	48.39	2.930
2.940	0.3665	0.0298	0.0813	1.1256	0.3203	10.7504	1.2187	0.5129	0.1803	0.4788	0.3457	9.9175	11.6026	2.6091	48.59	2.940
2.950	0.3649	0.0293	0.0804	1.1270	0.3173	10.8102	1.2195	0.5145	0.1788	0.4782	0.3428	9.9863	11.6784	2.6206	48.78	2.950
2.960	0.3633	0.0289	0.0796	1.1284	0.3143	10.8702	1.2204	0.5160	0.1773	0.4776	0.3398	10.0552	11.7544	2.6322	48.98	2.960
2.970	0.3618	0.0285	0.0787	1.1298	0.3113	10.9304	1.2213	0.5176	0.1758	0.4770	0.3369	10.1244	11.8306	2.6439	49.18	2.970
2.980	0.3602	0.0281	0.0779	1.1312	0.3083	10.9908	1.2222	0.5191	0.1744	0.4764	0.3340	10.1938	11.9072	2.6555	49.37	2.980
2.990	0.3587	0.0276	0.0770	1.1325	0.3054	11.0514	1.2230	0.5206	0.1729	0.4758	0.3312	10.2635	11.9839	2.6673	49.56	2.990
3.000	0.3571	0.0272	0.0762	1.1339	0.3025	11.1122	1.2239	0.5222	0.1715	0.4752	0.3283	10.3333	12.0610	2.6790	49.76	3.000

GAS FLOW TABLES ($\gamma=1.333$): SUBSONIC FLOW

M	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{p_0}$
0.010	1.0000	0.9999	1.0000	0.0058	0.0231	0.0231	43.2958	7493.200	0.0001
0.020	0.9999	0.9997	0.9998	0.0115	0.0462	0.0462	21.6560	1868.007	0.0003
0.030	0.9999	0.9994	0.9996	0.0173	0.0693	0.0693	14.4464	826.7890	0.0006
0.040	0.9997	0.9989	0.9992	0.0231	0.0923	0.0924	10.8442	462.6179	0.0011
0.050	0.9996	0.9983	0.9988	0.0288	0.1153	0.1155	8.6851	294.2161	0.0017
0.060	0.9994	0.9976	0.9982	0.0346	0.1383	0.1386	7.2475	202.8455	0.0024
0.070	0.9992	0.9967	0.9976	0.0404	0.1612	0.1618	6.2222	147.8292	0.0033
0.080	0.9989	0.9957	0.9968	0.0461	0.1841	0.1849	5.4546	112.1800	0.0042
0.090	0.9987	0.9946	0.9960	0.0519	0.2069	0.2080	4.8587	87.7848	0.0054
0.100	0.9983	0.9934	0.9950	0.0577	0.2297	0.2312	4.3831	70.3719	0.0066
0.110	0.9980	0.9920	0.9940	0.0634	0.2523	0.2544	3.9949	57.5186	0.0080
0.120	0.9976	0.9905	0.9928	0.0692	0.2749	0.2775	3.6724	47.7680	0.0095
0.130	0.9972	0.9888	0.9916	0.0749	0.2974	0.3007	3.4003	40.2012	0.0111
0.140	0.9967	0.9870	0.9903	0.0807	0.3197	0.3239	3.1678	34.2155	0.0129
0.150	0.9963	0.9851	0.9888	0.0864	0.3420	0.3471	2.9670	29.4027	0.0148
0.160	0.9958	0.9831	0.9873	0.0921	0.3641	0.3704	2.7920	25.4777	0.0168
0.170	0.9952	0.9810	0.9857	0.0979	0.3861	0.3936	2.6383	22.2372	0.0189
0.180	0.9946	0.9787	0.9840	0.1036	0.4080	0.4169	2.5022	19.5326	0.0211
0.190	0.9940	0.9763	0.9822	0.1093	0.4298	0.4402	2.3809	17.2536	0.0235
0.200	0.9934	0.9738	0.9803	0.1150	0.4514	0.4635	2.2724	15.3166	0.0260
0.210	0.9927	0.9711	0.9783	0.1207	0.4728	0.4869	2.1747	13.6578	0.0285
0.220	0.9920	0.9684	0.9762	0.1264	0.4941	0.5102	2.0863	12.2273	0.0312
0.230	0.9913	0.9655	0.9740	0.1321	0.5152	0.5336	2.0061	10.9859	0.0340
0.240	0.9905	0.9625	0.9717	0.1378	0.5362	0.5570	1.9330	9.9026	0.0370
0.250	0.9897	0.9594	0.9694	0.1435	0.5569	0.5805	1.8662	8.9522	0.0400
0.260	0.9889	0.9562	0.9669	0.1492	0.5775	0.6040	1.8049	8.1146	0.0431
0.270	0.9880	0.9529	0.9644	0.1549	0.5979	0.6275	1.7486	7.3731	0.0463
0.280	0.9871	0.9494	0.9618	0.1605	0.6181	0.6510	1.6966	6.7140	0.0496
0.290	0.9862	0.9459	0.9591	0.1662	0.6380	0.6746	1.6486	6.1261	0.0530
0.300	0.9852	0.9422	0.9563	0.1718	0.6578	0.6982	1.6042	5.5998	0.0565
0.310	0.9843	0.9384	0.9534	0.1775	0.6774	0.7218	1.5629	5.1272	0.0601
0.320	0.9832	0.9346	0.9505	0.1831	0.6967	0.7455	1.5245	4.7016	0.0638
0.330	0.9822	0.9306	0.9475	0.1887	0.7158	0.7692	1.4888	4.3173	0.0675
0.340	0.9811	0.9265	0.9444	0.1943	0.7347	0.7929	1.4555	3.9693	0.0714
0.350	0.9800	0.9224	0.9412	0.1999	0.7533	0.8167	1.4244	3.6535	0.0753
0.360	0.9789	0.9181	0.9379	0.2055	0.7717	0.8405	1.3953	3.3663	0.0793
0.370	0.9777	0.9137	0.9346	0.2111	0.7898	0.8644	1.3680	3.1046	0.0834
0.380	0.9765	0.9093	0.9311	0.2167	0.8077	0.8883	1.3425	2.8655	0.0875
0.390	0.9753	0.9047	0.9276	0.2223	0.8253	0.9122	1.3185	2.6469	0.0917
0.400	0.9741	0.9001	0.9241	0.2278	0.8427	0.9362	1.2959	2.4466	0.0960
0.410	0.9728	0.8954	0.9204	0.2334	0.8598	0.9603	1.2747	2.2627	0.1003
0.420	0.9715	0.8906	0.9167	0.2389	0.8766	0.9843	1.2548	2.0937	0.1047
0.430	0.9701	0.8857	0.9130	0.2444	0.8932	1.0085	1.2360	1.9382	0.1091
0.440	0.9688	0.8807	0.9091	0.2499	0.9095	1.0326	1.2183	1.7949	0.1136
0.450	0.9674	0.8757	0.9052	0.2554	0.9255	1.0569	1.2016	1.6627	0.1182
0.460	0.9660	0.8706	0.9012	0.2609	0.9412	1.0811	1.1858	1.5405	0.1228
0.470	0.9645	0.8654	0.8972	0.2664	0.9567	1.1055	1.1710	1.4276	0.1274
0.480	0.9631	0.8601	0.8931	0.2718	0.9718	1.1299	1.1569	1.3231	0.1321
0.490	0.9616	0.8548	0.8890	0.2773	0.9867	1.1543	1.1436	1.2263	0.1368
0.500	0.9600	0.8494	0.8847	0.2827	1.0012	1.1788	1.1310	1.1365	0.1415

$\gamma=1.333$

M	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m}\sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m}\sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m}\sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2}\rho V^2}{p_0}$
0.510	0.9585	0.8439	0.8805	0.2881	1.0155	1.2033	1.1192	1.0532	0.1463
0.520	0.9569	0.8384	0.8761	0.2935	1.0295	1.2279	1.1079	0.9759	0.1511
0.530	0.9553	0.8328	0.8717	0.2989	1.0431	1.2526	1.0973	0.9041	0.1559
0.540	0.9537	0.8271	0.8673	0.3043	1.0565	1.2773	1.0872	0.8373	0.1608
0.550	0.9520	0.8214	0.8628	0.3097	1.0696	1.3021	1.0777	0.7752	0.1656
0.560	0.9504	0.8157	0.8583	0.3150	1.0823	1.3269	1.0687	0.7174	0.1705
0.570	0.9487	0.8099	0.8537	0.3204	1.0948	1.3518	1.0601	0.6636	0.1754
0.580	0.9470	0.8040	0.8490	0.3257	1.1069	1.3768	1.0520	0.6136	0.1803
0.590	0.9452	0.7981	0.8443	0.3310	1.1188	1.4018	1.0444	0.5669	0.1852
0.600	0.9434	0.7921	0.8396	0.3363	1.1303	1.4269	1.0371	0.5235	0.1901
0.610	0.9417	0.7861	0.8348	0.3416	1.1415	1.4521	1.0303	0.4830	0.1950
0.620	0.9398	0.7801	0.8300	0.3469	1.1524	1.4773	1.0238	0.4452	0.1999
0.630	0.9380	0.7740	0.8252	0.3521	1.1630	1.5026	1.0176	0.4101	0.2048
0.640	0.9362	0.7679	0.8203	0.3573	1.1733	1.5280	1.0118	0.3773	0.2096
0.650	0.9343	0.7618	0.8153	0.3626	1.1833	1.5534	1.0063	0.3467	0.2145
0.660	0.9324	0.7556	0.8104	0.3678	1.1930	1.5789	1.0011	0.3183	0.2194
0.670	0.9305	0.7494	0.8054	0.3729	1.2023	1.6045	0.9962	0.2918	0.2242
0.680	0.9285	0.7431	0.8003	0.3781	1.2114	1.6301	0.9916	0.2671	0.2290
0.690	0.9266	0.7368	0.7953	0.3833	1.2201	1.6559	0.9872	0.2441	0.2338
0.700	0.9246	0.7306	0.7902	0.3884	1.2285	1.6817	0.9831	0.2227	0.2386
0.710	0.9226	0.7242	0.7850	0.3935	1.2367	1.7075	0.9792	0.2028	0.2433
0.720	0.9205	0.7179	0.7799	0.3986	1.2445	1.7335	0.9755	0.1843	0.2480
0.730	0.9185	0.7116	0.7747	0.4037	1.2520	1.7595	0.9721	0.1671	0.2527
0.740	0.9164	0.7052	0.7695	0.4088	1.2592	1.7856	0.9688	0.1512	0.2574
0.750	0.9144	0.6988	0.7643	0.4139	1.2661	1.8118	0.9658	0.1364	0.2620
0.760	0.9123	0.6924	0.7590	0.4189	1.2727	1.8381	0.9629	0.1227	0.2666
0.770	0.9102	0.6860	0.7537	0.4239	1.2790	1.8644	0.9603	0.1100	0.2711
0.780	0.9080	0.6796	0.7484	0.4289	1.2850	1.8908	0.9578	0.0983	0.2756
0.790	0.9059	0.6732	0.7431	0.4339	1.2907	1.9174	0.9554	0.0875	0.2800
0.800	0.9037	0.6668	0.7378	0.4389	1.2961	1.9440	0.9533	0.0776	0.2844
0.810	0.9015	0.6603	0.7325	0.4438	1.3013	1.9706	0.9513	0.0685	0.2888
0.820	0.8993	0.6539	0.7271	0.4487	1.3061	1.9974	0.9494	0.0601	0.2930
0.830	0.8971	0.6475	0.7217	0.4536	1.3107	2.0243	0.9477	0.0524	0.2973
0.840	0.8949	0.6411	0.7164	0.4585	1.3149	2.0512	0.9461	0.0454	0.3015
0.850	0.8926	0.6346	0.7110	0.4634	1.3189	2.0782	0.9446	0.0391	0.3056
0.860	0.8904	0.6282	0.7056	0.4683	1.3226	2.1053	0.9433	0.0333	0.3097
0.870	0.8881	0.6218	0.7002	0.4731	1.3260	2.1326	0.9420	0.0281	0.3137
0.880	0.8858	0.6154	0.6948	0.4779	1.3292	2.1599	0.9409	0.0235	0.3176
0.890	0.8835	0.6090	0.6893	0.4827	1.3321	2.1873	0.9399	0.0193	0.3215
0.900	0.8812	0.6026	0.6839	0.4875	1.3347	2.2147	0.9390	0.0156	0.3253
0.910	0.8788	0.5963	0.6785	0.4923	1.3370	2.2423	0.9383	0.0124	0.3291
0.920	0.8765	0.5899	0.6731	0.4970	1.3391	2.2700	0.9376	0.0096	0.3328
0.930	0.8741	0.5836	0.6676	0.5018	1.3410	2.2978	0.9370	0.0072	0.3364
0.940	0.8717	0.5773	0.6622	0.5065	1.3425	2.3256	0.9365	0.0052	0.3400
0.950	0.8694	0.5710	0.6568	0.5111	1.3439	2.3536	0.9360	0.0035	0.3435
0.960	0.8670	0.5647	0.6514	0.5158	1.3449	2.3817	0.9357	0.0022	0.3469
0.970	0.8646	0.5585	0.6459	0.5205	1.3458	2.4098	0.9354	0.0012	0.3502
0.980	0.8621	0.5522	0.6405	0.5251	1.3464	2.4381	0.9353	0.0005	0.3535
0.990	0.8597	0.5460	0.6351	0.5297	1.3467	2.4664	0.9351	0.0001	0.3567
1.000	0.8573	0.5398	0.6297	0.5343	1.3468	2.4949	0.9351	0.0000	0.3598

GAS FLOW TABLES ($\gamma=1.333$): SUPERSONIC FLOW

M	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4 c_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{p_0}$
1.010	0.8548	0.5337	0.6243	0.5389	1.3467	2.5234	0.9351	0.0001	0.3628
1.020	0.8524	0.5276	0.6189	0.5434	1.3464	2.5521	0.9352	0.0005	0.3658
1.030	0.8499	0.5215	0.6136	0.5479	1.3458	2.5809	0.9354	0.0011	0.3687
1.040	0.8474	0.5154	0.6082	0.5525	1.3450	2.6097	0.9356	0.0019	0.3715
1.050	0.8449	0.5093	0.6028	0.5569	1.3440	2.6387	0.9359	0.0029	0.3743
1.060	0.8424	0.5033	0.5975	0.5614	1.3428	2.6678	0.9363	0.0042	0.3769
1.070	0.8399	0.4974	0.5922	0.5659	1.3414	2.6970	0.9367	0.0056	0.3795
1.080	0.8374	0.4914	0.5869	0.5703	1.3397	2.7263	0.9371	0.0071	0.3820
1.090	0.8349	0.4855	0.5816	0.5747	1.3379	2.7557	0.9376	0.0089	0.3845
1.100	0.8323	0.4796	0.5763	0.5791	1.3359	2.7852	0.9381	0.0108	0.3868
1.110	0.8298	0.4738	0.5710	0.5835	1.3337	2.8148	0.9387	0.0128	0.3891
1.120	0.8272	0.4680	0.5658	0.5878	1.3313	2.8446	0.9394	0.0150	0.3913
1.130	0.8247	0.4622	0.5605	0.5922	1.3287	2.8744	0.9401	0.0173	0.3934
1.140	0.8221	0.4565	0.5553	0.5965	1.3259	2.9043	0.9408	0.0197	0.3954
1.150	0.8195	0.4508	0.5501	0.6008	1.3229	2.9344	0.9415	0.0223	0.3974
1.160	0.8170	0.4452	0.5449	0.6050	1.3198	2.9646	0.9424	0.0250	0.3993
1.170	0.8144	0.4396	0.5398	0.6093	1.3165	2.9949	0.9432	0.0277	0.4011
1.180	0.8118	0.4340	0.5347	0.6135	1.3131	3.0253	0.9441	0.0306	0.4028
1.190	0.8092	0.4285	0.5295	0.6177	1.3094	3.0558	0.9450	0.0335	0.4044
1.200	0.8066	0.4230	0.5245	0.6219	1.3057	3.0864	0.9459	0.0366	0.4060
1.210	0.8040	0.4176	0.5194	0.6261	1.3017	3.1172	0.9469	0.0397	0.4075
1.220	0.8014	0.4122	0.5143	0.6302	1.2976	3.1481	0.9479	0.0429	0.4089
1.230	0.7988	0.4068	0.5093	0.6344	1.2934	3.1791	0.9489	0.0462	0.4102
1.240	0.7962	0.4015	0.5043	0.6385	1.2890	3.2102	0.9500	0.0495	0.4115
1.250	0.7936	0.3963	0.4994	0.6426	1.2845	3.2414	0.9511	0.0529	0.4127
1.260	0.7909	0.3911	0.4944	0.6466	1.2798	3.2727	0.9522	0.0564	0.4138
1.270	0.7883	0.3859	0.4895	0.6507	1.2751	3.3042	0.9533	0.0599	0.4148
1.280	0.7857	0.3808	0.4846	0.6547	1.2701	3.3358	0.9545	0.0634	0.4158
1.290	0.7830	0.3757	0.4798	0.6587	1.2651	3.3675	0.9557	0.0670	0.4167
1.300	0.7804	0.3706	0.4749	0.6627	1.2599	3.3993	0.9569	0.0707	0.4175
1.310	0.7778	0.3657	0.4701	0.6667	1.2547	3.4313	0.9581	0.0744	0.4182
1.320	0.7751	0.3607	0.4654	0.6706	1.2493	3.4633	0.9594	0.0781	0.4189
1.330	0.7725	0.3558	0.4606	0.6746	1.2438	3.4955	0.9606	0.0819	0.4195
1.340	0.7698	0.3510	0.4559	0.6785	1.2382	3.5279	0.9619	0.0857	0.4200
1.350	0.7672	0.3462	0.4512	0.6824	1.2325	3.5603	0.9632	0.0895	0.4205
1.360	0.7646	0.3414	0.4465	0.6862	1.2266	3.5929	0.9645	0.0934	0.4209
1.370	0.7619	0.3367	0.4419	0.6901	1.2207	3.6256	0.9659	0.0973	0.4212
1.380	0.7593	0.3320	0.4373	0.6939	1.2147	3.6584	0.9672	0.1012	0.4215
1.390	0.7566	0.3274	0.4328	0.6977	1.2086	3.6914	0.9686	0.1051	0.4216
1.400	0.7540	0.3229	0.4282	0.7015	1.2025	3.7245	0.9700	0.1091	0.4218
1.410	0.7513	0.3183	0.4237	0.7053	1.1962	3.7577	0.9714	0.1130	0.4218
1.420	0.7487	0.3139	0.4192	0.7090	1.1899	3.7910	0.9728	0.1170	0.4218
1.430	0.7460	0.3094	0.4148	0.7127	1.1835	3.8245	0.9742	0.1210	0.4217
1.440	0.7434	0.3051	0.4104	0.7164	1.1770	3.8581	0.9756	0.1250	0.4216
1.450	0.7407	0.3007	0.4060	0.7201	1.1704	3.8918	0.9771	0.1290	0.4214
1.460	0.7381	0.2965	0.4017	0.7238	1.1638	3.9257	0.9785	0.1331	0.4212
1.470	0.7354	0.2922	0.3974	0.7275	1.1571	3.9597	0.9800	0.1371	0.4209
1.480	0.7328	0.2880	0.3931	0.7311	1.1504	3.9938	0.9815	0.1411	0.4205
1.490	0.7301	0.2839	0.3888	0.7347	1.1435	4.0281	0.9829	0.1452	0.4201
1.500	0.7275	0.2798	0.3846	0.7383	1.1367	4.0625	0.9844	0.1492	0.4196

Oblique Shock Tables ($\gamma = 1.4$)

M_1	θ	β	M_2	$\frac{P_{02}}{P_{01}}$	$\frac{\rho_2}{\rho_1}$	M_2	$\frac{T_2}{T_1}$
1.60	2.000	40.724	1.1046	1.0289	1.5323	0.99990	1.4534
	4.000	42.931	1.2189	1.1516	1.0584	1.4638	0.6547
	6.000	45.344	1.3446	1.2346	1.0891	1.3934	0.6467
	8.000	48.030	1.4843	1.3236	1.1215	1.3195	0.6421
	10.000	51.116	1.6430	1.4207	1.1565	1.2397	1.4578
	12.000	54.889	1.8320	1.5311	1.1965	1.483	0.99989
	14.000	60.537	2.0974	1.6777	1.2502	1.0232	1.0605
	14.652	65.828	2.3192	1.7929	1.2936	0.9188	1.0605
	14.000	70.895	2.5000	1.8824	1.3281	0.8320	1.0918
	12.000	75.900	2.6428	1.9504	1.3550	0.7611	1.1244
	10.000	79.102	2.7132	1.9831	1.3682	0.7250	1.1244
	8.000	81.691	2.7576	2.0035	1.3764	0.7018	1.1244
	6.000	83.967	2.7870	2.0168	1.3819	0.6862	1.1244
	4.000	86.061	2.8059	2.0254	1.3854	0.6761	1.1244
	2.000	88.054	2.8166	2.0302	1.3873	0.6703	1.1244
	1.65	39.267	1.1058	1.0744	1.0292	1.5823	0.99990
	4.000	41.377	1.2212	1.1531	1.0590	1.5140	0.99919
	6.000	43.665	1.3475	1.2365	1.0898	1.4444	0.99730
	8.000	46.181	1.4869	1.3252	1.1221	1.3720	0.99367
	10.000	49.007	1.6429	1.4206	1.1565	1.2952	0.98766
	12.000	52.312	1.8224	1.5257	1.1945	1.2104	0.97837
	14.000	56.541	2.0441	1.6490	1.2396	1.1090	0.96384
	15.855	65.547	2.4653	1.8655	1.3215	0.9184	0.92915
	14.000	73.864	2.7642	2.0065	1.3776	0.7782	0.90073
	12.000	77.411	2.8587	2.0491	1.3951	0.7317	0.89132
	10.000	80.102	2.9157	2.0744	1.4056	0.7029	0.88557
	8.000	82.389	2.9539	2.0911	1.4126	0.68333	0.88169
	6.000	84.446	2.9798	2.1024	1.4174	0.6697	0.87904
	4.000	86.364	2.9968	2.1097	1.4205	0.66207	0.87730
	2.000	88.200	3.0065	2.1139	1.4222	0.6556	0.87631
	1.70	37.927	1.1072	1.0754	1.0295	1.6320	0.99989
	4.000	39.957	1.2239	1.1550	1.0597	1.5638	0.99916
	6.000	42.145	1.3514	1.2390	1.0907	1.4946	0.99722
	8.000	44.528	1.4914	1.3280	1.1231	1.4232	0.99333
	10.000	47.167	1.6466	1.4228	1.1573	1.3482	0.98750
	12.000	50.168	1.8216	1.5252	1.1943	1.2674	0.97841
	14.000	53.771	2.0273	1.6399	1.2362	1.1757	0.96504
	16.000	58.794	2.2999	1.7831	1.2898	1.0569	0.94369
	17.012	65.319	2.6171	1.9383	1.3502	0.9185	0.91502
	16.000	71.426	2.8629	2.0510	1.3959	0.8077	0.89090
	14.000	75.670	2.9984	2.1104	1.4208	0.7439	0.87713
	12.000	78.555	3.0722	2.1421	1.4342	0.7080	0.86983
	10.000	80.906	3.1208	2.1626	1.4431	0.68338	0.86450
	8.000	82.965	3.1544	2.1767	1.4492	0.66667	0.86100

Oblique Shock Tables ($\gamma = 1.4$)

M_1	θ	β	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	M_2	$\frac{p_{02}}{p_{01}}$	M_1	θ	β	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	M_2	$\frac{p_{02}}{p_{01}}$
1.85	2.000	34.466	1.1121	1.0788	1.0309	1.7805	0.99988	1.95	2.000	32.528	1.1160	1.0319	0.99987
	4.000	36.323	1.2343	1.1619	1.0623	1.7114	0.99905	4.000	4.000	34.304	1.2424	1.0643	1.0045
	6.000	38.302	1.3672	1.2499	1.0945	1.6418	0.99689	6.000	6.000	36.191	1.3801	1.2575	1.0975
	8.000	40.424	1.5123	1.3409	1.1278	1.5711	0.99284	8.000	8.000	38.204	1.5302	1.3521	1.1318
	10.000	42.717	1.6709	1.4373	1.1625	1.4983	0.98638	10.000	10.000	40.360	1.6938	1.4509	1.1674
	12.000	45.223	1.8453	1.5388	1.1992	1.4224	0.97701	12.000	12.000	42.688	1.8726	1.5542	1.1518
	14.000	48.014	2.0395	1.6465	1.2387	1.3415	0.96417	14.000	14.000	45.230	2.0693	1.2446	1.4396
	16.000	51.232	2.2607	1.7631	1.2822	1.2524	0.94697	16.000	16.000	48.059	2.2879	1.2875	1.3553
	18.000	55.227	2.5275	1.8956	1.3333	1.1476	0.92345	18.000	18.000	51.320	2.5368	1.9001	1.3351
	20.000	62.099	2.9519	2.0902	1.4123	0.9818	0.88118	20.000	20.000	55.381	2.8378	2.0397	1.1520
	20.198	64.872	3.1062	2.1565	1.4404	0.9205	0.86601	22.000	22.000	62.860	3.3464	2.2553	0.89342
	20.000	67.544	3.2437	2.2136	1.4653	0.8648	0.85167	22.092	22.092	64.716	3.4603	2.3003	0.84087
	18.000	73.440	3.5019	2.3165	1.5117	0.7560	0.82446	22.000	22.000	66.523	3.5655	2.3410	0.9229
	16.000	76.511	3.6090	2.3576	1.5308	0.7085	0.81314	20.000	20.000	72.926	3.8872	2.4601	1.5231
	14.000	78.861	3.6772	2.3833	1.5429	0.6773	0.80593	18.000	18.000	75.964	4.0086	2.5030	1.5801
	12.000	80.844	3.7252	2.4011	1.5514	0.6548	0.80088	16.000	16.000	78.253	4.0857	2.5297	1.6015
	10.000	82.606	3.7601	2.4140	1.5576	0.6381	0.79719	14.000	14.000	80.165	4.1401	2.5484	1.6151
	8.000	84.222	3.7858	2.4234	1.5622	0.6257	0.78449	12.000	12.000	81.849	4.1804	2.5620	1.6246
	6.000	85.740	3.8042	2.4301	1.5655	0.6166	0.79255	10.000	10.000	83.381	4.2106	2.5722	1.6317
	4.000	87.193	3.8167	2.4346	1.5677	0.6105	0.79124	8.000	8.000	84.808	4.2333	2.5798	1.6370
	2.000	88.606	3.8239	2.4373	1.5689	0.6069	0.79048	6.000	6.000	86.163	4.2497	2.5853	1.6409
1.90	2.000	33.466	1.1140	1.0801	1.0314	1.8298	0.99987	2.000	2.000	31.647	1.1180	1.0829	1.0324
	4.000	35.279	1.2382	1.1646	1.0633	1.7600	0.99901	4.000	4.000	33.390	1.2468	1.1702	1.0654
	6.000	37.209	1.3735	1.2533	1.0959	1.6901	0.99675	2.00	2.00	35.241	1.3871	1.2620	1.0991
	8.000	39.272	1.5209	1.3463	1.1297	1.6191	0.99254	6.000	6.000	37.210	1.5400	1.3581	1.1339
	10.000	41.490	1.6818	1.4438	1.1649	1.5464	0.98586	8.000	8.000	39.314	1.7066	1.4584	1.1702
	12.000	43.898	1.8582	1.5460	1.2019	1.4709	0.97624	12.000	12.000	41.575	1.8884	1.5631	1.2081
	14.000	46.550	2.0530	1.6538	1.2414	1.3913	0.96319	10.000	10.000	44.029	2.0876	1.6724	1.2483
	16.000	49.544	2.2718	1.7688	1.2844	1.3052	0.94605	14.000	14.000	46.731	2.3076	1.7870	1.2913
	18.000	53.095	2.5263	1.8951	1.3331	1.2077	0.92356	12.000	12.000	49.785	2.5546	1.9086	1.3384
	20.000	57.900	2.8557	2.0477	1.3946	1.0835	0.89162	16.000	16.000	53.423	2.8429	2.0420	1.3922
	21.167	64.783	3.2805	2.2286	1.4720	0.9216	0.84781	18.000	18.000	58.457	3.2228	2.2051	1.4616
	20.000	71.057	3.6012	2.3546	1.5294	0.7935	0.81397	20.000	20.000	58.974	3.6458	2.3715	1.5373
	18.000	74.861	3.7578	2.4131	1.5572	0.7274	0.79744	16.000	16.000	64.669	4.0876	2.6274	1.6262
	16.000	77.463	3.8466	2.4455	1.5729	0.6884	0.78810	22.000	22.000	70.332	4.3777	2.4991	1.7217
	14.000	79.565	3.9068	2.4671	1.5836	0.6611	0.78178	20.000	20.000	74.270	4.1570	2.5541	1.6276
	12.000	81.383	3.9504	2.4826	1.5913	0.6409	0.77721	18.000	18.000	76.862	4.2589	2.5883	1.6278
	10.000	83.020	3.9828	2.4940	1.5970	0.6257	0.77383	16.000	16.000	78.921	4.3277	2.6110	1.6574
	8.000	84.534	4.0068	2.5024	1.6012	0.6142	0.77133	14.000	14.000	80.684	4.3777	2.6274	1.6662
	6.000	85.965	4.0241	2.5084	1.6042	0.6058	0.76953	12.000	12.000	82.257	4.4153	2.6396	1.6727
	4.000	87.338	4.0359	2.5125	1.6063	0.6001	0.76830	10.000	10.000	83.700	4.4438	2.6487	1.6777
	2.000	88.677	4.0428	2.5149	1.6075	0.5967	0.76759	8.000	8.000	85.052	4.4653	2.6556	1.6854
								6.000	6.000	86.339	4.4810	2.6606	1.6842
								4.000	4.000	87.582	4.4917	2.6640	1.6861
								2.000	2.000	88.798	4.4979	2.6660	1.6871

Oblique Shock Tables ($\gamma = 1.4$)

M_1	θ	β	$\frac{p_2}{p_1}$	$\frac{M_2}{M_1}$	$\frac{p_{02}}{p_{01}}$	$\frac{T_2}{T_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{p_2}{p_1}$	$\frac{M_2}{M_1}$	$\frac{p_{02}}{p_{01}}$
2.05	2.000	30.816	1.0843	1.0330	0.99985	1.9771	0.99985	2.10	4.000	87.778
	4.000	32.532	1.2512	1.1732	1.0665	1.1008	1.8330	2.00	2.000	88.894
	6.000	34.350	1.3943	1.2696	1.1362	1.7605	0.99148			
	8.000	36.281	1.5502	1.4664	1.1730	1.6868	0.98396	2.15	2.000	29.293
	10.000	38.341	1.7201	1.4664	1.5726	1.2116	1.6111			
	12.000	40.547	1.9053	1.5726	1.2116	1.53226	0.95914			
	14.000	42.928	2.1076	1.6831	1.2522	1.4500	0.94112			
	16.000	45.528	2.3300	1.7983	1.2956	1.3614	0.91878			
	18.000	48.428	2.5774	1.9195	1.3427	1.2630	0.89120			
	20.000	51.785	2.8600	2.0497	1.3953	1.1444	0.85565			
	22.000	56.032	3.2057	2.1980	1.4585	1.0444	0.81913			
	23.814	64.638	3.8367	2.4419	1.5712	0.9257	0.78913			
	22.000	72.193	4.2777	2.5946	1.6487	0.7626	0.74336			
	20.000	75.324	4.4215	2.6416	1.6738	0.7056	0.72876			
	18.000	77.614	4.5107	2.6700	1.6894	0.6688	0.71981			
	16.000	79.498	4.5734	2.6898	1.7003	0.6422	0.71356			
	14.000	81.138	4.6199	2.7043	1.7084	0.6219	0.70894			
	12.000	82.617	4.6553	2.7152	1.7145	0.6062	0.70545			
	10.000	83.983	4.6824	2.7236	1.7192	0.5939	0.70278			
	8.000	85.289	4.7029	2.7299	1.7228	0.58486	0.70077			
	6.000	86.497	4.7179	2.7344	1.7254	0.5776	0.69930			
	4.000	87.665	4.7283	2.7376	1.7272	0.5728	0.69829			
	2.000	88.849	4.7343	2.7394	1.7282	0.5700	0.69770			
2.10	2.000	30.033	1.1222	1.0858	1.0335	1.0984	0.99984	2.0260	8.000	86.639
	4.000	31.723	1.2558	1.1763	1.0676	1.9530	0.99880	6.000	86.767	5.2091
	6.000	33.513	1.4017	1.2714	1.1025	1.8801	0.99609	4.000	87.862	5.2187
	8.000	35.412	1.5608	1.3709	1.1386	1.8069	0.99108	2.000	88.936	5.2244
	10.000	37.433	1.7342	1.4746	1.1760	1.7325	0.98324			
	12.000	39.592	1.9230	1.5825	1.2152	1.6564	0.97216			
	14.000	41.912	2.1290	1.6944	1.2565	1.5777	0.95750			
	16.000	44.430	2.3547	1.8107	1.3004	1.4954	0.93899			
	18.000	47.210	2.6041	1.9322	1.3478	1.4078	0.91626			
	20.000	50.385	2.8848	2.0607	1.3999	1.3122	0.88870			
	22.000	54.169	3.2152	2.2019	1.4602	1.2019	0.85466			
	24.000	59.767	3.6739	2.3820	1.5424	1.0493	0.80628			
	24.614	64.621	4.0332	2.5116	1.6058	0.92273	0.76858			
	24.000	69.104	4.3238	2.6098	1.6568	0.8245	0.73867			
	22.000	73.521	4.5644	2.6870	1.6987	0.7345	0.71445			
	20.000	76.189	4.6852	2.7244	1.7197	0.6870	0.70251			
	18.000	78.257	4.7652	2.7488	1.7336	0.6543	0.69468			
	16.000	80.001	4.8232	2.7662	1.7436	0.6299	0.68906			
	14.000	81.539	4.8669	2.7792	1.7512	0.6111	0.68484			
	12.000	82.938	4.9006	2.7892	1.7570	0.5964	0.68162			
	10.000	84.237	4.9264	2.7968	1.7615	0.5849	0.67914			
	8.000	85.463	4.9461	2.8025	1.7649	0.5760	0.67726			
	6.000	86.638	4.9606	2.8068	1.7674	0.5694	0.67588			
2.1237	2.000	30.033	1.1222	1.0858	1.0335	1.0984	0.99984	2.0260	8.000	86.639
	4.000	31.723	1.2558	1.1763	1.0676	1.9530	0.99880	6.000	86.767	5.2091
	6.000	33.513	1.4017	1.2714	1.1025	1.8801	0.99609	4.000	87.862	5.2187
	8.000	35.412	1.5608	1.3709	1.1386	1.8069	0.99108	2.000	88.936	5.2244
	10.000	37.433	1.7342	1.4746	1.1760	1.7325	0.98324			
	12.000	39.592	1.9230	1.5825	1.2152	1.6564	0.97216			
	14.000	41.912	2.1290	1.6944	1.2565	1.5777	0.95750			
	16.000	44.430	2.3547	1.8107	1.3004	1.4954	0.93899			
	18.000	47.210	2.6041	1.9322	1.3478	1.4078	0.91626			
	20.000	50.385	2.8848	2.0607	1.3999	1.3122	0.88870			
	22.000	54.169	3.2152	2.2019	1.4602	1.2019	0.85466			
	24.000	59.767	3.6739	2.3820	1.5424	1.0493	0.80628			
	24.614	64.621	4.0332	2.5116	1.6058	0.92273	0.76858			
	24.000	69.104	4.3238	2.6098	1.6568	0.8245	0.73867			
	22.000	73.521	4.5644	2.6870	1.6987	0.7345	0.71445			
	20.000	76.189	4.6852	2.7244	1.7197	0.6870	0.70251			
	18.000	78.257	4.7652	2.7488	1.7336	0.6543	0.69468			
	16.000	80.001	4.8232	2.7662	1.7436	0.6299	0.68906			
	14.000	81.539	4.8669	2.7792	1.7512	0.6111	0.68484			
	12.000	82.938	4.9006	2.7892	1.7570	0.5964	0.68162			
	10.000	84.237	4.9264	2.7968	1.7615	0.5849	0.67914			
	8.000	85.463	4.9461	2.8025	1.7649	0.5760	0.67726			
	6.000	86.638	4.9606	2.8068	1.7674	0.5694	0.67588			

Oblique Shock Tables ($\gamma = 1.4$)

M_1	θ	β	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	M_2	$\frac{p_{02}}{p_{01}}$	M_1	θ	β	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	M_2	$\frac{p_{02}}{p_{01}}$	
2.20	20.000	77.549	5.2175	2.8799	1.8117	0.6568	0.65185	2.30	16.000	40.816	2.4701	1.8678	1.6676	0.92872
	18.000	79.308	5.2856	2.8987	1.8234	0.6296	0.64562		18.000	43.299	2.7380	1.9936	1.5804	0.90351
	16.000	80.839	5.3369	2.9127	1.8323	0.6086	0.64096		20.000	46.007	3.0276	2.1280	1.4261	0.87413
	14.000	82.216	5.3764	2.9235	1.8391	0.5921	0.63739		22.000	49.026	3.3514	2.2573	1.4847	0.84035
	12.000	83.483	5.4073	2.9318	1.8444	0.5789	0.63462		24.000	52.536	3.7216	2.3998	1.5508	0.80125
	10.000	84.670	5.4313	2.9382	1.8485	0.5686	0.63247		26.000	57.077	4.1819	2.5625	1.6319	0.75319
	8.000	85.798	5.4497	2.9431	1.8517	0.5605	0.63083		27.454	64.653	4.8759	2.7813	1.7524	0.9338
	6.000	86.883	5.4633	2.9468	1.8540	0.5545	0.62982		26.000	71.264	5.3682	2.9212	1.8377	0.7743
	4.000	87.938	5.4727	2.9493	1.8556	0.5503	0.62879		24.000	74.512	5.5649	2.9736	1.8714	0.7060
	2.000	88.973	5.4782	2.9507	1.8565	0.5479	0.62830		22.000	76.770	5.6817	3.0059	1.8915	0.6635
									20.000	78.582	5.7631	3.0246	1.9054	0.6328
									18.000	80.133	5.8238	3.0399	1.9158	0.6092
									16.000	81.509	5.8705	3.0515	1.9238	0.5906
									14.000	82.764	5.9071	3.0606	1.9301	0.5757
									12.000	83.928	5.9360	3.0677	1.9350	0.5638
									10.000	85.026	5.9586	3.0732	1.9389	0.5543
									8.000	86.074	5.9761	3.0775	1.9419	0.5469
									6.000	87.085	5.9890	3.0807	1.9441	0.5413
									4.000	88.070	5.9980	3.0828	1.9456	0.5374
									2.000	89.039	6.0033	3.0841	1.9465	0.5352
														0.58344
2.25	2.000	27.926	1.1288	1.0903	1.0353	2.1725	0.99982				2.35	2.000	2.6692	1.0355
	4.000	29.555	1.2703	1.1859	1.0712	2.0962	0.99861						2.000	2.8289
	6.000	31.277	1.4254	1.2864	1.1080	2.0203	0.99548						1.75736	1.2904
	8.000	33.102	1.5949	1.3916	1.1461	1.9443	0.98973						1.1118	0.99502
	10.000	35.034	1.7798	1.5011	1.1856	1.8674	0.98079						1.1513	0.98872
	12.000	37.088	1.9812	1.6147	1.2270	1.7891	0.96827						1.1924	0.97895
	14.000	39.277	2.2004	1.7319	1.2705	1.7088	0.95189						1.2354	0.96534
	16.000	41.623	2.4392	1.8527	1.3166	1.6257	0.93152						1.2807	0.94765
	18.000	44.161	2.7000	1.9770	1.3657	1.5398	0.90703						1.7834	0.92580
	20.000	46.948	2.9871	2.1055	1.4187	1.4466	0.87829						1.7089	0.90833
	22.000	50.091	3.3085	2.2400	1.4770	1.3464	0.84448						1.3285	0.89980
	24.000	53.837	3.6830	2.3854	1.5440	1.2318	0.80532						1.0736	0.89846
	26.000	59.122	4.1839	2.5632	1.6323	1.0792	0.75298						1.1911	0.99502
	26.795	64.633	4.6556	2.7153	1.7145	0.9821	0.70542						1.2119	0.99502
	26.000	69.627	5.0238	2.8250	1.7783	0.8115	0.66991						1.2970	0.98872
	24.000	73.634	5.2707	2.8946	1.8209	0.7254	0.64688						1.4420	0.98872
	22.000	76.145	5.4009	2.9301	1.8433	0.6775	0.63519						1.3462	0.98872
	20.000	78.098	5.4884	2.9534	1.8583	0.6841	0.62739						1.4062	0.98872
	18.000	79.744	5.5523	2.9703	1.8693	0.6189	0.62175						1.4420	0.98872
	16.000	81.192	5.6011	2.9830	1.8776	0.5993	0.61749						1.4814	0.98872
	14.000	82.504	5.6391	2.9929	1.8842	0.5836	0.61418						1.5199	0.98872
	12.000	83.716	5.6688	3.0006	1.8893	0.5711	0.61161						1.6060	0.98872
	10.000	84.856	5.6921	3.0065	1.8932	0.56112	0.60960						1.6212	0.98872
	8.000	85.942	5.7100	3.0111	1.8963	0.5535	0.60806						1.6212	0.98872
	6.000	86.988	5.7233	3.0145	1.8986	0.5477	0.60692						1.6212	0.98872
	4.000	88.007	5.7324	3.0168	1.9002	0.5437	0.60614						1.6212	0.98872
	2.000	89.008	5.7378	3.0182	1.9011	0.5413	0.60588						1.6212	0.98872
													1.6212	0.98872
2.30	2.000	27.294	1.1311	1.0919	1.0359	2.2212	0.99981						1.0365	1.2698
	4.000	28.906	1.2753	1.1892	1.0724	2.1437	0.99854						1.0736	1.2698
	6.000	30.611	1.4336	1.2916	1.1099	2.0867	0.99526						1.1911	1.2698
	8.000	32.415	1.6068	1.3988	1.1487	1.9896	0.98923						1.2911	1.2698
	10.000	34.326	1.7959	1.5104	1.1890	1.97989	0.97989						1.4111	1.2698
	12.000	36.354	2.0019	1.6260	1.2311	1.8825	0.96684						1.5311	1.2698
	14.000	38.510	2.2261	1.7452	1.2755	1.94982	0.94982						1.6511	1.2698

Oblique Shock Tables ($\gamma = 1.4$)

M_1	θ	β	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	M_2	$\frac{p_{02}}{p_{01}}$	M_1	θ	β	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	M_2	$\frac{p_{02}}{p_{01}}$	
2.35	6.000	87.174	6.2606	3.1453	1.9904	0.5353	0.56272	2.45	26.000	53.045	4.3053	2.6037	1.2861	0.74055
	4.000	88.129	6.2694	3.1474	1.9919	0.5315	0.56203		28.000	57.780	4.8455	2.7729	1.1385	0.66691
	2.000	89.068	6.2745	3.1486	1.9928	0.5293	0.56162		29.253	64.744	5.5614	2.9727	1.8708	0.9386
									20.000	70.828	6.0810	3.1029	1.9598	0.7837
2.40	2.000	26.120	1.1358	1.0951	1.0371	2.3184	0.99979		24.000	74.185	6.3161	1.5882	1.9999	0.7082
	4.000	27.702	1.2856	1.1960	1.0749	2.2383	0.99839		22.000	78.236	6.4516	3.1891	2.0230	0.55836
	6.000	29.377	1.4505	1.3023	1.1138	2.1589	0.99478		18.000	81.089	6.6682	3.2372	2.0599	0.57709
	8.000	31.149	1.6314	1.4137	1.1540	2.0794	0.98818		16.000	82.299	6.7105	3.2464	2.0671	0.55836
	10.000	33.023	1.8292	1.5295	1.1959	1.9994	0.97797		14.000	83.416	6.7442	3.2536	2.0728	0.54787
	12.000	35.007	2.0450	1.6495	1.2398	1.9181	0.96377		12.000	84.462	6.7710	3.2594	2.0398	0.6294
	14.000	37.112	2.2798	1.7729	1.2860	1.8350	0.94538		10.000	85.455	6.7923	3.2640	2.0508	0.6042
	16.000	39.351	2.5351	1.8993	1.3348	1.7497	0.92274		8.000	86.408	6.8088	3.2675	2.0810	0.53555
	18.000	41.748	2.8128	2.0285	1.3866	1.6613	0.89592		6.000	87.331	6.8211	3.2701	2.0838	0.5842
	20.000	44.336	3.1155	2.1604	1.4421	1.5689	0.86505		4.000	88.232	6.8296	3.2719	2.0859	0.52129
	22.000	47.174	3.4480	2.2955	1.5021	1.4709	0.83015		2.000	89.119	6.8346	3.2730	2.0882	0.5186
	24.000	50.371	3.8196	2.4357	1.5682	1.3644	0.79093							
	26.000	54.184	4.2521	2.5861	1.6442	1.2426	0.74598							
	28.000	59.656	4.8382	2.7707	1.7462	1.0779	0.68761							
	28.681	64.710	5.3269	2.9100	1.8305	0.9370	0.64187	2.50	2.000	25.050	1.0984	1.0384	0.99977	
	28.000	69.291	5.7130	3.0119	1.8968	0.8201	0.60781		4.000	26.609	1.2961	1.2029	1.0775	0.99822
	28.000	73.400	6.0048	3.0845	1.9468	0.7260	0.58331		6.000	28.259	1.4679	1.3133	1.1177	0.99427
	24.000	75.889	6.1539	3.1203	1.9722	0.6751	0.57121		8.000	30.005	1.6568	1.4289	1.1595	0.98703
	22.000	77.803	6.2534	3.1436	1.9892	0.6397	0.56329		10.000	31.851	1.8639	1.5493	1.2031	0.97589
	20.000	79.402	6.3260	3.1605	2.0016	0.6129	0.55758		12.000	33.802	2.0900	1.6737	1.2488	0.0022
	18.000	80.800	6.3816	3.1732	2.0111	0.5919	0.55326		14.000	34.866	2.3364	1.8015	1.2969	0.94057
	16.000	82.059	6.4251	3.1831	2.0185	0.5751	0.54990		16.000	38.057	2.6042	1.9322	1.3478	1.8295
	14.000	83.217	6.4596	3.1909	2.0244	0.5615	0.54726		18.000	40.389	2.8949	2.0652	1.4018	1.7394
	12.000	84.299	6.4870	3.1971	2.0290	0.5505	0.54517		20.000	42.890	3.2109	2.2002	1.4594	1.6458
	10.000	85.324	6.5087	3.2019	2.0327	0.5416	0.54352		22.000	45.602	3.5558	2.3373	1.5213	1.5475
	8.000	86.306	6.5254	3.2057	2.0356	0.5348	0.54225		24.000	48.600	3.9361	2.4775	1.5887	1.4426
	6.000	87.255	6.5379	3.2085	2.0377	0.5296	0.54131		26.000	52.036	4.3657	2.6235	1.6641	1.3268
	4.000	88.182	6.5466	3.2104	2.0392	0.5260	0.54065		28.000	56.335	4.8844	2.7844	1.7542	1.1888
	2.000	89.094	6.5517	3.2115	2.0400	0.5238	0.54027		29.797	64.782	5.8014	3.0342	1.9120	0.9402
									28.000	71.949	6.4249	3.1831	2.0185	0.7573
	2.45	2.000	25.572	1.1381	1.0377	2.3670	0.99978		24.000	74.856	6.6273	3.2282	2.0529	0.53460
	4.000	27.143	1.2908	1.1994	1.0762	2.2855	0.99831		22.000	76.939	6.7526	3.2555	2.0742	0.6509
	6.000	28.805	1.4591	1.3078	1.1557	2.2048	0.99453		20.000	80.070	6.9082	3.2885	2.1007	0.5962
	8.000	30.563	1.6440	1.4212	1.1567	2.1241	0.98761		18.000	81.353	6.9602	3.2994	2.1095	0.5770
	10.000	32.422	1.8463	1.5393	1.1994	2.0428	0.97695		16.000	82.518	7.0014	3.3080	2.1165	0.5616
	12.000	34.388	2.0672	1.6615	1.2442	1.9603	0.96215		14.000	83.598	7.0343	3.3148	2.1221	0.5489
	14.000	36.472	2.3078	1.7871	1.2914	1.8762	0.94302		12.000	84.612	7.0607	3.3202	2.1266	0.5387
	16.000	38.685	2.5692	1.9156	1.3412	1.7898	0.91955		10.000	85.576	7.0816	3.3245	2.1301	0.5304
	18.000	41.047	2.8632	2.0466	1.3941	1.7006	0.89187		8.000	86.502	7.0979	3.3278	2.1329	0.5240
	20.000	43.588	3.1623	2.1800	1.4506	1.6077	0.86018		6.000	87.400	7.1100	3.3303	2.1350	0.5191
	22.000	46.358	3.5007	2.3160	1.5115	1.5097	0.82459		4.000	88.277	7.1184	3.3320	2.1364	0.5157
	24.000	49.445	3.8759	2.4560	1.5781	1.4042	0.78502		2.000	89.142	7.1234	3.3330	2.1372	0.5137

Oblique Shock Tables ($\gamma = 1.4$)

M_1	θ	β	$\frac{p_2}{p_1}$	$\frac{M_2}{p_01}$	M_1	M_2	$\frac{p_02}{p_01}$	$\frac{\rho_2}{\rho_1}$	$\frac{p_2}{p_1}$	M_2	$\frac{T_2}{T_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{p_02}{p_01}$
2.55	2.000	1.1429	1.1001	1.0390	1.0788	2.4639	0.99976	2.60	30.814	64.866	6.2972	3.1538	0.55984
	4.000	24.550	1.3015	1.2065	1.3189	1.1198	2.3796	0.99814	30.00	69.778	6.7777	3.2609	0.8111
	6.000	27.739	1.4768	1.0788	1.4367	1.1623	2.2961	0.99939	28.00	73.590	7.0906	3.3263	0.52354
	8.000	29.474	1.6699	1.1699	1.5593	1.2067	2.2128	0.98642	26.00	75.955	7.2555	3.3596	0.50138
	10.000	31.307	1.8817	1.8817	2.1133	1.6861	2.1288	0.97479	24.00	77.778	7.3665	3.3815	0.49015
	12.000	33.244	2.3656	2.3656	2.3591	1.8162	2.0438	0.95871	22.00	79.299	7.4481	3.3974	0.48276
	14.000	35.293	37.463	36.130	39.770	2.6399	1.9490	1.3545	20.00	80.626	7.5108	3.4095	0.47742
	16.000	39.770	51.130	4.4319	2.9378	2.0840	1.4097	1.7776	18.00	81.815	7.5602	3.4189	0.47336
	20.000	42.236	55.131	4.9401	3.2611	2.2207	1.6885	1.88333	16.00	82.906	7.5997	3.4264	0.47020
	22.000	44.899	61.449	5.6866	3.0051	2.3591	1.5315	1.6832	14.00	83.922	7.6316	3.4324	0.46768
	24.000	47.822	64.823	6.0466	3.9995	2.4998	1.5999	1.5845	12.00	84.879	7.6572	3.4372	0.46566
	26.000	51.130	72.844	6.7595	3.2569	2.0754	1.4797	1.77209	10.00	85.792	7.6775	3.4411	0.46277
	28.000	55.131	75.440	6.9402	3.2952	2.1061	1.6756	1.3655	8.00	86.671	7.6934	3.4440	0.46178
	30.000	61.449	77.380	7.0575	3.1935	2.1260	1.6405	0.50368	6.00	87.524	7.7053	3.4462	0.46104
	30.317	64.823	67.966	6.3519	3.1664	2.0060	0.8568	0.55557	4.00	88.359	7.7135	3.4478	0.46053
	30.000	72.844	78.978	7.1423	3.3368	2.1404	0.6115	0.49783	2.00	89.183	7.7184	3.4487	0.46022
	24.000	75.440	80.360	7.2068	3.3499	2.1514	0.5887	0.49343	2.65	2.00	23.613	1.1479	1.0403
	22.000	78.978	81.594	7.2575	3.3600	2.1600	0.5703	0.49002	6.00	26.766	1.4950	1.3124	0.99796
	18.000	82.720	82.720	7.2978	3.3680	2.1668	0.5554	0.48732	8.00	28.482	1.6966	1.3802	0.99341
	14.000	83.766	84.750	7.3301	3.3744	2.1723	0.5432	0.48517	10.00	30.295	1.9182	1.5798	2.3007
	12.000	85.688	86.590	7.3767	3.3795	2.1767	0.5333	0.48345	12.00	32.210	2.1610	1.7113	0.98514
	10.000	87.464	88.320	7.4047	3.3835	2.1802	0.5253	0.48209	14.00	34.232	2.4260	1.8462	2.0370
	8.000	88.320	89.163	7.4131	3.3906	2.1849	0.5190	0.48104	16.00	36.368	2.7141	1.9835	1.9459
	6.000	89.163	89.163	7.4180	3.3916	2.1872	0.5142	0.48025	18.00	38.632	3.0267	2.1226	0.90566
	4.000	90.000	90.000	88.320	3.3906	2.1864	0.5109	0.47971	20.00	41.043	3.3657	2.1424	0.87423
	2.000	90.000	90.000	89.163	3.3916	2.1872	0.5090	0.47939	22.00	43.627	3.7355	2.4873	0.83384
	2.60	2.000	24.071	1.1017	1.0396	0.99975	30.00	57.877	24.00	46.433	2.4042	1.6559	0.80000
	4.000	25.611	1.3070	1.0801	2.4265	0.99805	31.288	64.910	1.3437	2.5465	1.6237	1.5507	0.75806
	6.000	27.241	1.4858	1.3245	2.3416	0.99371	30.00	70.983	4.5776	2.6911	1.7010	1.4380	0.71313
	8.000	28.966	1.6831	1.4445	1.1651	2.2568	0.98579	28.00	74.230	7.4211	2.1877	0.7039	0.47918
	10.000	30.789	1.8998	1.5695	1.2105	2.1715	0.97365	26.00	76.415	7.742	3.4216	0.68585	0.46830
	12.000	32.714	2.1369	1.6986	1.2580	2.0852	0.95690	24.00	78.138	7.9801	3.4415	2.2316	0.46262
	14.000	34.749	2.3955	1.8311	1.3082	1.9973	0.93541	22.00	79.592	7.7589	3.4562	2.2449	0.45771
	16.000	36.901	2.6767	1.9662	1.3613	1.9075	0.90930	20.00	80.870	7.8200	3.4674	2.2553	0.45752
	18.000	39.185	2.9817	2.1032	1.4177	1.8152	0.87884	18.00	82.020	7.8684	3.4763	2.2634	0.45592
	20.000	41.621	3.3126	2.2417	1.4778	1.7199	0.84443	16.00	83.079	7.9073	3.4833	2.2700	0.45442
	22.000	44.242	3.6723	2.3814	1.5421	1.6205	0.80645	14.00	84.066	7.9387	3.4890	2.2753	0.44677
	24.000	47.102	4.0658	2.5229	1.6116	1.5157	0.76520	12.00	84.998	7.9640	3.4935	2.2796	0.44526
	26.000	50.305	4.5028	2.6675	1.6880	1.4025	0.72060	10.00	85.888	7.9841	3.4972	2.2830	0.5158
	28.000	54.088	5.0067	2.8201	1.7754	1.2744	0.67151	8.00	86.746	7.9999	3.5000	2.2857	0.5098
	30.000	59.352	5.6706	3.0010	1.0622	1.01645	1.01145	1.0000	6.00	87.579	8.0116	3.5021	2.2877

Oblique Shock Tables ($\gamma = 1.4$)

M_1	θ	β	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	M_2	$\frac{p_{02}}{p_{01}}$	M_1	θ	β	$\frac{p_2}{p_1}$	M_2	$\frac{p_{02}}{p_{01}}$
2.65	4.000	88.396	8.0198	3.5035	2.2891	0.5021	0.44194	2.75	24.000	45.225	4.2794	2.5951
	2.000	89.200	8.0247	3.5044	2.2899	0.5003	0.44165		26.000	48.206	4.7375	2.7404
									28.000	51.579	5.2490	2.8886
									30.000	55.674	5.8507	3.0466
									32.000	62.549	6.7812	3.2616
									32.173	65.002	7.0807	3.3243
									32.000	67.3223	7.3448	3.3773
									30.000	72.678	7.8741	3.4773
									28.000	75.285	8.0870	3.5154
									26.000	77.202	8.2233	3.5393
									24.000	78.766	8.3214	3.5561
									22.000	80.110	8.3960	3.5688
									20.000	81.303	8.4545	3.5786
									18.000	82.386	8.5014	3.5864
									16.000	83.387	8.5392	3.5927
									14.000	84.324	8.5699	3.5978
									12.000	85.212	8.5948	3.6019
									10.000	86.062	8.6146	3.6051
									8.000	86.882	8.6301	3.6077
									6.000	87.680	8.6418	3.6301
									4.000	88.462	8.6499	3.6109
									2.000	89.234	8.6547	3.6117
2.70	2.000	23.1173	1.1503	1.1051	1.0409	2.6090	0.99972	2.75	24.000	45.225	4.2794	2.5951
									26.000	48.206	4.7375	2.7404
									30.000	51.579	5.2490	2.8886
									32.000	55.674	6.7812	3.2616
									32.173	65.002	7.0807	3.3243
									32.000	67.3223	7.3448	3.3773
									30.000	72.678	7.8741	3.4773
									28.000	75.285	8.0870	3.5154
									26.000	77.202	8.2233	3.5393
									24.000	78.766	8.3214	3.5561
									22.000	80.110	8.3960	3.5688
									20.000	81.303	8.4545	3.5786
									18.000	82.386	8.5014	3.5864
									16.000	83.387	8.5392	3.5927
									14.000	84.324	8.5699	3.5978
									12.000	85.212	8.5948	3.6019
									10.000	86.062	8.6146	3.6051
									8.000	86.882	8.6301	3.6077
									6.000	87.680	8.6418	3.6301
									4.000	88.462	8.6499	3.6109
									2.000	89.234	8.6547	3.6117
2.75	2.000	22.750	1.1528	1.1068	1.0415	2.6573	0.99971	2.80	2.000	22.344	2.7056	0.99969
									4.000	23.854	1.3292	1.2246
									6.000	25.455	1.5230	1.3476
									8.000	27.150	1.7379	1.4768
									10.000	28.940	1.9751	1.6113
									12.000	30.830	2.2357	1.7502
									14.000	32.822	2.5205	1.8923
									16.000	34.923	2.8309	2.0367
									18.000	37.141	3.1677	2.1822
									20.000	39.490	3.5324	2.3283
									22.000	41.990	3.9271	2.4743
									24.000	44.676	4.3550	2.6200
									26.000	47.604	4.8219	2.7658
									28.000	50.887	5.3398	2.9135
									30.000	54.786	5.9387	3.0683
									32.000	60.433	6.7529	3.2555
									32.587	65.050	7.3524	3.3788
									32.000	69.211	7.8278	3.4689
									30.000	73.328	8.2272	3.5399
									28.000	75.728	8.4241	3.5735
									26.000	77.543	8.5544	3.5952
									24.000	79.042	8.6461	3.6409
									22.000	80.339	8.7224	3.6227
									20.000	81.496	8.7800	3.6319

Oblique Shock Tables ($\gamma = 1.4$)

M_1	θ	β	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	M_2	$\frac{p_{v2}}{p_{01}}$	M_1	θ	β	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	M_2	$\frac{p_{v2}}{p_{01}}$
2.80	18.000	82.550	8.8262	3.6393	2.4252	0.5425	0.39731	2.90	6.000	24.666	1.3594	1.1344	2.6117
	16.000	83.525	8.8637	3.6453	2.4316	0.5297	0.39538		8.000	26.350	1.4933	1.1828	2.5175
	14.000	84.440	8.8942	3.6501	2.4367	0.5191	0.39382		10.000	28.129	2.0143	1.6528	2.4229
	12.000	85.308	8.9188	3.6540	2.4409	0.5103	0.39256		12.000	30.007	2.2873	1.7767	2.3273
	10.000	86.140	8.9385	3.6571	2.4442	0.5033	0.39156		14.000	31.985	2.5863	1.9238	0.94475
	8.000	86.943	8.9540	3.6595	2.4468	0.4977	0.39078		16.000	34.069	2.9123	1.3444	2.2304
	6.000	87.725	8.9656	3.6613	2.4487	0.4935	0.39019		18.000	36.264	3.2863	2.0729	1.4050
	4.000	88.492	8.9737	3.6626	2.4501	0.4905	0.38978		20.000	38.584	3.6496	2.3729	1.5380
	2.000	89.248	8.9784	3.6633	2.4509	0.4887	0.38954		22.000	41.044	4.0638	2.5222	1.6112
2.85	2.000		21.954	1.1579	1.1103	1.0429	2.7537	0.99868		24.000	43.672	4.5119	2.6704
	4.000		23.457	1.3349	1.2283	1.0868	2.6598	0.99755		26.000	46.515	4.9884	2.8177
	6.000		25.052	1.5325	1.3535	1.1323	2.5670	0.99213		28.000	53.274	5.5328	2.9852
	8.000		26.742	1.7520	1.4850	1.1798	2.4744	0.98230		30.000	57.931	6.1364	3.1161
	10.000		28.526	1.9946	1.6220	1.2297	2.3815	0.96735		32.000	65.145	7.9116	3.2824
	12.000		30.410	2.2613	1.7634	1.2824	2.2876	0.94692		34.000	71.287	8.6550	3.4841
	14.000		32.394	2.5532	1.9080	1.3382	2.1923	0.92105		36.000	74.392	9.3947	3.6955
	16.000		34.486	2.8712	2.0547	1.3974	2.0953	0.89006		38.000	76.490	9.1095	3.8836
	18.000		36.692	3.2165	2.2025	1.4604	1.9964	0.85451		40.000	78.142	9.2307	3.7020
	20.000		39.025	3.5904	2.3505	1.5275	1.8950	0.81511		42.000	79.533	9.3212	3.7156
	22.000		41.505	3.9948	2.4982	1.5991	1.7906	0.77258		44.000	80.750	9.5597	3.7506
	24.000		44.160	4.4325	2.6451	1.6757	1.6825	0.72766		46.000	81.843	9.4475	3.7343
	26.000		47.042	4.9089	2.7916	1.7585	1.5692	0.68081		48.000	82.845	9.4928	3.7409
	28.000		50.247	5.4345	2.9391	1.8490	1.4481	0.63219		50.000	83.775	9.5296	3.7462
	30.000		53.992	6.0344	3.0917	1.9518	1.3127	0.58089		52.000	84.651	9.5597	3.7506
	32.000		59.037	6.8013	3.2659	2.0825	1.1407	0.52183		54.000	85.484	9.5842	3.7541
	32.984		65.097	7.6294	3.4320	2.2230	0.9503	0.46580		56.000	86.283	9.6038	3.7570
	32.000		70.389	8.2421	3.5425	2.3266	0.8001	0.42903		58.000	87.055	9.6191	3.7592
	30.000		73.893	8.5802	3.5995	2.3837	0.7107	0.41030		60.000	87.808	9.6506	3.7608
	28.000		76.127	8.7648	3.6295	2.4149	0.6588	0.40050		62.000	88.546	9.6887	3.7620
	26.000		77.855	8.8902	3.6495	2.4360	0.6220	0.39402		64.000	89.275	9.6434	3.7636
	24.000		79.297	8.9827	3.6640	2.4516	0.5938	0.38893		66.000			
	22.000		80.552	9.0543	3.6751	2.4637	0.5713	0.38574	2.95	2.000			
	20.000		81.676	9.1110	3.6838	2.4733	0.5530	0.38294		22.000	22.708		
	18.000		82.702	9.1567	3.6908	2.4810	0.5379	0.38069		24.000	24.294		
	16.000		83.655	9.1938	3.6964	2.4872	0.5253	0.37788		26.000	25.974		
	14.000		84.549	9.2241	3.7010	2.4923	0.5150	0.37741		28.000	27.749		
	12.000		85.399	9.2486	3.7047	2.4964	0.5064	0.37623		30.000	29.621		
	10.000		86.213	9.2683	3.7077	2.4998	0.4995	0.37528		32.000	31.593		
	8.000		87.001	9.2836	3.7100	2.5023	0.4940	0.37454		34.000	33.670		
	6.000		87.768	9.2952	3.7117	2.5043	0.4899	0.37399		36.000	35.856		
	4.000		88.520	9.3033	3.7129	2.5057	0.4870	0.37360		38.000	38.164		
	2.000		89.262	9.3080	3.7136	2.5065	0.4853	0.37338		40.000	40.607		
2.90	2.000		21.578	1.1604	1.1120	1.0435	2.8019	0.99966		42.000	43.211		
	4.000		23.076	1.3406	1.2320	1.0882	2.7062	0.99744		44.000	46.018		

Oblique Shock Tables ($\gamma = 1.4$)

M_1	θ	β	$\frac{p_2}{p_1}$	M_2	$\frac{p_{02}}{p_{01}}$	M_1	θ	β	$\frac{p_2}{p_1}$	M_2	$\frac{p_{02}}{p_{01}}$
2.95	32.000	56.997	6.9741	3.3023	2.1119	1.2199	0.50950	3.00	14.000	84.837	10.2483
	33.726	65.193	8.1990	3.5350	2.3194	0.9528	0.43150	12.000	85.638	10.2726	3.8459
	32.000	72.020	9.0188	3.6696	2.4577	0.7585	0.38752	10.000	86.408	10.2921	3.8491
	30.000	74.838	9.2917	3.7112	2.5037	0.6877	0.37416	8.000	87.154	10.3074	3.8517
	28.000	76.821	9.4585	3.7359	2.5318	0.6420	0.36628	6.000	87.881	10.3190	3.8537
	26.000	78.407	9.5762	3.7530	2.5516	0.6084	0.36086	4.000	88.594	10.3270	3.8553
	24.000	79.752	9.6649	3.7657	2.5666	0.5821	0.35684	2.000	89.299	10.3318	3.8563
	22.000	80.935	9.7342	3.7755	2.5782	0.5610	0.35374		3.8569	2.6787	0.4757
	20.000	82.000	9.7896	3.7834	2.5875	0.5437	0.35128				0.32841
	18.000	82.978	9.8345	3.7986	2.5951	0.5293	0.34931				
	16.000	83.889	9.8712	3.7947	2.6013	0.5173	0.34771				
	14.000	84.747	9.9012	3.7989	2.6063	0.5074	0.34641				
	12.000	85.563	9.9255	3.8023	2.6104	0.4992	0.34536				
	10.000	86.348	9.9450	3.8050	2.6137	0.4925	0.34452				
	8.000	87.106	9.9604	3.8071	2.6163	0.4872	0.34386				
	6.000	87.845	9.9719	3.8087	2.6182	0.4832	0.34336				
	4.000	88.571	9.9799	3.8098	2.6196	0.4804	0.34302				
	2.000	89.288	9.9847	3.8104	2.6204	0.4788	0.34282				
3.00	2.000	20.867	1.1656	1.1155	1.0449	2.8981	0.99963	24.000	42.361	47.796	1.74317
	4.000	22.355	1.3522	1.2395	1.0909	2.7988	0.99721	26.000	45.110	47.607	1.8039
	6.000	23.936	1.5616	1.3714	1.1387	2.7008	0.99105	28.000	48.102	52.806	1.8226
	8.000	25.611	1.7953	1.5101	1.1888	2.6031	0.97993	30.000	51.455	5.8462	1.8455
	10.000	27.383	2.0545	1.6546	1.2417	2.5050	0.96308	32.000	55.456	6.4722	1.8738
	12.000	29.251	2.3404	1.8036	1.2977	2.4060	0.94022	34.000	61.505	6.71967	1.9066
	14.000	31.218	2.6540	1.9556	1.3571	2.3056	0.91148	34.407	65.288	6.8795	1.9352
	16.000	33.288	2.9964	2.1095	1.4204	2.2037	0.87734	34.000	68.742	9.2596	1.9552
	18.000	35.467	3.3685	2.2641	1.4878	2.1000	0.83855	32.000	73.184	9.7779	1.98514
	20.000	37.764	3.7713	2.4181	1.5596	1.9941	0.79602	30.000	75.604	10.0154	2.1497
	22.000	40.192	4.2064	2.5708	1.6362	1.8858	0.75068	28.000	77.406	10.1703	2.3222
	24.000	42.775	4.6761	2.7216	1.7181	1.7744	0.70340	26.000	78.880	10.2825	2.4190
	26.000	45.552	5.1844	3.0706	1.8060	1.6589	0.65491	24.000	80.145	10.3683	2.5064
	28.000	48.586	5.7388	3.0184	1.9012	1.5374	0.60560	22.000	81.267	10.4361	2.6963
	30.000	52.014	6.3559	3.1673	2.0067	1.4059	0.55256	20.000	82.284	10.4966	2.8146
	32.000	56.182	7.0810	3.3244	2.1300	1.2541	0.50205	18.000	83.221	10.5350	2.9776
	34.000	63.673	8.2682	3.5470	2.3310	1.0029	0.42755	16.000	84.095	10.5714	3.0879
	34.073	65.241	8.4917	3.5848	2.3688	0.9540	0.41510	14.000	84.921	10.6012	3.8917
	34.000	66.749	8.6971	3.6186	2.4035	0.9083	0.40406	12.000	85.709	10.6255	3.8948
	32.000	72.642	9.3988	3.7271	2.5217	0.7428	0.36908	10.000	86.466	10.6450	3.8973
	30.000	75.239	9.6517	3.7638	2.5643	0.6779	0.35743	8.000	87.199	10.6603	3.8992
	28.000	77.126	9.8121	3.7865	2.5913	0.6345	0.35029	6.000	87.914	10.6719	3.9007
	26.000	78.652	9.9268	3.8024	2.6106	0.6022	0.34530	4.000	88.617	10.6799	3.9017
	24.000	79.956	10.0139	3.8144	2.6253	0.5768	0.34157	2.000	89.310	10.6847	3.9023
	22.000	81.106	10.0824	3.8237	2.6368	0.5563	0.33868				0.4728
	20.000	82.147	10.1373	3.8311	2.6460	0.5394	0.33638				0.31456
	18.000	83.103	10.1819	3.8371	2.6536	0.5253	0.33453	3.10	2.000	21.684	1.2471
	16.000	83.996	10.2184	3.8420				4.000			0.99696

Oblique Shock Tables ($\gamma = 1.4$)

M_1	θ	β	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	M_2	$\frac{p_{02}}{p_{01}}$	M_1	θ	β	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	M_2	$\frac{p_{02}}{p_{01}}$		
3.10	6.000	1.3835	1.1431	2.7894	0.99027	3.15	28.000	47.216	6.0688	3.1000	1.9577	1.6194	0.57808		
	8.000	1.5815	1.8249	1.5271	2.6881	0.97822	30.000	50.449	6.7158	3.2475	2.0680	1.4886	0.52806		
	10.000	2.2558	24.927	1.0956	1.2499	2.5884	0.96004	54.201	7.4487	3.3975	2.1924	1.3441	0.47738		
	12.000	26.692	28.554	2.3949	1.8308	1.3081	2.4887	0.93546	34.000	59.196	8.3736	2.3489	1.1632	0.42162	
	14.000	30.513	39.421	2.7236	1.9879	1.3701	2.3798	0.90473	35.033	65.382	9.4008	3.7274	2.5221	0.9575	
	16.000	32.574	41.968	3.0831	2.1467	1.4362	2.2743	0.86841	34.000	70.719	10.1474	3.8325	2.6478	0.7974	
	18.000	34.739	34.740	3.4740	2.3057	1.5067	2.1672	0.82741	32.000	74.089	10.5396	3.8839	2.7137	0.7064	
	20.000	37.017	38.8973	2.4637	1.5819	2.0581	0.78278	30.000	76.244	10.7550	3.9111	2.7499	0.6531	0.31190	
	22.000	39.421	43.543	2.6198	1.6621	1.9468	0.73556	28.000	77.906	10.9014	3.9292	2.7745	0.6152	0.30644	
	24.000	41.968	48.470	2.7733	1.7477	1.8529	0.68676	26.000	79.289	11.0097	3.9424	2.7927	0.5860	0.30248	
	26.000	44.692	53.788	2.9241	1.8395	1.7154	0.63718	24.000	80.490	11.0936	3.9524	2.80688	0.5627	0.29947	
	28.000	47.646	59.563	3.0563	3.07225	1.9385	1.5928	0.58731	22.000	81.560	11.1602	3.9604	2.8180	0.5436	0.29710
	30.000	50.935	65.922	3.2205	2.0470	1.4620	0.53722	20.000	82.535	11.2142	3.9668	2.8270	0.5278	0.29520	
	32.000	54.800	7.3197	3.3723	2.1705	1.3157	0.48586	18.000	83.436	11.2583	3.9720	2.8344	0.5145	0.29366	
	34.000	60.205	82.768	3.5485	2.3525	1.1241	0.42706	16.000	84.279	16.000	3.9762	2.8405	0.5035	0.29240	
	34.726	65.335	9.0925	3.6810	2.4701	0.9564	0.38835	14.000	85.076	11.3243	3.9797	2.8455	0.4942	0.29138	
	34.000	69.872	9.7174	3.7732	2.5754	0.8203	0.35449	12.000	85.838	11.3486	3.9825	2.8496	0.4865	0.29054	
	32.000	73.661	10.1577	3.8339	2.6495	0.7171	0.33553	10.000	86.571	11.3682	3.9848	2.8529	0.4803	0.28987	
	30.000	75.938	10.3831	3.8636	2.6874	0.66007	0.32664	8.000	87.281	11.3835	3.9866	2.8555	0.4754	0.28935	
	28.000	77.666	10.5334	3.8831	2.7126	0.62112	0.32040	6.000	87.976	11.3951	3.9879	2.8574	0.4716	0.28895	
	26.000	79.091	10.6435	3.8971	2.7311	0.5911	0.31614	4.000	88.687	11.4032	3.9889	2.8588	0.4690	0.28867	
	24.000	80.324	10.7282	3.9077	2.7454	0.5671	0.31291	2.000	89.330	11.4080	3.9894	2.8596	0.4674	0.28851	
	22.000	81.419	10.7954	3.9161	2.7567	0.5476	0.31038								
	20.000	82.413	10.8496	3.9228	2.7658	0.5314	0.30836								
	18.000	83.331	10.8938	3.9282	2.7732	0.5179	0.30672								
	16.000	84.189	10.9301	3.9327	2.7793	0.5067	0.30559								
	14.000	85.001	10.9599	3.9363	2.7843	0.4973	0.30430								
	12.000	85.775	10.9842	3.9393	2.7884	0.4895	0.30341								
	10.000	86.520	11.0037	3.9416	2.7917	0.4832	0.30270								
	8.000	87.242	11.0190	3.9435	2.7942	0.4781	0.30215								
	6.000	87.945	11.0306	3.9449	2.7962	0.4743	0.30173								
	4.000	88.637	11.0387	3.9458	2.7975	0.4716	0.30144								
	2.000	89.321	11.0434	3.9464	2.7983	0.4701	0.30127								
	3.15	2.000	1.1734	1.2510	1.3896	1.1453	1.5357	1.1981	2.7304	0.97734	3.020	2.00	19.587	1.1226	
		19.891	21.366	22.937	24.603	1.8399	1.8878	1.9811	2.6267	0.95846	32.000	41.238	4.000	1.0475	
									0.9330	0.90123	43.920	52.045	6.000	1.0965	
									0.90123	0.87297	46.811	55.816	6.000	1.0965	
									0.86382	0.83902	49.994	58.350	6.000	1.0965	
									0.82003	0.79851	53.651	62.780	6.000	1.0965	
									0.77734	0.75304	56.811	65.428	6.000	1.0965	
									0.73041	0.70670	59.714	68.842	6.000	1.0965	
									0.68676	0.66089	62.6670	71.582	6.000	1.0965	
									0.64654	0.62199	72.1377	81.090	6.000	1.0965	
									0.60646	0.58185	74.5607	83.508	6.000	1.0965	
									0.56680	0.54233	75.952	84.906	6.000	1.0965	
									0.52683	0.50687	78.0206	87.048	6.000	1.0965	
									0.48673	0.46873	79.8743	88.964	6.000	1.0965	
									0.44654	0.42683	80.3437	89.454	6.000	1.0965	
									0.40646	0.38685	81.7229	90.858	6.000	1.0965	
									0.36636	0.34685	82.9055	92.025	6.000	1.0965	
									0.32636	0.30685	83.9718	93.090	6.000	1.0965	
									0.28625	0.26690	84.8252	94.048	6.000	1.0965	
									0.24623	0.22690	85.780	95.061	6.000	1.0965	
									0.20611	0.18693	86.6888	96.901	6.000	1.0965	
									0.16601	0.146873	87.7743	97.985	6.000	1.0965	
									0.12591	0.10680	88.7743	99.097	6.000	1.0965	
									0.08580	0.06573	89.7743	100.2152	6.000	1.0965	
									0.04571	0.02560	90.7743	101.795	6.000	1.0965	
									0.00560	0.00560	91.7743	102.714	6.000	1.0965	

Oblique Shock Tables ($\gamma = 1.4$)

M_1	θ	β	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	M_2	$\frac{P_{02}}{P_{01}}$	M_1	θ	β	$\frac{P_2}{P_1}$	$\frac{\rho_2}{\rho_1}$	M_2	$\frac{P_{02}}{P_{01}}$		
3.20	22.000	81.694	11.5307	4.0035	2.8802	0.5398	0.28438	3.30	2.000	19.009	1.1812	3.1858	0.99953		
	20.000	82.649	11.5844	4.0096	2.8892	0.5243	0.28260		4.000	20.475	1.3880	3.0748	0.99642		
	18.000	83.533	11.6285	4.0146	2.8966	0.5113	0.28115		6.000	22.039	1.6222	1.4082	2.9663		
	16.000	84.363	11.6647	4.0187	2.9026	0.5004	0.27996		8.000	23.699	1.8859	1.5617	2.8563		
	14.000	85.147	11.6945	4.0220	2.9076	0.4913	0.27899		10.000	25.457	2.1807	1.7216	2.7468		
	12.000	85.897	11.7188	4.0247	2.9117	0.4837	0.27820		12.000	27.310	2.5078	1.8861	2.6384		
	10.000	86.619	11.7385	4.0269	2.9150	0.4776	0.27757		14.000	29.261	2.8688	2.0536	2.5248		
	8.000	87.320	11.7539	4.0286	2.9176	0.4727	0.27707		16.000	31.308	3.2640	2.2219	2.4118		
	6.000	88.003	11.7655	4.0299	2.9196	0.4690	0.27669		18.000	33.456	3.6947	2.3888	2.8494		
	4.000	88.675	11.7736	4.0308	2.9209	0.4664	0.27643		20.000	35.710	4.1617	2.5557	3.0409		
	2.000	89.340	11.7784	4.0313	2.9217	0.4649	0.27628		22.000	38.077	4.6655	2.7184	2.1813		
									24.000	40.573	5.2081	2.8773	2.0636		
									26.000	43.222	5.7918	3.0318	1.903		
									28.000	46.062	6.4212	3.1822	1.8215		
									30.000	49.163	7.1057	3.3294	1.7108		
									32.000	52.667	7.8868	3.4758	1.6018		
										34.000	56.963	8.7622	3.6291	1.5020	
										35.882	65.518	10.3564	3.8602	1.4064	
										34.000	72.501	11.3896	3.9873	1.2974	
										32.000	75.148	11.7036	4.0230	1.1916	
										30.000	77.029	11.8983	4.0445	1.0482	
										28.000	78.535	12.0364	4.0595	0.9593	
										26.000	79.812	12.1408	4.0706	0.8606	
										24.000	80.932	12.2227	4.0793	0.7502	
										22.000	81.938	12.2884	4.0862	0.6507	
										20.000	82.859	12.3420	4.0964	0.5507	
										18.000	83.714	12.3860	4.1077	0.4516	
										16.000	84.517	12.4223	4.1101	0.3525	
										14.000	85.278	12.4523	4.1132	0.2535	
										12.000	86.007	12.4767	4.1157	0.1538	
											20.000	82.059	12.4964	4.1077	0.4725
											18.000	83.714	12.5120	4.1093	0.3448
											16.000	84.517	12.5237	4.1105	0.2403
											14.000	85.278	12.5319	4.1114	0.1416
											12.000	86.007	12.5387	4.1119	0.0481
											20.000	89.357	12.5387	4.1119	0.4601
											18.000	87.357	12.5387	4.1119	0.2932
											16.000	88.000	12.5387	4.1119	0.1917
											14.000	88.665	12.5387	4.1119	0.0951
											12.000	89.326	12.5387	4.1119	0.0090
											10.000	90.000	12.5387	4.1119	0.0000
											8.000	90.665	12.5387	4.1119	0.0000
											6.000	91.330	12.5387	4.1119	0.0000
											4.000	92.000	12.5387	4.1119	0.0000
											2.000	92.665	12.5387	4.1119	0.0000
											0.000	93.330	12.5387	4.1119	0.0000

Oblique Shock Tables ($\gamma = 1.4$)

M_1	θ	β	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	M_2	$\frac{p_{02}}{p_{01}}$	M_1	θ	β	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	M_2	$\frac{p_{02}}{p_{01}}$				
3.35	24.000	40.264	2.9033	1.8263	1.9704	0.64409	3.40	34.000	73.352	12.2131	2.9946	0.7279	0.26279				
	26.000	42.898	5.8998	3.0588	1.9288	0.59200		32.000	75.717	12.4992	4.1080	3.0426	0.25440				
	28.000	45.716	6.5433	3.2097	2.0386	1.7198	0.54090		30.000	77.467	12.6849	4.1268	3.0738	0.24914			
	30.000	48.782	7.2416	3.3568	2.1573	1.5874	0.49109		28.000	78.891	12.8193	4.1402	3.0963	0.24542			
	32.000	52.225	8.0134	3.5024	2.2880	1.4458	0.44232		26.000	80.110	12.9221	4.1503	3.1135	0.24263			
	34.000	56.375	8.9114	3.6528	2.4396	1.2844	0.39294		24.000	81.185	13.0033	4.1582	3.1271	0.24046			
	36.000	63.380	10.2976	3.8524	2.6730	1.0339	0.32979		22.000	82.156	13.0688	4.1645	3.1381	0.23872			
	36.143	65.562	10.6853	3.9023	2.7382	0.9616	0.31454		20.000	83.047	13.1224	4.1697	3.1471	0.23732			
	36.000	67.623	11.0286	3.9446	2.7958	0.8957	0.30180		18.000	83.876	13.1665	4.1739	3.1545	0.23617			
	34.000	72.950	11.8006	4.0338	2.9255	0.7384	0.27557		16.000	84.656	13.2030	4.1774	3.1606	0.23522			
	32.000	75.444	12.0992	4.0662	2.9755	0.6723	0.26624		14.000	85.396	13.2331	4.1802	3.1657	0.23445			
	30.000	77.255	12.2891	4.0863	3.0074	0.6279	0.26053		12.000	86.105	13.2578	4.1826	3.1698	0.23381			
	28.000	78.719	12.4252	4.1004	3.0302	0.5946	0.25653		10.000	86.789	13.2777	4.1844	3.1731	0.23330			
	26.000	79.965	12.5287	4.1110	3.0476	0.5684	0.25355		8.000	87.453	13.2994	4.1859	3.1757	0.23290			
	24.000	81.062	12.6102	4.1193	3.0612	0.5471	0.25124		6.000	88.103	13.3052	4.1870	3.1777	0.23260			
	22.000	82.050	12.6758	4.1259	3.0722	0.5295	0.24939		4.000	88.741	13.3135	4.1878	3.1791	0.23239			
	20.000	82.956	12.7293	4.1313	3.0812	0.5148	0.24790		2.000	89.372	13.3184	4.1883	3.1799	0.23227			
	18.000	83.798	12.7734	4.1357	3.0886	0.5024	0.246668										
	16.000	84.588	12.8098	4.1393	3.0947	0.4920	0.24568										
	14.000	85.339	12.8398	4.1422	3.0997	0.4832	0.24486	3.45	2.000	18.209	1.1892	1.1316	1.0509	3.3292			
	12.000	86.057	12.8644	4.1446	3.1038	0.4780	0.24420		4.000	19.668	1.4063	1.2743	1.1036	0.99947			
	10.000	86.750	12.8842	4.1466	3.1072	0.4701	0.24366		6.000	21.226	1.6536	1.4270	1.1588	0.99597			
	8.000	87.422	12.8998	4.1481	3.1098	0.4654	0.24324		8.000	22.884	1.9331	1.5881	1.2172	0.98718			
	6.000	88.080	12.9116	4.1493	3.1118	0.4618	0.24292		10.000	24.639	2.2468	1.7559	1.2796	0.9809			
	4.000	88.726	12.9198	4.1501	3.1131	0.4593	0.24270		12.000	26.491	2.5962	1.9284	1.3463	0.97149			
	2.000	89.365	12.9246	4.1506	3.1140	0.4578	0.24256		14.000	28.438	2.9823	2.1035	1.4178	0.97149			
									16.000	30.481	3.4063	2.2791	1.4946	2.5118	0.83456		
									18.000	32.821	3.8688	2.4535	1.5789	2.3915	0.78577		
									20.000	34.863	4.3706	2.6251	1.6649	2.2698	0.73391		
										22.000	37.213	4.9123	2.7926	1.7590	2.1468	0.68049	
										24.000	39.883	5.4951	2.9552	1.8595	2.0224	0.62680	
										26.000	42.292	6.1211	3.1125	1.9666	1.8960	0.57385	
										28.000	45.073	6.7941	3.2644	2.0813	1.7667	0.52235	
										30.000	48.080	7.5215	3.4115	2.2047	1.6329	0.47267	
											32.000	51.420	8.3194	3.5558	2.3397	1.4914	0.42456
											34.000	55.344	9.2294	3.7018	2.4932	1.3339	0.37715
											36.000	60.903	10.4958	3.8705	2.6962	1.1265	0.32424
											38.000	65.647	11.3584	3.9837	2.8512	0.9634	0.29020
											39.000	69.850	12.0718	4.0633	2.9709	0.8302	0.26708
											40.000	73.716	12.6278	4.1211	3.0642	0.7184	0.25074
											42.000	75.970	12.9035	4.1485	3.1104	0.6589	0.24313
											44.000	77.665	13.0858	4.1662	3.1410	0.6175	0.23828
											46.000	79.054	13.2189	4.1789	3.1633	0.5860	0.23481
											48.000	80.246	13.3210	4.1885	3.1804	0.5609	0.23220
											50.000	81.302	13.4020	4.1961	3.1939	0.5404	0.23016
											52.000	82.256	13.4675	4.2021	3.2049	0.5234	0.22852
											54.000	83.134	13.5211	4.2071	3.2139	0.5091	0.22719
											56.000	83.951	13.5654	4.2111	3.2213	0.4971	0.22611

Oblique Shock Tables ($\gamma = 1.4$)

M_1	θ	β	$\frac{p_2}{p_1}$	M_2	$\frac{p_{02}}{p_{01}}$	M_1	θ	β	$\frac{p_2}{p_1}$	$\frac{\rho_2}{\rho_1}$	M_2	$\frac{T_2}{T_1}$	$\frac{p_{02}}{p_{01}}$
3.70	20.000	83.507	4.3797	3.5621	0.4969	0.18289	3.75	2.000	89.416	16.2379	4.4261	3.6687	0.4428
	18.000	84.274	15.6008	4.3831	3.5696	0.4856	0.18206	0.18138	3.80	2.000	16.600	1.2083	1.1445
	16.000	84.998	15.6460	4.3859	3.5759	0.4760	0.18082	0.18035	4.000	18.048	1.4503	1.3022	1.1137
	14.000	85.687	15.7147	4.3882	3.5811	0.4680	0.17998	0.17958	6.000	19.602	1.7294	1.4718	1.1750
	12.000	86.348	15.7402	4.3901	3.5854	0.4613	0.17947	0.17932	8.000	21.258	2.0480	1.6511	1.2404
	10.000	86.988	15.7609	4.3916	3.5889	0.4558	0.17998	0.17958	10.000	23.016	2.4088	1.8377	1.3108
	8.000	87.610	15.7772	4.3928	3.5916	0.4515	0.17969	0.17922	12.000	24.872	2.8134	2.0288	1.3867
	6.000	88.219	15.7896	4.3937	3.5937	0.4481	0.17947	0.17900	14.000	26.821	3.2631	2.2216	1.4688
	4.000	88.817	15.7982	4.3944	3.5951	0.4458	0.17932	0.17887	16.000	28.864	3.7592	2.4137	1.5575
	2.000	89.411	15.8033	4.3947	3.5960	0.4444	0.17922	0.17844	18.000	31.000	4.3021	2.6026	1.6530
3.75	2.000	16.810	1.2055	1.1426	1.0551	3.6149	0.99933	0.99497	20.000	33.229	4.8923	2.7867	1.7556
	4.000	18.260	1.4440	1.2982	1.1123	3.4840	0.99405	0.98405	22.000	35.556	5.5299	2.9644	2.4644
	6.000	19.814	1.7184	1.4654	1.1727	3.3550	0.98405	0.96476	24.000	37.989	6.2157	3.1348	2.3283
	8.000	21.470	2.0312	1.6420	1.2370	3.2264	0.96476	0.94476	26.000	40.542	6.9510	3.2975	2.1919
	10.000	23.227	2.3849	1.8258	1.3062	3.0974	0.93634	0.89905	28.000	43.234	7.7378	3.4523	2.2414
	12.000	25.081	2.7813	2.0142	1.3808	2.9674	0.89905	0.85397	30.000	46.105	8.5816	3.5997	2.3840
	14.000	27.030	3.2217	2.2046	1.4614	2.8363	0.85397	0.80280	32.000	49.218	9.4923	3.7408	2.5375
	16.000	29.072	3.7069	2.3943	1.5482	2.7042	0.80280	0.75199	34.000	52.702	10.4940	3.8780	2.7060
	18.000	31.207	4.2379	2.5813	1.6417	2.5712	0.74744	0.6964	36.000	56.894	11.6543	4.0175	3.2978
	20.000	33.498	4.8148	2.7637	1.7422	2.4376	0.68987	0.63185	38.000	64.192	13.4871	4.2039	3.2082
	22.000	35.767	5.4382	2.9401	1.8497	2.3034	0.63185	0.57486	38.092	65.921	13.8756	4.2390	3.2733
	24.000	38.204	6.1086	3.1095	1.9645	2.1688	0.57486	0.51996	38.000	67.568	14.2269	4.2696	3.3321
	26.000	40.762	6.8272	3.2714	2.0869	2.0333	0.51996	0.46786	36.000	73.114	15.2586	4.3536	3.5048
	28.000	43.464	7.5969	3.4259	2.2175	1.8964	0.46786	0.35772	34.000	75.572	15.6341	4.3822	3.5676
	30.000	46.350	8.4228	3.5733	2.3572	1.7570	0.41888	0.31200	32.000	77.342	15.8710	4.3997	3.6073
	32.000	49.486	9.3159	3.7148	2.5078	1.6129	0.37300	0.27300	30.000	78.762	16.0402	4.4120	3.6356
	34.000	53.014	10.3013	3.8529	2.6736	1.4594	0.32964	0.23700	28.000	79.967	16.1687	4.4212	3.6561
	36.000	57.310	11.4538	3.9947	2.8672	1.2839	0.28696	0.19683	26.000	81.022	16.2697	4.4284	3.6740
	37.906	65.884	13.5007	4.2052	3.2105	0.9683	0.22770	0.18053	24.000	81.969	16.3542	4.4341	3.6876
	36.000	72.794	14.8041	4.3176	3.4287	0.7481	0.19834	0.15129	22.000	82.833	16.4178	4.4387	3.6988
	34.000	75.361	15.1917	4.3484	3.4936	0.6755	0.19061	0.12129	20.000	83.634	16.4729	4.4426	3.7080
	32.000	77.180	15.4318	4.3669	3.5338	0.6280	0.18602	0.10129	18.000	84.383	16.5186	4.4457	3.7156
	30.000	78.631	15.6021	4.3798	3.5623	0.5926	0.18286	0.08129	16.000	85.092	16.5567	4.4484	3.7220
	28.000	79.856	15.7307	4.3894	3.5838	0.5649	0.18053	0.06129	14.000	85.767	16.5882	4.4505	3.7273
	26.000	80.927	15.8316	4.3968	3.6007	0.5423	0.17872	0.04129	12.000	86.415	16.6141	4.4523	3.7316
	24.000	81.887	15.9128	4.4028	3.6143	0.5237	0.17728	0.02129	10.000	87.043	16.6352	4.4537	3.7351
	22.000	82.762	15.9792	4.4076	3.6254	0.5080	0.17612	0.00129	8.000	87.653	16.6518	4.4549	3.7379
	20.000	83.572	16.0339	4.4115	3.6345	0.4948	0.17517	0.00129	6.000	88.251	16.6643	4.4557	3.7400
	18.000	84.380	16.0794	4.4148	3.6422	0.4836	0.17439	0.00129	4.000	88.839	16.6731	4.4563	3.7414
	16.000	85.045	16.1172	4.4175	3.6485	0.4741	0.17374	0.00129	2.000	89.421	16.6783	4.4567	3.7423
	14.000	85.727	16.1485	4.4198	3.6537	0.4662	0.17321	0.00129	0.000	89.421	16.6783	4.4412	3.7427
	12.000	86.382	16.1743	4.4216	3.6580	0.4595	0.17277	0.00129	0.000	89.421	16.6783	4.4412	3.7427
	10.000	87.016	16.1951	4.4231	3.6615	0.4541	0.17242	0.00129	0.000	89.421	16.6783	4.4412	3.7427
	8.000	87.662	16.2116	4.4242	3.6643	0.4498	0.17214	0.00129	0.000	89.421	16.6783	4.4412	3.7427
	6.000	88.235	16.2240	4.4251	3.6663	0.4465	0.17193	0.00129	0.000	89.421	16.6783	4.4412	3.7427
	4.000	88.829	16.2327	4.4257	3.6678	0.4441	0.17178	0.00129	0.000	89.421	16.6783	4.4412	3.7427

Oblique Shock Tables ($\gamma = 1.4$)

M_1	θ	β	$\frac{p_2}{p_1}$	$\frac{M_2}{M_1}$	$\frac{p_{02}}{p_{01}}$	$\frac{T_2}{T_1}$	$\frac{\rho_2}{\rho_1}$	M_2	$\frac{T_2}{T_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{p_{02}}{p_{01}}$
3.85	10.000	22.812	2.4328	1.3153	3.1734	0.93209	3.90	24.000	37.584	6.4345	3.1853
	12.000	24.668	2.8456	2.0432	3.0927	0.89264	26.000	40.126	7.2035	3.3492	2.1598
	14.000	26.619	3.3050	2.2386	1.4764	2.9028	0.84523	28.000	42.802	8.0258	3.5046
	16.000	28.664	3.8121	2.4330	1.5668	2.7661	0.79172	30.000	45.646	9.0559	3.6519
	18.000	30.799	4.3670	2.6239	1.6643	2.6287	0.73428	32.000	48.716	9.8536	3.7923
	20.000	33.028	4.9706	2.8097	1.7691	2.4909	0.67493	34.000	52.126	10.8901	3.9278
	22.000	35.353	5.6230	2.9887	1.8814	2.3529	0.61558	36.000	56.149	12.0723	4.0683
	24.000	37.783	6.3245	3.1601	2.0013	2.2146	0.55770	38.000	62.087	13.6897	4.2224
	26.000	40.330	7.0764	3.3234	2.1293	2.0760	0.50236	38.445	65.991	14.6407	4.3043
	28.000	43.014	7.8808	3.4785	2.2656	1.9364	0.45026	38.000	69.501	15.4023	4.3847
	30.000	45.871	8.7425	3.6259	2.4111	1.7948	0.40167	36.000	73.678	16.1758	4.4248
	32.000	48.961	9.6715	3.7666	2.5677	1.6493	0.35654	34.000	75.956	16.5334	4.4468
	34.000	52.407	10.6904	3.9030	2.7390	1.4957	0.31434	32.000	77.640	16.7653	4.4626
	36.000	56.508	11.8605	4.0404	2.9385	1.3239	0.27366	30.000	79.006	16.9330	4.4738
	38.000	62.939	13.5472	4.2095	3.2183	1.0767	0.22655	28.000	80.172	17.0613	4.4823
	38.272	65.956	14.2556	4.2721	3.3389	0.9697	0.21003	26.000	81.199	17.1629	4.4890
	38.000	68.733	14.8512	4.3214	3.4366	0.8764	0.19738	24.000	82.121	17.2449	4.4943
	36.000	73.407	15.7160	4.3883	3.5814	0.7314	0.18079	22.000	82.966	17.3122	4.4986
	34.000	75.770	16.0813	4.4150	3.6425	0.6649	0.17436	20.000	83.749	17.3680	4.5022
	32.000	77.495	16.3155	4.4316	3.6816	0.6198	0.17036	18.000	84.483	17.4143	4.5052
	30.000	78.888	16.4839	4.4433	3.7098	0.5859	0.16762	16.000	85.177	17.4529	4.5076
	28.000	80.072	16.6122	4.4522	3.7313	0.55591	0.16555	14.000	85.840	17.4850	4.5097
	26.000	81.112	16.7135	4.4591	3.7482	0.5372	0.16394	12.000	86.477	17.5113	4.5114
	24.000	82.047	16.7952	4.4646	3.7619	0.5190	0.16266	10.000	87.093	17.5327	4.5127
	22.000	82.901	16.8522	4.4691	3.7731	0.5037	0.16162	8.000	87.693	17.5496	4.5138
	20.000	83.692	16.9175	4.4728	3.7823	0.4907	0.16076	6.000	88.280	17.5623	4.5146
	18.000	84.434	16.9636	4.4758	3.7900	0.4798	0.16006	4.000	88.858	17.5713	4.5151
	16.000	85.136	17.0019	4.4784	3.7964	0.4705	0.15947	2.000	89.430	17.5766	4.5155
	14.000	85.804	17.0337	4.4805	3.8017	0.4627	0.15899				
	12.000	86.447	17.0598	4.4822	3.8061	0.4561	0.15859				
	10.000	87.068	17.0810	4.4836	3.8097	0.4508	0.15827				
	8.000	87.674	17.0978	4.4847	3.8125	0.4465	0.15782				
	6.000	88.266	17.1104	4.4855	3.8146	0.4433	0.15783				
	4.000	88.849	17.1193	4.4861	3.8161	0.4410	0.15770				
	2.000	89.426	17.1245	4.4865	3.8169	0.4397	0.15762				
3.90	2.000	16.196	1.2138	1.1482	1.0571	3.7573	0.99926	3.95	2.000	16.001	1.2166
	4.000	17.642	1.4633	1.3104	1.1167	3.6191	0.99441			17.447	1.4697
	6.000	19.196	1.7517	1.4849	1.1797	3.4893	0.98232			19.001	1.7630
	8.000	20.854	2.0821	1.6694	1.2472	3.3473	0.96105			20.000	1.8125
	10.000	22.614	2.4570	1.8614	1.3200	3.2111	0.92990			22.000	2.0724
	12.000	24.472	2.8783	2.0578	1.3987	3.0739	0.88935			24.000	2.3234
	14.000	26.424	3.3474	2.2557	1.4840	2.9357	0.84077			26.000	2.3931
	16.000	28.469	3.8655	2.4523	1.5763	2.7967	0.78611			28.000	2.483
	18.000	30.605	4.4329	2.6452	1.6758	2.6570	0.72761			30.000	2.5131
	20.000	32.834	5.0501	2.8826	1.7828	2.5171	0.66743			32.000	2.5859
	22.000	35.157	5.7171	3.0129	1.8975	2.3771	0.60746			36.000	35.812

Oblique Shock Tables ($\gamma = 1.4$)

M_1	θ	β	$\frac{P_2}{P_1}$	M_2	$\frac{P_{02}}{P_{01}}$	M_1	θ	β	$\frac{P_2}{P_1}$	M_2	$\frac{T_2}{T_1}$	$\frac{P_{02}}{P_{01}}$
3.95	38.000	61.406	13.8667	4.2383	3.2718	1.1389	0.21889	4.00	28.000	80.359	17.9765	4.5402
	38.612	66.026	15.0309	4.3358	3.4667	0.9711	0.19376	26.000	81.359	18.0787	4.5464	3.9594
	38.000	70.101	15.9275	4.4038	3.6167	0.8345	0.17703	24.000	82.261	18.1615	4.5514	0.5392
	36.000	73.928	16.6412	4.4541	3.7361	0.7172	0.16509	22.000	83.087	18.2296	4.5555	0.5126
	34.000	76.131	16.9804	4.4776	3.7945	0.6554	0.15965	20.000	83.854	18.2861	4.5688	0.4017
	32.000	77.777	17.2203	4.4927	3.8330	0.6125	0.15620	18.000	84.574	18.3331	4.5616	0.4111
	30.000	79.120	17.3877	4.5035	3.8609	0.5798	0.15375	16.000	85.256	18.3723	4.5659	0.4087
	28.000	80.288	17.5161	4.5117	3.8824	0.5537	0.15191	14.000	85.907	18.4049	4.5659	0.4037
	26.000	81.281	17.6179	4.5181	3.8994	0.5324	0.15047	12.000	86.533	18.4317	4.5674	0.3810
	24.000	82.192	17.7003	4.5232	3.9132	0.5147	0.14932	10.000	87.139	18.4535	4.5687	0.3855
	22.000	83.028	17.7580	4.5274	3.9245	0.4997	0.14838	8.000	87.730	18.4707	4.5697	0.3934
	20.000	83.803	17.8241	4.5309	3.9339	0.4870	0.14761	6.000	88.307	18.4837	4.5705	0.4042
	18.000	84.529	17.8708	4.5338	3.9417	0.4762	0.14698	4.000	88.876	18.4928	4.5710	0.3996
	16.000	85.218	17.9097	4.5362	3.9482	0.4671	0.14645	2.000	89.439	18.4982	4.5713	0.3885
	14.000	85.874	17.9420	4.5381	3.9536	0.4594	0.14601				4.0466	0.4354
	12.000	86.505	17.9886	4.5393	3.9581	0.4530	0.14566				0.4421	0.13878
	10.000	87.116	17.9902	4.5411	3.9617	0.4477	0.14537				0.4390	
	8.000	87.711	18.0072	4.5421	3.9645	0.4435	0.14514				0.4386	
	6.000	88.294	18.0201	4.5429	3.9667	0.4404	0.14497				0.4370	
	4.000	88.888	18.0291	4.5434	3.9682	0.4381	0.14485				0.4354	
	2.000	89.435	18.0345	4.5438	3.9691	0.4368	0.14478				0.4346	
4.00	2.000	15.813	1.2194	1.2159	1.0586	1.1196	3.8521	0.99920				
	4.000	17.258	1.4763	1.3185	1.0586	1.1844	3.7089	0.99401				
	6.000	18.812	1.7743	1.4980	1.0586	1.2540	3.5679	0.98110				
	8.000	20.471	2.1166	1.6879	1.0586	1.3293	3.4273	0.95845				
	10.000	22.234	2.5061	1.8853	1.0586	2.0870	3.2860	0.92542				
	12.000	24.095	2.9445	2.0870	1.4109	3.1439	3.08264	0.88264				
	14.000	26.050	3.4334	2.2898	1.4994	3.0009	0.83170					
	16.000	28.098	3.9741	2.4909	1.5954	2.8570	0.77474					
	18.000	30.236	4.5667	2.6877	1.6991	2.7128	0.71422					
	20.000	32.464	5.2116	2.8782	1.8107	2.5686	0.65240					
	22.000	34.786	5.9090	3.0611	1.9304	2.4246	0.59123					
	24.000	37.208	6.6592	3.22352	2.0583	2.2809	0.53224					
	26.000	39.740	7.4225	3.4002	2.1947	2.1374	0.47648					
	28.000	42.402	8.3215	3.5561	2.3401	1.9985	0.42453					
	30.000	45.224	9.2297	3.7034	2.4949	1.8485	0.37666					
	32.000	48.258	10.2259	3.8430	2.6609	1.7006	0.33272					
	34.000	51.605	11.2295	3.9768	2.8413	1.5463	0.29223					
	36.000	55.495	12.5100	4.1091	3.0444	1.3776	0.25409					
	38.000	60.827	14.0647	4.2556	3.3049	1.1637	0.21432					
	38.774	66.059	15.4261	4.3685	3.5329	0.9717	0.18613					
	38.000	70.601	16.4407	4.4403	3.7026	0.8196	0.16833					
	36.000	74.161	17.1095	4.4855	3.8144	0.7109	0.15785					
	34.000	76.297	17.4525	4.5076	3.8718	0.6511	0.15282					
	32.000	77.908	17.6608	4.5220	3.9099	0.6090	0.14959					
	30.000	79.227	17.8479	4.5324	3.9379	0.5769	0.14729					