

ENGINEERING TRIPOS PART IIB

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Wednesday 23 April 2008 2.30 to 4

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Module 4A3

TURBOMACHINERY I

*Answer not more than two questions.*

*All questions carry the same number of marks.*

*The **approximate** percentage of marks allocated to each part of a question is indicated in the right margin.*

*Attachment:*

*Compressible Flow Data Book (38 pages).*

STATIONERY REQUIREMENTS

Single-sided script paper

SPECIAL REQUIREMENTS

Engineering Data Book

CUED approved calculator allowed

**You may not start to read the questions printed on the subsequent pages of this question paper until instructed that you may do so by the Invigilator**

- 1 (a) For an axial flow turbine with repeating stages derive the relationship

$$A = 1 - \frac{\psi}{2} \left( \frac{\tan \alpha_2 + \tan \alpha_1}{\tan \alpha_2 - \tan \alpha_1} \right)$$

where  $A$  is the reaction,  $\psi$  the stage loading coefficient,  $\alpha_1$  the stator inlet flow angle and  $\alpha_2$  the stator exit flow angle. [25%]

(b) A low-pressure turbine within a turbofan jet engine consists of 5 repeating stages. The turbine inlet stagnation temperature is 1200 K and the inlet stagnation pressure is 213 kPa. It operates with a mass flow of  $15 \text{ kg s}^{-1}$  and generates 6.64 MW of mechanical power. The stator in each turbine stage turns the flow from  $-15^\circ$  at stator inlet to  $70^\circ$  at stator outlet. The turbine mean radius is 0.46 m and the rotational shaft speed is 5600 rpm.

(i) Calculate the turbine stage loading coefficient and hence show that the reaction is 0.5. Determine the flow coefficient and sketch the velocity triangles for one complete stage. [30%]

(ii) Calculate the annulus area at inlet to the turbine. Use this to estimate the blade height and the hub-to-tip radius ratio for the stator in the first turbine stage. [20%]

(c) The turbine is redesigned to have 4 repeating stages. The reaction, mean radius and the flow angles at inlet and outlet from each stator are all kept the same as the 5-stage design described in part (b). The inlet stagnation conditions, mass flow rate and power output are also the same. Calculate the new rotational shaft speed and the inlet annulus area required for the redesign. With reference to these results, comment on the potential disadvantages of the redesign relative to the 5-stage design. [25%]

Take  $\gamma = 1.333$ ,  $R = 287.2 \text{ J kg}^{-1} \text{ K}^{-1}$  and  $c_p = 1150 \text{ J kg}^{-1} \text{ K}^{-1}$  for the gas flowing through both designs of turbine.

2 (a) A single-stage compressor operates with axial flow at inlet. The inlet absolute stagnation temperature is 288 K and the inlet absolute stagnation pressure is 101 kPa. The relative flow angle at inlet to the rotor is  $-45^\circ$  and the inlet relative Mach number is 0.9. Calculate the rotor blade speed and the inlet relative stagnation pressure. [20%]

(b) The mean radius and the mass flow rate per unit annulus area are constant through the rotor. The rotor stagnation pressure loss coefficient is given by

$$Y_p = \frac{P_{01,rel} - P_{02,rel}}{P_{01,rel} - P_1} = 0.068$$

where  $P_{01,rel}$  is the inlet relative stagnation pressure,  $P_{02,rel}$  is the exit relative stagnation pressure and  $P_1$  is the inlet static pressure. If the rotor exit relative Mach number is 0.5, show that the rotor exit relative flow angle is  $-15^\circ$  and determine the static pressure ratio across the compressor rotor. [25%]

(c) Sketch the enthalpy-entropy diagram for the flow passing through the compressor rotor. Include curves of constant pressure for the static and relative stagnation conditions at inlet and exit. Show that for small values of the stagnation pressure loss coefficient, the entropy increase across the rotor can be approximated by

$$\Delta s = R \left( 1 - \frac{P_1}{P_{01,rel}} \right) Y_p$$

where  $R$  is the gas constant. [20%]

(d) Show that the absolute stagnation temperature and pressure at entry to the stator are 322 K and 145 kPa, respectively. Using the result from part (c) determine the total-to-total isentropic efficiency of the compressor stage if the stagnation pressure loss coefficient for the stator is 0.04. [35%]

In all parts of the question, take  $\gamma = 1.4$ ,  $R = 287.15 \text{ J kg}^{-1} \text{ K}^{-1}$  and  $c_p = 1005 \text{ J kg}^{-1} \text{ K}^{-1}$  for the air flowing through the compressor.

(TURN OVER)

3 (a) The Olympus 593 turbojet engine was designed for Concord to cruise at a flight Mach number of 2.0 at an altitude where the ambient static temperature is 216 K. At the design point the pressure ratio in each of the low-pressure and high-pressure compressors is 3.4 and the turbine entry temperature is 1500 K. Both compressors and both turbines have polytropic efficiencies of 0.85. The increase in mass flow caused by the addition of fuel can be neglected.

(i) Assuming that the pressure losses in the intake, the combustor and the propulsive nozzle are negligible, calculate the jet velocity and the propulsive efficiency of the engine at the design condition. [30%]

(ii) Find the pressure ratio in each compressor when the engine is tested at sea-level where the ambient temperature is 288 K and the turbine entry temperature is 1800 K. State clearly any assumptions that are required for the calculations. [30%]

Take  $\gamma = 1.4$ ,  $R = 287.15 \text{ J kg}^{-1} \text{ K}^{-1}$  and  $c_p = 1005 \text{ J kg}^{-1} \text{ K}^{-1}$  for both un-combusted air and the products of combustion.

(b) For a high speed multi-stage compressor describe how the location of initiation of aerodynamic instability (stall or surge) varies with the compressor shaft speed. Explain why this variation occurs. [20%]

(c) For a compressor having insufficient stability margin at part speed, it is possible to make it more stable by re-staggering the blades. Explain how this can be done, and discuss the potential adverse effects of this remedy on the compressor performance. [20%]

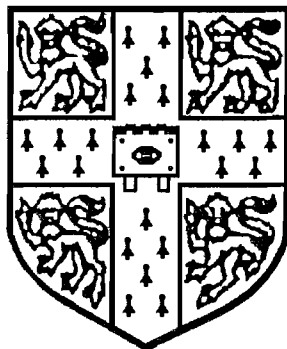
**END OF PAPER**

# Compressible Flow

# Data Book

for Part II of the  
Engineering Tripos

**2004 Edition**



Cambridge University Engineering Department

# PERFECT GAS RELATIONS FOR COMPRESSIBLE FLOW

## Ratios of stagnation to static quantities

$$\frac{T}{T_0} = \left(1 + \frac{\gamma-1}{2} M^2\right)^{-1}$$

$$\frac{p}{p_0} = \left(1 + \frac{\gamma-1}{2} M^2\right)^{-\frac{\gamma}{\gamma-1}}$$

$$\frac{\rho}{\rho_0} = \left(1 + \frac{\gamma-1}{2} M^2\right)^{-\frac{1}{\gamma-1}}$$

*Notes:*

(1)  $T_0 = \text{const.}$  in adiabatic flow with no shaft work

(2) If flow is isentropic,  $p_0 = \text{const.}$  and  $\rho_0 = \text{const.}$  when  $T_0 = \text{const.}$

## Mach number relations (see tables)

$$\frac{V}{\sqrt{c_p T_0}} = \sqrt{\gamma-1} M \left(1 + \frac{\gamma-1}{2} M^2\right)^{-\frac{1}{2}}$$

$$\frac{\dot{m} \sqrt{c_p T_0}}{A p_0} = \frac{\gamma}{\sqrt{\gamma-1}} M \left(1 + \frac{\gamma-1}{2} M^2\right)^{-\frac{1}{2} \left(\frac{\gamma+1}{\gamma-1}\right)}$$

$$\frac{\dot{m} \sqrt{c_p T_0}}{A p} = \frac{\gamma}{\sqrt{\gamma-1}} M \left(1 + \frac{\gamma-1}{2} M^2\right)^{\frac{1}{2}}$$

$$\frac{F}{\dot{m} \sqrt{c_p T_0}} = \frac{\sqrt{\gamma-1}}{\gamma} \frac{1 + \gamma M^2}{M} \left(1 + \frac{\gamma-1}{2} M^2\right)^{-\frac{1}{2}} \quad \text{where } F = (p + \rho V^2) A$$

$$\frac{\frac{1}{2} \rho V^2}{p_0} = \frac{1}{2} \gamma M^2 \left(1 + \frac{\gamma-1}{2} M^2\right)^{-\frac{\gamma}{\gamma-1}}$$

# ONE-DIMENSIONAL FLOW OF A PERFECT GAS

## Isentropic flow

$$\frac{A}{A^*} = \frac{1}{M} \left\{ \frac{2}{\gamma+1} \left( 1 + \frac{\gamma-1}{2} M^2 \right) \right\}^{\frac{1}{2} \left( \frac{\gamma+1}{\gamma-1} \right)}$$

## Adiabatic constant area flow

$$\frac{4c_f L_{\max}}{D} = \frac{1-M^2}{\gamma M^2} + \frac{\gamma+1}{2\gamma} \ln \left( \frac{(\gamma+1)M^2}{2 \left( 1 + \frac{\gamma-1}{2} M^2 \right)} \right)$$

## Normal shock waves in perfect gases

$$VV_s = a^{*2}$$

$$M_s = \left( \frac{1 + \frac{\gamma-1}{2} M^2}{\gamma M^2 - \frac{\gamma-1}{2}} \right)^{\frac{1}{2}}$$

$$\frac{p_{0s}}{p_0} = \left( \frac{\frac{\gamma+1}{2} M^2}{1 + \frac{\gamma-1}{2} M^2} \right)^{\frac{\gamma}{\gamma-1}} \left( \frac{2\gamma}{\gamma+1} M^2 - \frac{\gamma-1}{\gamma+1} \right)^{\frac{1}{1-\gamma}}$$

$$\frac{p_s}{p} = 1 + \frac{2\gamma}{\gamma+1} (M^2 - 1)$$

$$\frac{p_{0s}}{p} = \left( \frac{\gamma+1}{2} M^2 \right)^{\frac{\gamma}{\gamma-1}} \left( \frac{2\gamma}{\gamma+1} M^2 - \frac{\gamma-1}{\gamma+1} \right)^{\frac{1}{1-\gamma}}$$

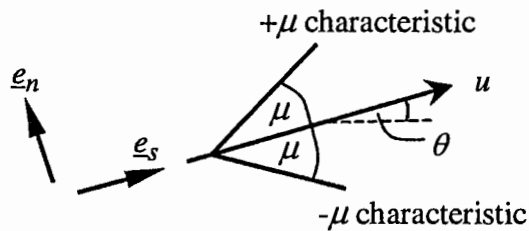
$$\frac{T_s}{T} = \frac{\gamma-1}{(\gamma+1)^2} \frac{2}{M^2} \left( 1 + \frac{\gamma-1}{2} M^2 \right) \left( \frac{2\gamma}{\gamma-1} M^2 - 1 \right)$$

$$\frac{\rho_s}{\rho} = \frac{(\gamma+1)M^2}{2 \left( 1 + \frac{\gamma-1}{2} M^2 \right)}$$

# TWO DIMENSIONAL SUPERSONIC FLOW

## Method of Characteristics for 2-D supersonic flow

Applicable to adiabatic ( $h_0 = \text{constant}$ ), isentropic flow



Mach Number

$$M = u/c$$

Mach angle

$$\mu = \sin^{-1}\left(\frac{1}{M}\right)$$

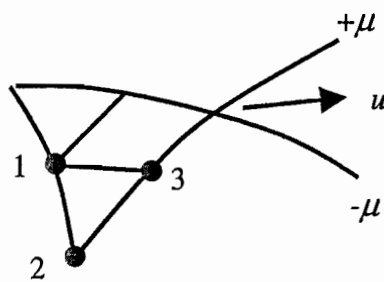
Prandtl-Meyer function

$$v = \int_1^M \sqrt{M^2 - 1} \frac{du}{u}$$

$$v = \sqrt{\frac{\gamma+1}{\gamma-1}} \tan^{-1} \sqrt{\frac{\gamma-1}{\gamma+1} (M^2 - 1)} - \tan^{-1} \sqrt{M^2 - 1} \quad \text{for a perfect gas}$$

## Calculations

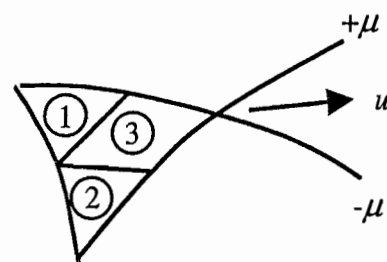
Lattice Method



$$v_3 - \theta_3 = v_2 - \theta_2 \quad \text{along } +\mu$$

$$v_3 + \theta_3 = v_1 + \theta_1 \quad \text{along } -\mu$$

Field (or wave) method

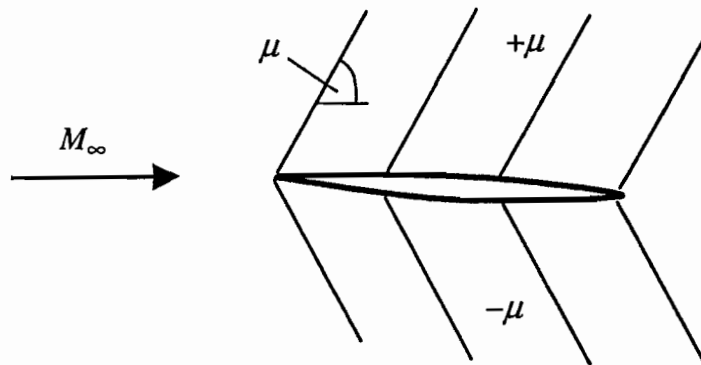


$$v_3 + \theta_3 = v_1 + \theta_1 \quad \text{across } +\mu$$

$$v_3 - \theta_3 = v_2 - \theta_2 \quad \text{across } -\mu$$



### Linearised Method of Characteristics (thin film theory)

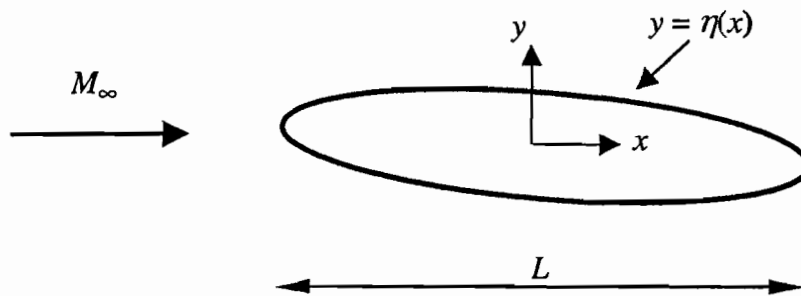


$$\mu \approx \sin^{-1}(1/M_{\infty})$$

$$\Delta p \approx \pm \frac{\rho_{\infty} u_{\infty}^2 \Delta \theta}{\sqrt{M_{\infty}^2 - 1}} \quad \text{across } \pm \mu \text{ waves}$$

$$\text{Pressure coefficient } c_p = \frac{p - p_{\infty}}{\frac{1}{2} \rho_{\infty} u_{\infty}^2} = \pm \frac{2\theta}{\sqrt{M_{\infty}^2 - 1}} \quad \text{on upper/lower surface}$$

### Prandtl-Glauert rule for linearised potential flow past geometrically similar bodies



$$\text{Pressure coefficient } c_p = \frac{p - p_{\infty}}{\frac{1}{2} \rho_{\infty} u_{\infty}^2}$$

For geometrically similar bodies with  $\frac{\eta}{L} = f\left(\frac{x}{L}\right)$  and  $c_p(M_{\infty} = 0) = c_{p0}$ ,

$$c_p = \frac{c_{p0}}{\sqrt{1 - M_{\infty}^2}} \quad \text{in subsonic flow}$$

$$c_p \propto \frac{1}{\sqrt{M_{\infty}^2 - 1}} \quad \text{in supersonic flow}$$

## Oblique Shock Relations (see tables)

$$\frac{p_2}{p_1} = 1 + \frac{2\gamma}{\gamma+1} (M_1^2 \sin^2 \beta - 1)$$

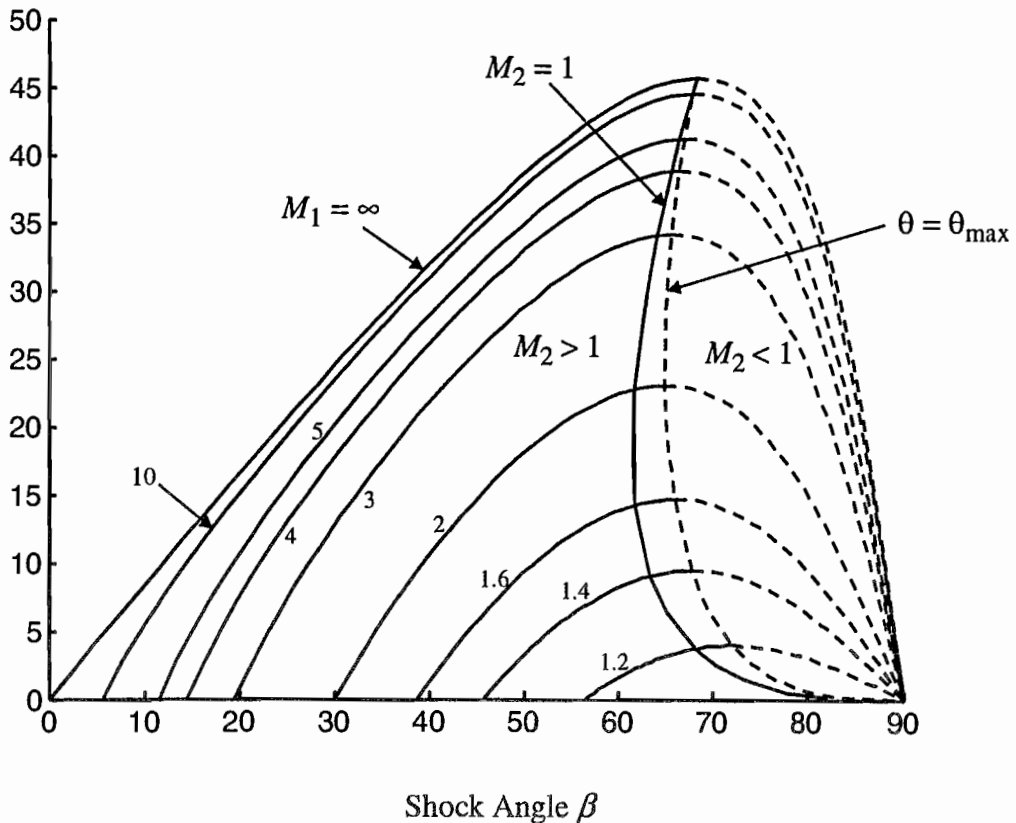
$$\frac{T_2}{T_1} = \frac{\gamma-1}{(\gamma+1)^2} \frac{2}{M_1^2 \sin^2 \beta} \left( 1 + \frac{\gamma-1}{2} M_1^2 \sin^2 \beta \right) \left( \frac{2\gamma}{\gamma-1} M_1^2 \sin^2 \beta - 1 \right)$$

$$\frac{\rho_2}{\rho_1} = \frac{(\gamma+1)M_1^2 \sin^2 \beta}{2 \left[ 1 + \frac{\gamma-1}{2} M_1^2 \sin^2 \beta \right]}$$

$$M_2 \sin(\beta - \theta) = \left[ \frac{1 + \frac{\gamma-1}{2} M_1^2 \sin^2 \beta}{\gamma M_1^2 \sin^2 \beta - \frac{\gamma-1}{2}} \right]^{\frac{1}{2}}$$

$$\frac{p_{02}}{p_{01}} = \left( \frac{\frac{\gamma+1}{2} M_1^2 \sin^2 \beta}{1 + \frac{\gamma-1}{2} M_1^2 \sin^2 \beta} \right)^{\frac{\gamma}{\gamma-1}} \left( \frac{2\gamma}{\gamma+1} M_1^2 \sin^2 \beta - \frac{\gamma-1}{\gamma+1} \right)^{\frac{1}{1-\gamma}}$$

$$\tan \theta = \frac{2 \cot \beta (M_1^2 \sin^2 \beta - 1)}{(\gamma+1)M_1^2 - 2(M_1^2 \sin^2 \beta - 1)}$$



# GAS FLOW TABLES ( $\gamma=1.400$ ): SUBSONIC FLOW

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m}\sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m}\sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m}\sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{1}{2}\rho V^2$ $p_0$
0.010	1.0000	0.9999	1.0000	0.0063	0.0221	0.0221	45.1813	7134.405	0.0001
0.020	0.9999	0.9997	0.9998	0.0126	0.0443	0.0443	22.5994	1778.450	0.0003
0.030	0.9998	0.9994	0.9996	0.0190	0.0664	0.0664	15.0761	787.0814	0.0006
0.040	0.9997	0.9989	0.9992	0.0253	0.0885	0.0886	11.3173	440.3522	0.0011
0.050	0.9995	0.9983	0.9988	0.0316	0.1105	0.1107	9.0644	280.0203	0.0017
0.060	0.9993	0.9975	0.9982	0.0379	0.1325	0.1329	7.5645	193.0311	0.0025
0.070	0.9990	0.9966	0.9976	0.0443	0.1545	0.1550	6.4947	140.6550	0.0034
0.080	0.9987	0.9955	0.9968	0.0506	0.1764	0.1772	5.6939	106.7182	0.0045
0.090	0.9984	0.9944	0.9960	0.0569	0.1983	0.1994	5.0723	83.4961	0.0056
0.100	0.9980	0.9930	0.9950	0.0632	0.2200	0.2216	4.5762	66.9216	0.0070
0.110	0.9976	0.9916	0.9940	0.0695	0.2417	0.2438	4.1714	54.6879	0.0084
0.120	0.9971	0.9900	0.9928	0.0758	0.2633	0.2660	3.8350	45.4080	0.0100
0.130	0.9966	0.9883	0.9916	0.0821	0.2849	0.2883	3.5513	38.2070	0.0117
0.140	0.9961	0.9864	0.9903	0.0884	0.3063	0.3105	3.3089	32.5113	0.0135
0.150	0.9955	0.9844	0.9888	0.0947	0.3276	0.3328	3.0996	27.9320	0.0155
0.160	0.9949	0.9823	0.9873	0.1009	0.3488	0.3551	2.9172	24.1978	0.0176
0.170	0.9943	0.9800	0.9857	0.1072	0.3699	0.3774	2.7569	21.1152	0.0198
0.180	0.9936	0.9776	0.9840	0.1135	0.3908	0.3997	2.6151	18.5427	0.0222
0.190	0.9928	0.9751	0.9822	0.1197	0.4116	0.4221	2.4889	16.3752	0.0246
0.200	0.9921	0.9725	0.9803	0.1260	0.4323	0.4445	2.3758	14.5333	0.0272
0.210	0.9913	0.9697	0.9783	0.1322	0.4528	0.4669	2.2740	12.9560	0.0299
0.220	0.9904	0.9668	0.9762	0.1385	0.4731	0.4893	2.1820	11.5961	0.0328
0.230	0.9895	0.9638	0.9740	0.1447	0.4933	0.5118	2.0985	10.4161	0.0357
0.240	0.9886	0.9607	0.9718	0.1509	0.5133	0.5343	2.0225	9.3865	0.0387
0.250	0.9877	0.9575	0.9694	0.1571	0.5332	0.5568	1.9530	8.4834	0.0419
0.260	0.9867	0.9541	0.9670	0.1633	0.5528	0.5794	1.8892	7.6876	0.0451
0.270	0.9856	0.9506	0.9645	0.1695	0.5723	0.6020	1.8306	6.9832	0.0485
0.280	0.9846	0.9470	0.9619	0.1757	0.5915	0.6246	1.7766	6.3572	0.0520
0.290	0.9835	0.9433	0.9592	0.1819	0.6106	0.6473	1.7267	5.7989	0.0555
0.300	0.9823	0.9395	0.9564	0.1881	0.6295	0.6700	1.6805	5.2993	0.0592
0.310	0.9811	0.9355	0.9535	0.1942	0.6481	0.6928	1.6377	4.8507	0.0629
0.320	0.9799	0.9315	0.9506	0.2003	0.6666	0.7156	1.5978	4.4467	0.0668
0.330	0.9787	0.9274	0.9476	0.2065	0.6848	0.7384	1.5608	4.0821	0.0707
0.340	0.9774	0.9231	0.9445	0.2126	0.7027	0.7613	1.5262	3.7520	0.0747
0.350	0.9761	0.9188	0.9413	0.2187	0.7205	0.7842	1.4939	3.4525	0.0788
0.360	0.9747	0.9143	0.9380	0.2248	0.7380	0.8072	1.4637	3.1801	0.0829
0.370	0.9733	0.9098	0.9347	0.2309	0.7553	0.8302	1.4354	2.9320	0.0872
0.380	0.9719	0.9052	0.9313	0.2369	0.7723	0.8532	1.4090	2.7054	0.0915
0.390	0.9705	0.9004	0.9278	0.2430	0.7891	0.8763	1.3841	2.4983	0.0959
0.400	0.9690	0.8956	0.9243	0.2490	0.8056	0.8995	1.3608	2.3085	0.1003
0.410	0.9675	0.8907	0.9207	0.2551	0.8219	0.9227	1.3388	2.1344	0.1048
0.420	0.9659	0.8857	0.9170	0.2611	0.8379	0.9460	1.3182	1.9744	0.1094
0.430	0.9643	0.8807	0.9132	0.2671	0.8536	0.9693	1.2988	1.8272	0.1140
0.440	0.9627	0.8755	0.9094	0.2730	0.8691	0.9927	1.2804	1.6915	0.1186
0.450	0.9611	0.8703	0.9055	0.2790	0.8843	1.0161	1.2632	1.5664	0.1234
0.460	0.9594	0.8650	0.9016	0.2850	0.8992	1.0396	1.2469	1.4509	0.1281
0.470	0.9577	0.8596	0.8976	0.2909	0.9138	1.0631	1.2315	1.3441	0.1329
0.480	0.9559	0.8541	0.8935	0.2968	0.9282	1.0867	1.2170	1.2453	0.1378
0.490	0.9542	0.8486	0.8894	0.3027	0.9423	1.1104	1.2033	1.1539	0.1426
0.500	0.9524	0.8430	0.8852	0.3086	0.9561	1.1341	1.1903	1.0691	0.1475

$$\gamma=1.400$$

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m}\sqrt{c_p T_0}}{A p_0}$	$\frac{\dot{m}\sqrt{c_p T_0}}{A p}$	$\frac{F}{\dot{m}\sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{1}{2}\frac{\rho V^2}{p_0}$
0.510	0.9506	0.8374	0.8809	0.3145	0.9696	1.1579	1.1781	0.9904	0.1525
0.520	0.9487	0.8317	0.8766	0.3203	0.9828	1.1818	1.1665	0.9174	0.1574
0.530	0.9468	0.8259	0.8723	0.3262	0.9958	1.2057	1.1556	0.8496	0.1624
0.540	0.9449	0.8201	0.8679	0.3320	1.0084	1.2297	1.1452	0.7866	0.1674
0.550	0.9430	0.8142	0.8634	0.3378	1.0208	1.2538	1.1354	0.7281	0.1724
0.560	0.9410	0.8082	0.8589	0.3436	1.0328	1.2779	1.1261	0.6736	0.1774
0.570	0.9390	0.8022	0.8544	0.3493	1.0446	1.3021	1.1173	0.6229	0.1825
0.580	0.9370	0.7962	0.8498	0.3551	1.0561	1.3264	1.1090	0.5757	0.1875
0.590	0.9349	0.7901	0.8451	0.3608	1.0672	1.3507	1.1011	0.5317	0.1925
0.600	0.9328	0.7840	0.8405	0.3665	1.0781	1.3751	1.0937	0.4908	0.1976
0.610	0.9307	0.7778	0.8357	0.3722	1.0887	1.3996	1.0867	0.4527	0.2026
0.620	0.9286	0.7716	0.8310	0.3779	1.0990	1.4242	1.0800	0.4172	0.2076
0.630	0.9265	0.7654	0.8262	0.3835	1.1090	1.4489	1.0737	0.3841	0.2127
0.640	0.9243	0.7591	0.8213	0.3891	1.1186	1.4736	1.0678	0.3533	0.2177
0.650	0.9221	0.7528	0.8164	0.3948	1.1280	1.4984	1.0621	0.3246	0.2226
0.660	0.9199	0.7465	0.8115	0.4003	1.1371	1.5233	1.0568	0.2979	0.2276
0.670	0.9176	0.7401	0.8066	0.4059	1.1459	1.5483	1.0518	0.2730	0.2326
0.680	0.9153	0.7338	0.8016	0.4115	1.1544	1.5733	1.0471	0.2498	0.2375
0.690	0.9131	0.7274	0.7966	0.4170	1.1626	1.5984	1.0426	0.2282	0.2424
0.700	0.9107	0.7209	0.7916	0.4225	1.1705	1.6237	1.0384	0.2081	0.2473
0.710	0.9084	0.7145	0.7865	0.4280	1.1782	1.6490	1.0344	0.1895	0.2521
0.720	0.9061	0.7080	0.7814	0.4335	1.1855	1.6744	1.0307	0.1721	0.2569
0.730	0.9037	0.7016	0.7763	0.4389	1.1925	1.6999	1.0272	0.1561	0.2617
0.740	0.9013	0.6951	0.7712	0.4443	1.1993	1.7254	1.0239	0.1411	0.2664
0.750	0.8989	0.6886	0.7660	0.4497	1.2058	1.7511	1.0208	0.1273	0.2711
0.760	0.8964	0.6821	0.7609	0.4551	1.2119	1.7768	1.0179	0.1145	0.2758
0.770	0.8940	0.6756	0.7557	0.4605	1.2178	1.8027	1.0152	0.1026	0.2804
0.780	0.8915	0.6691	0.7505	0.4658	1.2234	1.8286	1.0126	0.0917	0.2849
0.790	0.8890	0.6625	0.7452	0.4711	1.2288	1.8547	1.0103	0.0816	0.2894
0.800	0.8865	0.6560	0.7400	0.4764	1.2338	1.8808	1.0081	0.0723	0.2939
0.810	0.8840	0.6495	0.7347	0.4817	1.2386	1.9070	1.0060	0.0638	0.2983
0.820	0.8815	0.6430	0.7295	0.4869	1.2431	1.9333	1.0041	0.0559	0.3026
0.830	0.8789	0.6365	0.7242	0.4921	1.2474	1.9598	1.0024	0.0488	0.3069
0.840	0.8763	0.6300	0.7189	0.4973	1.2514	1.9863	1.0008	0.0423	0.3112
0.850	0.8737	0.6235	0.7136	0.5025	1.2551	2.0129	0.9993	0.0363	0.3153
0.860	0.8711	0.6170	0.7083	0.5077	1.2585	2.0396	0.9979	0.0310	0.3195
0.870	0.8685	0.6106	0.7030	0.5128	1.2617	2.0665	0.9967	0.0261	0.3235
0.880	0.8659	0.6041	0.6977	0.5179	1.2646	2.0934	0.9956	0.0218	0.3275
0.890	0.8632	0.5977	0.6924	0.5230	1.2673	2.1204	0.9946	0.0179	0.3314
0.900	0.8606	0.5913	0.6870	0.5280	1.2698	2.1476	0.9937	0.0145	0.3352
0.910	0.8579	0.5849	0.6817	0.5331	1.2719	2.1748	0.9929	0.0115	0.3390
0.920	0.8552	0.5785	0.6764	0.5381	1.2739	2.2021	0.9922	0.0089	0.3427
0.930	0.8525	0.5721	0.6711	0.5431	1.2756	2.2296	0.9916	0.0067	0.3464
0.940	0.8498	0.5658	0.6658	0.5481	1.2770	2.2572	0.9911	0.0048	0.3499
0.950	0.8471	0.5595	0.6604	0.5530	1.2783	2.2848	0.9907	0.0033	0.3534
0.960	0.8444	0.5532	0.6551	0.5579	1.2793	2.3126	0.9903	0.0021	0.3569
0.970	0.8416	0.5469	0.6498	0.5628	1.2800	2.3405	0.9901	0.0011	0.3602
0.980	0.8389	0.5407	0.6445	0.5677	1.2806	2.3685	0.9899	0.0005	0.3635
0.990	0.8361	0.5345	0.6392	0.5725	1.2809	2.3966	0.9898	0.0001	0.3667
1.000	0.8333	0.5283	0.6339	0.5774	1.2810	2.4249	0.9897	0.0000	0.3698

# GAS FLOW TABLES ( $\gamma=1.400$ ): SUPERSONIC FLOW

$M$	$\frac{T}{T_0}$	$\frac{P}{P_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p}$	$F$	$\frac{4c_f L_{\max}}{D}$	$\frac{1}{2} \rho V^2$	$M_s$	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	$V$	$M$
1.010	0.8306	0.5221	0.6287	0.5821	1.2809	2.4532	0.9898	0.0001	0.3728	0.9901	1.0000	1.0235	1.9152	1.0066	0.04	1.010
1.020	0.8278	0.5160	0.6234	0.5869	1.2806	2.4817	0.9899	0.0005	0.3758	0.9805	1.0000	1.0471	1.9379	1.0132	0.13	1.020
1.030	0.8250	0.5099	0.6181	0.5917	1.2801	2.5103	0.9900	0.0010	0.3787	0.9712	1.0000	1.0711	1.9610	1.0198	0.23	1.030
1.040	0.8222	0.5039	0.6129	0.5964	1.2793	2.5390	0.9903	0.0018	0.3815	0.9620	0.9999	1.0952	1.9844	1.0263	0.35	1.040
1.050	0.8193	0.4979	0.6077	0.6011	1.2784	2.5678	0.9905	0.0027	0.3842	0.9531	0.9999	1.1196	2.0083	1.0328	0.49	1.050
1.060	0.8165	0.4919	0.6024	0.6058	1.2773	2.5967	0.9909	0.0038	0.3869	0.9444	0.9998	1.1442	2.0325	1.0393	0.64	1.060
1.070	0.8137	0.4860	0.5972	0.6104	1.2760	2.6258	0.9913	0.0051	0.3895	0.9360	0.9996	1.1691	2.0570	1.0458	0.80	1.070
1.080	0.8108	0.4800	0.5920	0.6151	1.2745	2.6549	0.9917	0.0066	0.3919	0.9277	0.9994	1.1941	2.0819	1.0522	0.97	1.080
1.090	0.8080	0.4742	0.5869	0.6197	1.2728	2.6842	0.9922	0.0082	0.3944	0.9196	0.9992	1.2195	2.1072	1.0586	1.15	1.090
1.100	0.8052	0.4684	0.5817	0.6243	1.2709	2.7136	0.9928	0.0099	0.3967	0.9118	0.9989	1.2450	2.1328	1.0649	1.34	1.100
1.110	0.8023	0.4626	0.5766	0.6288	1.2689	2.7432	0.9934	0.0118	0.3990	0.9041	0.9986	1.2708	2.1588	1.0713	1.53	1.110
1.120	0.7994	0.4568	0.5714	0.6333	1.2667	2.7728	0.9940	0.0138	0.4011	0.8966	0.9982	1.2968	2.1851	1.0776	1.74	1.120
1.130	0.7966	0.4511	0.5663	0.6379	1.2643	2.8026	0.9947	0.0159	0.4032	0.8892	0.9978	1.3231	2.2118	1.0840	1.94	1.130
1.140	0.7937	0.4455	0.5612	0.6423	1.2618	2.8325	0.9954	0.0182	0.4052	0.8820	0.9973	1.3495	2.2388	1.0903	2.16	1.140
1.150	0.7908	0.4398	0.5562	0.6468	1.2590	2.8626	0.9961	0.0205	0.4072	0.8750	0.9967	1.3763	2.2661	1.0966	2.38	1.150
1.160	0.7879	0.4343	0.5511	0.6512	1.2562	2.8927	0.9969	0.0230	0.4090	0.8682	0.9961	1.4032	2.2937	1.1029	2.61	1.160
1.170	0.7851	0.4287	0.5461	0.6556	1.2531	2.9230	0.9978	0.0255	0.4108	0.8615	0.9953	1.4304	2.3217	1.1092	2.84	1.170
1.180	0.7822	0.4232	0.5411	0.6600	1.2500	2.9534	0.9986	0.0281	0.4125	0.8549	0.9946	1.4578	2.3500	1.1154	3.07	1.180
1.190	0.7793	0.4178	0.5361	0.6644	1.2466	2.9840	0.9995	0.0309	0.4141	0.8485	0.9937	1.4855	2.3786	1.1217	3.31	1.190
1.200	0.7764	0.4124	0.5311	0.6687	1.2432	3.0147	1.0004	0.0336	0.4157	0.8422	0.9928	1.5133	2.4075	1.1280	3.56	1.200
1.210	0.7735	0.4070	0.5262	0.6730	1.2396	3.0455	1.0014	0.0365	0.4171	0.8360	0.9918	1.5415	2.4367	1.1343	3.81	1.210
1.220	0.7706	0.4017	0.5213	0.6773	1.2358	3.0764	1.0024	0.0394	0.4185	0.8300	0.9907	1.5698	2.4663	1.1405	4.06	1.220
1.230	0.7677	0.3964	0.5164	0.6816	1.2319	3.1075	1.0034	0.0424	0.4198	0.8241	0.9896	1.5984	2.4961	1.1468	4.31	1.230
1.240	0.7648	0.3912	0.5115	0.6858	1.2279	3.1387	1.0045	0.0455	0.4211	0.8183	0.9884	1.6272	2.5263	1.1531	4.57	1.240
1.250	0.7619	0.3861	0.5067	0.6901	1.2238	3.1700	1.0055	0.0486	0.4223	0.8126	0.9871	1.6563	2.5568	1.1594	4.83	1.250

$\gamma=1.400$

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{m\sqrt{c_p T_0}}{A p_0}$	$\frac{m\sqrt{c_p T_0}}{A p}$	$F$	$\frac{4c_f L_{\max}}{D}$	$\frac{1}{2}\rho V^2$	$M_s$	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	$v$	$M$
1.260	0.7590	0.3809	0.5019	0.6943	1.2195	3.2015	1.0066	0.0517	0.4233	0.8071	0.9857	1.6855	2.5875	1.1657	5.09	1.260
1.270	0.7561	0.3759	0.4971	0.6984	1.2152	3.2331	1.0077	0.0549	0.4244	0.8016	0.9842	1.7151	2.6186	1.1720	5.36	1.270
1.280	0.7532	0.3708	0.4923	0.7026	1.2107	3.2648	1.0089	0.0582	0.4253	0.7963	0.9827	1.7448	2.6500	1.1783	5.63	1.280
1.290	0.7503	0.3658	0.4876	0.7067	1.2061	3.2967	1.0100	0.0615	0.4262	0.7911	0.9811	1.7748	2.6816	1.1846	5.90	1.290
1.300	0.7474	0.3609	0.4829	0.7108	1.2014	3.3287	1.0112	0.0648	0.4270	0.7860	0.9794	1.8050	2.7136	1.1909	6.17	1.300
1.310	0.7445	0.3560	0.4782	0.7149	1.1965	3.3608	1.0124	0.0682	0.4277	0.7809	0.9776	1.8355	2.7459	1.1972	6.44	1.310
1.320	0.7416	0.3512	0.4736	0.7189	1.1916	3.3931	1.0136	0.0716	0.4283	0.7760	0.9758	1.8661	2.7784	1.2035	6.72	1.320
1.330	0.7387	0.3464	0.4690	0.7229	1.1866	3.4255	1.0149	0.0750	0.4289	0.7712	0.9738	1.8971	2.8112	1.2099	7.00	1.330
1.340	0.7358	0.3417	0.4644	0.7270	1.1815	3.4581	1.0161	0.0785	0.4294	0.7664	0.9718	1.9282	2.8444	1.2162	7.28	1.340
1.350	0.7329	0.3370	0.4598	0.7309	1.1763	3.4907	1.0174	0.0820	0.4299	0.7618	0.9697	1.9596	2.8778	1.2226	7.56	1.350
1.360	0.7300	0.3323	0.4553	0.7349	1.1710	3.5236	1.0187	0.0855	0.4303	0.7572	0.9676	1.9912	2.9115	1.2290	7.84	1.360
1.370	0.7271	0.3277	0.4508	0.7388	1.1656	3.5566	1.0200	0.0890	0.4306	0.7527	0.9653	2.0231	2.9455	1.2354	8.13	1.370
1.380	0.7242	0.3232	0.4463	0.7427	1.1601	3.5897	1.0213	0.0926	0.4308	0.7483	0.9630	2.0551	2.9798	1.2418	8.41	1.380
1.390	0.7213	0.3187	0.4418	0.7466	1.1546	3.6229	1.0226	0.0962	0.4310	0.7440	0.9607	2.0875	3.0144	1.2482	8.70	1.390
1.400	0.7184	0.3142	0.4374	0.7505	1.1490	3.6563	1.0240	0.0997	0.4311	0.7397	0.9582	2.1200	3.0492	1.2547	8.99	1.400
1.410	0.7155	0.3098	0.4330	0.7543	1.1433	3.6899	1.0253	0.1033	0.4312	0.7355	0.9557	2.1528	3.0844	1.2612	9.28	1.410
1.420	0.7126	0.3055	0.4287	0.7581	1.1375	3.7236	1.0267	0.1069	0.4312	0.7314	0.9531	2.1858	3.1198	1.2676	9.57	1.420
1.430	0.7097	0.3012	0.4244	0.7619	1.1317	3.7574	1.0281	0.1106	0.4311	0.7274	0.9504	2.2191	3.1555	1.2741	9.86	1.430
1.440	0.7069	0.2969	0.4201	0.7657	1.1258	3.7914	1.0295	0.1142	0.4310	0.7235	0.9476	2.2525	3.1915	1.2807	10.15	1.440
1.450	0.7040	0.2927	0.4158	0.7694	1.1198	3.8255	1.0308	0.1178	0.4308	0.7196	0.9448	2.2863	3.2278	1.2872	10.44	1.450
1.460	0.7011	0.2886	0.4116	0.7732	1.1138	3.8598	1.0323	0.1215	0.4306	0.7157	0.9420	2.3202	3.2643	1.2938	10.73	1.460
1.470	0.6982	0.2845	0.4074	0.7769	1.1077	3.8942	1.0337	0.1251	0.4303	0.7120	0.9390	2.3544	3.3011	1.3003	11.02	1.470
1.480	0.6954	0.2804	0.4032	0.7805	1.1016	3.9287	1.0351	0.1288	0.4299	0.7083	0.9360	2.3888	3.3382	1.3069	11.32	1.480
1.490	0.6925	0.2764	0.3991	0.7842	1.0954	3.9634	1.0365	0.1324	0.4295	0.7047	0.9329	2.4235	3.3756	1.3136	11.61	1.490
1.500	0.6897	0.2724	0.3950	0.7878	1.0891	3.9983	1.0379	0.1361	0.4290	0.7011	0.9298	2.4583	3.4133	1.3202	11.91	1.500

$\gamma=1.400$

$M$	$\frac{T}{T_0}$	$\frac{P}{P_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{m\sqrt{c_p T_0}}{A P_0}$	$\frac{m\sqrt{c_p T_0}}{A P}$	$F$	$\frac{4c_f L_{\max}}{D}$	$\frac{1}{2}\rho V^2$	$M_s$	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	$V$	$M$
1.510	0.6868	0.2685	0.3909	0.7914	1.0829	4.0333	1.0394	0.1397	0.4285	0.6976	0.9266	2.4935	3.4512	1.3269	12.20	1.510
1.520	0.6840	0.2646	0.3869	0.7950	1.0765	4.0684	1.0408	0.1433	0.4279	0.6941	0.9233	2.5288	3.4894	1.3336	12.49	1.520
1.530	0.6811	0.2608	0.3829	0.7986	1.0702	4.1037	1.0423	0.1470	0.4273	0.6907	0.9200	2.5644	3.5279	1.3403	12.79	1.530
1.540	0.6783	0.2570	0.3789	0.8021	1.0638	4.1392	1.0437	0.1506	0.4266	0.6874	0.9166	2.6002	3.5667	1.3470	13.09	1.540
1.550	0.6754	0.2533	0.3750	0.8057	1.0573	4.1748	1.0452	0.1543	0.4259	0.6841	0.9132	2.6363	3.6057	1.3538	13.38	1.550
1.560	0.6726	0.2496	0.3710	0.8092	1.0508	4.2105	1.0467	0.1579	0.4252	0.6809	0.9097	2.6725	3.6450	1.3606	13.68	1.560
1.570	0.6698	0.2459	0.3672	0.8126	1.0443	4.2464	1.0481	0.1615	0.4243	0.6777	0.9062	2.7091	3.6846	1.3674	13.97	1.570
1.580	0.6670	0.2423	0.3633	0.8161	1.0378	4.2825	1.0496	0.1651	0.4235	0.6746	0.9026	2.7458	3.7244	1.3742	14.27	1.580
1.590	0.6642	0.2388	0.3595	0.8195	1.0312	4.3187	1.0511	0.1688	0.4226	0.6715	0.8989	2.7828	3.7646	1.3811	14.56	1.590
1.600	0.6614	0.2353	0.3557	0.8230	1.0246	4.3551	1.0526	0.1724	0.4216	0.6684	0.8952	2.8200	3.8050	1.3880	14.86	1.600
1.610	0.6586	0.2318	0.3520	0.8263	1.0180	4.3916	1.0541	0.1760	0.4206	0.6655	0.8915	2.8575	3.8456	1.3949	15.16	1.610
1.620	0.6558	0.2284	0.3483	0.8297	1.0114	4.4282	1.0555	0.1795	0.4196	0.6625	0.8877	2.8951	3.8866	1.4018	15.45	1.620
1.630	0.6530	0.2250	0.3446	0.8331	1.0047	4.4651	1.0570	0.1831	0.4185	0.6596	0.8838	2.9331	3.9278	1.4088	15.75	1.630
1.640	0.6502	0.2217	0.3409	0.8364	0.9980	4.5020	1.0585	0.1867	0.4174	0.6568	0.8799	2.9712	3.9693	1.4158	16.04	1.640
1.650	0.6475	0.2184	0.3373	0.8397	0.9913	4.5392	1.0600	0.1902	0.4162	0.6540	0.8760	3.0096	4.0110	1.4228	16.34	1.650
1.660	0.6447	0.2151	0.3337	0.8430	0.9846	4.5765	1.0615	0.1938	0.4150	0.6512	0.8720	3.0482	4.0531	1.4299	16.63	1.660
1.670	0.6419	0.2119	0.3302	0.8462	0.9779	4.6139	1.0630	0.1973	0.4138	0.6485	0.8680	3.0871	4.0953	1.4369	16.93	1.670
1.680	0.6392	0.2088	0.3266	0.8495	0.9712	4.6515	1.0645	0.2008	0.4125	0.6458	0.8639	3.1261	4.1379	1.4440	17.22	1.680
1.690	0.6364	0.2057	0.3232	0.8527	0.9644	4.6892	1.0660	0.2043	0.4112	0.6431	0.8599	3.1655	4.1807	1.4512	17.52	1.690
1.700	0.6337	0.2026	0.3197	0.8559	0.9577	4.7272	1.0674	0.2078	0.4098	0.6405	0.8557	3.2050	4.2238	1.4583	17.81	1.700
1.710	0.6310	0.1996	0.3163	0.8591	0.9509	4.7652	1.0689	0.2113	0.4085	0.6380	0.8516	3.2448	4.2672	1.4655	18.10	1.710
1.720	0.6283	0.1966	0.3129	0.8622	0.9442	4.8035	1.0704	0.2147	0.4071	0.6355	0.8474	3.2848	4.3108	1.4727	18.40	1.720
1.730	0.6256	0.1936	0.3095	0.8654	0.9374	4.8418	1.0719	0.2182	0.4056	0.6330	0.8431	3.3251	4.3547	1.4800	18.69	1.730
1.740	0.6229	0.1907	0.3062	0.8685	0.9307	4.8804	1.0734	0.2216	0.4041	0.6305	0.8389	3.3655	4.3989	1.4873	18.98	1.740
1.750	0.6202	0.1878	0.3029	0.8716	0.9239	4.9191	1.0749	0.2250	0.4026	0.6281	0.8346	3.4063	4.4433	1.4946	19.27	1.750

$\gamma=1.400$

$M$	$\frac{T}{T_0}$	$\frac{P}{P_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{m\sqrt{c_p T_0}}{A P_0}$	$\frac{m\sqrt{c_p T_0}}{A P}$	$F$	$\frac{4c_f L_{\max}}{D}$	$\frac{1}{2}\rho V^2$	$M_s$	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	$V$	$M$
1.760	0.6175	0.1850	0.2996	0.8747	0.9172	4.9580	1.0764	0.2284	0.4011	0.6257	0.8302	3.4472	4.4880	1.5019	19.56	1.760
1.770	0.6148	0.1822	0.2964	0.8777	0.9104	4.9970	1.0779	0.2318	0.3996	0.6234	0.8259	3.4884	4.5330	1.5093	19.86	1.770
1.780	0.6121	0.1794	0.2931	0.8808	0.9037	5.0362	1.0793	0.2352	0.3980	0.6210	0.8215	3.5298	4.5782	1.5167	20.15	1.780
1.790	0.6095	0.1767	0.2900	0.8838	0.8970	5.0755	1.0808	0.2385	0.3964	0.6188	0.8171	3.5715	4.6237	1.5241	20.44	1.790
1.800	0.6068	0.1740	0.2868	0.8868	0.8902	5.1150	1.0823	0.2419	0.3947	0.6165	0.8127	3.6133	4.6695	1.5316	20.73	1.800
1.810	0.6041	0.1714	0.2837	0.8898	0.8835	5.1547	1.0838	0.2452	0.3931	0.6143	0.8082	3.6555	4.7155	1.5391	21.01	1.810
1.820	0.6015	0.1688	0.2806	0.8927	0.8768	5.1945	1.0852	0.2485	0.3914	0.6121	0.8038	3.6978	4.7618	1.5466	21.30	1.820
1.830	0.5989	0.1662	0.2776	0.8957	0.8701	5.2345	1.0867	0.2518	0.3897	0.6099	0.7993	3.7404	4.8084	1.5541	21.59	1.830
1.840	0.5963	0.1637	0.2745	0.8986	0.8634	5.2747	1.0882	0.2551	0.3879	0.6078	0.7948	3.7832	4.8552	1.5617	21.88	1.840
1.850	0.5936	0.1612	0.2715	0.9015	0.8568	5.3150	1.0896	0.2583	0.3862	0.6057	0.7902	3.8263	4.9023	1.5693	22.16	1.850
1.860	0.5910	0.1587	0.2686	0.9044	0.8501	5.3555	1.0911	0.2616	0.3844	0.6036	0.7857	3.8695	4.9497	1.5770	22.45	1.860
1.870	0.5884	0.1563	0.2656	0.9072	0.8435	5.3962	1.0926	0.2648	0.3826	0.6016	0.7811	3.9131	4.9973	1.5847	22.73	1.870
1.880	0.5859	0.1539	0.2627	0.9101	0.8368	5.4370	1.0940	0.2680	0.3808	0.5996	0.7765	3.9568	5.0452	1.5924	23.02	1.880
1.890	0.5833	0.1516	0.2598	0.9129	0.8302	5.4780	1.0955	0.2712	0.3790	0.5976	0.7720	4.0008	5.0934	1.6001	23.30	1.890
1.900	0.5807	0.1492	0.2570	0.9157	0.8237	5.5191	1.0969	0.2743	0.3771	0.5956	0.7674	4.0450	5.1418	1.6079	23.59	1.900
1.910	0.5782	0.1470	0.2542	0.9185	0.8171	5.5604	1.0984	0.2775	0.3753	0.5937	0.7627	4.0895	5.1905	1.6157	23.87	1.910
1.920	0.5756	0.1447	0.2514	0.9213	0.8106	5.6019	1.0998	0.2806	0.3734	0.5918	0.7581	4.1341	5.2394	1.6236	24.15	1.920
1.930	0.5731	0.1425	0.2486	0.9240	0.8041	5.6435	1.1012	0.2837	0.3715	0.5899	0.7535	4.1791	5.2886	1.6314	24.43	1.930
1.940	0.5705	0.1403	0.2459	0.9268	0.7976	5.6853	1.1027	0.2868	0.3696	0.5880	0.7488	4.2242	5.3381	1.6394	24.71	1.940
1.950	0.5680	0.1381	0.2432	0.9295	0.7911	5.7273	1.1041	0.2899	0.3677	0.5862	0.7442	4.2696	5.3878	1.6473	24.99	1.950
1.960	0.5655	0.1360	0.2405	0.9322	0.7846	5.7695	1.1055	0.2929	0.3657	0.5844	0.7395	4.3152	5.4378	1.6553	25.27	1.960
1.970	0.5630	0.1339	0.2378	0.9349	0.7782	5.8118	1.1069	0.2960	0.3638	0.5826	0.7349	4.3611	5.4881	1.6633	25.55	1.970
1.980	0.5605	0.1318	0.2352	0.9375	0.7718	5.8542	1.1084	0.2990	0.3618	0.5808	0.7302	4.4071	5.5386	1.6713	25.83	1.980
1.990	0.5580	0.1298	0.2326	0.9402	0.7655	5.8969	1.1098	0.3020	0.3598	0.5791	0.7255	4.4535	5.5894	1.6794	26.10	1.990
2.000	0.5556	0.1278	0.2300	0.9428	0.7591	5.9397	1.1112	0.3050	0.3579	0.5774	0.7209	4.5000	5.6404	1.6875	26.38	2.000



$\gamma=1.400$

$M$	$\frac{T}{T_0}$	$\frac{P}{P_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{1}{2} \rho V^2$	$M_s$	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	$V$	$M$
2.010	0.5531	0.1258	0.2275	0.9454	0.7528	5.9827	1.1126	0.3080	0.3559	0.5757	0.7162	4.5468	5.6918	1.6956	26.66	2.010
2.020	0.5506	0.1239	0.2250	0.9480	0.7465	6.0258	1.1140	0.3109	0.3539	0.5740	0.7115	4.5938	5.7433	1.7038	26.93	2.020
2.030	0.5482	0.1220	0.2225	0.9506	0.7403	6.0692	1.1154	0.3138	0.3518	0.5723	0.7069	4.6411	5.7952	1.7120	27.20	2.030
2.040	0.5458	0.1201	0.2200	0.9531	0.7340	6.1126	1.1167	0.3168	0.3498	0.5707	0.7022	4.6885	5.8473	1.7203	27.48	2.040
2.050	0.5433	0.1182	0.2176	0.9557	0.7279	6.1563	1.1181	0.3197	0.3478	0.5691	0.6975	4.7363	5.8996	1.7285	27.75	2.050
2.060	0.5409	0.1164	0.2152	0.9582	0.7217	6.2001	1.1195	0.3225	0.3458	0.5675	0.6928	4.7842	5.9523	1.7369	28.02	2.060
2.070	0.5385	0.1146	0.2128	0.9607	0.7156	6.2441	1.1209	0.3254	0.3437	0.5659	0.6882	4.8324	6.0051	1.7452	28.29	2.070
2.080	0.5361	0.1128	0.2104	0.9632	0.7095	6.2883	1.1222	0.3282	0.3417	0.5643	0.6835	4.8808	6.0583	1.7536	28.56	2.080
2.090	0.5337	0.1111	0.2081	0.9657	0.7034	6.3326	1.1236	0.3310	0.3396	0.5628	0.6789	4.9295	6.1117	1.7620	28.83	2.090
2.100	0.5313	0.1094	0.2058	0.9681	0.6974	6.3772	1.1250	0.3339	0.3376	0.5613	0.6742	4.9783	6.1654	1.7705	29.10	2.100
2.110	0.5290	0.1077	0.2035	0.9706	0.6914	6.4218	1.1263	0.3366	0.3355	0.5598	0.6696	5.0275	6.2193	1.7789	29.36	2.110
2.120	0.5266	0.1060	0.2013	0.9730	0.6854	6.4667	1.1276	0.3394	0.3334	0.5583	0.6649	5.0768	6.2735	1.7875	29.63	2.120
2.130	0.5243	0.1043	0.1990	0.9754	0.6795	6.5117	1.1290	0.3422	0.3314	0.5568	0.6603	5.1264	6.3280	1.7960	29.90	2.130
2.140	0.5219	0.1027	0.1968	0.9778	0.6736	6.5569	1.1303	0.3449	0.3293	0.5554	0.6557	5.1762	6.3827	1.8046	30.16	2.140
2.150	0.5196	0.1011	0.1946	0.9802	0.6677	6.6023	1.1317	0.3476	0.3272	0.5540	0.6511	5.2263	6.4377	1.8132	30.43	2.150
2.160	0.5173	0.0996	0.1925	0.9825	0.6619	6.6478	1.1330	0.3503	0.3252	0.5525	0.6464	5.2765	6.4929	1.8219	30.69	2.160
2.170	0.5150	0.0980	0.1903	0.9849	0.6561	6.6936	1.1343	0.3530	0.3231	0.5511	0.6419	5.3271	6.5484	1.8306	30.95	2.170
2.180	0.5127	0.0965	0.1882	0.9872	0.6503	6.7395	1.1356	0.3556	0.3210	0.5498	0.6373	5.3778	6.6042	1.8393	31.21	2.180
2.190	0.5104	0.0950	0.1861	0.9895	0.6446	6.7855	1.1369	0.3583	0.3189	0.5484	0.6327	5.4288	6.6602	1.8481	31.47	2.190
2.200	0.5081	0.0935	0.1841	0.9918	0.6389	6.8318	1.1382	0.3609	0.3169	0.5471	0.6281	5.4800	6.7165	1.8569	31.73	2.200
2.210	0.5059	0.0921	0.1820	0.9941	0.6333	6.8782	1.1395	0.3635	0.3148	0.5457	0.6236	5.5315	6.7730	1.8657	31.99	2.210
2.220	0.5036	0.0906	0.1800	0.9964	0.6277	6.9248	1.1408	0.3661	0.3127	0.5444	0.6191	5.5831	6.8298	1.8746	32.25	2.220
2.230	0.5014	0.0892	0.1780	0.9986	0.6221	6.9715	1.1421	0.3687	0.3106	0.5431	0.6145	5.6351	6.8869	1.8835	32.51	2.230
2.240	0.4991	0.0878	0.1760	1.0009	0.6165	7.0185	1.1434	0.3712	0.3085	0.5418	0.6100	5.6872	6.9442	1.8924	32.76	2.240
2.250	0.4969	0.0865	0.1740	1.0031	0.6110	7.0656	1.1446	0.3738	0.3065	0.5406	0.6055	5.7396	7.0018	1.9014	33.02	2.250

$\gamma=1.400$

$M$	$\frac{T}{T_0}$	$\frac{P}{P_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{m\sqrt{c_p T_0}}{A p_0}$	$\frac{m\sqrt{c_p T_0}}{A p}$	$\frac{F}{m\sqrt{c_p T_0}}$	$\frac{4cfL_{max}}{D}$	$\frac{1}{2}\rho V^2$	$M_s$	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	$V$	$M$
2.260	0.4947	0.0851	0.1721	1.0053	0.6056	7.1129	1.1459	0.3763	0.3044	0.5393	0.6011	5.7922	7.0587	1.9104	33.27	2.260
2.270	0.4925	0.0838	0.1702	1.0075	0.6002	7.1603	1.1472	0.3788	0.3023	0.5381	0.5966	5.8451	7.1178	1.9194	33.53	2.270
2.280	0.4903	0.0825	0.1683	1.0097	0.5948	7.2080	1.1484	0.3813	0.3003	0.5368	0.5921	5.8981	7.1762	1.9285	33.78	2.280
2.290	0.4881	0.0812	0.1664	1.0118	0.5894	7.2558	1.1497	0.3838	0.2982	0.5356	0.5877	5.9515	7.2348	1.9376	34.03	2.290
2.300	0.4859	0.0800	0.1646	1.0140	0.5841	7.3038	1.1509	0.3862	0.2961	0.5344	0.5833	6.0050	7.2937	1.9468	34.28	2.300
2.310	0.4837	0.0787	0.1628	1.0161	0.5788	7.3520	1.1521	0.3887	0.2941	0.5332	0.5789	6.0588	7.3528	1.9560	34.53	2.310
2.320	0.4816	0.0775	0.1609	1.0182	0.5736	7.4003	1.1534	0.3911	0.2920	0.5321	0.5745	6.1128	7.4122	1.9652	34.78	2.320
2.330	0.4794	0.0763	0.1592	1.0204	0.5684	7.4488	1.1546	0.3935	0.2900	0.5309	0.5702	6.1671	7.4719	1.9745	35.03	2.330
2.340	0.4773	0.0751	0.1574	1.0224	0.5632	7.4975	1.1558	0.3959	0.2879	0.5297	0.5658	6.2215	7.5319	1.9838	35.28	2.340
2.350	0.4752	0.0740	0.1556	1.0245	0.5581	7.5464	1.1570	0.3983	0.2859	0.5286	0.5615	6.2763	7.5920	1.9931	35.53	2.350
2.360	0.4731	0.0728	0.1539	1.0266	0.5530	7.5955	1.1582	0.4006	0.2839	0.5275	0.5572	6.3312	7.6525	2.0025	35.77	2.360
2.370	0.4709	0.0717	0.1522	1.0286	0.5480	7.6447	1.1595	0.4030	0.2818	0.5264	0.5529	6.3864	7.7132	2.0119	36.02	2.370
2.380	0.4688	0.0706	0.1505	1.0307	0.5430	7.6941	1.1606	0.4053	0.2798	0.5253	0.5486	6.4418	7.7742	2.0213	36.26	2.380
2.390	0.4668	0.0695	0.1488	1.0327	0.5380	7.7437	1.1618	0.4076	0.2778	0.5242	0.5444	6.4975	7.8354	2.0308	36.50	2.390
2.400	0.4647	0.0684	0.1472	1.0347	0.5331	7.7935	1.1630	0.4099	0.2758	0.5231	0.5401	6.5533	7.8969	2.0403	36.75	2.400
2.410	0.4626	0.0673	0.1456	1.0367	0.5282	7.8434	1.1642	0.4122	0.2738	0.5221	0.5359	6.6095	7.9587	2.0499	36.99	2.410
2.420	0.4606	0.0663	0.1439	1.0387	0.5233	7.8935	1.1654	0.4144	0.2718	0.5210	0.5317	6.6658	8.0207	2.0595	37.23	2.420
2.430	0.4585	0.0653	0.1424	1.0407	0.5185	7.9438	1.1665	0.4167	0.2698	0.5200	0.5276	6.7224	8.0830	2.0691	37.47	2.430
2.440	0.4565	0.0643	0.1408	1.0426	0.5137	7.9943	1.1677	0.4189	0.2678	0.5189	0.5234	6.7792	8.1455	2.0788	37.71	2.440
2.450	0.4544	0.0633	0.1392	1.0446	0.5090	8.0450	1.1689	0.4211	0.2658	0.5179	0.5193	6.8363	8.2083	2.0885	37.95	2.450
2.460	0.4524	0.0623	0.1377	1.0465	0.5043	8.0958	1.1700	0.4233	0.2639	0.5169	0.5152	6.8935	8.2713	2.0982	38.18	2.460
2.470	0.4504	0.0613	0.1362	1.0484	0.4996	8.1468	1.1712	0.4255	0.2619	0.5159	0.5111	6.9511	8.3346	2.1080	38.42	2.470
2.480	0.4484	0.0604	0.1346	1.0503	0.4950	8.1980	1.1723	0.4277	0.2599	0.5149	0.5071	7.0088	8.3982	2.1178	38.66	2.480
2.490	0.4464	0.0594	0.1332	1.0522	0.4904	8.2494	1.1734	0.4298	0.2580	0.5140	0.5030	7.0668	8.4620	2.1276	38.89	2.490
2.500	0.4444	0.0585	0.1317	1.0541	0.4858	8.3010	1.1746	0.4320	0.2561	0.5130	0.4990	7.1250	8.5261	2.1375	39.12	2.500

$\gamma=1.400$

$M$	$\frac{T}{T_0}$	$\frac{P}{P_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p}$	$F$	$\frac{4 c_f L_{\max}}{2 D}$	$\frac{1}{2} \frac{\rho V^2}{p_0}$	$M_s$	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	$v$	$M$
2.510	0.4425	0.0576	0.1302	1.0560	0.4813	8.3527	1.1757	0.4341	0.2541	0.5120	0.4950	7.1835	8.5905	2.1474	39.36	2.510
2.520	0.4405	0.0567	0.1288	1.0578	0.4768	8.4046	1.1768	0.4362	0.2522	0.5111	0.4911	7.2421	8.6551	2.1574	39.59	2.520
2.530	0.4386	0.0559	0.1274	1.0597	0.4724	8.4567	1.1779	0.4383	0.2503	0.5102	0.4871	7.3011	8.7200	2.1674	39.82	2.530
2.540	0.4366	0.0550	0.1260	1.0615	0.4680	8.5090	1.1790	0.4404	0.2484	0.5092	0.4832	7.3602	8.7851	2.1774	40.05	2.540
2.550	0.4347	0.0542	0.1246	1.0633	0.4636	8.5615	1.1801	0.4425	0.2465	0.5083	0.4793	7.4196	8.8505	2.1875	40.28	2.550
2.560	0.4328	0.0533	0.1232	1.0651	0.4593	8.6141	1.1812	0.4445	0.2446	0.5074	0.4754	7.4792	8.9161	2.1976	40.51	2.560
2.570	0.4309	0.0525	0.1218	1.0669	0.4550	8.6670	1.1823	0.4466	0.2427	0.5065	0.4715	7.5391	8.9820	2.2077	40.74	2.570
2.580	0.4289	0.0517	0.1205	1.0687	0.4507	8.7200	1.1834	0.4486	0.2409	0.5056	0.4677	7.5991	9.0482	2.2179	40.96	2.580
2.590	0.4271	0.0509	0.1192	1.0705	0.4465	8.7732	1.1844	0.4506	0.2390	0.5047	0.4639	7.6595	9.1146	2.2281	41.19	2.590
2.600	0.4252	0.0501	0.1179	1.0722	0.4423	8.8265	1.1855	0.4526	0.2371	0.5039	0.4601	7.7200	9.1813	2.2383	41.41	2.600
2.610	0.4233	0.0493	0.1166	1.0740	0.4382	8.8801	1.1866	0.4546	0.2353	0.5030	0.4564	7.7808	9.2483	2.2486	41.64	2.610
2.620	0.4214	0.0486	0.1153	1.0757	0.4341	8.9338	1.1876	0.4565	0.2335	0.5022	0.4526	7.8418	9.3155	2.2590	41.86	2.620
2.630	0.4196	0.0478	0.1140	1.0774	0.4300	8.9877	1.1887	0.4585	0.2317	0.5013	0.4489	7.9031	9.3829	2.2693	42.09	2.630
2.640	0.4177	0.0471	0.1128	1.0791	0.4260	9.0418	1.1897	0.4604	0.2298	0.5005	0.4452	7.9645	9.4506	2.2797	42.31	2.640
2.650	0.4159	0.0464	0.1115	1.0808	0.4220	9.0961	1.1908	0.4624	0.2280	0.4996	0.4416	8.0263	9.5186	2.2902	42.53	2.650
2.660	0.4141	0.0457	0.1103	1.0825	0.4180	9.1506	1.1918	0.4643	0.2262	0.4988	0.4379	8.0882	9.5869	2.3006	42.75	2.660
2.670	0.4122	0.0450	0.1091	1.0842	0.4141	9.2052	1.1928	0.4662	0.2245	0.4980	0.4343	8.1504	9.6554	2.3111	42.97	2.670
2.680	0.4104	0.0443	0.1079	1.0859	0.4102	9.2601	1.1939	0.4681	0.2227	0.4972	0.4307	8.2128	9.7241	2.3217	43.19	2.680
2.690	0.4086	0.0436	0.1067	1.0875	0.4063	9.3151	1.1949	0.4700	0.2209	0.4964	0.4271	8.2755	9.7931	2.3323	43.40	2.690
2.700	0.4068	0.0430	0.1056	1.0892	0.4025	9.3703	1.1959	0.4718	0.2192	0.4956	0.4236	8.3383	9.8624	2.3429	43.62	2.700
2.710	0.4051	0.0423	0.1044	1.0908	0.3987	9.4257	1.1969	0.4737	0.2174	0.4949	0.4201	8.4015	9.9319	2.3536	43.84	2.710
2.720	0.4033	0.0417	0.1033	1.0924	0.3949	9.4812	1.1979	0.4755	0.2157	0.4941	0.4166	8.4648	10.0017	2.3642	44.05	2.720
2.730	0.4015	0.0410	0.1022	1.0941	0.3912	9.5370	1.1989	0.4773	0.2140	0.4933	0.4131	8.5284	10.0718	2.3750	44.27	2.730
2.740	0.3998	0.0404	0.1010	1.0957	0.3875	9.5929	1.1999	0.4791	0.2123	0.4926	0.4097	8.5922	10.1421	2.3858	44.48	2.740
2.750	0.3980	0.0398	0.0999	1.0973	0.3838	9.6490	1.2009	0.4809	0.2106	0.4918	0.4062	8.6563	10.2127	2.3966	44.69	2.750

$\gamma=1.400$

$M$	$\frac{T}{T_0}$	$\frac{P}{P_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{m\sqrt{c_p T_0}}{A P_0}$	$\frac{m\sqrt{c_p T_0}}{A P}$	$F$	$\frac{4c_f L_{\max}}{D}$	$\frac{1}{2}\rho V^2$	$M_s$	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	$V$	$M$
2.760	0.3963	0.0392	0.0989	1.0988	0.3802	9.7053	1.2019	0.4827	0.2089	0.4911	0.4028	8.7205	10.2835	2.4074	44.91	2.760
2.770	0.3945	0.0386	0.0978	1.1004	0.3766	9.7618	1.2029	0.4845	0.2072	0.4903	0.3994	8.7851	10.3546	2.4183	45.12	2.770
2.780	0.3928	0.0380	0.0967	1.1020	0.3730	9.8185	1.2038	0.4863	0.2055	0.4896	0.3961	8.8498	10.4259	2.4292	45.33	2.780
2.790	0.3911	0.0374	0.0957	1.1035	0.3695	9.8753	1.2048	0.4880	0.2039	0.4889	0.3928	8.9148	10.4975	2.4402	45.54	2.790
2.800	0.3894	0.0368	0.0946	1.1051	0.3660	9.9324	1.2058	0.4898	0.2022	0.4882	0.3895	8.9800	10.5694	2.4512	45.75	2.800
2.810	0.3877	0.0363	0.0936	1.1066	0.3625	9.9896	1.2067	0.4915	0.2006	0.4875	0.3862	9.0455	10.6415	2.4622	45.95	2.810
2.820	0.3860	0.0357	0.0926	1.1081	0.3591	10.0470	1.2077	0.4932	0.1990	0.4868	0.3829	9.1111	10.7139	2.4733	46.16	2.820
2.830	0.3844	0.0352	0.0916	1.1096	0.3557	10.1046	1.2086	0.4949	0.1973	0.4861	0.3797	9.1771	10.7865	2.4844	46.37	2.830
2.840	0.3827	0.0347	0.0906	1.1111	0.3523	10.1624	1.2095	0.4966	0.1957	0.4854	0.3765	9.2432	10.8594	2.4955	46.57	2.840
2.850	0.3810	0.0341	0.0896	1.1126	0.3490	10.2204	1.2105	0.4983	0.1941	0.4847	0.3733	9.3096	10.9326	2.5067	46.78	2.850
2.860	0.3794	0.0336	0.0886	1.1141	0.3457	10.2785	1.2114	0.5000	0.1926	0.4840	0.3701	9.3762	11.0060	2.5179	46.98	2.860
2.870	0.3777	0.0331	0.0877	1.1156	0.3424	10.3368	1.2123	0.5016	0.1910	0.4833	0.3670	9.4431	11.0797	2.5292	47.19	2.870
2.880	0.3761	0.0326	0.0867	1.1171	0.3392	10.3954	1.2132	0.5033	0.1894	0.4827	0.3639	9.5101	11.1536	2.5405	47.39	2.880
2.890	0.3745	0.0321	0.0858	1.1185	0.3359	10.4541	1.2142	0.5049	0.1879	0.4820	0.3608	9.5775	11.2278	2.5518	47.59	2.890
2.900	0.3729	0.0317	0.0849	1.1199	0.3328	10.5130	1.2151	0.5065	0.1863	0.4814	0.3577	9.6450	11.3022	2.5632	47.79	2.900
2.910	0.3712	0.0312	0.0840	1.1214	0.3296	10.5720	1.2160	0.5081	0.1848	0.4807	0.3547	9.7128	11.3770	2.5746	47.99	2.910
2.920	0.3696	0.0307	0.0831	1.1228	0.3265	10.6313	1.2169	0.5097	0.1833	0.4801	0.3517	9.7808	11.4519	2.5861	48.19	2.920
2.930	0.3681	0.0302	0.0822	1.1242	0.3234	10.6908	1.2178	0.5113	0.1818	0.4795	0.3487	9.8491	11.5271	2.5976	48.39	2.930
2.940	0.3665	0.0298	0.0813	1.1256	0.3203	10.7504	1.2187	0.5129	0.1803	0.4788	0.3457	9.9175	11.6026	2.6091	48.59	2.940
2.950	0.3649	0.0293	0.0804	1.1270	0.3173	10.8102	1.2195	0.5145	0.1788	0.4782	0.3428	9.9863	11.6784	2.6206	48.78	2.950
2.960	0.3633	0.0289	0.0796	1.1284	0.3143	10.8702	1.2204	0.5160	0.1773	0.4776	0.3398	10.0552	11.7544	2.6322	48.98	2.960
2.970	0.3618	0.0285	0.0787	1.1298	0.3113	10.9304	1.2213	0.5176	0.1758	0.4770	0.3369	10.1244	11.8306	2.6439	49.18	2.970
2.980	0.3602	0.0281	0.0779	1.1312	0.3083	10.9908	1.2222	0.5191	0.1744	0.4764	0.3340	10.1938	11.9072	2.6555	49.37	2.980
2.990	0.3587	0.0276	0.0770	1.1325	0.3054	11.0514	1.2230	0.5206	0.1729	0.4758	0.3312	10.2635	11.9839	2.6673	49.56	2.990
3.000	0.3571	0.0272	0.0762	1.1339	0.3025	11.1122	1.2239	0.5222	0.1715	0.4752	0.3283	10.3333	12.0610	2.6790	49.76	3.000

# GAS FLOW TABLES ( $\gamma=1.333$ ): SUBSONIC FLOW

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m}\sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m}\sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m}\sqrt{c_p T_0}}$	$\frac{4c_f L_{max}}{D}$	$\frac{1}{2}\frac{\rho V^2}{p_0}$
0.010	1.0000	0.9999	1.0000	0.0058	0.0231	0.0231	43.2958	7493.200	0.0001
0.020	0.9999	0.9997	0.9998	0.0115	0.0462	0.0462	21.6560	1868.007	0.0003
0.030	0.9999	0.9994	0.9996	0.0173	0.0693	0.0693	14.4464	826.7890	0.0006
0.040	0.9997	0.9989	0.9992	0.0231	0.0923	0.0924	10.8442	462.6179	0.0011
0.050	0.9996	0.9983	0.9988	0.0288	0.1153	0.1155	8.6851	294.2161	0.0017
0.060	0.9994	0.9976	0.9982	0.0346	0.1383	0.1386	7.2475	202.8455	0.0024
0.070	0.9992	0.9967	0.9976	0.0404	0.1612	0.1618	6.2222	147.8292	0.0033
0.080	0.9989	0.9957	0.9968	0.0461	0.1841	0.1849	5.4546	112.1800	0.0042
0.090	0.9987	0.9946	0.9960	0.0519	0.2069	0.2080	4.8587	87.7848	0.0054
0.100	0.9983	0.9934	0.9950	0.0577	0.2297	0.2312	4.3831	70.3719	0.0066
0.110	0.9980	0.9920	0.9940	0.0634	0.2523	0.2544	3.9949	57.5186	0.0080
0.120	0.9976	0.9905	0.9928	0.0692	0.2749	0.2775	3.6724	47.7680	0.0095
0.130	0.9972	0.9888	0.9916	0.0749	0.2974	0.3007	3.4003	40.2012	0.0111
0.140	0.9967	0.9870	0.9903	0.0807	0.3197	0.3239	3.1678	34.2155	0.0129
0.150	0.9963	0.9851	0.9888	0.0864	0.3420	0.3471	2.9670	29.4027	0.0148
0.160	0.9958	0.9831	0.9873	0.0921	0.3641	0.3704	2.7920	25.4777	0.0168
0.170	0.9952	0.9810	0.9857	0.0979	0.3861	0.3936	2.6383	22.2372	0.0189
0.180	0.9946	0.9787	0.9840	0.1036	0.4080	0.4169	2.5022	19.5326	0.0211
0.190	0.9940	0.9763	0.9822	0.1093	0.4298	0.4402	2.3809	17.2536	0.0235
0.200	0.9934	0.9738	0.9803	0.1150	0.4514	0.4635	2.2724	15.3166	0.0260
0.210	0.9927	0.9711	0.9783	0.1207	0.4728	0.4869	2.1747	13.6578	0.0285
0.220	0.9920	0.9684	0.9762	0.1264	0.4941	0.5102	2.0863	12.2273	0.0312
0.230	0.9913	0.9655	0.9740	0.1321	0.5152	0.5336	2.0061	10.9859	0.0340
0.240	0.9905	0.9625	0.9717	0.1378	0.5362	0.5570	1.9330	9.9026	0.0370
0.250	0.9897	0.9594	0.9694	0.1435	0.5569	0.5805	1.8662	8.9522	0.0400
0.260	0.9889	0.9562	0.9669	0.1492	0.5775	0.6040	1.8049	8.1146	0.0431
0.270	0.9880	0.9529	0.9644	0.1549	0.5979	0.6275	1.7486	7.3731	0.0463
0.280	0.9871	0.9494	0.9618	0.1605	0.6181	0.6510	1.6966	6.7140	0.0496
0.290	0.9862	0.9459	0.9591	0.1662	0.6380	0.6746	1.6486	6.1261	0.0530
0.300	0.9852	0.9422	0.9563	0.1718	0.6578	0.6982	1.6042	5.5998	0.0565
0.310	0.9843	0.9384	0.9534	0.1775	0.6774	0.7218	1.5629	5.1272	0.0601
0.320	0.9832	0.9346	0.9505	0.1831	0.6967	0.7455	1.5245	4.7016	0.0638
0.330	0.9822	0.9306	0.9475	0.1887	0.7158	0.7692	1.4888	4.3173	0.0675
0.340	0.9811	0.9265	0.9444	0.1943	0.7347	0.7929	1.4555	3.9693	0.0714
0.350	0.9800	0.9224	0.9412	0.1999	0.7533	0.8167	1.4244	3.6535	0.0753
0.360	0.9789	0.9181	0.9379	0.2055	0.7717	0.8405	1.3953	3.3663	0.0793
0.370	0.9777	0.9137	0.9346	0.2111	0.7898	0.8644	1.3680	3.1046	0.0834
0.380	0.9765	0.9093	0.9311	0.2167	0.8077	0.8883	1.3425	2.8655	0.0875
0.390	0.9753	0.9047	0.9276	0.2223	0.8253	0.9122	1.3185	2.6469	0.0917
0.400	0.9741	0.9001	0.9241	0.2278	0.8427	0.9362	1.2959	2.4466	0.0960
0.410	0.9728	0.8954	0.9204	0.2334	0.8598	0.9603	1.2747	2.2627	0.1003
0.420	0.9715	0.8906	0.9167	0.2389	0.8766	0.9843	1.2548	2.0937	0.1047
0.430	0.9701	0.8857	0.9130	0.2444	0.8932	1.0085	1.2360	1.9382	0.1091
0.440	0.9688	0.8807	0.9091	0.2499	0.9095	1.0326	1.2183	1.7949	0.1136
0.450	0.9674	0.8757	0.9052	0.2554	0.9255	1.0569	1.2016	1.6627	0.1182
0.460	0.9660	0.8706	0.9012	0.2609	0.9412	1.0811	1.1858	1.5405	0.1228
0.470	0.9645	0.8654	0.8972	0.2664	0.9567	1.1055	1.1710	1.4276	0.1274
0.480	0.9631	0.8601	0.8931	0.2718	0.9718	1.1299	1.1569	1.3231	0.1321
0.490	0.9616	0.8548	0.8890	0.2773	0.9867	1.1543	1.1436	1.2263	0.1368
0.500	0.9600	0.8494	0.8847	0.2827	1.0012	1.1788	1.1310	1.1365	0.1415

$$\gamma=1.333$$

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m}\sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m}\sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m}\sqrt{c_p T_0}}$	$\frac{4c_f L_{max}}{D}$	$\frac{1}{2}\frac{\rho V^2}{p_0}$
0.510	0.9585	0.8439	0.8805	0.2881	1.0155	1.2033	1.1192	1.0532	0.1463
0.520	0.9569	0.8384	0.8761	0.2935	1.0295	1.2279	1.1079	0.9759	0.1511
0.530	0.9553	0.8328	0.8717	0.2989	1.0431	1.2526	1.0973	0.9041	0.1559
0.540	0.9537	0.8271	0.8673	0.3043	1.0565	1.2773	1.0872	0.8373	0.1608
0.550	0.9520	0.8214	0.8628	0.3097	1.0696	1.3021	1.0777	0.7752	0.1656
0.560	0.9504	0.8157	0.8583	0.3150	1.0823	1.3269	1.0687	0.7174	0.1705
0.570	0.9487	0.8099	0.8537	0.3204	1.0948	1.3518	1.0601	0.6636	0.1754
0.580	0.9470	0.8040	0.8490	0.3257	1.1069	1.3768	1.0520	0.6136	0.1803
0.590	0.9452	0.7981	0.8443	0.3310	1.1188	1.4018	1.0444	0.5669	0.1852
0.600	0.9434	0.7921	0.8396	0.3363	1.1303	1.4269	1.0371	0.5235	0.1901
0.610	0.9417	0.7861	0.8348	0.3416	1.1415	1.4521	1.0303	0.4830	0.1950
0.620	0.9398	0.7801	0.8300	0.3469	1.1524	1.4773	1.0238	0.4452	0.1999
0.630	0.9380	0.7740	0.8252	0.3521	1.1630	1.5026	1.0176	0.4101	0.2048
0.640	0.9362	0.7679	0.8203	0.3573	1.1733	1.5280	1.0118	0.3773	0.2096
0.650	0.9343	0.7618	0.8153	0.3626	1.1833	1.5534	1.0063	0.3467	0.2145
0.660	0.9324	0.7556	0.8104	0.3678	1.1930	1.5789	1.0011	0.3183	0.2194
0.670	0.9305	0.7494	0.8054	0.3729	1.2023	1.6045	0.9962	0.2918	0.2242
0.680	0.9285	0.7431	0.8003	0.3781	1.2114	1.6301	0.9916	0.2671	0.2290
0.690	0.9266	0.7368	0.7953	0.3833	1.2201	1.6559	0.9872	0.2441	0.2338
0.700	0.9246	0.7306	0.7902	0.3884	1.2285	1.6817	0.9831	0.2227	0.2386
0.710	0.9226	0.7242	0.7850	0.3935	1.2367	1.7075	0.9792	0.2028	0.2433
0.720	0.9205	0.7179	0.7799	0.3986	1.2445	1.7335	0.9755	0.1843	0.2480
0.730	0.9185	0.7116	0.7747	0.4037	1.2520	1.7595	0.9721	0.1671	0.2527
0.740	0.9164	0.7052	0.7695	0.4088	1.2592	1.7856	0.9688	0.1512	0.2574
0.750	0.9144	0.6988	0.7643	0.4139	1.2661	1.8118	0.9658	0.1364	0.2620
0.760	0.9123	0.6924	0.7590	0.4189	1.2727	1.8381	0.9629	0.1227	0.2666
0.770	0.9102	0.6860	0.7537	0.4239	1.2790	1.8644	0.9603	0.1100	0.2711
0.780	0.9080	0.6796	0.7484	0.4289	1.2850	1.8908	0.9578	0.0983	0.2756
0.790	0.9059	0.6732	0.7431	0.4339	1.2907	1.9174	0.9554	0.0875	0.2800
0.800	0.9037	0.6668	0.7378	0.4389	1.2961	1.9440	0.9533	0.0776	0.2844
0.810	0.9015	0.6603	0.7325	0.4438	1.3013	1.9706	0.9513	0.0685	0.2888
0.820	0.8993	0.6539	0.7271	0.4487	1.3061	1.9974	0.9494	0.0601	0.2930
0.830	0.8971	0.6475	0.7217	0.4536	1.3107	2.0243	0.9477	0.0524	0.2973
0.840	0.8949	0.6411	0.7164	0.4585	1.3149	2.0512	0.9461	0.0454	0.3015
0.850	0.8926	0.6346	0.7110	0.4634	1.3189	2.0782	0.9446	0.0391	0.3056
0.860	0.8904	0.6282	0.7056	0.4683	1.3226	2.1053	0.9433	0.0333	0.3097
0.870	0.8881	0.6218	0.7002	0.4731	1.3260	2.1326	0.9420	0.0281	0.3137
0.880	0.8858	0.6154	0.6948	0.4779	1.3292	2.1599	0.9409	0.0235	0.3176
0.890	0.8835	0.6090	0.6893	0.4827	1.3321	2.1873	0.9399	0.0193	0.3215
0.900	0.8812	0.6026	0.6839	0.4875	1.3347	2.2147	0.9390	0.0156	0.3253
0.910	0.8788	0.5963	0.6785	0.4923	1.3370	2.2423	0.9383	0.0124	0.3291
0.920	0.8765	0.5899	0.6731	0.4970	1.3391	2.2700	0.9376	0.0096	0.3328
0.930	0.8741	0.5836	0.6676	0.5018	1.3410	2.2978	0.9370	0.0072	0.3364
0.940	0.8717	0.5773	0.6622	0.5065	1.3425	2.3256	0.9365	0.0052	0.3400
0.950	0.8694	0.5710	0.6568	0.5111	1.3439	2.3536	0.9360	0.0035	0.3435
0.960	0.8670	0.5647	0.6514	0.5158	1.3449	2.3817	0.9357	0.0022	0.3469
0.970	0.8646	0.5585	0.6459	0.5205	1.3458	2.4098	0.9354	0.0012	0.3502
0.980	0.8621	0.5522	0.6405	0.5251	1.3464	2.4381	0.9353	0.0005	0.3535
0.990	0.8597	0.5460	0.6351	0.5297	1.3467	2.4664	0.9351	0.0001	0.3567
1.000	0.8573	0.5398	0.6297	0.5343	1.3468	2.4949	0.9351	0.0000	0.3598

# GAS FLOW TABLES ( $\gamma=1.333$ ): SUPERSONIC FLOW

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m}\sqrt{c_p T_0}}{A p_0}$	$\frac{\dot{m}\sqrt{c_p T_0}}{A p}$	$\frac{F}{\dot{m}\sqrt{c_p T_0}}$	$\frac{4c_f L_{max}}{D}$	$\frac{1}{2}\frac{\rho V^2}{p_0}$
1.010	0.8548	0.5337	0.6243	0.5389	1.3467	2.5234	0.9351	0.0001	0.3628
1.020	0.8524	0.5276	0.6189	0.5434	1.3464	2.5521	0.9352	0.0005	0.3658
1.030	0.8499	0.5215	0.6136	0.5479	1.3458	2.5809	0.9354	0.0011	0.3687
1.040	0.8474	0.5154	0.6082	0.5525	1.3450	2.6097	0.9356	0.0019	0.3715
1.050	0.8449	0.5093	0.6028	0.5569	1.3440	2.6387	0.9359	0.0029	0.3743
1.060	0.8424	0.5033	0.5975	0.5614	1.3428	2.6678	0.9363	0.0042	0.3769
1.070	0.8399	0.4974	0.5922	0.5659	1.3414	2.6970	0.9367	0.0056	0.3795
1.080	0.8374	0.4914	0.5869	0.5703	1.3397	2.7263	0.9371	0.0071	0.3820
1.090	0.8349	0.4855	0.5816	0.5747	1.3379	2.7557	0.9376	0.0089	0.3845
1.100	0.8323	0.4796	0.5763	0.5791	1.3359	2.7852	0.9381	0.0108	0.3868
1.110	0.8298	0.4738	0.5710	0.5835	1.3337	2.8148	0.9387	0.0128	0.3891
1.120	0.8272	0.4680	0.5658	0.5878	1.3313	2.8446	0.9394	0.0150	0.3913
1.130	0.8247	0.4622	0.5605	0.5922	1.3287	2.8744	0.9401	0.0173	0.3934
1.140	0.8221	0.4565	0.5553	0.5965	1.3259	2.9043	0.9408	0.0197	0.3954
1.150	0.8195	0.4508	0.5501	0.6008	1.3229	2.9344	0.9415	0.0223	0.3974
1.160	0.8170	0.4452	0.5449	0.6050	1.3198	2.9646	0.9424	0.0250	0.3993
1.170	0.8144	0.4396	0.5398	0.6093	1.3165	2.9949	0.9432	0.0277	0.4011
1.180	0.8118	0.4340	0.5347	0.6135	1.3131	3.0253	0.9441	0.0306	0.4028
1.190	0.8092	0.4285	0.5295	0.6177	1.3094	3.0558	0.9450	0.0335	0.4044
1.200	0.8066	0.4230	0.5245	0.6219	1.3057	3.0864	0.9459	0.0366	0.4060
1.210	0.8040	0.4176	0.5194	0.6261	1.3017	3.1172	0.9469	0.0397	0.4075
1.220	0.8014	0.4122	0.5143	0.6302	1.2976	3.1481	0.9479	0.0429	0.4089
1.230	0.7988	0.4068	0.5093	0.6344	1.2934	3.1791	0.9489	0.0462	0.4102
1.240	0.7962	0.4015	0.5043	0.6385	1.2890	3.2102	0.9500	0.0495	0.4115
1.250	0.7936	0.3963	0.4994	0.6426	1.2845	3.2414	0.9511	0.0529	0.4127
1.260	0.7909	0.3911	0.4944	0.6466	1.2798	3.2727	0.9522	0.0564	0.4138
1.270	0.7883	0.3859	0.4895	0.6507	1.2751	3.3042	0.9533	0.0599	0.4148
1.280	0.7857	0.3808	0.4846	0.6547	1.2701	3.3358	0.9545	0.0634	0.4158
1.290	0.7830	0.3757	0.4798	0.6587	1.2651	3.3675	0.9557	0.0670	0.4167
1.300	0.7804	0.3706	0.4749	0.6627	1.2599	3.3993	0.9569	0.0707	0.4175
1.310	0.7778	0.3657	0.4701	0.6667	1.2547	3.4313	0.9581	0.0744	0.4182
1.320	0.7751	0.3607	0.4654	0.6706	1.2493	3.4633	0.9594	0.0781	0.4189
1.330	0.7725	0.3558	0.4606	0.6746	1.2438	3.4955	0.9606	0.0819	0.4195
1.340	0.7698	0.3510	0.4559	0.6785	1.2382	3.5279	0.9619	0.0857	0.4200
1.350	0.7672	0.3462	0.4512	0.6824	1.2325	3.5603	0.9632	0.0895	0.4205
1.360	0.7646	0.3414	0.4465	0.6862	1.2266	3.5929	0.9645	0.0934	0.4209
1.370	0.7619	0.3367	0.4419	0.6901	1.2207	3.6256	0.9659	0.0973	0.4212
1.380	0.7593	0.3320	0.4373	0.6939	1.2147	3.6584	0.9672	0.1012	0.4215
1.390	0.7566	0.3274	0.4328	0.6977	1.2086	3.6914	0.9686	0.1051	0.4216
1.400	0.7540	0.3229	0.4282	0.7015	1.2025	3.7245	0.9700	0.1091	0.4218
1.410	0.7513	0.3183	0.4237	0.7053	1.1962	3.7577	0.9714	0.1130	0.4218
1.420	0.7487	0.3139	0.4192	0.7090	1.1899	3.7910	0.9728	0.1170	0.4218
1.430	0.7460	0.3094	0.4148	0.7127	1.1835	3.8245	0.9742	0.1210	0.4217
1.440	0.7434	0.3051	0.4104	0.7164	1.1770	3.8581	0.9756	0.1250	0.4216
1.450	0.7407	0.3007	0.4060	0.7201	1.1704	3.8918	0.9771	0.1290	0.4214
1.460	0.7381	0.2965	0.4017	0.7238	1.1638	3.9257	0.9785	0.1331	0.4212
1.470	0.7354	0.2922	0.3974	0.7275	1.1571	3.9597	0.9800	0.1371	0.4209
1.480	0.7328	0.2880	0.3931	0.7311	1.1504	3.9938	0.9815	0.1411	0.4205
1.490	0.7301	0.2839	0.3888	0.7347	1.1435	4.0281	0.9829	0.1452	0.4201
1.500	0.7275	0.2798	0.3846	0.7383	1.1367	4.0625	0.9844	0.1492	0.4196

$$\gamma=1.333$$

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m}\sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m}\sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m}\sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{1}{2}\rho V^2$ $p_0$
1.510	0.7248	0.2758	0.3804	0.7419	1.1298	4.0970	0.9859	0.1532	0.4191
1.520	0.7222	0.2718	0.3763	0.7454	1.1228	4.1317	0.9874	0.1573	0.4185
1.530	0.7195	0.2678	0.3722	0.7489	1.1158	4.1665	0.9889	0.1613	0.4178
1.540	0.7169	0.2639	0.3681	0.7524	1.1087	4.2014	0.9905	0.1654	0.4171
1.550	0.7143	0.2600	0.3641	0.7559	1.1016	4.2365	0.9920	0.1694	0.4164
1.560	0.7116	0.2562	0.3600	0.7594	1.0945	4.2717	0.9935	0.1734	0.4156
1.570	0.7090	0.2524	0.3561	0.7629	1.0873	4.3070	0.9950	0.1775	0.4147
1.580	0.7064	0.2487	0.3521	0.7663	1.0801	4.3425	0.9966	0.1815	0.4138
1.590	0.7038	0.2450	0.3482	0.7697	1.0729	4.3782	0.9981	0.1855	0.4129
1.600	0.7011	0.2414	0.3443	0.7731	1.0656	4.4139	0.9997	0.1895	0.4119
1.610	0.6985	0.2378	0.3405	0.7765	1.0583	4.4498	1.0012	0.1935	0.4109
1.620	0.6959	0.2343	0.3367	0.7799	1.0510	4.4859	1.0028	0.1975	0.4098
1.630	0.6933	0.2308	0.3329	0.7832	1.0436	4.5220	1.0043	0.2015	0.4087
1.640	0.6907	0.2273	0.3291	0.7865	1.0363	4.5584	1.0059	0.2055	0.4075
1.650	0.6881	0.2239	0.3254	0.7898	1.0289	4.5948	1.0075	0.2094	0.4063
1.660	0.6855	0.2206	0.3217	0.7931	1.0215	4.6314	1.0090	0.2134	0.4051
1.670	0.6829	0.2172	0.3181	0.7964	1.0141	4.6682	1.0106	0.2173	0.4038
1.680	0.6803	0.2139	0.3145	0.7996	1.0066	4.7051	1.0122	0.2213	0.4025
1.690	0.6777	0.2107	0.3109	0.8028	0.9992	4.7421	1.0137	0.2252	0.4011
1.700	0.6751	0.2075	0.3074	0.8061	0.9918	4.7793	1.0153	0.2291	0.3997
1.710	0.6726	0.2044	0.3039	0.8093	0.9843	4.8166	1.0169	0.2330	0.3983
1.720	0.6700	0.2012	0.3004	0.8124	0.9769	4.8541	1.0184	0.2369	0.3968
1.730	0.6674	0.1982	0.2969	0.8156	0.9694	4.8917	1.0200	0.2407	0.3953
1.740	0.6649	0.1951	0.2935	0.8187	0.9620	4.9294	1.0216	0.2446	0.3938
1.750	0.6623	0.1922	0.2901	0.8218	0.9545	4.9673	1.0232	0.2484	0.3922
1.760	0.6597	0.1892	0.2868	0.8249	0.9471	5.0054	1.0247	0.2522	0.3906
1.770	0.6572	0.1863	0.2835	0.8280	0.9396	5.0435	1.0263	0.2560	0.3890
1.780	0.6546	0.1834	0.2802	0.8311	0.9322	5.0819	1.0279	0.2598	0.3874
1.790	0.6521	0.1806	0.2770	0.8341	0.9248	5.1204	1.0294	0.2636	0.3857
1.800	0.6496	0.1778	0.2737	0.8372	0.9173	5.1590	1.0310	0.2673	0.3840
1.810	0.6471	0.1751	0.2706	0.8402	0.9099	5.1978	1.0326	0.2711	0.3822
1.820	0.6445	0.1723	0.2674	0.8432	0.9025	5.2367	1.0341	0.2748	0.3805
1.830	0.6420	0.1697	0.2643	0.8461	0.8951	5.2758	1.0357	0.2785	0.3787
1.840	0.6395	0.1670	0.2612	0.8491	0.8878	5.3150	1.0373	0.2822	0.3769
1.850	0.6370	0.1644	0.2581	0.8521	0.8804	5.3544	1.0388	0.2858	0.3751
1.860	0.6345	0.1619	0.2551	0.8550	0.8731	5.3939	1.0404	0.2895	0.3732
1.870	0.6320	0.1593	0.2521	0.8579	0.8658	5.4336	1.0419	0.2931	0.3714
1.880	0.6295	0.1568	0.2491	0.8608	0.8585	5.4734	1.0435	0.2967	0.3695
1.890	0.6271	0.1544	0.2462	0.8636	0.8512	5.5134	1.0450	0.3003	0.3676
1.900	0.6246	0.1520	0.2433	0.8665	0.8439	5.5535	1.0466	0.3039	0.3656
1.910	0.6221	0.1496	0.2404	0.8693	0.8367	5.5938	1.0481	0.3074	0.3637
1.920	0.6197	0.1472	0.2376	0.8722	0.8295	5.6342	1.0497	0.3110	0.3617
1.930	0.6172	0.1449	0.2348	0.8750	0.8223	5.6748	1.0512	0.3145	0.3598
1.940	0.6148	0.1426	0.2320	0.8778	0.8152	5.7155	1.0527	0.3180	0.3578
1.950	0.6123	0.1404	0.2292	0.8805	0.8081	5.7564	1.0543	0.3215	0.3558
1.960	0.6099	0.1382	0.2265	0.8833	0.8010	5.7974	1.0558	0.3249	0.3537
1.970	0.6075	0.1360	0.2238	0.8860	0.7939	5.8386	1.0573	0.3284	0.3517
1.980	0.6051	0.1338	0.2212	0.8888	0.7869	5.8800	1.0588	0.3318	0.3497
1.990	0.6026	0.1317	0.2185	0.8915	0.7799	5.9215	1.0603	0.3352	0.3476
2.000	0.6002	0.1296	0.2159	0.8942	0.7729	5.9631	1.0619	0.3386	0.3455



### Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$
1.05	0.558	79.937	1.0803	1.0567	1.0223	0.9845	0.99995	1.40	8.000	75.893	1.9842	1.6163	1.2276	0.8184	0.96806
1.10	1.515	76.297	1.1658	1.1157	1.0449	0.9711	0.99963				2.0575	1.6562	1.2423	0.7762	0.96286
1.15	2.000	67.003	1.1408	1.0986	1.0384	1.0434	0.99977	1.45	2.000	46.004	1.1028	1.0723	1.0284	1.3808	0.99990
	2.671	73.822	1.2565	1.1767	1.0678	0.9598	0.99879	4.000	4.000	48.679	1.2169	1.1503	1.0579	1.3091	0.99923
	2.000	81.173	1.3399	1.2316	1.0880	0.9007	0.99745	6.000	6.000	51.755	1.3463	1.2357	1.0895	1.2325	0.99733
1.20	2.000	61.050	1.1197	1.0841	1.0329	1.1113	0.99985	8.000	8.000	55.517	1.5000	1.3333	1.1250	1.1460	0.99325
	3.944	71.977	1.3525	1.2397	1.0910	0.9502	0.99720	10.000	10.785	61.046	1.7114	1.4613	1.1712	1.0317	0.98440
	2.000	83.861	1.4941	1.3297	1.1237	0.8551	0.99344	10.000	8.000	67.097	1.9147	1.5779	1.2135	0.9235	0.97269
1.25	2.000	56.844	1.1110	1.0780	1.0306	1.1696	0.99988	4.000	4.000	72.994	2.0764	1.6664	1.2461	0.8366	0.96147
	4.000	61.986	1.2541	1.1752	1.0672	1.0721	0.99882	6.000	6.000	78.197	2.1836	1.7232	1.2672	0.7777	0.95324
	5.286	70.540	1.4539	1.3045	1.1146	0.9423	0.99468	8.000	8.000	81.733	2.2355	1.7501	1.2774	0.7485	0.94905
	4.000	79.385	1.5944	1.3913	1.1459	0.8525	0.98975	4.000	4.000	84.702	2.2653	1.7654	1.2832	0.7316	0.94659
	2.000	85.211	1.6435	1.4210	1.1566	0.8209	0.98763	2.000	2.000	87.406	2.2812	1.7736	1.2862	0.7225	0.94526
1.30	2.000	53.474	1.1065	1.0749	1.0294	1.2244	0.99989	1.50	2.000	44.065	1.1030	1.0725	1.0284	1.4316	0.99990
	4.000	57.423	1.2334	1.1613	1.0621	1.1398	0.99906	4.000	4.000	46.543	1.2165	1.1500	1.0578	1.3615	0.99923
	6.000	63.459	1.4113	1.2775	1.1048	1.0274	0.99585	6.000	6.000	49.326	1.3433	1.2337	1.0888	1.2879	0.99739
	6.662	69.395	1.5608	1.3709	1.1386	0.9359	0.99108	8.000	8.000	52.571	1.4887	1.3263	1.1224	1.2079	0.99362
	6.000	75.372	1.6793	1.4423	1.1643	0.8636	0.98598	10.000	10.000	56.679	1.6662	1.4345	1.1615	1.1144	0.98660
	4.000	81.649	1.7634	1.4917	1.1822	0.8118	0.98169	12.000	12.000	64.359	1.9668	1.6068	1.2241	0.9607	0.96925
	2.000	86.058	1.7957	1.5103	1.1889	0.7918	0.97990	12.113	12.113	66.589	2.0439	1.6489	1.2396	0.9213	0.96385
1.35	2.000	50.634	1.1042	1.0733	1.0287	1.2774	0.99990	12.000	10.000	68.790	2.1147	1.6869	1.2537	0.8849	0.95860
	4.000	53.965	1.2238	1.1549	1.0596	1.1994	0.99916	10.000	8.000	75.995	2.3046	1.7855	1.2908	0.7854	0.94329
	6.000	58.232	1.3702	1.2512	1.0952	1.1099	0.99682	8.000	8.000	79.712	2.3746	1.8207	1.3042	0.7476	0.93725
	8.000	66.914	1.6327	1.4145	1.1543	0.9543	0.98812	6.000	6.000	82.662	2.4155	1.8410	1.3121	0.7250	0.93363
	8.048	68.470	1.6732	1.4387	1.1630	0.9307	0.98627	4.000	4.000	85.256	2.4404	1.8533	1.3168	0.7112	0.93141
	8.000	70.023	1.7114	1.4613	1.1712	0.9085	0.98440	2.000	2.000	87.668	2.4540	1.8599	1.3194	0.7035	0.93018
	6.000	78.660	1.8774	1.5569	1.2058	0.8111	0.97506	2.000	2.000	42.315	1.1036	1.0729	1.0286	1.4821	0.99990
	4.000	83.028	1.9283	1.5854	1.2163	0.8070	0.97182	4.000	4.000	44.642	1.2173	1.1505	1.0580	1.4130	0.99923
	2.000	86.644	1.9523	1.5988	1.2211	0.7662	0.97023	6.000	6.000	47.214	1.3430	1.2336	1.0887	1.3414	0.99739
1.40	2.000	48.173	1.1030	1.0725	1.0284	1.3295	0.99990	1.55	2.000	42.315	1.1036	1.0729	1.0286	1.4821	0.99990
	4.000	51.117	1.2189	1.1516	1.0584	1.2553	0.99921	4.000	4.000	44.642	1.2173	1.1505	1.0580	1.4130	0.99923
	6.000	54.633	1.3539	1.2406	1.0913	1.1737	0.99717	6.000	6.000	47.214	1.3430	1.2336	1.0887	1.3414	0.99739
	8.000	59.367	1.5263	1.3496	1.1309	1.0744	0.99235	8.000	8.000	50.131	1.4845	1.3236	1.1215	1.2651	0.99375
	9.427	67.716	1.7912	1.5077	1.1880	0.9266	0.98016	10.000	10.000	53.598	1.6491	1.4243	1.1578	1.1804	0.98738
								12.000	12.000	58.240	1.8597	1.5469	1.2022	1.0758	0.97615
								13.403	13.403	66.171	2.1787	1.7206	1.2663	0.9198	0.95362
								12.000	12.000	73.688	2.4151	1.8408	1.3120	0.8014	0.93367
								10.000	10.000	77.804	2.5112	1.8877	1.3302	0.7515	0.92496
								8.000	8.000	80.825	2.5650	1.9136	1.3404	0.7229	0.91995
								6.000	6.000	83.385	2.5991	1.9298	1.3468	0.7045	0.91673
								4.000	4.000	85.699	2.6205	1.9399	1.3508	0.6928	0.91470
								2.000	2.000	87.879	2.6324	1.9455	1.3531	0.6862	0.91356

### Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$\frac{P_{02}}{P_{01}}$	$M_2$	$\frac{P_{02}}{P_{01}}$
1.60	2.000	40.724	1.1046	1.0289	1.5323	0.99990	1.70	6.000	84.848	3.1778	1.4534	2.1865	0.6547	0.85856
	4.000	42.931	1.2189	1.0584	1.4638	0.99921		4.000	86.619	3.1933	1.4562	2.1929	0.6467	0.85695
	6.000	45.344	1.3446	1.0891	1.3934	0.99736		2.000	88.325	3.2021	1.4578	2.1965	0.6421	0.85602
	8.000	48.030	1.4843	1.1215	1.3195	0.99376								
	10.000	51.116	1.6430	1.1565	1.2397	0.98766								
	12.000	54.889	1.8320	1.1965	1.1483	0.97781	1.75	2.000	36.689	1.1087	1.0300	1.0765	1.6816	0.99989
	14.000	60.537	2.0974	1.2502	1.0232	0.95990		4.000	38.651	1.2271	1.0605	1.1571	1.6133	0.99913
	14.652	65.828	2.3192	1.2936	0.9188	0.94204		6.000	40.756	1.3561	1.0918	1.2421	1.5441	0.99713
	14.000	70.895	2.5000	1.3281	0.8320	0.92598		8.000	43.034	1.4973	1.1244	1.3317	1.4733	0.99334
	12.000	75.900	2.6428	1.3550	0.7611	0.91256		10.000	45.531	1.6529	1.1586	1.4266	1.3995	0.98721
1.65	2.000	79.102	2.7132	1.3682	0.7250	0.90574		12.000	48.319	1.8263	1.1953	1.5279	1.3210	0.97814
	8.000	81.691	2.7576	1.3764	0.7018	0.90139		14.000	51.547	2.0245	1.2357	1.6384	1.2348	0.96524
	6.000	83.967	2.7870	1.3819	0.6862	0.89848		16.000	55.589	2.2652	1.2831	1.7654	1.1329	0.94660
	4.000	86.061	2.8059	1.3854	0.6761	0.89660		18.000	62.944	2.6670	1.3595	1.9617	0.9645	0.91023
	2.000	88.054	2.8166	1.3873	0.6703	0.89554		18.121	65.134	2.7745	1.3795	2.0112	0.9189	0.89972
								18.000	67.269	2.8728	1.3977	1.3977	0.8766	0.88991
								16.000	73.757	3.1267	1.4441	1.4441	0.7635	0.86389
	2.000	39.267	1.1058	1.0292	1.5823	0.99990		14.000	76.988	3.2251	1.4620	1.4620	0.7175	0.85362
	4.000	41.377	1.2212	1.0590	1.5140	0.99919		12.000	79.465	3.2868	1.4731	1.4731	0.6878	0.84714
	6.000	43.665	1.3475	1.0898	1.4444	0.99730		10.000	81.570	3.3295	1.4808	1.4808	0.6669	0.84266
1.70	8.000	46.181	1.4869	1.1221	1.3720	0.99367		8.000	83.451	3.3598	1.4862	2.2606	0.6518	0.83947
	10.000	49.007	1.6429	1.1565	1.2952	0.98766		6.000	85.190	3.3811	1.4901	2.2691	0.6409	0.83722
	12.000	52.312	1.8224	1.1945	1.2104	0.98384		4.000	86.838	3.3954	1.4926	2.2748	0.6337	0.83571
	14.000	56.541	2.0441	1.2396	1.1090	0.96384								
	15.855	65.547	2.4653	1.3215	0.9184	0.92915								
	14.000	73.864	2.7642	1.3776	0.7782	0.90073								
	12.000	77.411	2.8587	1.3951	0.7317	0.89132	1.80	2.000	35.538	1.1104	1.1104	1.0776	1.7312	0.99988
	10.000	80.102	2.9157	1.4056	0.7029	0.88557		4.000	37.444	1.2306	1.2306	1.1594	1.6624	0.99909
	8.000	82.389	2.9539	1.4126	0.6833	0.88169		6.000	39.481	1.3615	1.3615	1.2455	1.5932	0.99701
	6.000	84.446	2.9798	1.4174	0.6697	0.87904		8.000	41.673	1.5044	1.5044	1.3360	1.5225	0.99310
4.000	86.364	2.9968	1.4205	0.6607	0.87730		10.000	44.057	1.6611	1.6611	1.4315	1.4494	0.98683	
2.000	88.200	3.0065	1.4222	0.6556	0.87631		12.000	46.686	1.8345	1.8345	1.5326	1.3725	0.97766	
1.70														
2.000	37.927	1.1072	1.0295	1.6320	0.99969		16.000	53.198	2.2568	1.2815	1.2815	1.2896	0.96489	
4.000	39.957	1.2239	1.0597	1.5638	0.99916		18.000	57.995	2.5516	1.3379	1.3379	1.1958	0.94729	
6.000	42.145	1.3514	1.0907	1.4946	0.99722		18.000	64.987	2.9376	1.4096	1.4096	1.0766	0.92120	
8.000	44.528	1.4914	1.1231	1.4232	0.99353		18.000	71.424	3.2297	1.4628	1.4628	0.9195	0.88335	
10.000	47.167	1.6466	1.1573	1.3482	0.98750		16.000	75.324	3.3707	1.4882	1.4882	0.7956	0.85313	
12.000	50.168	1.8216	1.1943	1.2674	0.97841		14.000	78.020	3.4505	1.5025	1.5025	0.6958	0.82990	
14.000	53.771	2.0273	1.2362	1.1757	0.96504		12.000	80.214	3.5041	1.5121	1.5121	0.6703	0.82423	
16.000	58.794	2.2999	1.2898	1.0569	0.94369		10.000	82.128	3.5424	1.5189	1.5189	0.6518	0.82018	
17.012	65.319	2.6171	1.3502	0.9185	0.91502		8.000	83.865	3.5702	1.5239	1.5239	0.6381	0.81725	
16.000	71.426	2.8629	1.3959	0.8077	0.89090		6.000	85.485	3.5899	1.5274	1.5274	0.6283	0.81516	
14.000	75.670	2.9984	1.4208	0.7439	0.87713		4.000	87.028	3.6032	1.5298	1.5298	0.6216	0.81376	
12.000	78.555	3.0722	1.4342	0.7080	0.86953		2.000	88.525	3.6108	1.5311	1.5311	0.6178	0.81295	
10.000	80.906	3.1208	1.4431	0.6838	0.86450									
8.000	82.965	3.1544	1.4492	0.6667	0.86100									

### Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$
1.85	2.000	34.466	1.1121	1.0309	1.7805	0.99988	1.95	2.000	32.528	1.1160	1.0319	1.0815	1.0319	1.8790	0.99987
	4.000	36.323	1.2343	1.0623	1.7114	0.99905		4.000	34.304	1.2424	1.0643	1.1674	1.0643	1.8085	0.99896
	6.000	38.302	1.3672	1.0945	1.6418	0.99689		6.000	36.191	1.3801	1.0975	1.2575	1.0975	1.7380	0.99660
	8.000	40.424	1.5123	1.1278	1.5711	0.99284		8.000	38.204	1.5302	1.1318	1.3521	1.1318	1.6666	0.99221
	10.000	42.717	1.6709	1.1625	1.4983	0.98638		10.000	40.360	1.6938	1.1674	1.4509	1.1674	1.5938	0.98528
	12.000	45.223	1.8453	1.1992	1.4224	0.97701		12.000	42.688	1.8726	1.2049	1.5542	1.2049	1.5185	0.97535
	14.000	48.014	2.0395	1.2387	1.3415	0.96417		14.000	45.230	2.0693	1.2446	1.6625	1.2446	1.4396	0.96200
	16.000	51.232	2.2607	1.2822	1.2524	0.94697		16.000	48.059	2.2879	1.2875	1.7770	1.2875	1.3553	0.94470
	18.000	55.227	2.5275	1.3333	1.1476	0.92345		18.000	51.320	2.5368	1.3351	1.9001	1.3351	1.2622	0.92258
	20.000	62.099	2.9519	1.4123	0.9818	0.88189		20.000	55.381	2.8378	1.3913	2.0397	1.3913	1.1520	0.89342
	20.198	64.872	3.1062	1.4404	0.9205	0.86601		22.000	62.860	3.3464	1.4838	2.2553	1.4838	0.9655	0.84087
	20.000	67.544	3.2437	1.4653	0.8648	0.85167		22.092	64.716	3.4603	1.5043	2.3003	1.5043	0.9229	0.82885
	18.000	73.440	3.5019	1.5117	0.7560	0.82446		22.000	66.523	3.5655	1.5231	2.3410	1.5231	0.8829	0.81774
	16.000	76.511	3.6090	1.5308	0.7085	0.81314		20.000	72.926	3.8872	1.5801	2.4601	1.5801	0.7555	0.78384
	14.000	78.861	3.6772	1.5429	0.6773	0.80593		18.000	75.964	4.0086	1.6015	2.5030	1.6015	0.7045	0.77114
	12.000	80.844	3.7252	1.5514	0.6548	0.80088		16.000	78.253	4.0857	1.6151	2.5297	1.6151	0.6710	0.76313
	10.000	82.606	3.7601	1.5576	0.6381	0.79719		14.000	80.165	4.1401	1.6246	2.5484	1.6246	0.6467	0.75750
	8.000	84.222	3.7858	1.5622	0.6257	0.79449		12.000	81.849	4.1804	1.6317	2.5620	1.6317	0.6283	0.75335
	6.000	85.740	3.8042	1.5655	0.6166	0.79255		10.000	83.381	4.2106	1.6370	2.5722	1.6370	0.6142	0.75024
	4.000	87.193	3.8167	1.5677	0.6105	0.79124		8.000	84.808	4.2333	1.6409	2.5798	1.6409	0.6036	0.74791
	2.000	88.606	3.8239	1.5689	0.6069	0.79048		6.000	86.163	4.2497	1.6438	2.5853	1.6438	0.5957	0.74623
								4.000	87.467	4.2609	1.6458	2.5890	1.6458	0.5904	0.74508
								2.000	88.741	4.2674	1.6469	2.5912	1.6469	0.5872	0.74441
1.90	2.000	33.466	1.1140	1.0314	1.8298	0.99987	2.00	2.000	31.647	1.1180	1.0324	1.0829	1.0324	1.9280	0.99986
	4.000	35.279	1.2382	1.0633	1.7600	0.99901		4.000	33.390	1.2468	1.0654	1.1702	1.0654	1.8568	0.99891
	6.000	37.209	1.3735	1.0959	1.6901	0.99675		6.000	35.241	1.3871	1.0991	1.2620	1.0991	1.7856	0.99644
	8.000	39.272	1.5209	1.1297	1.6191	0.99254		8.000	37.210	1.5400	1.1339	1.3581	1.1339	1.7138	0.99186
	10.000	41.490	1.6818	1.1649	1.5464	0.98586		10.000	39.314	1.7066	1.1702	1.4584	1.1702	1.6405	0.98464
	12.000	43.898	1.8582	1.2019	1.4709	0.97624		12.000	41.575	1.8884	1.2081	1.5631	1.2081	1.5651	0.97437
	14.000	46.550	2.0530	1.2414	1.3913	0.96319		14.000	44.029	2.0876	1.2483	1.6724	1.2483	1.4866	0.96064
	16.000	49.544	2.2718	1.2844	1.3052	0.94605		16.000	46.731	2.3076	1.2913	1.7870	1.2913	1.4034	0.94304
	18.000	53.095	2.5263	1.3331	1.2077	0.92356		18.000	49.785	2.5546	1.3384	1.9086	1.3384	1.3131	0.92092
	20.000	57.900	2.8557	1.3946	1.0835	0.89162		20.000	53.423	2.8429	1.3922	2.0420	1.3922	1.2102	0.89291
	21.167	64.783	3.2805	1.4720	0.9216	0.84781		22.000	58.457	3.2228	1.4616	2.2051	1.4616	1.0760	0.85385
	18.000	74.861	3.7578	1.5572	0.7724	0.79744		22.974	64.669	3.6458	1.5373	2.3715	1.5373	0.9243	0.80926
	16.000	77.463	3.8466	1.5729	0.6884	0.78810		22.000	70.332	3.9714	1.5950	2.4899	1.5950	0.8017	0.77503
	14.000	79.565	3.9068	1.5836	0.6611	0.78178		20.000	74.270	4.1570	1.6276	2.5541	1.6276	0.7278	0.75576
	12.000	81.383	3.9504	1.5913	0.6409	0.77723		18.000	76.862	4.2589	1.6454	2.5883	1.6454	0.6854	0.74529
	10.000	83.020	3.9828	1.5970	0.6257	0.77381		16.000	78.921	4.3277	1.6574	2.6110	1.6574	0.6558	0.73827
	8.000	84.534	4.0068	1.6012	0.6142	0.77133		14.000	80.684	4.3777	1.6662	2.6274	1.6662	0.6337	0.73319
	6.000	85.965	4.0241	1.6042	0.6058	0.76953		12.000	82.257	4.4153	1.6727	2.6396	1.6727	0.6168	0.72939
	4.000	87.338	4.0359	1.6063	0.6001	0.76830		10.000	83.700	4.4438	1.6777	2.6487	1.6777	0.6037	0.72652
	2.000	88.677	4.0428	1.6075	0.5967	0.76759		8.000	85.052	4.4653	1.6815	2.6556	1.6815	0.5937	0.72436
								6.000	86.339	4.4810	1.6842	2.6606	1.6842	0.5864	0.72278
								4.000	87.582	4.4917	1.6861	2.6640	1.6861	0.5813	0.72171
								2.000	88.798	4.4979	1.6871	2.6660	1.6871	0.5783	0.72108

### Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	
2.05	2.000	30.816	1.1200	1.0843	1.0330	1.9771	0.99985	2.10	4.000	87.778	4.9706	2.8097	1.7691	0.5648	0.67494	
	4.000	32.532	1.2512	1.1732	1.0665	1.9050	0.99885	2.000	2.000	88.894	4.9764	2.8113	1.7701	0.5622	0.67438	
	6.000	34.350	1.3943	1.2666	1.1008	1.8330	0.99627									
	8.000	36.281	1.5502	1.3644	1.1362	1.7605	0.99148									
	10.000	38.341	1.7201	1.4664	1.1730	1.6868	0.98396									
	12.000	40.547	1.9053	1.5726	1.2116	1.6111	0.97330									
	14.000	42.928	2.1076	1.6831	1.2522	1.5326	0.95914									
	16.000	45.528	2.3300	1.7983	1.2956	1.4500	0.94112		2.000	29.293		1.1243	1.0872	1.0341	2.0749	0.99984
	18.000	48.428	2.5774	1.9195	1.3427	1.3614	0.91878		4.000	30.960		1.2606	1.1794	1.0688	2.0008	0.99874
	20.000	51.785	2.8600	2.0497	1.3953	1.2630	0.89120		6.000	32.725		1.4094	1.2763	1.1043	1.9271	0.99590
2.000	23.814	64.638	3.2057	2.1980	1.4444	1.1444	0.85565		8.000	34.596		1.5719	1.3777	1.1410	1.8529	0.99065
	22.000	72.193	4.2777	2.4419	1.5712	0.9257	0.78913		10.000	36.584		1.7490	1.4833	1.1791	1.7778	0.98246
	20.000	75.324	4.4215	2.5946	1.6487	0.7626	0.74336		12.000	38.702		1.9417	1.5929	1.2190	1.7011	0.97093
	18.000	77.614	4.5107	2.6700	1.6738	0.7056	0.72876		14.000	40.971		2.1518	1.7065	1.2610	1.6221	0.95574
	16.000	79.498	4.5734	2.6898	1.6903	0.6688	0.71981		16.000	43.422		2.3813	1.8241	1.3055	1.5397	0.93666
	14.000	81.138	4.6199	2.7043	1.7084	0.6219	0.70894		18.000	46.104		2.6337	1.9461	1.3533	1.4527	0.91943
	12.000	82.617	4.6553	2.7152	1.7145	0.6062	0.70545		20.000	49.106		2.9150	2.0740	1.4055	1.3588	0.89564
	10.000	83.983	4.6824	2.7236	1.7192	0.5939	0.70278		22.000	52.618		3.2384	2.2115	1.4644	1.2534	0.85222
	8.000	85.269	4.7029	2.7299	1.7228	0.5846	0.70077		24.000	57.212		3.6452	2.3712	1.5372	1.1223	0.80932
	6.000	86.497	4.7179	2.7344	1.7254	0.5776	0.69930		25.376	64.616		4.2352	2.5804	1.6413	0.9289	0.74772
2.10	4.000	87.685	4.7283	2.7376	1.7272	0.5728	0.69829		24.000	71.164		4.6641	2.7180	1.7160	0.7794	0.70458
	2.000	88.849	4.7343	2.7394	1.7282	0.5700	0.69770		22.000	74.564		4.8442	2.7725	1.7472	0.7122	0.68703
									18.000	78.817		5.0234	2.8249	1.7782	0.6709	0.67689
									16.000	80.444		5.0776	2.8405	1.7876	0.6418	0.66994
									14.000	81.896		5.1191	2.8523	1.7947	0.6012	0.66097
									12.000	83.224		5.1512	2.8613	1.8003	0.5674	0.65798
									10.000	84.464		5.1761	2.8683	1.8046	0.5765	0.65568
									8.000	85.639		5.1951	2.8736	1.8078	0.5680	0.65392
									6.000	86.767		5.2091	2.8775	1.8103	0.5617	0.65263
									4.000	87.862		5.2187	2.8802	1.8119	0.5574	0.65174
								2.000	88.936		5.2244	2.8818	1.8129	0.5548	0.65122	
2.10	2.000	30.033	1.1222	1.0858	1.0335	2.0260	0.99984	2.20	2.000	28.592	1.1266	1.0888	1.0347	2.1237	0.99983	
	4.000	31.723	1.2558	1.1763	1.0676	1.9530	0.99880	4.000	4.000	30.238	1.2654	1.1826	1.0700	2.0485	0.99867	
	6.000	33.513	1.4017	1.2714	1.1025	1.8801	0.99609	6.000	6.000	31.981	1.4173	1.2813	1.1061	1.9738	0.99569	
	8.000	35.412	1.5608	1.3709	1.1386	1.8069	0.99108	8.000	8.000	33.827	1.5832	1.3845	1.1435	1.8987	0.99020	
	10.000	37.433	1.7342	1.4746	1.1760	1.7325	0.98324	10.000	10.000	35.785	1.7641	1.4921	1.1823	1.8228	0.98165	
	12.000	39.592	1.9230	1.5825	1.2152	1.6564	0.97216	12.000	12.000	37.869	1.9611	1.6036	1.2229	1.7454	0.96964	
	14.000	41.912	2.1290	1.6944	1.2565	1.5777	0.95750	14.000	14.000	40.095	2.1756	1.7190	1.2656	1.6657	0.95387	
	16.000	44.430	2.3547	1.8107	1.3004	1.4954	0.93899	16.000	16.000	42.489	2.4095	1.8380	1.3109	1.5831	0.93417	
	18.000	47.210	2.6041	1.9322	1.3478	1.4078	0.91626	18.000	18.000	45.092	2.6658	1.9611	1.3593	1.4963	0.91035	
	20.000	50.365	2.8848	2.0607	1.3999	1.3122	0.88870	20.000	20.000	47.975	2.9494	2.0891	1.4118	1.4035	0.88867	
2.000	24.614	64.621	3.2152	2.2019	1.4602	1.2019	0.85466	22.000	22.000	51.277	3.2704	2.2245	1.4701	1.3013	0.84887	
	24.000	69.104	4.0332	2.5116	1.6058	0.9273	0.76858	24.000	24.000	55.356	3.6552	2.3750	1.5390	1.1805	0.80826	
	22.000	73.521	4.5644	2.6988	1.6568	0.8245	0.73867	26.000	26.000	62.695	4.2918	2.5992	1.6512	0.9795	0.74193	
	20.000	76.189	4.6852	2.7244	1.6987	0.7345	0.71445	26.103	26.103	64.620	4.4426	2.6484	1.6775	0.9305	0.72663	
	18.000	78.257	4.7652	2.7488	1.7197	0.6870	0.69468	26.000	26.000	66.480	4.5807	2.6921	1.7015	0.8849	0.71283	
	16.000	80.001	4.8232	2.7662	1.7436	0.6543	0.670251	24.000	24.000	72.560	4.8728	2.8103	1.7695	0.7490	0.67473	
	14.000	81.539	4.8669	2.7792	1.7512	0.6299	0.64968	22.000	22.000	75.420	5.1222	2.8531	1.7953	0.6936	0.66088	
	12.000	82.938	4.9006	2.7892	1.7570	0.6111	0.6299	20.000	20.000							
	10.000	84.237	4.9264	2.7968	1.7615	0.5964	0.6162	18.000	18.000							
	8.000	85.463	4.9461	2.8025	1.7649	0.5849	0.60714	16.000	16.000							
6.000	86.638	4.9606	2.8068	1.7674	0.5694	0.5694	14.000	14.000								

### Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{P_2}{P_1}$	$\frac{P_2}{P_1}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_2$	$\frac{T_2}{T_1}$	$\frac{P_{02}}{P_{01}}$
2.20	20.00	77.549	5.2175	2.8799	1.8117	2.30	16.000	40.816	2.4701	1.8678	1.3224	1.6676	0.92872	1.6676	1.3224	0.92872
	16.000	79.308	5.2856	2.8987	1.8234		18.000	43.299	2.7360	1.9936	1.3724	1.5804	0.90351	1.5804	1.3724	0.90351
	18.000	80.839	5.3369	2.9127	1.8323		20.000	46.007	3.0276	2.1230	1.4261	1.4885	0.87413	1.4885	1.4261	0.87413
	14.000	82.216	5.3764	2.9235	1.8391		22.000	49.026	3.3514	2.2573	1.4847	1.3894	0.84035	1.3894	1.4847	0.84035
	12.000	83.483	5.4073	2.9318	1.8444		24.000	52.536	3.7216	2.3998	1.5508	1.2788	0.80125	1.2788	1.5508	0.80125
	10.000	84.670	5.4313	2.9382	1.8485		26.000	57.077	4.1819	2.5625	1.6319	1.1425	0.75319	1.1425	1.6319	0.75319
	8.000	85.798	5.4497	2.9431	1.8517		27.454	64.653	4.8739	2.7813	1.7524	0.9338	0.68417	0.9338	1.7524	0.68417
	6.000	86.883	5.4633	2.9468	1.8540		26.000	71.264	5.3682	2.9212	1.8377	0.7743	0.63813	0.7743	1.8377	0.63813
	4.000	87.938	5.4727	2.9493	1.8556		24.000	74.512	5.5649	2.9736	1.8714	0.7060	0.62065	0.7060	1.8714	0.62065
	2.000	88.973	5.4782	2.9507	1.8565		22.000	76.770	5.6817	3.0039	1.8915	0.6635	0.61049	0.6635	1.8915	0.61049
2.25	2.000	27.926	1.1288	1.0903	1.0353		20.000	78.582	5.7631	3.0246	1.9054	0.6328	0.60352	0.6328	1.9054	0.60352
	4.000	29.555	1.2703	1.1859	1.0712		18.000	80.133	5.8238	3.0399	1.9158	0.6092	0.59838	0.6092	1.9158	0.59838
	6.000	31.277	1.4254	1.2864	1.1080		16.000	81.509	5.8705	3.0515	1.9238	0.5906	0.59445	0.5906	1.9238	0.59445
	8.000	33.102	1.5949	1.3916	1.1461		14.000	82.764	5.9071	3.0606	1.9301	0.5757	0.59139	0.5757	1.9301	0.59139
	10.000	35.034	1.7798	1.5011	1.1856		12.000	83.928	5.9360	3.0677	1.9350	0.5638	0.58899	0.5638	1.9350	0.58899
	12.000	37.088	1.9812	1.6147	1.2270		10.000	85.026	5.9586	3.0732	1.9389	0.5543	0.58712	0.5543	1.9389	0.58712
	14.000	39.277	2.2004	1.7319	1.2705		8.000	86.074	5.9861	3.0775	1.9419	0.5469	0.58568	0.5469	1.9419	0.58568
	16.000	41.623	2.4392	1.8527	1.3166		6.000	87.085	5.9980	3.0807	1.9441	0.5413	0.58461	0.5413	1.9441	0.58461
	18.000	44.161	2.7000	1.9770	1.3657		4.000	88.070	5.9980	3.0828	1.9456	0.5374	0.58387	0.5374	1.9456	0.58387
	20.000	46.948	2.9871	2.1055	1.4187		2.000	89.039	6.0033	3.0841	1.9465	0.5352	0.58344	0.5352	1.9465	0.58344
2.30	2.000	50.091	3.3085	2.2400	1.4770	2.35	2.000	26.692	1.1334	1.0935	1.0365	2.2698	0.99980	2.2698	1.0365	0.99980
	4.000	53.837	3.6830	2.3854	1.5440		4.000	28.289	1.2804	1.1926	1.0736	2.1911	0.98846	2.1911	1.0736	0.98846
	6.000	58.122	4.1839	2.5632	1.6323		6.000	29.979	1.4420	1.2970	1.1118	2.1129	0.99502	2.1129	1.1118	0.99502
	8.000	64.633	4.6556	2.7153	1.7145		8.000	31.765	1.6189	1.4062	1.1513	2.0346	0.98872	2.0346	1.1513	0.98872
	10.000	69.627	5.0238	2.8250	1.7783		10.000	33.657	1.8124	1.5199	1.1924	1.9557	0.97895	1.9557	1.1924	0.97895
	12.000	73.634	5.2707	2.8946	1.8209		12.000	35.662	2.0232	1.6376	1.2354	1.8755	0.96534	1.8755	1.2354	0.96534
	14.000	76.145	5.4009	2.9301	1.8433		14.000	37.790	2.2526	1.7589	1.2807	1.7934	0.94765	1.7934	1.2807	0.94765
	16.000	78.098	5.4884	2.9534	1.8583		16.000	40.060	2.5021	1.8833	1.3285	1.7089	0.92580	1.7089	1.3285	0.92580
	18.000	79.744	5.5523	2.9703	1.8693		18.000	42.497	2.7736	2.0108	1.3794	1.6212	0.89981	1.6212	1.3794	0.89981
	20.000	81.192	5.6011	2.9830	1.8776		20.000	45.140	3.0705	2.1413	1.4339	1.5291	0.86971	1.5291	1.4339	0.86971
2.30	2.000	82.504	5.6391	2.9929	1.8842		22.000	48.059	3.3981	2.2759	1.4931	1.4308	0.83542	1.4308	1.4931	0.83542
	4.000	84.856	5.6921	3.0066	1.8893		24.000	51.393	3.7677	2.4168	1.5590	1.3227	0.79639	1.3227	1.5590	0.79639
	6.000	85.942	5.7100	3.0111	1.8932		26.000	55.500	4.2092	2.5717	1.6367	1.1954	0.75038	1.1954	1.6367	0.75038
	8.000	86.988	5.7233	3.0145	1.8963		28.000	62.973	4.9459	2.8024	1.7648	0.9810	0.67729	0.9810	1.7648	0.67729
	10.000	88.007	5.7324	3.0168	1.8986		28.082	64.679	5.0977	2.8462	1.7911	0.9354	0.66296	0.9354	1.7911	0.66296
	12.000	89.008	5.7378	3.0182	1.9002		28.000	66.328	5.2377	2.8855	1.8152	0.8927	0.65000	0.8927	1.8152	0.65000
	2.000	27.294	1.1311	1.0919	1.0359		26.000	72.454	5.6907	3.0062	1.8930	0.7474	0.60972	0.7474	1.8930	0.60972
	4.000	28.906	1.2753	1.1892	1.0724		24.000	75.251	5.8587	3.0486	1.9218	0.6895	0.59544	0.6895	1.9218	0.59544
	6.000	30.611	1.4336	1.2916	1.1099		22.000	77.317	5.9657	3.0750	1.9401	0.6510	0.58653	0.6510	1.9401	0.58653
	8.000	32.415	1.6068	1.3988	1.1487		20.000	79.014	6.0423	3.0936	1.9532	0.6224	0.58024	0.6224	1.9532	0.58024
10.000	34.326	1.7959	1.5104	1.1890		18.000	80.483	6.1001	3.1075	1.9631	0.6002	0.57554	0.6002	1.9631	0.57554	
12.000	36.354	2.0019	1.6260	1.2311		16.000	81.798	6.1451	3.1182	1.9707	0.5826	0.57191	0.5826	1.9707	0.57191	
14.000	38.510	2.2261	1.7452	1.2755		14.000	83.001	6.1806	3.1266	1.9768	0.5683	0.56907	0.5683	1.9768	0.56907	
						12.000	84.122	6.2087	3.1332	1.9816	0.5569	0.56683	0.5569	1.9816	0.56683	
						10.000	85.182	6.2308	3.1384	1.9854	0.5478	0.56508	0.5478	1.9854	0.56508	
						8.000	86.195	6.2479	3.1424	1.9883	0.5406	0.56372	0.5406	1.9883	0.56372	

### Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$\frac{P_2}{P_1}$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$
2.35	6.000	87.174	6.2606	1.9904	0.5353	0.56272	2.45	26.000	53.045	4.3053	1.6535	2.6037	1.6535	1.2861	0.74055	
	4.000	88.129	6.2694	1.9919	0.5315	0.56203		28.000	57.780	4.8455	1.7475	1.7475	2.7729	1.7475	1.1385	0.68691
	2.000	89.068	6.2745	1.9928	0.5293	0.56162		29.253	64.744	5.5614	1.8708	1.8708	2.9727	1.8708	0.9386	0.62095
2.40	2.000	26.120	1.1358	1.0371	2.3184	0.99979	24.000	76.446	6.4516	6.3161	1.9598	3.1029	3.1029	0.7837	0.57709	
	4.000	27.702	1.2856	1.0749	2.2383	0.99839	22.000	78.236	6.5451	6.4516	1.9999	3.1582	3.1582	0.7082	0.55836	
	6.000	29.377	1.4505	1.1138	2.1589	0.99478	20.000	79.752	6.6146	6.6146	2.0230	3.1891	3.1891	0.6623	0.54787	
	8.000	31.149	1.6314	1.1540	2.0794	0.98818	18.000	81.089	6.6682	6.6682	2.0389	3.2101	3.2101	0.6294	0.54076	
	10.000	33.023	1.8292	1.1959	1.9994	0.97797	16.000	82.299	6.7105	6.7105	2.0508	3.2254	3.2254	0.6042	0.53555	
	12.000	35.007	2.0450	1.2398	1.9181	0.96377	14.000	83.416	6.7442	6.7442	2.0599	3.2372	3.2372	0.5842	0.53157	
	14.000	37.112	2.2798	1.2860	1.8350	0.94538	12.000	84.462	6.7710	6.7710	2.0671	3.2464	3.2464	0.5681	0.52845	
	16.000	39.351	2.5351	1.3348	1.7497	0.92274	10.000	85.455	6.7923	6.7923	2.0728	3.2536	3.2536	0.5550	0.52599	
	18.000	41.748	2.8128	1.3866	1.6613	0.89592	8.000	86.408	6.8088	6.8088	2.0774	3.2594	3.2594	0.5444	0.52403	
	20.000	44.336	3.1155	1.4421	1.5689	0.86505	6.000	87.331	6.8211	6.8211	2.0810	3.2640	3.2640	0.5359	0.52249	
	22.000	47.174	3.4480	1.5021	1.4709	0.83015	4.000	88.232	6.8296	6.8296	2.0838	3.2675	3.2675	0.5292	0.52129	
	24.000	50.371	3.8196	1.5682	1.3644	0.79093	2.000	89.119	6.8346	6.8346	2.0859	3.2701	3.2701	0.5242	0.52041	
	26.000	54.184	4.2521	1.6442	1.2426	0.74598					2.0873	3.2719	3.2719	0.5207	0.51979	
	28.000	59.656	4.8382	1.7462	1.0779	0.68761					2.0882	3.2730	3.2730	0.5186	0.51943	
	28.681	64.710	5.3269	1.8305	0.9370	0.64187										
28.000	69.291	5.7130	3.0119	1.8968	0.8201	0.60781	2.50	2.000	25.050	1.1405	1.0384	1.0984	1.0384	2.4155	0.99977	
26.000	73.400	6.0048	3.0845	1.9468	0.7260	0.58331		4.000	26.609	1.2961	1.0775	1.2029	1.2029	2.3326	0.99822	
24.000	75.889	6.1539	3.1203	1.9722	0.6751	0.57121		6.000	28.259	1.4679	1.1177	1.3133	1.3133	2.2505	0.99427	
22.000	77.803	6.2534	3.1436	1.9892	0.6397	0.56329		8.000	30.005	1.6568	1.0595	1.4289	1.4289	2.1685	0.98703	
20.000	79.402	6.3260	3.1605	1.9992	0.6129	0.55758		10.000	31.851	1.8639	1.2031	1.5493	1.5493	2.0859	0.97589	
18.000	80.800	6.3816	3.1732	2.0111	0.5919	0.55326		12.000	33.802	2.0900	1.2488	1.6737	1.6737	2.0022	0.96046	
16.000	82.059	6.4251	3.1831	2.0185	0.5751	0.54990		14.000	35.866	2.3364	1.2969	1.7969	1.7969	1.9169	0.94057	
14.000	83.217	6.4596	3.1909	2.0244	0.5615	0.54726		16.000	38.057	2.6042	1.3478	1.9322	1.9322	1.8295	0.91625	
12.000	84.299	6.4870	3.1971	2.0290	0.5505	0.54517		18.000	40.389	2.8949	1.4018	2.0652	2.0652	1.7394	0.88767	
10.000	85.324	6.5087	3.2019	2.0327	0.5416	0.54352		20.000	42.890	3.2109	1.4594	2.2002	2.2002	1.6458	0.85510	
8.000	86.306	6.5254	3.2057	2.0356	0.5348	0.54225		22.000	45.602	3.5558	1.5213	2.3373	2.3373	1.5475	0.81877	
6.000	87.255	6.5379	3.2085	2.0377	0.5296	0.54131		24.000	48.600	3.9361	1.5887	2.4775	2.4775	1.4426	0.77871	
4.000	88.182	6.5466	3.2104	2.0392	0.5260	0.54065		26.000	52.036	4.3657	1.6641	2.6235	2.6235	1.3268	0.73441	
2.000	89.094	6.5517	3.2115	2.0400	0.5238	0.54027		28.000	56.335	4.8844	1.7542	2.7844	2.7844	1.1888	0.68317	
								29.797	64.782	5.8014	3.0342	3.0342	3.0342	0.9402	0.60027	
							28.000	71.949	6.4249	3.1831	2.0185	3.1831	2.0185	0.7573	0.54992	
							26.000	74.856	6.6273	3.2282	2.0529	3.2282	2.0529	0.6928	0.53460	
							24.000	76.939	6.7526	3.2555	2.0742	3.2555	2.0742	0.6509	0.52537	
							22.000	78.625	6.8414	3.2744	2.0893	3.2744	2.0893	0.6201	0.51894	
							20.000	80.070	6.9082	3.2885	2.1007	3.2885	2.1007	0.5962	0.51417	
							18.000	81.353	6.9602	3.2994	2.1095	3.2994	2.1095	0.5770	0.51048	
							16.000	82.518	7.0014	3.3080	2.1165	3.3080	2.1165	0.5616	0.50759	
							14.000	83.598	7.0343	3.3148	2.1221	3.3148	2.1221	0.5489	0.50528	
							12.000	84.612	7.0607	3.3202	2.1266	3.3202	2.1266	0.5387	0.50345	
							10.000	85.576	7.0816	3.3245	2.1301	3.3245	2.1301	0.5304	0.50200	
							8.000	86.502	7.0979	3.3278	2.1329	3.3278	2.1329	0.5240	0.50088	
							6.000	87.400	7.1104	3.3303	2.1350	3.3303	2.1350	0.5191	0.50005	
							4.000	88.277	7.1184	3.3320	2.1364	3.3320	2.1364	0.5157	0.49947	
							2.000	89.142	7.1234	3.3330	2.1372	3.3330	2.1372	0.5137	0.49913	
2.45	2.000	25.572	1.1381	1.0377	2.3670	0.99978	2.45	2.000	25.572	1.0988	1.0377	1.0988	1.0377	2.3670	0.99978	
	4.000	27.143	1.2908	1.0762	2.2855	0.99831		4.000	27.143	1.1994	1.1994	1.0762	1.1994	1.0762	2.2855	0.99831
	6.000	28.805	1.4591	1.1157	2.2048	0.99453		6.000	28.805	1.3078	1.3078	1.1157	1.3078	1.1157	2.2048	0.99453
	8.000	30.563	1.6463	1.1567	2.1241	0.98761		8.000	30.563	1.4212	1.4212	1.1567	1.4212	1.1567	2.1241	0.98761
	10.000	32.422	1.8463	1.1994	2.0428	0.97695		10.000	32.422	1.5393	1.5393	1.1994	1.5393	1.1994	2.0428	0.97695
	12.000	34.388	2.0672	1.2442	1.9603	0.96215		12.000	34.388	1.6615	1.6615	1.2442	1.6615	1.2442	1.9603	0.96215
	14.000	36.472	2.3078	1.2914	1.8762	0.94502		14.000	36.472	1.7871	1.7871	1.2914	1.7871	1.2914	1.8762	0.94502
	16.000	38.685	2.5692	1.3412	1.7898	0.91955		16.000	38.685	1.9156	1.9156	1.3412	1.9156	1.3412	1.7898	0.91955
	18.000	41.047	2.8532	1.3941	1.7006	0.89187		18.000	41.047	2.0466	2.0466	1.3941	2.0466	1.3941	1.7006	0.89187
	20.000	43.588	3.1623	1.4506	1.6077	0.86018		20.000	43.588	2.1800	2.1800	1.4506	2.1800	1.4506	1.6077	0.86018
	22.000	46.358	3.5007	1.5115	1.5097	0.82459		22.000	46.358	2.3160	2.3160	1.5115	2.3160	1.5115	1.5097	0.82459
	24.000	49.445	3.8759	1.5781	1.4042	0.78502		24.000	49.445	2.4560	2.4560	1.5781	2.4560	1.5781	1.4042	0.78502



### Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$
2.55	2.00	24.550	1.1429	1.1001	1.0390	2.4639	0.99976	2.60	30.814	64.866	6.2972	3.1538	1.9967	0.9433	0.55984
	4.00	26.099	1.3015	1.2065	1.0788	2.3796	0.99814		30.000	69.778	6.7777	3.2609	2.0785	0.8111	0.52354
	6.00	27.739	1.4768	1.3189	1.1198	2.2961	0.99399		28.000	75.590	7.0906	3.3263	2.1317	0.7189	0.50138
	8.00	29.474	1.6699	1.4367	1.1623	2.2128	0.98642		26.000	81.955	7.2555	3.3596	2.1596	0.6673	0.49015
	10.00	31.307	1.8817	1.5593	1.2067	2.1288	0.97479		24.000	87.778	7.3665	3.3815	2.1785	0.6311	0.48276
	12.00	33.244	2.1133	1.6861	1.2534	2.0438	0.95871		22.000	92.999	7.4481	3.3974	2.1923	0.6035	0.47742
	14.00	35.293	2.3656	1.8162	1.3025	1.9573	0.93803		20.000	97.626	7.5108	3.4095	2.2029	0.5817	0.47336
	16.00	37.463	2.6399	1.9490	1.3545	1.8687	0.91283		18.000	101.815	7.5602	3.4189	2.2113	0.5641	0.47020
	18.00	39.770	2.9378	2.0840	1.4097	1.7776	0.88333		16.000	105.506	7.5997	3.4264	2.2180	0.5497	0.46768
	20.00	42.236	3.2611	2.2207	1.4685	1.6832	0.84985		14.000	108.922	7.6316	3.4324	2.2234	0.5378	0.46566
	22.00	44.899	3.6130	2.3591	1.5315	1.5845	0.81272		12.000	112.072	7.6572	3.4372	2.2277	0.5282	0.46405
	24.00	47.822	3.9995	2.4998	1.5999	1.4797	0.77209		10.000	114.922	7.6775	3.4411	2.2312	0.5204	0.46277
	26.00	51.130	4.4319	2.6449	1.6756	1.3655	0.72772		8.000	117.524	7.6934	3.4440	2.2338	0.5143	0.46178
	28.00	55.131	4.9401	2.8007	1.7638	1.2334	0.67784		6.000	119.852	7.7053	3.4462	2.2359	0.5096	0.46104
	30.00	61.449	5.6866	3.0051	1.8923	1.0385	0.61007		4.000	121.935	7.7135	3.4478	2.2372	0.5064	0.46053
	30.317	64.823	6.0466	3.0946	1.9539	0.9418	0.57989		2.000	123.813	7.7184	3.4487	2.2381	0.5045	0.46022
	30.00	67.966	6.3519	3.1664	2.0060	0.8568	0.55557								
	28.00	72.844	6.7595	3.2569	2.0754	0.7364	0.52487								
	26.00	75.440	6.9402	3.2952	2.1061	0.6793	0.51190								
	24.00	77.380	7.0575	3.3195	2.1260	0.6405	0.50368								
	22.00	78.978	7.1423	3.3368	2.1404	0.6115	0.49783	2.65	2.000	23.613	1.1479	1.1034	1.0403	2.5607	0.99973
	20.00	80.360	7.2068	3.3499	2.1514	0.5887	0.49343		4.000	25.144	1.3124	1.2136	1.0814	2.4734	0.99796
	18.00	81.594	7.2575	3.3600	2.1600	0.5703	0.49002		6.000	26.766	1.4950	1.3302	1.1239	2.3969	0.99341
	16.00	82.720	7.2978	3.3680	2.1668	0.5554	0.48732		8.000	28.482	1.6966	1.4525	1.1680	2.3007	0.98514
	14.00	83.766	7.3301	3.3744	2.1723	0.5432	0.48517		10.000	30.295	1.9182	1.5798	1.2142	2.2139	0.97247
	12.00	84.750	7.3561	3.3795	2.1767	0.5333	0.48345		12.000	32.210	2.1610	1.7113	1.2628	2.1262	0.95502
	10.00	85.688	7.3767	3.3835	2.1802	0.5253	0.48209		14.000	34.232	2.4260	1.8462	1.3141	2.0370	0.93270
	8.000	86.590	7.3927	3.3866	2.1829	0.5190	0.48104		16.000	36.368	2.7141	1.9835	1.3683	1.9459	0.90566
	6.000	87.464	7.4047	3.3890	2.1849	0.5142	0.48025		18.000	38.632	3.0267	2.1226	1.4259	1.8524	0.87423
	4.000	88.320	7.4131	3.3906	2.1864	0.5109	0.47971		20.000	41.043	3.3657	2.2630	1.4873	1.7560	0.83884
	2.000	89.163	7.4180	3.3916	2.1872	0.5090	0.47939		22.000	43.627	3.7335	2.4042	1.5529	1.6559	0.80000
									24.000	46.433	4.1347	2.5465	1.6237	1.5507	0.75806
									26.000	49.549	4.5776	2.6911	1.7010	1.4380	0.71313
									28.000	53.164	5.0815	2.8416	1.7883	1.3126	0.66448
									30.000	57.877	5.7097	3.0110	1.8963	1.1576	0.60809
2.60	2.000	24.071	1.1454	1.1017	1.0396	2.5123	0.99975		31.288	64.910	6.5531	3.2118	2.0403	0.9447	0.54016
	4.000	25.611	1.3070	1.2100	1.0801	2.4265	0.99805		30.000	70.983	7.1564	3.3397	2.1428	0.7814	0.49687
	6.000	27.241	1.4858	1.3245	1.1218	2.3416	0.99371		28.000	74.230	7.4211	3.3922	2.1877	0.7039	0.47918
	8.000	28.966	1.6831	1.4445	1.1651	2.2568	0.98579		26.000	76.415	7.5742	3.4216	2.2137	0.6565	0.46930
	10.000	30.789	1.8998	1.5695	1.2105	2.1715	0.97365		24.000	78.138	7.6801	3.4415	2.2316	0.6224	0.46262
	12.000	32.714	2.1369	1.6986	1.2580	2.0852	0.95690		22.000	79.592	7.7589	3.4562	2.2449	0.5962	0.45771
	14.000	34.749	2.3955	1.8311	1.3082	1.9973	0.93541		20.000	80.870	7.8200	3.4674	2.2553	0.5752	0.45396
	16.000	36.901	2.6767	1.9662	1.3613	1.9075	0.90930		18.000	82.020	7.8684	3.4763	2.2634	0.5582	0.45101
	18.000	39.185	2.9817	2.1032	1.4177	1.8152	0.87884		16.000	83.079	7.9073	3.4833	2.2700	0.5442	0.44866
	20.000	41.621	3.3126	2.2417	1.4778	1.7199	0.84443		14.000	84.066	7.9387	3.4890	2.2753	0.5327	0.44677
	22.000	44.242	3.6723	2.3814	1.5421	1.6205	0.80645		12.000	84.998	7.9640	3.4935	2.2796	0.5234	0.44526
	24.000	47.102	4.0658	2.5229	1.6116	1.5157	0.76520		10.000	85.888	7.9841	3.4972	2.2830	0.5158	0.44406
	26.000	50.305	4.5028	2.6675	1.6880	1.4025	0.72060		8.000	86.746	7.9999	3.5000	2.2857	0.5098	0.44312
	28.000	54.088	5.0067	2.8201	1.7754	1.2744	0.67151		6.000	87.579	8.0116	3.5021	2.2877	0.5053	0.44242
	30.000	59.352	5.6706	3.0010	1.8896	1.1062	0.61145								

### Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$
2.65	4.000	88.396	8.0198	3.5035	2.2891	0.5021	0.44194	2.75	24.000	45.225	4.2794	2.5951	1.6490	1.6181	0.74319
	2.000	89.200	8.0247	3.5044	2.2899	0.5003	0.44165		26.000	48.206	4.7375	2.7404	2.7404	1.7288	1.5056
2.70	2.000	23.173	1.1503	1.1051	1.0409	2.6090	0.99972	30.000	55.674	5.5674	5.8507	3.0466	1.9204	1.2416	0.59611
	4.000	24.696	1.3179	1.2172	1.0827	2.5201	0.99786	32.000	62.549	6.7812	7.0807	3.2616	2.0791	1.0209	0.52329
	6.000	26.311	1.5042	1.3360	1.1260	2.4321	0.99311	32.173	65.002	7.1748	7.3448	3.3243	2.1300	0.9476	0.50207
	8.000	28.019	1.7102	1.4605	1.1709	2.3444	0.98446	32.000	67.323	7.3448	7.3448	3.3773	2.1748	0.8812	0.48420
	10.000	29.824	1.9369	1.5902	1.2180	2.2561	0.97125	30.000	72.678	7.8741	7.8741	3.4773	2.2644	0.7401	0.45066
	12.000	31.728	2.1855	1.7241	1.2676	2.1669	0.95309	28.000	75.285	8.0870	8.0870	3.5154	2.3004	0.6789	0.43799
	14.000	33.739	2.4569	1.8614	1.3199	2.0763	0.92991	24.000	77.202	8.2233	8.2233	3.5393	2.3235	0.6378	0.43010
	16.000	35.862	2.7523	2.0010	1.3754	1.9838	0.90191	22.000	78.766	8.3214	8.3214	3.5561	2.3400	0.6071	0.42454
	18.000	38.109	3.0727	2.1423	1.4343	1.8890	0.86948	20.000	80.110	8.3960	8.3960	3.5688	2.3526	0.5829	0.42037
	20.000	40.496	3.4200	2.2845	1.4970	1.7915	0.83311	18.000	81.303	8.4545	8.4545	3.5786	2.3625	0.5634	0.41714
	22.000	43.049	3.7964	2.4273	1.5641	1.6905	0.79337	16.000	82.386	8.5014	8.5014	3.5864	2.3704	0.5474	0.41457
	24.000	45.809	4.2059	2.5706	1.6362	1.5848	0.75072	14.000	83.387	8.5392	8.5392	3.5927	2.3768	0.5343	0.41251
	26.000	48.852	4.6560	2.7155	1.7146	1.4723	0.70538	12.000	84.324	8.5699	8.5699	3.5978	2.3820	0.5234	0.41085
	28.000	52.334	5.1626	2.8645	1.8022	1.3488	0.65692	10.000	85.212	8.5948	8.5948	3.6019	2.3862	0.5145	0.40951
	30.000	56.687	5.7730	3.0271	1.9071	1.2018	0.60268	8.000	86.062	8.6146	8.6146	3.6051	2.3895	0.5072	0.40845
	31.741	64.956	6.8143	3.2687	2.0847	0.9462	0.47286	6.000	86.882	8.6301	8.6301	3.6077	2.3922	0.5015	0.40762
	30.000	71.913	7.5186	3.4110	2.2042	0.7587	0.42850	4.000	87.680	8.6418	8.6418	3.6096	2.3941	0.4972	0.40700
	28.000	74.790	7.7529	3.4551	2.2439	0.6907	0.45808	2.000	88.462	8.6499	8.6499	3.6109	2.3955	0.4942	0.40656
	26.000	76.828	7.8967	3.4814	2.2682	0.6468	0.44930		89.234	8.6547	8.6547	3.6117	2.3963	0.4924	0.40631
	24.000	78.466	7.9983	3.4997	2.2854	0.6145	0.44321								
22.000	79.862	8.0748	3.5133	2.2984	0.5893	0.43870	2.80	2.000	22.344	1.1553	1.1085	1.0422	2.7056	0.99969	
20.000	81.095	8.1345	3.5238	2.3085	0.5691	0.43522	4.000	4.000	23.854	1.3292	1.2246	1.0854	2.6133	0.99766	
18.000	82.210	8.1821	3.5321	2.3165	0.5527	0.43247	6.000	6.000	25.455	1.5230	1.3476	1.1302	2.5222	0.99246	
16.000	83.238	8.2204	3.5388	2.3230	0.5391	0.43027	8.000	8.000	27.150	1.7379	1.4768	1.1768	2.4313	0.98304	
14.000	84.199	8.2515	3.5441	2.3282	0.5279	0.42850	10.000	10.000	28.940	1.9751	1.6113	1.2257	2.3399	0.96869	
12.000	85.109	8.2765	3.5484	2.3324	0.5188	0.42708	12.000	12.000	30.830	2.2357	1.7502	1.2774	2.2476	0.94903	
10.000	85.978	8.2965	3.5518	2.3358	0.5114	0.42595	14.000	14.000	32.822	2.5205	1.8923	1.3320	2.1540	0.92409	
8.000	86.816	8.3121	3.5545	2.3385	0.5056	0.42506	16.000	16.000	34.923	2.8309	2.0367	1.3900	2.0585	0.89411	
6.000	87.631	8.3238	3.5565	2.3404	0.5012	0.42441	18.000	18.000	37.141	3.1677	2.1822	1.4516	1.9610	0.85962	
4.000	88.430	8.3319	3.5579	2.3418	0.4981	0.42395	20.000	20.000	39.490	3.5324	2.3283	1.5172	1.8610	0.82123	
2.000	89.218	8.3367	3.5587	2.3426	0.4962	0.42368	22.000	22.000	41.990	3.9271	2.4743	1.5872	1.7578	0.77965	
								24.000	44.676	4.3550	2.6200	1.6622	1.6622	1.6506	0.73549
								26.000	47.604	4.8219	2.7658	1.7434	1.7434	1.5379	0.68919
2.75	2.000	22.750	1.1528	1.1068	1.0415	2.6573	0.99971	28.000	50.887	5.3398	2.9135	1.8328	1.8328	1.4163	0.64070
4.000	24.267	1.3236	1.2209	1.0841	1.0841	2.5667	0.99776	30.000	54.786	5.9387	3.0683	1.9355	1.9355	1.2783	0.58877
6.000	25.873	1.5135	1.3417	1.1280	1.1280	2.4772	0.99279	32.000	60.433	6.7529	3.2555	2.0743	2.0743	1.0909	0.52535
8.000	27.575	1.7239	1.4686	1.1738	1.1738	2.3879	0.98377	32.587	65.050	7.3524	3.3788	2.1761	2.1761	0.9490	0.48369
10.000	29.372	1.9558	1.6007	1.2219	1.2219	2.2982	0.96999	32.000	69.211	7.8278	3.4689	2.2566	2.2566	0.8307	0.45348
12.000	31.269	2.2104	1.7371	1.2724	1.2724	2.2074	0.95109	30.000	73.328	8.2272	3.5399	2.3241	2.3241	0.7243	0.42988
14.000	33.269	2.4885	1.8768	1.3259	1.3259	2.1153	0.92704	28.000	75.728	8.4241	3.5735	2.3574	2.3574	0.6684	0.41882
16.000	35.381	2.7912	2.0188	1.3826	1.3826	2.0213	0.89806	26.000	77.543	8.5544	3.5952	2.3794	2.3794	0.6296	0.41169
18.000	37.612	3.1197	2.1622	1.4429	1.4429	1.9253	0.86461	24.000	79.042	8.6495	3.6108	2.3954	2.3954	0.6002	0.40659
20.000	39.980	3.4757	2.3063	1.5070	1.5070	1.8265	0.82724	22.000	80.339	8.7224	3.6227	2.4077	2.4077	0.5769	0.40273
22.000	42.504	3.8610	2.4506	1.5755	1.5755	1.7245	0.78659	20.000	81.496	8.7800	3.6319	2.4174	2.4174	0.5580	0.39971



### Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$
2.80	18.000	82.550	8.8262	3.6393	2.4252	0.5425	0.39731	2.90	6.000	24.666	1.5421	1.3594	1.1344	2.6117	0.99178
	16.000	83.525	8.8637	3.6453	2.4316	0.5297	0.39538		8.000	26.350	1.7663	1.4933	1.1828	2.5175	0.98153
	14.000	84.440	8.8942	3.6501	2.4367	0.5191	0.39382		10.000	28.129	2.0143	1.6328	1.2336	2.4229	0.96597
	12.000	85.308	8.9188	3.6540	2.4409	0.5103	0.39256		12.000	30.007	2.2873	1.7767	1.2874	2.3273	0.94475
	10.000	86.140	8.9385	3.6571	2.4442	0.5033	0.39156		14.000	31.985	2.5863	1.9238	1.3444	2.2304	0.91794
	8.000	86.943	8.9540	3.6595	2.4468	0.4977	0.39078		16.000	34.069	2.9123	2.0729	1.4050	2.1318	0.88591
	6.000	87.725	8.9656	3.6613	2.4487	0.4935	0.39019		18.000	36.264	3.2466	2.2229	1.4694	2.0313	0.84930
	4.000	88.492	8.9737	3.6626	2.4501	0.4905	0.38978		20.000	38.584	3.6496	2.3729	1.5380	1.9285	0.80886
	2.000	89.248	8.9784	3.6633	2.4509	0.4887	0.38954		22.000	41.044	4.0638	2.5222	1.6112	1.8229	0.76540
									24.000	43.672	4.5119	2.6704	1.6896	1.7138	0.71969
2.85	2.000	21.954	1.1579	1.1103	1.0429	2.7537	0.99968		26.000	46.515	4.9984	2.8177	1.7739	1.5999	0.67230
	4.000	23.457	1.3349	1.2283	1.0868	2.6598	0.99755		28.000	49.655	5.5328	2.9652	1.8659	1.4788	0.62347
	6.000	25.052	1.5325	1.3535	1.1323	2.5670	0.99213		30.000	53.274	6.1364	3.1161	1.9692	1.3453	0.57262
	8.000	26.742	1.7520	1.4850	1.1798	2.4744	0.98230		32.000	57.931	6.8791	3.2824	2.0957	1.1827	0.51624
	10.000	28.526	1.9946	1.6220	1.2297	2.3815	0.96735		33.363	65.145	7.9116	3.4841	2.2708	0.9516	0.44840
	12.000	30.410	2.2613	1.7634	1.2824	2.2876	0.94692		32.000	74.392	8.6350	3.6085	2.3930	0.7771	0.40736
	14.000	32.394	2.5532	1.9080	1.3382	2.1923	0.92105		30.000	76.490	9.1095	3.6836	2.4730	0.6500	0.38301
	16.000	34.486	2.8712	2.0547	1.3974	2.0953	0.89006		26.000	78.142	9.2307	3.7020	2.4934	0.6149	0.37709
	18.000	36.692	3.2165	2.2025	1.4604	1.9964	0.85451		24.000	79.533	9.3212	3.7156	2.5087	0.5878	0.37275
	20.000	39.025	3.5904	2.3505	1.5275	1.8950	0.81511		22.000	80.750	9.3915	3.7260	2.5205	0.5660	0.36942
2.90	2.000	41.505	3.9948	2.4982	1.5991	1.7906	0.77258		18.000	81.843	9.4475	3.7343	2.5300	0.5482	0.36680
	4.000	44.160	4.4325	2.6451	1.6757	1.6825	0.72766		16.000	82.845	9.4928	3.7409	2.5376	0.5335	0.36469
	6.000	47.042	4.9089	2.7916	1.7585	1.5692	0.68081		14.000	83.775	9.5296	3.7462	2.5438	0.5212	0.36299
	8.000	50.247	5.4345	2.9391	1.8490	1.4481	0.63219		12.000	84.651	9.5597	3.7506	2.5489	0.5111	0.36161
	10.000	53.992	6.0344	3.0917	1.9518	1.3127	0.58089		10.000	85.484	9.5842	3.7541	2.5530	0.5027	0.36049
	12.000	59.037	6.8013	3.2659	2.0825	1.1407	0.52183		8.000	86.283	9.6038	3.7570	2.5563	0.4959	0.35960
	14.000	65.097	7.6294	3.4320	2.2230	0.9503	0.46580		6.000	87.055	9.6191	3.7592	2.5588	0.4906	0.35890
	16.000	70.389	8.2421	3.5425	2.3266	0.8001	0.42903		4.000	87.808	9.6306	3.7608	2.5608	0.4865	0.35838
	18.000	73.893	8.5802	3.5995	2.3837	0.7107	0.41030		2.000	88.546	9.6387	3.7620	2.5621	0.4836	0.35802
	20.000	76.127	8.7648	3.6295	2.4149	0.6588	0.40050			89.275	9.6434	3.7626	2.5629	0.4819	0.35780
2.95	2.000	77.855	8.8902	3.6495	2.4360	0.6220	0.39402								
	4.000	79.297	8.9827	3.6640	2.4516	0.5938	0.38933								
	6.000	80.552	9.0543	3.6751	2.4637	0.5713	0.38574		2.000	21.216	1.1630	1.1138	1.0442	2.8500	0.99965
	8.000	81.676	9.1110	3.6838	2.4733	0.5530	0.38294		4.000	22.708	1.3464	1.2357	1.0895	2.7526	0.99732
	10.000	82.702	9.1567	3.6908	2.4810	0.5379	0.38069		6.000	24.294	1.5518	1.3654	1.1366	2.6563	0.99142
	12.000	83.655	9.1938	3.6964	2.4872	0.5253	0.37888		8.000	25.974	1.7807	1.5017	1.1858	2.5604	0.98074
	14.000	84.549	9.2241	3.7010	2.4923	0.5150	0.37741		10.000	27.749	2.0343	1.6437	1.2377	2.4640	0.96454
	16.000	85.399	9.2486	3.7047	2.4964	0.5064	0.37623		12.000	29.621	2.3137	1.7901	1.2925	2.3668	0.94252
	18.000	86.213	9.2683	3.7077	2.4998	0.4995	0.37528		14.000	31.593	2.6199	1.9396	1.3507	2.2682	0.91475
	20.000	87.001	9.2836	3.7100	2.5023	0.4940	0.37454		16.000	33.670	2.9540	2.0911	1.4126	2.1679	0.88168
2.99	2.000	87.768	9.2952	3.7117	2.5043	0.4899	0.37399		18.000	35.856	3.3169	2.2434	1.4785	2.0658	0.84398
	4.000	88.520	9.3033	3.7129	2.5057	0.4870	0.37360		20.000	38.164	3.7098	2.3954	1.5487	1.9615	0.80249
	6.000	89.262	9.3080	3.7136	2.5065	0.4853	0.37338		22.000	40.607	4.1344	2.5464	1.6236	1.8546	0.75809
									24.000	43.211	4.5930	2.6959	1.7037	1.7444	0.71160
	2.000	21.578	1.1604	1.1120	1.0435	2.8019	0.99966		26.000	46.018	5.0902	2.8441	1.7898	1.6297	0.66366
	4.000	23.076	1.3406	1.2320	1.0882	2.7062	0.99744		28.000	49.102	5.6343	2.9916	1.8833	1.5085	0.61460
									30.000	52.618	6.2438	3.1414	1.9876	1.3762	0.56404



### Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	
3.10	6.000	23.258	1.5815	1.3835	1.1431	2.7894	0.99027	3.15	28.000	47.216	6.0688	3.1000	1.9577	1.6194	0.57808	
	8.000	24.927	1.8249	1.5271	1.1950	2.6881	0.97822		30.000	50.449	6.7158	3.2475	2.0680	1.4886	0.52806	
	10.000	26.692	2.0956	1.6767	1.2499	2.5864	0.96004		32.000	54.201	7.4487	3.3975	2.1924	1.3441	0.47738	
	12.000	28.554	2.3949	1.8308	1.3081	2.4837	0.93546		34.000	59.196	8.3736	3.5650	2.3489	1.1632	0.42162	
	14.000	30.513	2.7236	1.9879	1.3701	2.3798	0.90473		35.033	65.382	9.4008	3.7274	2.5221	0.9575	0.36898	
	16.000	32.574	3.0831	2.1467	1.4362	2.2743	0.86841		34.000	70.719	10.1474	3.8325	2.6478	0.7974	0.33596	
	18.000	34.739	3.4740	2.3057	1.5067	2.1672	0.82741		32.000	74.089	10.5396	3.8839	2.7137	0.7064	0.32016	
	20.000	37.017	3.8973	2.4637	1.5819	2.0581	0.78278		30.000	76.244	10.7550	3.9111	2.7499	0.6531	0.31190	
	22.000	39.421	4.3543	2.6198	1.6621	1.9468	0.73556		28.000	77.906	10.9014	3.9292	2.7745	0.6152	0.30644	
	24.000	41.968	4.8470	2.7733	1.7477	1.8329	0.68676		26.000	79.289	11.0097	3.9424	2.7927	0.5860	0.30248	
	26.000	44.692	5.3788	2.9241	1.8395	1.7154	0.63718		24.000	80.490	11.0936	3.9524	2.8068	0.5627	0.29947	
	28.000	47.646	5.9563	3.0727	1.9385	1.5928	0.58731		22.000	81.560	11.1602	3.9604	2.8180	0.5436	0.29710	
	30.000	50.935	6.5922	3.2205	2.0470	1.4620	0.53722		20.000	82.535	11.2142	3.9668	2.8270	0.5278	0.29520	
	32.000	54.800	7.3197	3.3723	2.1705	1.3157	0.48586		18.000	83.436	11.2583	3.9720	2.8344	0.5145	0.29366	
	34.000	60.205	8.2768	3.5485	2.3325	1.1241	0.42706		16.000	84.279	11.2945	3.9762	2.8405	0.5035	0.29240	
	36.000	65.335	9.0925	3.6810	2.4701	0.9564	0.36385		14.000	85.076	11.3243	3.9797	2.8455	0.4942	0.29138	
	38.000	69.872	9.7174	3.7732	2.5754	0.8203	0.30449		12.000	85.838	11.3486	3.9825	2.8496	0.4865	0.29054	
	40.000	73.661	10.1577	3.8339	2.6495	0.7171	0.25353		10.000	86.571	11.3682	3.9848	2.8529	0.4803	0.28987	
	42.000	75.938	10.3831	3.8636	2.6874	0.6607	0.20634		8.000	87.281	11.3835	3.9866	2.8555	0.4754	0.28935	
	44.000	77.666	10.5334	3.8831	2.7126	0.6212	0.16164		6.000	87.976	11.3951	3.9879	2.8574	0.4716	0.28895	
46.000	79.091	10.6435	3.8971	2.7311	0.5911	0.12191		4.000	88.657	11.4032	3.9889	2.8588	0.4690	0.28867		
48.000	80.324	10.7282	3.9077	2.7454	0.5671	0.08291		2.000	89.330	11.4080	3.9894	2.8596	0.4674	0.28851		
50.000	81.419	10.7954	3.9161	2.7567	0.5476	0.05108										
52.000	82.413	10.8496	3.9228	2.7658	0.5314	0.03036										
54.000	83.331	10.8938	3.9282	2.7732	0.5179	0.02072		3.20	2.000	19.587	1.1760	1.1226	1.0475	3.0901	0.99957	
56.000	84.189	10.9301	3.9327	2.7793	0.5067	0.01539		4.000	21.059	21.059	1.3759	1.2548	1.0965	2.9831	0.99670	
58.000	85.001	10.9599	3.9363	2.7843	0.4973	0.01040		6.000	22.628	22.628	1.6017	1.3958	1.1475	2.8776	0.98944	
60.000	85.775	10.9842	3.9393	2.7884	0.4895	0.00741		8.000	24.292	24.292	1.8552	1.5443	1.2013	2.7725	0.97642	
62.000	86.520	11.0037	3.9416	2.7917	0.4832	0.00520		10.000	26.052	26.052	2.1377	1.6990	1.2582	2.6670	0.96684	
64.000	87.242	11.0190	3.9435	2.7942	0.4781	0.00365		12.000	27.909	27.909	2.4507	1.8583	1.3188	2.5605	0.93048	
66.000	87.945	11.0306	3.9449	2.7962	0.4743	0.00253		14.000	29.863	29.863	2.7952	2.0206	1.3834	2.4528	0.89766	
68.000	88.637	11.0387	3.9458	2.7975	0.4716	0.00174		16.000	31.915	31.915	3.1723	2.1842	1.4524	2.3437	0.85914	
70.000	89.321	11.0434	3.9464	2.7983	0.4701	0.00127		18.000	34.071	34.071	3.5828	2.3476	1.5261	2.2329	0.81591	
								20.000	36.335	36.335	4.0273	2.5095	1.6048	2.1205	0.76919	
								22.000	38.718	38.718	4.5073	2.6690	1.6888	2.0061	0.72014	
								24.000	41.238	41.238	5.0245	2.8252	1.7784	1.8893	0.66984	
								26.000	43.920	43.920	5.5816	2.9780	1.8743	1.7695	0.61919	
								28.000	46.811	46.811	6.1840	3.1274	1.9774	1.6454	0.56880	
								30.000	49.994	49.994	6.8427	3.2747	2.0895	1.5144	0.51885	
								32.000	53.561	53.561	7.5832	3.4233	2.2152	1.3711	0.46873	
								34.000	58.350	58.350	8.4906	3.5846	2.3686	1.1976	0.41516	
								35.327	65.428	65.428	9.7141	3.7727	2.5748	0.9585	0.35463	
								34.000	71.408	71.408	10.5657	3.8872	2.7181	0.7791	0.31914	
								32.000	74.475	74.475	10.9242	3.9320	2.7783	0.6967	0.30560	
								30.000	76.526	76.526	11.1314	3.9570	2.8131	0.6461	0.29812	
								28.000	78.130	78.130	11.2746	3.9739	2.8372	0.6096	0.29310	
								26.000	79.475	79.475	11.3814	3.9864	2.8551	0.5812	0.28942	
								24.000	80.646	80.646	11.4644	3.9959	2.8690	0.5585	0.28660	

### Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$
3.20	22.000	81.694	11.5307	4.0035	2.8802	0.5398	0.28438	3.30	2.000	19.009	1.1812	1.1262	1.0489	3.1858	0.99953
	20.000	82.649	11.5844	4.0096	2.8892	0.5243	0.28260		4.000	20.475	1.3880	1.2626	1.0993	3.0748	0.99642
	18.000	83.533	11.6285	4.0146	2.8966	0.5113	0.28115		6.000	22.039	1.6222	1.4082	1.1520	2.9653	0.98858
	16.000	84.363	11.6647	4.0187	2.9026	0.5004	0.27996		8.000	23.699	1.8859	1.5617	1.2076	2.8563	0.97453
	14.000	85.147	11.6945	4.0220	2.9076	0.4913	0.27899		10.000	25.457	2.1807	1.7216	1.2666	2.7468	0.95347
	12.000	85.897	11.7188	4.0247	2.9117	0.4837	0.27820		12.000	27.310	2.5078	1.8861	1.3296	2.6364	0.92526
	10.000	86.619	11.7385	4.0269	2.9150	0.4776	0.27757		14.000	29.261	2.8688	2.0536	1.3970	2.5248	0.89031
	8.000	87.320	11.7539	4.0286	2.9176	0.4727	0.27707		16.000	31.308	3.2640	2.2219	1.4690	2.4118	0.84954
	6.000	88.003	11.7655	4.0299	2.9196	0.4690	0.27669		18.000	33.456	3.6947	2.3898	1.5460	2.2974	0.80409
	4.000	88.675	11.7736	4.0308	2.9209	0.4664	0.27643		20.000	35.710	4.1617	2.5557	1.6284	2.1813	0.75527
2.000	89.340	11.7784	4.0313	2.9217	0.4649	0.27628		22.000	38.077	4.6655	2.7184	1.7163	2.0636	0.70444	
3.25	2.000	19.293	1.1786	1.1244	1.0482	3.1380	0.99955		24.000	40.573	5.2081	2.8773	1.8101	1.9439	0.65272
	4.000	20.762	1.3818	1.2586	1.0979	3.0290	0.99656		26.000	43.222	5.7918	3.0318	1.9103	1.8215	0.60108
	6.000	22.328	1.6119	1.4019	1.1498	2.9215	0.98902		28.000	46.062	6.4212	3.1822	2.0178	1.6955	0.55020
	8.000	23.990	1.8704	1.5530	1.2044	2.8145	0.97549		30.000	49.163	7.1057	3.3294	2.1342	1.5638	0.50034
	10.000	25.749	2.1590	1.7103	1.2624	2.7070	0.95518		32.000	52.667	7.8658	3.4758	2.2630	1.4218	0.45116
	12.000	27.604	2.4791	1.8722	1.3242	2.5986	0.92789		34.000	56.963	8.7622	3.6291	2.4144	1.2575	0.40064
	14.000	29.556	2.8318	2.0370	1.3901	2.4889	0.89402		35.882	65.518	10.3564	3.8602	2.6829	0.9606	0.32741
	16.000	31.606	3.2179	2.2030	1.4607	2.3779	0.85437		34.000	72.501	11.3896	4.0230	2.9092	0.7502	0.27869
	18.000	33.757	3.6384	2.3687	1.5360	2.2653	0.81004		30.000	77.029	11.8983	4.0445	2.9418	0.6336	0.27247
	20.000	36.016	4.0940	2.5326	1.6165	2.1511	0.76227		28.000	78.535	12.0364	4.0595	2.9650	0.5993	0.26817
3.30	2.000	38.390	4.5858	2.6937	1.7024	2.0350	0.71232		26.000	79.812	12.1408	4.0706	2.9825	0.5725	0.26497
	4.000	40.898	5.1156	2.8513	1.7941	1.9168	0.66129		24.000	80.932	12.2227	4.0793	2.9963	0.5528	0.26251
	6.000	43.563	5.6858	3.0049	1.8922	1.7958	0.61015		22.000	81.938	12.2884	4.0862	3.0073	0.5328	0.26055
	8.000	46.426	6.3015	3.1548	1.9974	1.6707	0.55950		20.000	82.859	12.3420	4.0918	3.0163	0.5178	0.25896
	10.000	49.566	6.9727	3.3020	2.1116	1.5394	0.50960		18.000	83.714	12.3860	4.0964	3.0236	0.5052	0.25767
	12.000	53.141	7.7223	3.4494	2.2387	1.3970	0.45998		16.000	84.517	12.4223	4.1001	3.0297	0.4946	0.25662
	14.000	57.616	8.6213	3.6062	2.3907	1.2287	0.40809		14.000	85.278	12.4523	4.1032	3.0348	0.4858	0.25575
	16.000	65.473	9.6473	3.8170	2.6285	2.7875	0.34078		12.000	86.007	12.4767	4.1057	3.0389	0.4785	0.25504
	18.000	71.993	10.9786	3.9386	2.8434	2.6285	0.29180		10.000	86.708	12.4964	4.1077	3.0422	0.4725	0.25448
	20.000	74.827	11.3120	3.9783	2.8434	2.6285	0.26361		8.000	87.390	12.5120	4.1093	3.0448	0.4677	0.25403
3.35	2.000	76.787	11.5124	4.0014	2.8771	0.6396	0.28499		6.000	88.056	12.5237	4.1105	3.0467	0.4641	0.25369
	4.000	78.339	11.6529	4.0173	2.9007	0.6043	0.28035		4.000	88.710	12.5319	4.1114	3.0481	0.4616	0.25346
	6.000	79.649	11.7584	4.0291	2.9184	0.5767	0.27692		2.000	89.357	12.5367	4.1119	3.0489	0.4601	0.25332
	8.000	80.793	11.8408	4.0382	2.9322	0.5545	0.27429								
	10.000	81.819	11.9067	4.0454	2.9433	0.5362	0.27220								
	12.000	82.757	11.9604	4.0513	2.9523	0.5210	0.27052								
	14.000	83.626	12.0044	4.0560	2.9596	0.5082	0.26916								
	16.000	84.442	12.0407	4.0599	2.9657	0.4974	0.26804								
	18.000	85.214	12.0705	4.0631	2.9707	0.4885	0.26712								
	20.000	85.953	12.0949	4.0658	2.9748	0.4810	0.26637								
3.40	2.000	86.665	12.1145	4.0679	2.9781	0.4750	0.26577								
	4.000	87.356	12.1300	4.0695	2.9807	0.4702	0.26530								
	6.000	88.030	12.1417	4.0707	2.9827	0.4665	0.26495								
	8.000	88.693	12.1498	4.0716	2.9840	0.4639	0.26470								
	10.000	89.348	12.1547	4.0721	2.9848	0.4624	0.26455								
	12.000	90.000	12.1585	4.0726	2.9853	0.4610	0.26440								
	14.000	90.648	12.1613	4.0730	2.9857	0.4600	0.26430								
	16.000	91.285	12.1632	4.0733	2.9860	0.4593	0.26420								
	18.000	91.912	12.1643	4.0735	2.9862	0.4589	0.26415								
	20.000	92.530	12.1648	4.0736	2.9863	0.4587	0.26410								

### Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$\frac{P_2}{P_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$
3.35	24.000	40.264	5.3024	1.8263	1.9704	0.64409	3.40	34.000	73.352	12.2131	2.9946	4.0783	0.7279	0.26279
	26.000	42.898	5.8998	1.9288	1.8468	0.59200		32.000	75.717	12.4992	3.0426	4.1080	0.6653	0.25440
	28.000	45.716	6.5433	2.0386	1.7198	0.54090		30.000	77.467	12.6849	3.0738	4.1268	0.5902	0.24914
	30.000	48.782	7.2416	2.1573	1.5874	0.49109		28.000	78.891	12.8193	3.0963	4.1402	0.52452	0.24263
	32.000	52.225	8.0134	2.2880	1.4458	0.44232		26.000	80.110	12.9221	3.1135	4.1503	0.5646	0.24046
	34.000	56.375	8.9114	2.4396	1.2844	0.39294		24.000	81.185	13.0033	3.1271	4.1582	0.5437	0.23872
	36.000	63.380	10.2976	2.6730	1.0339	0.32979		22.000	82.156	13.0688	3.1381	4.1645	0.5264	0.23732
	36.143	65.562	10.6853	2.7382	0.9616	0.31454		20.000	83.047	13.1224	3.1471	4.1697	0.5119	0.23617
	36.000	67.623	11.0286	2.7958	0.8957	0.30180		18.000	83.876	13.1665	3.1545	4.1739	0.4997	0.23522
	34.000	72.950	11.8006	2.9255	0.7384	0.27557		16.000	84.656	13.2030	3.1606	4.1774	0.4894	0.23445
	32.000	75.444	12.0992	2.9755	0.6723	0.26624		14.000	85.396	13.2331	3.1657	4.1802	0.4808	0.23381
	30.000	77.255	12.2891	3.0074	0.6279	0.26053		12.000	86.105	13.2578	3.1698	4.1826	0.4736	0.23330
	28.000	78.719	12.4252	3.0302	0.5946	0.25653		10.000	86.789	13.2777	3.1731	4.1844	0.4678	0.23280
	26.000	79.965	12.5287	3.0476	0.5684	0.25355		8.000	87.453	13.2934	3.1757	4.1859	0.4632	0.23239
	24.000	81.062	12.6102	3.0612	0.5471	0.25124		6.000	88.103	13.3052	3.1777	4.1870	0.4596	0.23227
	22.000	82.050	12.6758	3.0722	0.5295	0.24939		4.000	88.741	13.3135	3.1791	4.1878	0.4572	0.23227
	20.000	82.956	12.7293	3.0812	0.5148	0.24790		2.000	89.372	13.3184	3.1799	4.1883	0.4557	0.23227
	18.000	83.798	12.7734	3.0886	0.5024	0.24668								
	16.000	84.588	12.8098	3.0947	0.4920	0.24568								
	14.000	85.339	12.8398	3.0997	0.4832	0.24486								
	12.000	86.057	12.8644	3.1038	0.4760	0.24420								
	10.000	86.750	12.8842	3.1072	0.4701	0.24366								
	8.000	87.422	12.8998	3.1098	0.4654	0.24324								
	6.000	88.080	12.9116	3.1118	0.4618	0.24292								
	4.000	88.726	12.9198	3.1131	0.4593	0.24270								
	2.000	89.365	12.9246	3.1140	0.4578	0.24256								
							3.45	2.000	18.209	1.1892	1.0509	1.1316	3.3292	0.99947
								4.000	19.668	1.4063	1.1036	1.2743	3.2118	0.99597
								6.000	21.226	1.6536	1.1588	1.4270	3.0962	0.98718
								8.000	22.884	1.9331	1.2172	1.5881	2.9809	0.97149
								10.000	24.639	2.2468	1.2796	1.7559	2.8653	0.94812
								12.000	26.491	2.5962	1.3463	1.9284	2.7486	0.91701
								14.000	28.438	2.9823	1.4178	2.1035	2.6309	0.87878
								16.000	30.481	3.4063	1.4946	2.2791	2.5118	0.83456
								18.000	32.621	3.8688	1.5769	2.4535	2.3915	0.78577
								20.000	34.863	4.3706	1.6649	2.6251	2.2698	0.73391
								22.000	37.213	4.9123	1.7590	2.7926	2.1468	0.68049
								24.000	39.683	5.4951	1.8595	2.9552	2.0224	0.62680
								26.000	42.292	6.1211	1.9666	3.1125	1.8960	0.57385
								28.000	45.073	6.7941	2.0813	3.2644	1.7667	0.52235
								30.000	48.080	7.5215	2.2047	3.4115	1.6329	0.47267
								32.000	51.420	8.3194	2.3397	3.5558	1.4914	0.42466
								34.000	55.344	9.2294	2.4932	3.7018	1.3339	0.37715
								36.000	60.903	10.4358	2.6962	3.8705	1.1265	0.32424
								36.635	65.647	11.3584	2.8512	3.9837	0.9634	0.29020
								36.000	69.850	12.0718	2.9709	4.0633	0.8302	0.26708
								34.000	73.716	12.6278	3.0642	4.1211	0.7184	0.25074
								32.000	75.970	12.9035	3.1104	4.1485	0.6589	0.24313
								30.000	77.665	13.0858	3.1633	4.1662	0.6175	0.23828
								28.000	79.054	13.2189	3.1804	4.1789	0.5860	0.23481
								26.000	80.246	13.3210	3.1939	4.1885	0.5609	0.23220
								24.000	81.302	13.4020	3.1939	4.1961	0.5404	0.23016
								22.000	82.256	13.4675	3.2049	4.2021	0.5234	0.22852
								20.000	83.134	13.5211	3.2139	4.2071	0.5091	0.22719
								18.000	83.951	13.5654	3.2213	4.2111	0.4971	0.22611
3.40	2.000	18.467	1.1866	1.0502	3.2814	0.99949								
	4.000	19.928	1.4001	1.1022	3.1662	0.99613								
	6.000	21.488	1.6430	1.1565	3.0527	0.98766								
	8.000	23.147	1.9173	1.2140	2.9395	0.97253								
	10.000	24.902	2.2245	1.2752	2.8260	0.94995								
	12.000	26.755	2.5664	1.3407	2.7115	0.91981								
	14.000	28.702	2.9440	1.4108	2.5958	0.88269								
	16.000	30.746	3.3583	1.4860	2.4788	0.83962								
	18.000	32.889	3.8100	1.5665	2.3604	0.79194								
	20.000	35.133	4.2998	1.6526	2.2407	0.74110								
	22.000	37.489	4.8289	1.7446	2.1195	0.68851								
	24.000	39.967	5.3980	1.8428	1.9966	0.63546								
	26.000	42.588	6.0096	1.9476	1.8716	0.58292								
	28.000	45.386	6.6675	2.0598	1.7435	0.53162								
	30.000	48.422	7.3802	2.1808	1.6105	0.48186								
	32.000	51.810	8.1645	2.3135	1.4690	0.43348								
	34.000	55.838	9.0673	2.4659	1.3098	0.38509								
	36.000	61.914	10.3308	2.6786	1.0874	0.32845								
	36.393	65.605	11.0193	2.7943	0.9625	0.30214								
	36.000	68.960	11.5817	2.8887	0.8560	0.28269								



### Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$
3.45	16.000	84.720	13.6020	4.2145	3.2275	0.4869	0.22521	3.55	2.000	17.715	1.1947	1.1353	1.0523	3.4246	0.99943
	14.000	85.451	13.6322	4.2172	3.2325	0.4784	0.22448		4.000	19.170	1.4187	1.2822	1.1065	3.3029	0.99566
	12.000	86.151	13.6570	4.2195	3.2367	0.4714	0.22388		6.000	20.726	1.6748	1.4396	1.1634	3.1829	0.98619
	10.000	86.826	13.6770	4.2213	3.2400	0.4656	0.22340		8.000	22.383	1.9653	1.6059	1.2238	3.0633	0.96935
	8.000	87.482	13.6928	4.2227	3.2427	0.4610	0.22300		10.000	24.138	2.2920	1.7791	1.2883	2.9435	0.94435
	6.000	88.125	13.7047	4.2238	3.2447	0.4575	0.22273		12.000	25.989	2.6566	1.9569	1.3576	2.8224	0.91123
	4.000	88.756	13.7130	4.2245	3.2461	0.4551	0.22253		14.000	27.936	3.0603	2.1370	1.4321	2.7003	0.87077
	2.000	89.379	13.7180	4.2250	3.2469	0.4536	0.22241		16.000	29.977	3.5040	2.3174	1.5121	2.5771	0.82424
									18.000	32.115	3.9887	2.4961	1.5980	2.4526	0.77322
									20.000	34.352	4.5148	2.6714	1.6901	2.3271	0.71939
3.50	2.000	17.958	1.1920	1.1335	1.0516	3.3769	0.99945		22.000	36.692	5.0827	2.8419	1.7885	2.2005	0.66437
	4.000	19.415	1.4125	1.2783	1.1050	3.2574	0.99582		24.000	39.149	5.6937	3.0069	1.8935	2.0727	0.60946
	6.000	20.972	1.6642	1.4333	1.1611	3.1396	0.98669		26.000	41.738	6.3495	3.1659	2.0056	1.9434	0.55575
	8.000	22.629	1.9491	1.5970	1.2205	3.0222	0.97044		28.000	44.488	7.0535	3.3187	2.1254	1.8117	0.50395
	10.000	24.384	2.2693	1.7675	1.2839	2.9044	0.94626		30.000	47.447	7.8120	3.4660	2.2539	1.6762	0.45445
	12.000	26.236	2.6262	1.9426	1.3519	2.7856	0.91415		32.000	50.705	8.6392	3.6092	2.3937	1.5342	0.40714
	14.000	28.182	3.0211	2.1202	1.4249	2.6657	0.87481		34.000	54.463	9.5691	3.7520	2.5504	1.3790	0.36118
	16.000	30.225	3.4549	2.2982	1.5033	2.5445	0.82942		36.000	59.399	10.7262	3.9075	2.7450	1.1885	0.31299
	18.000	32.363	3.9283	2.4747	1.5874	2.4222	0.77952		37.091	65.729	12.0520	4.0612	2.9676	0.9651	0.26768
	20.000	34.602	4.4421	2.6482	1.6774	2.2986	0.72668		38.000	71.121	13.4667	4.2021	3.2048	0.7943	0.24063
22.000	36.947	4.9969	2.8173	1.7737	2.1739	0.67245		39.000	74.353	13.7265	4.2257	3.2483	0.7473	0.22221	
24.000	39.410	5.5936	2.9811	1.8764	2.0478	0.61813		40.000	76.427	13.9033	4.2415	3.2779	0.6983	0.21803	
26.000	42.009	6.2345	3.1392	1.9860	1.9199	0.56478		41.000	78.025	14.0342	4.2530	3.2998	0.6582	0.21501	
28.000	44.774	6.9227	3.2916	2.1032	1.7894	0.51313		42.000	79.351	14.1355	4.2618	3.3168	0.6241	0.21271	
30.000	47.755	7.6554	3.4388	2.2291	1.6549	0.46353		43.000	80.497	14.2033	4.2687	3.3303	0.5943	0.21090	
32.000	51.053	8.4777	3.5825	2.3664	1.5131	0.41586		44.000	81.517	14.2463	4.2743	3.3413	0.5678	0.20944	
34.000	54.888	9.3968	3.7268	2.5214	1.3570	0.36917		45.000	82.442	14.2819	4.2789	3.3503	0.5431	0.20826	
36.000	60.090	10.5715	3.8879	2.7191	1.1594	0.31891		46.000	83.294	14.3358	4.2827	3.3578	0.5209	0.20729	
36.867	65.689	11.7027	4.0229	2.9090	0.9643	0.27872		47.000	84.090	14.3804	4.2858	3.3640	0.4922	0.20649	
38.000	70.545	12.5396	4.1121	3.0494	0.8105	0.25324		48.000	84.839	14.4173	4.2883	3.3691	0.4740	0.20583	
40.000	76.207	13.0455	4.1623	3.1342	0.7098	0.23934		49.000	85.552	14.4478	4.2904	3.3733	0.4671	0.20529	
42.000	77.851	13.4920	4.2044	3.1790	0.6529	0.23241		50.000	86.235	14.4729	4.2921	3.3767	0.4615	0.20485	
44.000	79.207	13.6238	4.2165	3.2311	0.5820	0.22468		51.000	86.895	14.4931	4.2934	3.3794	0.4570	0.20451	
46.000	80.375	13.7255	4.2256	3.2481	0.5574	0.22223		52.000	87.537	14.5091	4.2944	3.3814	0.4535	0.20425	
48.000	81.413	13.8064	4.2329	3.2617	0.5373	0.22031		53.000	88.165	14.5212	4.2951	3.3828	0.4511	0.20407	
50.000	82.352	13.8719	4.2387	3.2727	0.5205	0.21877		54.000	88.782	14.5296	4.2956	3.3836	0.4497	0.20397	
52.000	83.216	13.9256	4.2435	3.2817	0.5065	0.21751									
54.000	84.022	13.9700	4.2474	3.2891	0.4946	0.21649									
56.000	84.781	14.0067	4.2506	3.2952	0.4846	0.21564		3.60	17.479	1.1973	1.1371	1.0530	3.4722	0.99940	
58.000	85.503	14.0371	4.2532	3.3003	0.4762	0.21494		4.000	18.932	1.4250	1.2862	1.1079	3.3482	0.99549	
60.000	86.194	14.0620	4.2554	3.3045	0.4692	0.21438		6.000	20.488	1.6857	1.4461	1.1657	3.2260	0.98567	
62.000	86.862	14.0822	4.2572	3.3079	0.4635	0.21392		8.000	22.144	1.9816	1.6149	1.2271	3.1043	0.96824	
64.000	87.510	14.0980	4.2585	3.3105	0.4590	0.21356		10.000	23.899	2.3149	1.7907	1.2927	2.9821	0.94241	
66.000	88.145	14.1100	4.2596	3.3125	0.4555	0.21329		12.000	25.751	2.6873	1.9711	1.3633	2.8590	0.90827	
68.000	88.769	14.1184	4.2603	3.3139	0.4531	0.21310		14.000	27.698	3.0999	2.1538	1.4393	2.7347	0.86667	
70.000	89.386	14.1234	4.2607	3.3148	0.4516	0.21298		16.000	29.740	3.5540	2.3366	1.5210	2.6092	0.81895	
								18.000	31.876	4.0498	2.5174	1.6088	2.4827	0.76685	



### Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$
3.70	20.000	83.507	15.6008	3.5621	0.4969	0.18289	3.75	2.000	89.416	16.2379	3.6687	0.4428	0.17169
	18.000	84.274	15.6460	3.5696	0.4856	0.18206							
	16.000	84.998	15.6836	3.5759	0.4760	0.18138	3.80	2.000	16.600	1.2083	1.0558	3.6624	0.99931
	14.000	85.687	15.7147	3.5811	0.4680	0.18082		4.000	18.048	1.4503	1.1137	3.5291	0.99479
	12.000	86.348	15.7402	3.5854	0.4613	0.18035		6.000	19.602	1.7294	1.1750	3.3978	0.98349
	10.000	86.988	15.7609	3.5889	0.4558	0.17998		8.000	21.258	2.0480	1.2404	3.2669	0.96355
	8.000	87.610	15.7772	3.5916	0.4515	0.17969		10.000	23.016	2.4088	1.3108	3.1354	0.93423
	6.000	88.219	15.7896	3.5937	0.4481	0.17947		12.000	24.872	2.8134	1.3867	3.0031	0.89586
	4.000	88.817	15.7982	3.5951	0.4458	0.17932		14.000	26.821	3.2631	1.4688	2.8697	0.84963
	2.000	89.411	15.8033	3.5960	0.4444	0.17922		16.000	28.864	3.7592	1.5575	2.7353	0.79728
								18.000	31.000	4.3021	1.6530	2.6001	0.74088
3.75	2.000	16.810	1.2055	1.0551	3.6149	0.99933		33.229	2.7867	1.7556	1.7556	2.4644	0.68241
	4.000	18.260	1.4440	1.1123	3.4840	0.99497		35.556	2.9644	1.8654	1.8654	2.3283	0.62373
	6.000	19.814	1.7184	1.1727	3.3550	0.98405		37.989	3.1348	1.9828	1.9828	2.1919	0.56627
	8.000	21.470	2.0312	1.2370	3.2264	0.96476		40.542	3.2975	2.1080	2.1080	2.0548	0.51113
	10.000	23.227	2.3849	1.3062	3.0974	0.93634		43.234	3.4523	2.2414	1.9166	1.9166	0.45902
	12.000	25.081	2.7813	1.3808	2.9674	0.89905		46.105	3.5997	2.3840	1.7761	1.7761	0.41022
	14.000	27.030	3.2217	1.4614	2.8363	0.85397		49.218	3.7408	2.5375	1.6313	1.6313	0.36471
	16.000	29.072	3.7069	1.5482	2.7042	0.80280		52.702	3.8780	2.7060	1.4778	1.4778	0.32194
	18.000	31.207	4.2379	1.6417	2.5712	0.74744		56.894	4.0175	2.9009	1.3044	1.3044	0.28030
	20.000	33.438	4.8148	1.7422	2.4376	0.68987		64.192	4.2039	3.2082	1.0293	1.0293	0.22804
	22.000	35.767	5.4382	1.8497	2.3034	0.63185		65.921	4.2990	3.2733	0.9690	0.9690	0.21868
	24.000	38.204	6.1086	1.9645	2.1688	0.57486		67.568	4.2696	3.3321	0.9133	0.9133	0.21066
	26.000	40.762	6.8272	2.0869	2.0333	0.51996		73.114	4.2696	3.3822	0.8670	0.8670	0.20322
	28.000	43.464	7.5969	2.2175	1.8964	0.46786		75.572	4.3822	3.4397	0.8238	0.8238	0.19702
	30.000	46.350	8.4228	2.3572	1.7570	0.41888		78.762	4.4120	3.4927	0.7862	0.7862	0.19150
	32.000	49.486	9.3159	2.5078	1.6129	0.37300		79.967	4.4212	3.5421	0.7519	0.7519	0.18628
	34.000	53.014	10.3013	2.6736	1.4594	0.32964		81.022	4.4284	3.5876	0.7286	0.7286	0.18116
	36.000	57.310	11.4538	2.8672	1.2839	0.28696		81.969	4.4341	3.6356	0.7069	0.7069	0.17680
	37.906	65.884	13.5007	3.2105	1.0963	0.22770		82.833	4.4387	3.6876	0.6876	0.6876	0.17280
	36.000	72.794	14.8041	3.4287	0.7481	0.19834		83.634	4.4426	3.7351	0.6700	0.6700	0.16980
	34.000	75.361	15.1917	3.4936	0.6755	0.19061		84.383	4.4457	3.7820	0.6558	0.6558	0.16780
	32.000	77.180	15.4318	3.5338	0.6280	0.18602		85.092	4.4484	3.8220	0.6444	0.6444	0.16644
	30.000	78.631	15.6021	3.5623	0.5926	0.18286		85.767	4.4505	3.8673	0.6356	0.6356	0.16594
	28.000	79.856	15.7307	3.5838	0.5649	0.18053		86.415	4.4523	3.9161	0.6286	0.6286	0.16552
	26.000	80.927	15.8316	3.6007	0.5423	0.17872		87.043	4.4537	3.9688	0.6231	0.6231	0.16518
	24.000	81.887	15.9128	3.6143	0.5237	0.17728		87.653	4.4549	4.0259	0.6189	0.6189	0.16492
	22.000	82.762	15.9792	3.6254	0.5080	0.17612		88.251	4.4557	4.0883	0.6152	0.6152	0.16472
	18.000	83.572	16.0399	3.6345	0.4948	0.17517		88.839	4.4563	4.1557	0.6126	0.6126	0.16458
	16.000	84.330	16.0794	3.6422	0.4836	0.17439		89.421	4.4567	4.2283	0.6101	0.6101	0.16450
	14.000	85.045	16.1172	3.6485	0.4741	0.17374							
	12.000	85.727	16.1485	3.6537	0.4662	0.17321							
	10.000	86.382	16.1743	3.6580	0.4595	0.17277							
	8.000	87.016	16.1951	3.6615	0.4541	0.17242	3.85	2.000	16.395	1.2110	1.0564	3.7099	0.99928
	6.000	87.632	16.2116	3.6643	0.4498	0.17214		4.000	17.843	1.4568	1.1152	3.5741	0.99460
	4.000	88.235	16.2240	3.6663	0.4465	0.17193		6.000	19.396	1.7405	1.1783	3.4404	0.98291
		88.829	16.2327	3.6678	0.4441	0.17178		8.000	21.053	2.0650	1.2498	3.3071	0.96231



### Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	
3.85	10.00	22.812	2.4328	1.3153	3.1734	0.93209	3.90	24.000	37.584	6.4345	2.0201	3.1853	2.0201	2.2371	0.54918	
	12.00	24.668	2.8456	1.3927	3.0386	0.89264		26.000	40.126	7.2035	2.1508	3.3492	2.1508	2.0968	0.49366	
	14.00	26.619	3.3050	1.4764	2.9028	0.84523		28.000	42.802	8.0258	2.2901	3.5046	2.2901	1.9558	0.44158	
	16.00	28.664	3.8121	1.5668	2.7661	0.79172		30.000	45.646	8.9059	2.4387	3.6519	2.4387	1.8131	0.39322	
	18.00	30.799	4.3670	1.6643	2.6287	0.73428		32.000	48.716	9.8536	2.5983	3.7923	2.5983	1.6668	0.34848	
	20.00	33.028	4.9706	1.7691	2.4909	0.67493		34.000	52.126	10.8901	3.9278	2.7726	3.9278	1.5130	0.30686	
	22.00	35.353	5.6230	1.8814	2.3529	0.61558		36.000	56.149	12.0723	4.0633	2.9710	4.0633	1.3425	0.26706	
	24.00	37.783	6.3245	2.0013	2.2146	0.55770		38.000	62.087	13.6897	4.2224	3.2421	4.2224	1.1106	0.22309	
	26.00	40.330	7.0764	2.1293	2.0760	0.50236		38.445	65.991	14.6407	4.3043	3.4014	4.3043	0.9704	0.20173	
	28.00	43.014	7.8808	2.2656	1.9364	0.45026		38.000	69.501	15.4023	4.3647	3.5289	4.3647	0.8527	0.18658	
	30.00	45.871	8.7425	2.4111	1.7948	0.40167		36.000	73.678	16.1768	4.4218	3.6584	4.4218	0.7240	0.17273	
	32.00	48.961	9.6715	2.5677	1.6493	0.35654		34.000	75.956	16.5334	4.4688	3.7181	4.4688	0.6600	0.16682	
	34.00	52.407	10.6904	2.7390	1.4957	0.31434		32.000	77.640	16.7653	4.4626	3.7569	4.4626	0.6160	0.16313	
	36.00	56.508	11.8605	2.9355	1.3239	0.27366		30.000	79.006	16.9330	4.4738	3.7849	4.4738	0.5828	0.16052	
	38.00	62.939	13.5472	3.2183	1.0767	0.22655		28.000	80.172	17.0613	4.4823	3.8064	4.4823	0.5563	0.15857	
	38.272	65.956	14.2556	3.3369	0.9697	0.21003		26.000	81.199	17.1629	4.4890	3.8234	4.4890	0.5347	0.15705	
	38.000	68.733	14.8512	3.4366	0.8764	0.19738		24.000	82.121	17.2449	4.4943	3.8371	4.4943	0.5168	0.15583	
	36.000	73.407	15.7160	3.5814	0.7314	0.18079		22.000	82.966	17.3122	4.4986	3.8483	4.4986	0.5016	0.15485	
	34.000	75.770	16.0813	3.6425	0.6649	0.17436		20.000	83.749	17.3680	4.5022	3.8576	4.5022	0.4888	0.15404	
	32.000	77.495	16.3155	3.6816	0.6198	0.17039		18.000	84.483	17.4143	4.5052	3.8654	4.5052	0.4780	0.15337	
30.000	78.888	16.4839	3.7098	0.5859	0.16762		16.000	85.177	17.4529	4.5076	3.8718	4.5076	0.4688	0.15281		
28.000	80.072	16.6122	3.7313	0.5591	0.16555		14.000	85.840	17.4850	4.5097	3.8772	4.5097	0.4610	0.15235		
26.000	81.112	16.7135	3.7482	0.5372	0.16394		12.000	86.477	17.5113	4.5114	3.8816	4.5114	0.4545	0.15198		
24.000	82.047	16.7952	3.7619	0.5190	0.16266		10.000	87.093	17.5327	4.5127	3.8852	4.5127	0.4492	0.15167		
22.000	82.901	16.8622	3.7731	0.5037	0.16162		8.000	87.693	17.5496	4.5138	3.8880	4.5138	0.4450	0.15143		
20.000	83.692	16.9175	3.7823	0.4907	0.16076		6.000	88.280	17.5623	4.5146	3.8901	4.5146	0.4418	0.15125		
18.000	84.434	16.9636	3.7900	0.4798	0.16006		4.000	88.858	17.5713	4.5151	3.8916	4.5151	0.4395	0.15113		
16.000	85.136	17.0019	3.7964	0.4705	0.15947		2.000	89.430	17.5766	4.5155	3.8925	4.5155	0.4382	0.15105		
14.000	85.804	17.0337	3.8017	0.4627	0.15899											
12.000	86.447	17.0598	3.8061	0.4561	0.15859											
10.000	87.068	17.0810	3.8097	0.4508	0.15827		3.95	2.000	16.001	1.2166	1.0578	1.1500	1.0578	3.8047	0.99923	
8.000	87.674	17.0978	3.8125	0.4465	0.15802		4.000	17.447	17.447	1.4697	1.1182	1.3144	1.1182	3.6641	0.99421	
6.000	88.266	17.1104	3.8146	0.4433	0.15783		6.000	19.001	19.001	1.7630	1.1821	1.4915	1.1821	3.5255	0.98171	
4.000	88.849	17.1193	3.8161	0.4410	0.15770		8.000	20.660	20.660	2.0992	1.2506	1.6786	1.2506	3.3874	0.95977	
2.000	89.426	17.1245	3.8169	0.4397	0.15762		10.000	22.422	22.422	2.4815	1.3246	1.8734	1.3246	3.2486	0.92768	
							12.000	24.280	24.280	2.9112	1.4048	2.0724	1.4048	3.1090	0.88602	
							14.000	26.234	26.234	3.3902	1.4917	2.2727	1.4917	2.9684	0.83626	
3.90	2.000	16.196	1.2138	1.0571	0.99926		16.000	28.281	28.281	3.9194	1.5858	2.4716	1.5858	2.8270	0.78046	
4.000	17.642	17.642	1.4633	1.1167	0.99441		18.000	30.417	30.417	4.4992	1.6874	2.6664	1.6874	2.6851	0.72095	
6.000	19.196	19.196	1.7517	1.1797	0.98232		20.000	32.646	32.646	5.1304	1.7967	2.8554	1.7967	2.5430	0.65992	
8.000	20.854	20.854	2.0821	1.2472	0.96105		22.000	34.969	34.969	5.8125	1.9139	3.0370	1.9139	2.4010	0.59933	
10.000	22.614	22.614	2.4570	1.3200	0.92990		24.000	37.393	37.393	6.5462	2.0391	3.2103	2.0391	2.2591	0.54068	
12.000	24.472	24.472	2.8783	1.3987	0.88935		26.000	39.929	39.929	7.3323	2.1727	3.3748	2.1727	2.1172	0.48503	
14.000	26.424	26.424	3.3474	1.4840	0.84077		28.000	42.531	42.531	8.1726	2.3149	3.5304	2.3149	1.9748	0.43302	
16.000	28.469	28.469	3.8655	1.5763	0.79671		30.000	45.431	45.431	9.0717	2.4666	3.6778	2.4666	1.8310	0.38488	
18.000	30.605	30.605	4.4329	1.6758	0.72761		32.000	48.483	48.483	10.0386	2.6294	3.8178	2.6294	1.6838	0.34053	
20.000	32.834	32.834	5.0501	1.7828	0.66743		34.000	51.859	51.859	11.0931	2.8067	3.9524	2.8067	1.5299	0.29949	
22.000	35.157	35.157	5.7171	1.8975	0.60746		36.000	55.812	55.812	12.2888	3.0073	4.0863	3.0073	1.3604	0.26054	

### Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$
3.95	38.000	61.406	13.8667	4.2383	3.2718	1.1389	0.21889	4.00	28.000	80.359	17.9765	4.5402	3.9594	0.5513	0.14555
	38.612	66.026	15.0309	4.3358	3.4667	0.9711	0.19376		26.000	81.359	18.0787	4.5464	3.9765	0.5302	0.14419
	38.000	70.101	15.9275	4.4038	3.6167	0.8345	0.17703		24.000	82.261	18.1615	4.5514	3.9903	0.5126	0.14310
	36.000	73.928	16.6412	4.4541	3.7361	0.7172	0.16509		22.000	83.087	18.2296	4.5555	4.0017	0.4978	0.14221
	34.000	76.131	16.9904	4.4776	3.7945	0.6554	0.15965		20.000	83.854	18.2861	4.5588	4.0111	0.4852	0.14148
	32.000	77.777	17.2203	4.4927	3.8330	0.6125	0.15620		18.000	84.574	18.3331	4.5616	4.0190	0.4746	0.14087
	30.000	79.120	17.3877	4.5035	3.8609	0.5798	0.15375		16.000	85.256	18.3723	4.5639	4.0255	0.4655	0.14037
	28.000	80.268	17.5161	4.5117	3.8824	0.5537	0.15191		14.000	85.907	18.4049	4.5659	4.0310	0.4579	0.13996
	26.000	81.281	17.6179	4.5181	3.8994	0.5324	0.15047		12.000	86.533	18.4317	4.5674	4.0355	0.4515	0.13962
	24.000	82.192	17.7003	4.5232	3.9132	0.5147	0.14932		10.000	87.139	18.4535	4.5687	4.0391	0.4463	0.13934
	22.000	83.028	17.7680	4.5274	3.9245	0.4997	0.14838		8.000	87.730	18.4707	4.5697	4.0420	0.4421	0.13912
	20.000	83.803	17.8241	4.5309	3.9339	0.4870	0.14761		6.000	88.307	18.4837	4.5705	4.0442	0.4390	0.13896
	18.000	84.529	17.8708	4.5338	3.9417	0.4762	0.14698		4.000	88.876	18.4928	4.5710	4.0457	0.4367	0.13885
	16.000	85.218	17.9097	4.5362	3.9482	0.4671	0.14645		2.000	89.439	18.4982	4.5713	4.0466	0.4354	0.13878
	14.000	85.874	17.9420	4.5381	3.9536	0.4594	0.14601								
	12.000	86.505	17.9686	4.5398	3.9581	0.4530	0.14566								
	10.000	87.116	17.9902	4.5411	3.9617	0.4477	0.14537								
	8.000	87.711	18.0072	4.5421	3.9645	0.4435	0.14514								
	6.000	88.294	18.0201	4.5429	3.9667	0.4404	0.14497								
	4.000	88.868	18.0291	4.5434	3.9682	0.4381	0.14485								
	2.000	89.435	18.0345	4.5438	3.9691	0.4368	0.14478								
4.00	2.000	15.813	1.2194	1.1519	1.0586	3.8521	0.99920								
	4.000	17.258	1.4763	1.3185	1.1196	3.7089	0.99401								
	6.000	18.812	1.7743	1.4980	1.1844	3.5679	0.98110								
	8.000	20.471	2.1166	1.6879	1.2540	3.4273	0.95845								
	10.000	22.234	2.5061	1.8853	1.3293	3.2860	0.92542								
	12.000	24.095	2.9445	2.0870	1.4109	3.1439	0.88264								
	14.000	26.050	3.4334	2.2898	1.4994	3.0009	0.83170								
	16.000	28.098	3.9741	2.4909	1.5954	2.8570	0.77474								
	18.000	30.236	4.5667	2.6877	1.6991	2.7128	0.71422								
	20.000	32.464	5.2116	2.8782	1.8107	2.5686	0.65240								
	22.000	34.786	5.9090	3.0611	1.9304	2.4246	0.59123								
	24.000	37.208	6.6592	3.2352	2.0583	2.2809	0.53224								
	26.000	39.740	7.4625	3.4002	2.1947	2.1374	0.47648								
	28.000	42.402	8.3215	3.5561	2.3401	1.9935	0.42453								
	30.000	45.224	9.2397	3.7034	2.4949	1.8485	0.37666								
	32.000	48.258	10.2259	3.8430	2.6609	1.7006	0.33272								
	34.000	51.605	11.2995	3.9768	2.8413	1.5463	0.29223								
	36.000	55.495	12.5100	4.1091	3.0444	1.3776	0.25409								
	38.000	60.827	14.0647	4.2556	3.3049	1.1637	0.21432								
	38.774	66.059	15.4261	4.3665	3.5329	0.9717	0.18613								
	38.000	70.601	16.4407	4.4403	3.7026	0.8196	0.16833								
	36.000	74.161	17.1095	4.4855	3.8144	0.7109	0.15785								
	34.000	76.297	17.4525	4.5076	3.8718	0.6511	0.15282								
	32.000	77.908	17.6808	4.5220	3.9099	0.6090	0.14959								
	30.000	79.227	17.8479	4.5324	3.9379	0.5769	0.14729								