

## ENGINEERING TRIPoS PART IIB

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Friday 25 April 2008 9 to 10.30

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Module 4A11

### TURBOMACHINERY II

*Answer not more than two questions.*

*All questions carry the same number of marks.*

*The approximate percentage of marks allocated to each part of a question is indicated in the right margin.*

*Attachment: Compressible Flow Data Book (38 pages).*

#### STATIONERY

Single-sided script paper

#### SPECIAL REQUIREMENTS

Engineering Data Book

CUED approved calculator allowed

**You may not start to read the questions  
printed on the subsequent pages of this  
question paper until instructed that you  
may do so by the Invigilator**

1 (a) What is the streamline curvature term in the Meridional Streamline Curvature equation which governs axisymmetric hub-to-tip flow distributions in turbomachines? Explain carefully under which conditions this term can be neglected in the equation. Give two practical examples when the streamline curvature term is significant and cannot be regarded as negligible.

[30%]

(b) A low speed axial flow turbine stage has a cylindrical flow path with parallel hub and casing lines, with a casing diameter of 0.8 meters and a hub-to-tip ratio of 0.6. It is designed to have a forced vortex distribution at the stator exit,  $V_\theta = Kr$ , where  $K$  is a constant. The flow is uniform at the inlet to the turbine. Use the Simple Radial Equilibrium equation to find the radial distribution of  $V_x$  at the stator exit in terms of the axial velocity at the hub,  $V_{x,hub}$ , for a given uniform flow condition at the stage inlet. State clearly the assumptions you have used. Explain, without calculation, how the value of  $V_{x,hub}$  can be determined by reference to the uniform inlet conditions.

[40%]

(c) For a design which has  $(V_\theta/V_x)_{hub} = 2.0$ , discuss the validity of using the Simple Radial Equilibrium Equation for the turbine designed in (b). Comment on whether the radial distribution of the axial velocity at the stator exit would change if the streamline curvature effect is included and, if so, would the inclusion of the streamline curvature term make the axial velocity distribution more uniform or less uniform?

[30%]

2 (a) (i) Show that the entropy increase in a compressor blade row operating in air with an inlet Mach number  $M_{in}$  can be expressed approximately in terms of the stagnation pressure loss coefficient  $Y_p$  by

$$\Delta S = RY_p \left( 1 - \left( 1 + \frac{\gamma-1}{2} M_{in}^2 \right)^{-\gamma/(\gamma-1)} \right)$$

where  $R$  is the gas constant and  $\gamma$  is the ratio of specific heat capacities. State any assumptions you make in deriving this expression. [20%]

(ii) For a compressor, derive the expression for isentropic efficiency in terms of the entropy increase. State carefully the approximations involved. What is the equivalent expression for a turbine? [10%]

(iii) A transonic air compressor rotor section has a stagnation pressure ratio of 1.8 and an isentropic efficiency of 0.90 when operating at a relative inlet Mach number  $M_{rel,in} = 1.35$  and with a relative inlet stagnation temperature of 366K. The absolute stagnation temperature at the inlet is 288K. It is found that at this condition the section has a single passage shock wave normal to the flow with a shock upstream Mach number  $M_{rel,shock} = 1.5$ . Calculate the loss of efficiency due to the entropy generation across the shock wave and that due to viscous effects. What is the value of  $Y_p$  attributable to the viscous effects? Comment on your results. Take  $\gamma = 1.4$  and  $R = 287 \text{ J kg}^{-1}\text{K}^{-1}$  for air. [30%]

(b) (i) Describe the mechanisms for entropy generation in shrouded and unshrouded turbine blade tips. Discuss why the stage reaction is an important parameter when determining whether to use a shrouded or an unshrouded blade.

[15%]

(ii) Show that across an arbitrary shock wave in a compressor blade passage, the polytropic efficiency can be expressed in terms of the normal component of Mach number upstream of the shock wave  $M_n$  as:

$$\eta_p = \frac{\gamma-1}{\gamma} \frac{\ln \left( 1 + \frac{2\gamma}{\gamma+1} (M_n^2 - 1) \right)}{\ln \left( \frac{\gamma-1}{(\gamma+1)^2} \frac{2}{M_n^2} \left( 1 + \frac{\gamma-1}{2} M_n^2 \right) \left( \frac{2\gamma}{\gamma-1} M_n^2 - 1 \right) \right)}. \quad [25\%]$$

3 (a) A turbine blade at a sweep angle of  $45^\circ$  is shown in Fig.1. Explain, stating any assumptions, why this blade is expected to have a greater profile loss than an unswept design when both blades operate in the same annulus and produce the same turning. Why might a turbine designer need to use such a highly swept blade row?

[30%]

(b) (i) Contours of stagnation pressure loss coefficient at the exit of an unswept turbine blade between the hub and mid-span are shown in Fig.2 (a). Explain, using sketches where appropriate, why the distribution of loss takes this form. Sketch a plot showing the radial variation of the pitchwise-averaged yaw angle at this axial location.

[30%]

(ii) Figure 3 shows the mid-span blade surface pressure distribution for the swept blade of Fig.1. Sketch the expected blade surface pressure distributions close to the hub and close to the casing for this blade. Use your sketches to explain the differences between the exit loss distributions of Fig.2(a) and 2(b). Sketch the exit loss contours between mid-span and the casing for the swept blade.

[40%]

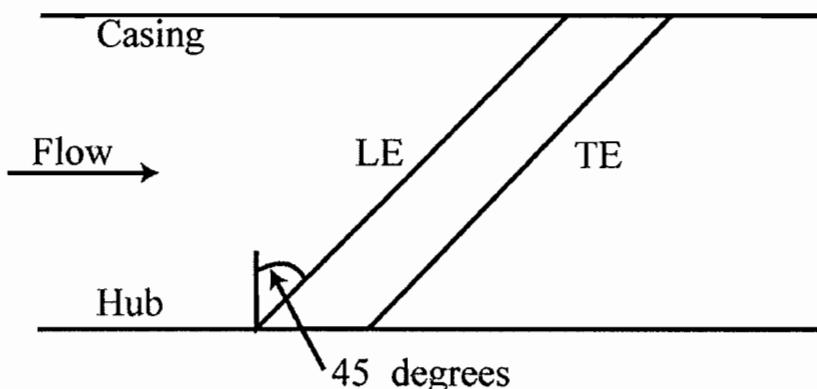


Fig.1: Blade swept at 45 degrees

(cont.)

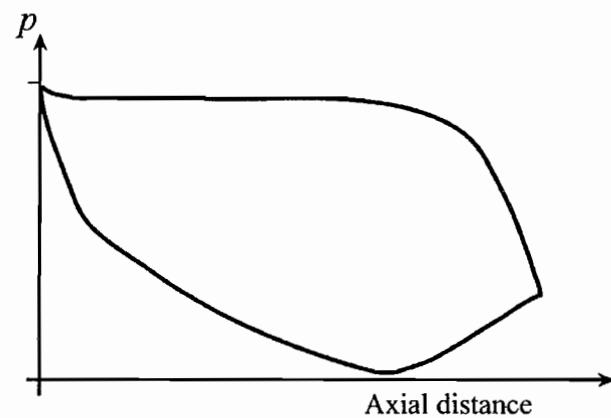
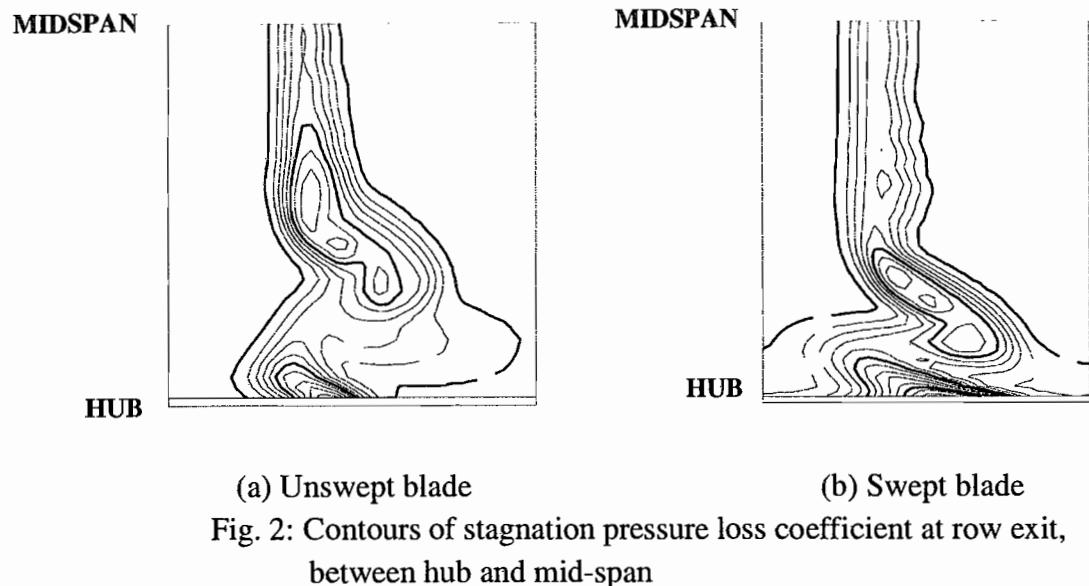


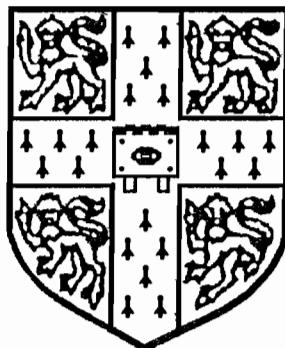
Fig. 3: Mid-span surface pressure distribution for the swept blade

**END OF PAPER**

# Compressible Flow Data Book

for Part II of the  
Engineering Tripos

**2004 Edition**



Cambridge University Engineering Department

# PERFECT GAS RELATIONS FOR COMPRESSIBLE FLOW

**Ratios of stagnation to static quantities**

$$\frac{T}{T_0} = \left(1 + \frac{\gamma-1}{2} M^2\right)^{-1}$$

$$\frac{p}{p_0} = \left(1 + \frac{\gamma-1}{2} M^2\right)^{-\frac{\gamma}{\gamma-1}}$$

$$\frac{\rho}{\rho_0} = \left(1 + \frac{\gamma-1}{2} M^2\right)^{-\frac{1}{\gamma-1}}$$

*Notes:*

- (1)  $T_0 = \text{const.}$  in adiabatic flow with no shaft work
- (2) If flow is isentropic,  $p_0 = \text{const.}$  and  $\rho_0 = \text{const.}$  when  $T_0 = \text{const.}$

**Mach number relations (see tables)**

$$\frac{V}{\sqrt{c_p T_0}} = \sqrt{\gamma-1} M \left(1 + \frac{\gamma-1}{2} M^2\right)^{-\frac{1}{2}}$$

$$\frac{\dot{m} \sqrt{c_p T_0}}{A p_0} = \frac{\gamma}{\sqrt{\gamma-1}} M \left(1 + \frac{\gamma-1}{2} M^2\right)^{-\frac{1}{2} \left(\frac{\gamma+1}{\gamma-1}\right)}$$

$$\frac{\dot{m} \sqrt{c_p T_0}}{A p} = \frac{\gamma}{\sqrt{\gamma-1}} M \left(1 + \frac{\gamma-1}{2} M^2\right)^{\frac{1}{2}}$$

$$\frac{F}{\dot{m} \sqrt{c_p T_0}} = \frac{\sqrt{\gamma-1}}{\gamma} \frac{1 + \gamma M^2}{M} \left(1 + \frac{\gamma-1}{2} M^2\right)^{-\frac{1}{2}} \quad \text{where} \quad F = (p + \rho V^2) A$$

$$\frac{\frac{1}{2} \rho V^2}{p_0} = \frac{1}{2} \gamma M^2 \left(1 + \frac{\gamma-1}{2} M^2\right)^{-\frac{\gamma}{\gamma-1}}$$

# ONE-DIMENSIONAL FLOW OF A PERFECT GAS

**Isentropic flow**

$$\frac{A}{A^*} = \frac{1}{M} \left\{ \frac{2}{\gamma+1} \left( 1 + \frac{\gamma-1}{2} M^2 \right) \right\}^{\frac{1}{2} \left( \frac{\gamma+1}{\gamma-1} \right)}$$

**Adiabatic constant area flow**

$$\frac{4c_f L_{\max}}{D} = \frac{1 - M^2}{\gamma M^2} + \frac{\gamma+1}{2\gamma} \ln \left( \frac{(\gamma+1)M^2}{2 \left( 1 + \frac{\gamma-1}{2} M^2 \right)} \right)$$

**Normal shock waves in perfect gases**

$$VV_s = a^*{}^2$$

$$M_s = \left( \frac{1 + \frac{\gamma-1}{2} M^2}{\gamma M^2 - \frac{\gamma-1}{2}} \right)^{\frac{1}{2}}$$

$$\frac{p_{0s}}{p_0} = \left( \frac{\frac{\gamma+1}{2} M^2}{1 + \frac{\gamma-1}{2} M^2} \right)^{\frac{\gamma}{\gamma-1}} \left( \frac{2\gamma}{\gamma+1} M^2 - \frac{\gamma-1}{\gamma+1} \right)^{\frac{1}{1-\gamma}}$$

$$\frac{p_s}{p} = 1 + \frac{2\gamma}{\gamma+1} (M^2 - 1)$$

$$\frac{p_{0s}}{p} = \left( \frac{\gamma+1}{2} M^2 \right)^{\frac{\gamma}{\gamma-1}} \left( \frac{2\gamma}{\gamma+1} M^2 - \frac{\gamma-1}{\gamma+1} \right)^{\frac{1}{1-\gamma}}$$

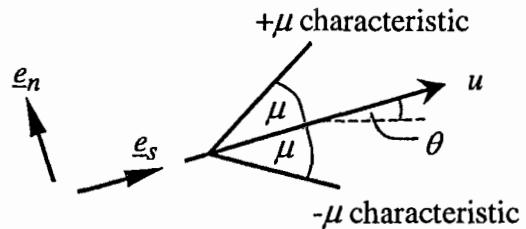
$$\frac{T_s}{T} = \frac{\gamma-1}{(\gamma+1)^2} \frac{2}{M^2} \left( 1 + \frac{\gamma-1}{2} M^2 \right) \left( \frac{2\gamma}{\gamma-1} M^2 - 1 \right)$$

$$\frac{\rho_s}{\rho} = \frac{(\gamma+1)M^2}{2 \left( 1 + \frac{\gamma-1}{2} M^2 \right)}$$

# TWO DIMENSIONAL SUPERSONIC FLOW

## Method of Characteristics for 2-D supersonic flow

Applicable to adiabatic ( $h_0 = \text{constant}$ ), isentropic flow



**Mach Number**

$$M = \frac{u}{c}$$

**Mach angle**

$$\mu = \sin^{-1} \left( \frac{1}{M} \right)$$

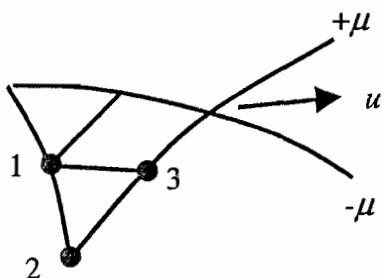
**Prandtl-Meyer function**

$$\nu = \int_1^M \sqrt{M^2 - 1} \frac{du}{u}$$

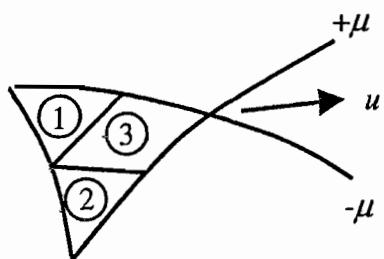
$$\nu = \sqrt{\frac{\gamma+1}{\gamma-1}} \tan^{-1} \sqrt{\frac{\gamma-1}{\gamma+1} (M^2 - 1)} - \tan^{-1} \sqrt{M^2 - 1} \quad \text{for a perfect gas}$$

## Calculations

Lattice Method



Field (or wave) method



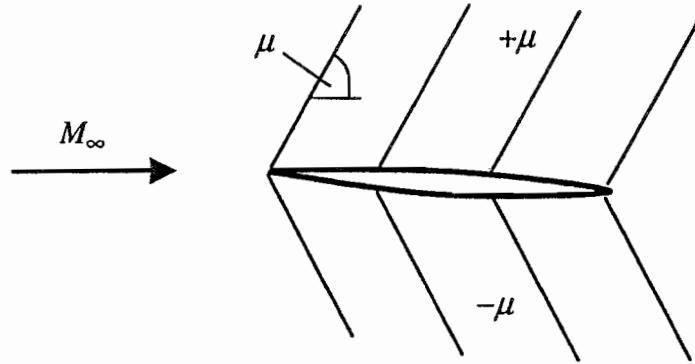
$$\nu_3 - \theta_3 = \nu_2 - \theta_2 \quad \text{along } +\mu$$

$$\nu_3 + \theta_3 = \nu_1 + \theta_1 \quad \text{along } -\mu$$

$$\nu_3 + \theta_3 = \nu_1 + \theta_1 \quad \text{across } +\mu$$

$$\nu_3 - \theta_3 = \nu_2 - \theta_2 \quad \text{across } -\mu$$

### Linearised Method of Characteristics (thin film theory)

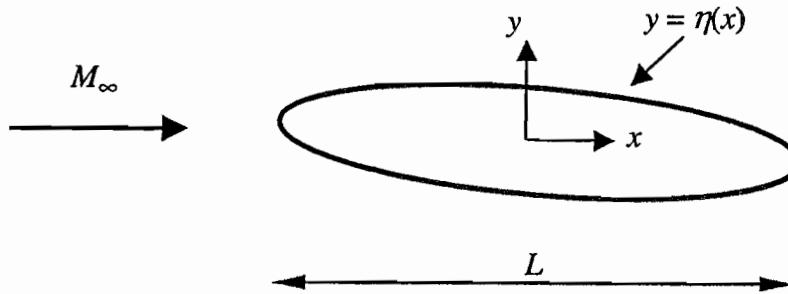


$$\mu \approx \sin^{-1}(1/M_\infty)$$

$$\Delta p \approx \pm \frac{\rho_\infty u_\infty^2 \Delta \theta}{\sqrt{M_\infty^2 - 1}} \quad \text{across } \pm \mu \text{ waves}$$

$$\text{Pressure coefficient } c_p = \frac{p - p_\infty}{\frac{1}{2} \rho_\infty u_\infty^2} = \pm \frac{2\theta}{\sqrt{M_\infty^2 - 1}} \quad \text{on upper/lower surface}$$

**Prandtl-Glauert rule for linearised potential flow past geometrically similar bodies**



$$\text{Pressure coefficient } c_p = \frac{p - p_\infty}{\frac{1}{2} \rho_\infty u_\infty^2}$$

For geometrically similar bodies with  $\frac{\eta}{L} = f\left(\frac{x}{L}\right)$  and  $c_p(M_\infty = 0) = c_{p0}$ ,

$$c_p = \frac{c_{p0}}{\sqrt{1 - M_\infty^2}} \quad \text{in subsonic flow}$$

$$c_p \propto \frac{1}{\sqrt{M_\infty^2 - 1}} \quad \text{in supersonic flow}$$

## Oblique Shock Relations (see tables)

$$\frac{p_2}{p_1} = 1 + \frac{2\gamma}{\gamma+1} (M_1^2 \sin^2 \beta - 1)$$

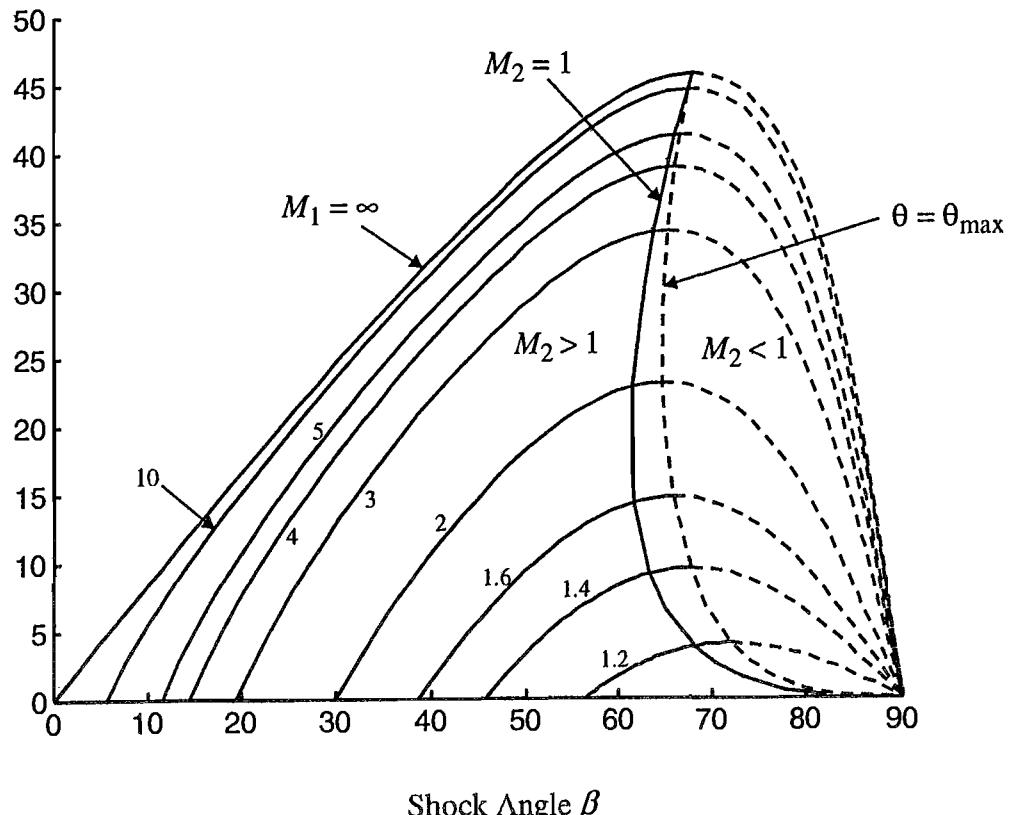
$$\frac{T_2}{T_1} = \frac{\gamma-1}{(\gamma+1)^2} \frac{2}{M_1^2 \sin^2 \beta} \left( 1 + \frac{\gamma-1}{2} M_1^2 \sin^2 \beta \right) \left( \frac{2\gamma}{\gamma-1} M_1^2 \sin^2 \beta - 1 \right)$$

$$\frac{\rho_2}{\rho_1} = \frac{(\gamma+1)M_1^2 \sin^2 \beta}{2 \left[ 1 + \frac{\gamma-1}{2} M_1^2 \sin^2 \beta \right]}$$

$$M_2 \sin(\beta - \theta) = \left[ \frac{2}{1 + \frac{\gamma-1}{2} M_1^2 \sin^2 \beta} \right]^{\frac{1}{2}}$$

$$\frac{p_{02}}{p_{01}} = \left( \frac{\frac{\gamma+1}{2} M_1^2 \sin^2 \beta}{1 + \frac{\gamma-1}{2} M_1^2 \sin^2 \beta} \right)^{\frac{\gamma}{\gamma-1}} \left( \frac{2\gamma}{\gamma+1} M_1^2 \sin^2 \beta - \frac{\gamma-1}{\gamma+1} \right)^{\frac{1}{1-\gamma}}$$

$$\tan \theta = \frac{2 \cot \beta (M_1^2 \sin^2 \beta - 1)}{(\gamma+1)M_1^2 - 2(M_1^2 \sin^2 \beta - 1)}$$



# GAS FLOW TABLES ( $\gamma=1.400$ ): SUBSONIC FLOW

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap}$	$\frac{\dot{m} \sqrt{c_p T_0}}{F}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{p_0}$
0.010	1.0000	0.9999	1.0000	0.0063	0.0221	0.0221	45.1813	7134.405	0.0001
0.020	0.9999	0.9997	0.9998	0.0126	0.0443	0.0443	22.5994	1778.450	0.0003
0.030	0.9998	0.9994	0.9996	0.0190	0.0664	0.0664	15.0761	787.0814	0.0006
0.040	0.9997	0.9989	0.9992	0.0253	0.0885	0.0886	11.3173	440.3522	0.0011
0.050	0.9995	0.9983	0.9988	0.0316	0.1105	0.1107	9.0644	280.0203	0.0017
0.060	0.9993	0.9975	0.9982	0.0379	0.1325	0.1329	7.5645	193.0311	0.0025
0.070	0.9990	0.9966	0.9976	0.0443	0.1545	0.1550	6.4947	140.6550	0.0034
0.080	0.9987	0.9955	0.9968	0.0506	0.1764	0.1772	5.6939	106.7182	0.0045
0.090	0.9984	0.9944	0.9960	0.0569	0.1983	0.1994	5.0723	83.4961	0.0056
0.100	0.9980	0.9930	0.9950	0.0632	0.2200	0.2216	4.5762	66.9216	0.0070
0.110	0.9976	0.9916	0.9940	0.0695	0.2417	0.2438	4.1714	54.6879	0.0084
0.120	0.9971	0.9900	0.9928	0.0758	0.2633	0.2660	3.8350	45.4080	0.0100
0.130	0.9966	0.9883	0.9916	0.0821	0.2849	0.2883	3.5513	38.2070	0.0117
0.140	0.9961	0.9864	0.9903	0.0884	0.3063	0.3105	3.3089	32.5113	0.0135
0.150	0.9955	0.9844	0.9888	0.0947	0.3276	0.3328	3.0996	27.9320	0.0155
0.160	0.9949	0.9823	0.9873	0.1009	0.3488	0.3551	2.9172	24.1978	0.0176
0.170	0.9943	0.9800	0.9857	0.1072	0.3699	0.3774	2.7569	21.1152	0.0198
0.180	0.9936	0.9776	0.9840	0.1135	0.3908	0.3997	2.6151	18.5427	0.0222
0.190	0.9928	0.9751	0.9822	0.1197	0.4116	0.4221	2.4889	16.3752	0.0246
0.200	0.9921	0.9725	0.9803	0.1260	0.4323	0.4445	2.3758	14.5333	0.0272
0.210	0.9913	0.9697	0.9783	0.1322	0.4528	0.4669	2.2740	12.9560	0.0299
0.220	0.9904	0.9668	0.9762	0.1385	0.4731	0.4893	2.1820	11.5961	0.0328
0.230	0.9895	0.9638	0.9740	0.1447	0.4933	0.5118	2.0985	10.4161	0.0357
0.240	0.9886	0.9607	0.9718	0.1509	0.5133	0.5343	2.0225	9.3865	0.0387
0.250	0.9877	0.9575	0.9694	0.1571	0.5332	0.5568	1.9530	8.4834	0.0419
0.260	0.9867	0.9541	0.9670	0.1633	0.5528	0.5794	1.8892	7.6876	0.0451
0.270	0.9856	0.9506	0.9645	0.1695	0.5723	0.6020	1.8306	6.9832	0.0485
0.280	0.9846	0.9470	0.9619	0.1757	0.5915	0.6246	1.7766	6.3572	0.0520
0.290	0.9835	0.9433	0.9592	0.1819	0.6106	0.6473	1.7267	5.7989	0.0555
0.300	0.9823	0.9395	0.9564	0.1881	0.6295	0.6700	1.6805	5.2993	0.0592
0.310	0.9811	0.9355	0.9535	0.1942	0.6481	0.6928	1.6377	4.8507	0.0629
0.320	0.9799	0.9315	0.9506	0.2003	0.6666	0.7156	1.5978	4.4467	0.0668
0.330	0.9787	0.9274	0.9476	0.2065	0.6848	0.7384	1.5608	4.0821	0.0707
0.340	0.9774	0.9231	0.9445	0.2126	0.7027	0.7613	1.5262	3.7520	0.0747
0.350	0.9761	0.9188	0.9413	0.2187	0.7205	0.7842	1.4939	3.4525	0.0788
0.360	0.9747	0.9143	0.9380	0.2248	0.7380	0.8072	1.4637	3.1801	0.0829
0.370	0.9733	0.9098	0.9347	0.2309	0.7553	0.8302	1.4354	2.9320	0.0872
0.380	0.9719	0.9052	0.9313	0.2369	0.7723	0.8532	1.4090	2.7054	0.0915
0.390	0.9705	0.9004	0.9278	0.2430	0.7891	0.8763	1.3841	2.4983	0.0959
0.400	0.9690	0.8956	0.9243	0.2490	0.8056	0.8995	1.3608	2.3085	0.1003
0.410	0.9675	0.8907	0.9207	0.2551	0.8219	0.9227	1.3388	2.1344	0.1048
0.420	0.9659	0.8857	0.9170	0.2611	0.8379	0.9460	1.3182	1.9744	0.1094
0.430	0.9643	0.8807	0.9132	0.2671	0.8536	0.9693	1.2988	1.8272	0.1140
0.440	0.9627	0.8755	0.9094	0.2730	0.8691	0.9927	1.2804	1.6915	0.1186
0.450	0.9611	0.8703	0.9055	0.2790	0.8843	1.0161	1.2632	1.5664	0.1234
0.460	0.9594	0.8650	0.9016	0.2850	0.8992	1.0396	1.2469	1.4509	0.1281
0.470	0.9577	0.8596	0.8976	0.2909	0.9138	1.0631	1.2315	1.3441	0.1329
0.480	0.9559	0.8541	0.8935	0.2968	0.9282	1.0867	1.2170	1.2453	0.1378
0.490	0.9542	0.8486	0.8894	0.3027	0.9423	1.1104	1.2033	1.1539	0.1426
0.500	0.9524	0.8430	0.8852	0.3086	0.9561	1.1341	1.1903	1.0691	0.1475

$\gamma=1.400$

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{p_0}$
0.510	0.9506	0.8374	0.8809	0.3145	0.9696	1.1579	1.1781	0.9904	0.1525
0.520	0.9487	0.8317	0.8766	0.3203	0.9828	1.1818	1.1665	0.9174	0.1574
0.530	0.9468	0.8259	0.8723	0.3262	0.9958	1.2057	1.1556	0.8496	0.1624
0.540	0.9449	0.8201	0.8679	0.3320	1.0084	1.2297	1.1452	0.7866	0.1674
0.550	0.9430	0.8142	0.8634	0.3378	1.0208	1.2538	1.1354	0.7281	0.1724
0.560	0.9410	0.8082	0.8589	0.3436	1.0328	1.2779	1.1261	0.6736	0.1774
0.570	0.9390	0.8022	0.8544	0.3493	1.0446	1.3021	1.1173	0.6229	0.1825
0.580	0.9370	0.7962	0.8498	0.3551	1.0561	1.3264	1.1090	0.5757	0.1875
0.590	0.9349	0.7901	0.8451	0.3608	1.0672	1.3507	1.1011	0.5317	0.1925
0.600	0.9328	0.7840	0.8405	0.3665	1.0781	1.3751	1.0937	0.4908	0.1976
0.610	0.9307	0.7778	0.8357	0.3722	1.0887	1.3996	1.0867	0.4527	0.2026
0.620	0.9286	0.7716	0.8310	0.3779	1.0990	1.4242	1.0800	0.4172	0.2076
0.630	0.9265	0.7654	0.8262	0.3835	1.1090	1.4489	1.0737	0.3841	0.2127
0.640	0.9243	0.7591	0.8213	0.3891	1.1186	1.4736	1.0678	0.3533	0.2177
0.650	0.9221	0.7528	0.8164	0.3948	1.1280	1.4984	1.0621	0.3246	0.2226
0.660	0.9199	0.7465	0.8115	0.4003	1.1371	1.5233	1.0568	0.2979	0.2276
0.670	0.9176	0.7401	0.8066	0.4059	1.1459	1.5483	1.0518	0.2730	0.2326
0.680	0.9153	0.7338	0.8016	0.4115	1.1544	1.5733	1.0471	0.2498	0.2375
0.690	0.9131	0.7274	0.7966	0.4170	1.1626	1.5984	1.0426	0.2282	0.2424
0.700	0.9107	0.7209	0.7916	0.4225	1.1705	1.6237	1.0384	0.2081	0.2473
0.710	0.9084	0.7145	0.7865	0.4280	1.1782	1.6490	1.0344	0.1895	0.2521
0.720	0.9061	0.7080	0.7814	0.4335	1.1855	1.6744	1.0307	0.1721	0.2569
0.730	0.9037	0.7016	0.7763	0.4389	1.1925	1.6999	1.0272	0.1561	0.2617
0.740	0.9013	0.6951	0.7712	0.4443	1.1993	1.7254	1.0239	0.1411	0.2664
0.750	0.8989	0.6886	0.7660	0.4497	1.2058	1.7511	1.0208	0.1273	0.2711
0.760	0.8964	0.6821	0.7609	0.4551	1.2119	1.7768	1.0179	0.1145	0.2758
0.770	0.8940	0.6756	0.7557	0.4605	1.2178	1.8027	1.0152	0.1026	0.2804
0.780	0.8915	0.6691	0.7505	0.4658	1.2234	1.8286	1.0126	0.0917	0.2849
0.790	0.8890	0.6625	0.7452	0.4711	1.2288	1.8547	1.0103	0.0816	0.2894
0.800	0.8865	0.6560	0.7400	0.4764	1.2338	1.8808	1.0081	0.0723	0.2939
0.810	0.8840	0.6495	0.7347	0.4817	1.2386	1.9070	1.0060	0.0638	0.2983
0.820	0.8815	0.6430	0.7295	0.4869	1.2431	1.9333	1.0041	0.0559	0.3026
0.830	0.8789	0.6365	0.7242	0.4921	1.2474	1.9598	1.0024	0.0488	0.3069
0.840	0.8763	0.6300	0.7189	0.4973	1.2514	1.9863	1.0008	0.0423	0.3112
0.850	0.8737	0.6235	0.7136	0.5025	1.2551	2.0129	0.9993	0.0363	0.3153
0.860	0.8711	0.6170	0.7083	0.5077	1.2585	2.0396	0.9979	0.0310	0.3195
0.870	0.8685	0.6106	0.7030	0.5128	1.2617	2.0665	0.9967	0.0261	0.3235
0.880	0.8659	0.6041	0.6977	0.5179	1.2646	2.0934	0.9956	0.0218	0.3275
0.890	0.8632	0.5977	0.6924	0.5230	1.2673	2.1204	0.9946	0.0179	0.3314
0.900	0.8606	0.5913	0.6870	0.5280	1.2698	2.1476	0.9937	0.0145	0.3352
0.910	0.8579	0.5849	0.6817	0.5331	1.2719	2.1748	0.9929	0.0115	0.3390
0.920	0.8552	0.5785	0.6764	0.5381	1.2739	2.2021	0.9922	0.0089	0.3427
0.930	0.8525	0.5721	0.6711	0.5431	1.2756	2.2296	0.9916	0.0067	0.3464
0.940	0.8498	0.5658	0.6658	0.5481	1.2770	2.2572	0.9911	0.0048	0.3499
0.950	0.8471	0.5595	0.6604	0.5530	1.2783	2.2848	0.9907	0.0033	0.3534
0.960	0.8444	0.5532	0.6551	0.5579	1.2793	2.3126	0.9903	0.0021	0.3569
0.970	0.8416	0.5469	0.6498	0.5628	1.2800	2.3405	0.9901	0.0011	0.3602
0.980	0.8389	0.5407	0.6445	0.5677	1.2806	2.3685	0.9899	0.0005	0.3635
0.990	0.8361	0.5345	0.6392	0.5725	1.2809	2.3966	0.9898	0.0001	0.3667
1.000	0.8333	0.5283	0.6339	0.5774	1.2810	2.4249	0.9897	0.0000	0.3698

# GAS FLOW TABLES ( $\gamma=1.400$ ): SUPERSONIC FLOW

$M$	$\frac{T}{T_0}$	$\frac{P}{P_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m}}{\dot{m}_0} \sqrt{c_p T_0}$	$\frac{\dot{m}}{\dot{m}_0} \sqrt{c_p T_0}$	$F$	$\frac{4c_f L_{\max}}{D} \frac{1}{2} \rho V^2$	$M_s$	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	$v$	$M$	
1.010	0.8306	0.5221	0.6287	0.5821	1.2809	2.4532	0.9898	0.0001	0.3728	0.9901	1.0000	1.0235	1.9152	1.0066	0.04	1.010
1.020	0.8278	0.5160	0.6234	0.5869	1.2806	2.4817	0.9899	0.0005	0.3758	0.9805	1.0000	1.0471	1.9379	1.0132	0.13	1.020
1.030	0.8250	0.5099	0.6181	0.5917	1.2801	2.5103	0.9900	0.0010	0.3787	0.9712	1.0000	1.0711	1.9610	1.0198	0.23	1.030
1.040	0.8222	0.5039	0.6129	0.5964	1.2793	2.5390	0.9903	0.0018	0.3815	0.9620	0.9999	1.0952	1.9844	1.0263	0.35	1.040
1.050	0.8193	0.4979	0.6077	0.6011	1.2784	2.5678	0.9905	0.0027	0.3842	0.9531	0.9999	1.1196	2.0083	1.0328	0.49	1.050
1.060	0.8165	0.4919	0.6024	0.6058	1.2773	2.5967	0.9909	0.0038	0.3869	0.9444	0.9998	1.1442	2.0325	1.0393	0.64	1.060
1.070	0.8137	0.4860	0.5972	0.6104	1.2760	2.6258	0.9913	0.0051	0.3895	0.9360	0.9996	1.1691	2.0570	1.0458	0.80	1.070
1.080	0.8108	0.4800	0.5920	0.6151	1.2745	2.6549	0.9917	0.0066	0.3919	0.9277	0.9994	1.1941	2.0819	1.0522	0.97	1.080
1.090	0.8080	0.4742	0.5869	0.6197	1.2728	2.6842	0.9922	0.0082	0.3944	0.9196	0.9992	1.2195	2.1072	1.0586	1.15	1.090
1.100	0.8052	0.4684	0.5817	0.6243	1.2709	2.7136	0.9928	0.0099	0.3967	0.9118	0.9989	1.2450	2.1328	1.0649	1.34	1.100
1.110	0.8023	0.4626	0.5766	0.6288	1.2689	2.7432	0.9934	0.0118	0.3990	0.9041	0.9986	1.2708	2.1588	1.0713	1.53	1.110
1.120	0.7994	0.4568	0.5714	0.6333	1.2667	2.7728	0.9940	0.0138	0.4011	0.8966	0.9982	1.2968	2.1851	1.0776	1.74	1.120
1.130	0.7966	0.4511	0.5663	0.6379	1.2643	2.8026	0.9947	0.0159	0.4032	0.8892	0.9978	1.3231	2.2118	1.0840	1.94	1.130
1.140	0.7937	0.4455	0.5612	0.6423	1.2618	2.8325	0.9954	0.0182	0.4052	0.8820	0.9973	1.3495	2.2388	1.0903	2.16	1.140
1.150	0.7908	0.4398	0.5562	0.6468	1.2590	2.8626	0.9961	0.0205	0.4072	0.8750	0.9967	1.3763	2.2661	1.0966	2.38	1.150
1.160	0.7879	0.4343	0.5511	0.6512	1.2562	2.8927	0.9969	0.0230	0.4090	0.8682	0.9961	1.4032	2.2937	1.1029	2.61	1.160
1.170	0.7851	0.4287	0.5461	0.6556	1.2531	2.9230	0.9978	0.0255	0.4108	0.8615	0.9953	1.4304	2.3217	1.1092	2.84	1.170
1.180	0.7822	0.4232	0.5411	0.6600	1.2500	2.9534	0.9986	0.0281	0.4125	0.8549	0.9946	1.4578	2.3500	1.1154	3.07	1.180
1.190	0.7793	0.4178	0.5361	0.6644	1.2466	2.9840	0.9995	0.0309	0.4141	0.8485	0.9937	1.4855	2.3786	1.1217	3.31	1.190
1.200	0.7764	0.4124	0.5311	0.6687	1.2432	3.0147	1.0004	0.0336	0.4157	0.8422	0.9928	1.5133	2.4075	1.1280	3.56	1.200
1.210	0.7735	0.4070	0.5262	0.6730	1.2396	3.0455	1.0014	0.0365	0.4171	0.8360	0.9918	1.5415	2.4367	1.1343	3.81	1.210
1.220	0.7706	0.4017	0.5213	0.6773	1.2358	3.0764	1.0024	0.0394	0.4185	0.8300	0.9907	1.5698	2.4663	1.1405	4.06	1.220
1.230	0.7677	0.3964	0.5164	0.6816	1.2319	3.1075	1.0034	0.0424	0.4198	0.8241	0.9896	1.5984	2.4961	1.1468	4.31	1.230
1.240	0.7648	0.3912	0.5115	0.6858	1.2279	3.1387	1.0045	0.0455	0.4211	0.8183	0.9884	1.6272	2.5263	1.1531	4.57	1.240
1.250	0.7619	0.3861	0.5067	0.6901	1.2238	3.1700	1.0055	0.0486	0.4223	0.8126	0.9871	1.6563	2.5568	1.1594	4.83	1.250

$\gamma=1.400$

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{1}{2} \rho V^2}{p_0}$	$M_s$	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	$v$	$M$
1.260	0.7590	0.3809	0.5019	0.6943	1.2195	3.2015	1.0066	0.0517	0.4233	0.8071	0.9857	1.6855	2.5875	1.1657	5.09	1.260
1.270	0.7561	0.3759	0.4971	0.6984	1.2152	3.2331	1.0077	0.0549	0.4244	0.8016	0.9842	1.7151	2.6186	1.1720	5.36	1.270
1.280	0.7532	0.3708	0.4923	0.7026	1.2107	3.2648	1.0089	0.0582	0.4253	0.7963	0.9827	1.7448	2.6500	1.1783	5.63	1.280
1.290	0.7503	0.3658	0.4876	0.7067	1.2061	3.2967	1.0100	0.0615	0.4262	0.7911	0.9811	1.7748	2.6816	1.1846	5.90	1.290
1.300	0.7474	0.3609	0.4829	0.7108	1.2014	3.3287	1.0112	0.0648	0.4270	0.7860	0.9794	1.8050	2.7136	1.1909	6.17	1.300
1.310	0.7445	0.3556	0.4782	0.7149	1.1965	3.3608	1.0124	0.0682	0.4277	0.7809	0.9776	1.8355	2.7459	1.1972	6.44	1.310
1.320	0.7416	0.3512	0.4736	0.7189	1.1916	3.3931	1.0136	0.0716	0.4283	0.7760	0.9758	1.8661	2.7784	1.2035	6.72	1.320
1.330	0.7387	0.3464	0.4690	0.7229	1.1866	3.4255	1.0149	0.0750	0.4289	0.7712	0.9738	1.8971	2.8112	1.2099	7.00	1.330
1.340	0.7358	0.3417	0.4644	0.7270	1.1815	3.4581	1.0161	0.0785	0.4294	0.7664	0.9718	1.9282	2.8444	1.2162	7.28	1.340
1.350	0.7329	0.3370	0.4598	0.7309	1.1763	3.4907	1.0174	0.0820	0.4299	0.7618	0.9697	1.9596	2.8778	1.2226	7.56	1.350
1.360	0.7300	0.3323	0.4553	0.7349	1.1710	3.5236	1.0187	0.0855	0.4303	0.7572	0.9676	1.9912	2.9115	1.2290	7.84	1.360
1.370	0.7271	0.3277	0.4508	0.7388	1.1656	3.5566	1.0200	0.0890	0.4306	0.7527	0.9653	2.0231	2.9455	1.2354	8.13	1.370
1.380	0.7242	0.3232	0.4463	0.7427	1.1601	3.5897	1.0213	0.0926	0.4308	0.7483	0.9630	2.0551	2.9798	1.2418	8.41	1.380
1.390	0.7213	0.3187	0.4418	0.7466	1.1546	3.6229	1.0226	0.0962	0.4310	0.7440	0.9607	2.0875	3.0144	1.2482	8.70	1.390
1.400	0.7184	0.3142	0.4374	0.7505	1.1490	3.6563	1.0240	0.0997	0.4311	0.7397	0.9582	2.1200	3.0492	1.2547	8.99	1.400
1.410	0.7155	0.3098	0.4330	0.7543	1.1433	3.6899	1.0253	0.1033	0.4312	0.7355	0.9557	2.1528	3.0844	1.2612	9.28	1.410
1.420	0.7126	0.3055	0.4287	0.7581	1.1375	3.7236	1.0267	0.1069	0.4312	0.7314	0.9531	2.1858	3.1198	1.2676	9.57	1.420
1.430	0.7097	0.3012	0.4244	0.7619	1.1317	3.7574	1.0281	0.1106	0.4311	0.7274	0.9504	2.2191	3.1555	1.2741	9.86	1.430
1.440	0.7069	0.2969	0.4201	0.7657	1.1258	3.7914	1.0295	0.1142	0.4310	0.7235	0.9476	2.2525	3.1915	1.2807	10.15	1.440
1.450	0.7040	0.2927	0.4158	0.7694	1.1198	3.8255	1.0308	0.1178	0.4308	0.7196	0.9448	2.2863	3.2278	1.2872	10.44	1.450
1.460	0.7011	0.2886	0.4116	0.7732	1.1138	3.8598	1.0323	0.1215	0.4306	0.7157	0.9420	2.3202	3.2643	1.2938	10.73	1.460
1.470	0.6982	0.2845	0.4074	0.7769	1.1077	3.8942	1.0337	0.1251	0.4303	0.7120	0.9390	2.3544	3.3011	1.3003	11.02	1.470
1.480	0.6954	0.2804	0.4032	0.7805	1.1016	3.9287	1.0351	0.1288	0.4299	0.7083	0.9360	2.3888	3.3382	1.3069	11.32	1.480
1.490	0.6925	0.2764	0.3991	0.7842	1.0954	3.9634	1.0365	0.1324	0.4295	0.7047	0.9329	2.4235	3.3756	1.3136	11.61	1.490
1.500	0.6897	0.2724	0.3950	0.7878	1.0891	3.9983	1.0379	0.1361	0.4290	0.7011	0.9298	2.4583	3.4133	1.3202	11.91	1.500

$\gamma=1.400$

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{1}{2} \rho V^2$	$M_s$	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	$V$	$M$
1.510	0.6868	0.2685	0.3999	0.7914	1.0829	4.0333	1.0394	0.1397	0.4285	0.6976	0.9266	2.4935	3.4512	1.3269	12.20	1.510
1.520	0.6840	0.2646	0.3869	0.7950	1.0765	4.0684	1.0408	0.1433	0.4279	0.6941	0.9233	2.5288	3.4894	1.3336	12.49	1.520
1.530	0.6811	0.2608	0.3829	0.7986	1.0702	4.1037	1.0423	0.1470	0.4273	0.6907	0.9200	2.5644	3.5279	1.3403	12.79	1.530
1.540	0.6783	0.2570	0.3789	0.8021	1.0638	4.1392	1.0437	0.1506	0.4266	0.6874	0.9166	2.6002	3.5667	1.3470	13.09	1.540
1.550	0.6754	0.2533	0.3750	0.8057	1.0573	4.1748	1.0452	0.1543	0.4259	0.6841	0.9132	2.6363	3.6057	1.3538	13.38	1.550
1.560	0.6726	0.2496	0.3710	0.8092	1.0508	4.2105	1.0467	0.1579	0.4252	0.6809	0.9097	2.6725	3.6450	1.3606	13.68	1.560
1.570	0.6698	0.2459	0.3672	0.8126	1.0443	4.2464	1.0481	0.1615	0.4243	0.6777	0.9062	2.7091	3.6846	1.3674	13.97	1.570
1.580	0.6670	0.2423	0.3633	0.8161	1.0378	4.2825	1.0496	0.1651	0.4235	0.6746	0.9026	2.7458	3.7244	1.3742	14.27	1.580
1.590	0.6642	0.2388	0.3595	0.8195	1.0312	4.3187	1.0511	0.1688	0.4226	0.6715	0.8989	2.7828	3.7646	1.3811	14.56	1.590
1.600	0.6614	0.2353	0.3557	0.8230	1.0246	4.3551	1.0526	0.1724	0.4216	0.6684	0.8952	2.8200	3.8050	1.3880	14.86	1.600
1.610	0.6586	0.2318	0.3520	0.8263	1.0180	4.3916	1.0541	0.1760	0.4206	0.6655	0.8915	2.8575	3.8456	1.3949	15.16	1.610
1.620	0.6558	0.2284	0.3483	0.8297	1.0114	4.4282	1.0555	0.1795	0.4196	0.6625	0.8877	2.8951	3.8866	1.4018	15.45	1.620
1.630	0.6530	0.2250	0.3446	0.8331	1.0047	4.4651	1.0570	0.1831	0.4185	0.6596	0.8838	2.9331	3.9278	1.4088	15.75	1.630
1.640	0.6502	0.2217	0.3409	0.8364	0.9980	4.5020	1.0585	0.1867	0.4174	0.6568	0.8799	2.9712	3.9693	1.4158	16.04	1.640
1.650	0.6475	0.2184	0.3373	0.8397	0.9913	4.5392	1.0600	0.1902	0.4162	0.6540	0.8760	3.0096	4.0110	1.4228	16.34	1.650
1.660	0.6447	0.2151	0.3337	0.8430	0.9846	4.5765	1.0615	0.1938	0.4150	0.6512	0.8720	3.0482	4.0531	1.4299	16.63	1.660
1.670	0.6419	0.2119	0.3302	0.8462	0.9779	4.6139	1.0630	0.1973	0.4138	0.6485	0.8680	3.0871	4.0953	1.4369	16.93	1.670
1.680	0.6392	0.2088	0.3266	0.8495	0.9712	4.6515	1.0645	0.2008	0.4125	0.6458	0.8639	3.1261	4.1379	1.4440	17.22	1.680
1.690	0.6364	0.2057	0.3232	0.8527	0.9644	4.6892	1.0660	0.2043	0.4112	0.6431	0.8599	3.1655	4.1807	1.4512	17.52	1.690
1.700	0.6337	0.2026	0.3197	0.8559	0.9577	4.7272	1.0674	0.2078	0.4098	0.6405	0.8557	3.2050	4.2238	1.4583	17.81	1.700
1.710	0.6310	0.1996	0.3163	0.8591	0.9509	4.7652	1.0689	0.2113	0.4085	0.6380	0.8516	3.2448	4.2672	1.4655	18.10	1.710
1.720	0.6283	0.1966	0.3129	0.8622	0.9442	4.8035	1.0704	0.2147	0.4071	0.6355	0.8474	3.2848	4.3108	1.4727	18.40	1.720
1.730	0.6256	0.1936	0.3095	0.8654	0.9374	4.8418	1.0719	0.2182	0.4056	0.6330	0.8431	3.3251	4.3547	1.4800	18.69	1.730
1.740	0.6229	0.1907	0.3062	0.8685	0.9307	4.8804	1.0734	0.2216	0.4041	0.6305	0.8389	3.3655	4.3989	1.4873	18.98	1.740
1.750	0.6202	0.1878	0.3029	0.8716	0.9239	4.9191	1.0749	0.2250	0.4026	0.6281	0.8346	3.4063	4.4433	1.4946	19.27	1.750

$\gamma=1.400$

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4 c_f L_{\max}}{D}$	$\frac{1}{2} \rho V^2$	$M_s$	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	$v$	$M$
1.760	0.6175	0.1850	0.2996	0.8747	0.9172	4.9580	1.0764	0.2284	0.4011	0.6257	0.8302	3.4472	4.4880	1.5019	19.56	1.760
1.770	0.6148	0.1822	0.2964	0.8777	0.9104	4.9970	1.0779	0.2318	0.3996	0.6234	0.8259	3.4884	4.5330	1.5093	19.86	1.770
1.780	0.6121	0.1794	0.2931	0.8808	0.9037	5.0362	1.0793	0.2252	0.3980	0.6210	0.8215	3.5298	4.5782	1.5167	20.15	1.780
1.790	0.6095	0.1767	0.2900	0.8838	0.8970	5.0755	1.0808	0.2285	0.3964	0.6188	0.8171	3.5715	4.6227	1.5241	20.44	1.790
1.800	0.6068	0.1740	0.2868	0.8868	0.8902	5.1150	1.0823	0.2419	0.3947	0.6165	0.8127	3.6133	4.6695	1.5316	20.73	1.800
1.810	0.6041	0.1714	0.2837	0.8898	0.8835	5.1547	1.0838	0.2452	0.3931	0.6143	0.8082	3.6555	4.7155	1.5391	21.01	1.810
1.820	0.6015	0.1688	0.2806	0.8927	0.8768	5.1945	1.0852	0.2485	0.3914	0.6121	0.8038	3.6978	4.7618	1.5466	21.30	1.820
1.830	0.5989	0.1662	0.2776	0.8957	0.8701	5.2345	1.0867	0.2518	0.3897	0.6099	0.7993	3.7404	4.8084	1.5541	21.59	1.830
1.840	0.5963	0.1637	0.2745	0.8986	0.8634	5.2747	1.0882	0.2551	0.3879	0.6078	0.7948	3.7882	4.8552	1.5617	21.88	1.840
1.850	0.5936	0.1612	0.2715	0.9015	0.8568	5.3150	1.0896	0.2583	0.3862	0.6057	0.7902	3.8263	4.9023	1.5693	22.16	1.850
1.860	0.5910	0.1587	0.2686	0.9044	0.8501	5.3555	1.0911	0.2616	0.3844	0.6036	0.7857	3.8695	4.9497	1.5770	22.45	1.860
1.870	0.5884	0.1563	0.2656	0.9072	0.8435	5.3962	1.0926	0.2648	0.3826	0.6016	0.7811	3.9131	4.9973	1.5847	22.73	1.870
1.880	0.5859	0.1539	0.2627	0.9101	0.8368	5.4370	1.0940	0.2680	0.3808	0.5996	0.7765	3.9568	5.0452	1.5924	23.02	1.880
1.890	0.5833	0.1516	0.2598	0.9129	0.8302	5.4780	1.0955	0.2712	0.3790	0.5976	0.7720	4.0008	5.0934	1.6001	23.30	1.890
1.900	0.5807	0.1492	0.2570	0.9157	0.8237	5.5191	1.0969	0.2743	0.3771	0.5956	0.7674	4.0450	5.1418	1.6079	23.59	1.900
1.910	0.5782	0.1470	0.2542	0.9185	0.8171	5.5604	1.0984	0.2775	0.3753	0.5937	0.7627	4.0895	5.1905	1.6157	23.87	1.910
1.920	0.5756	0.1447	0.2514	0.9213	0.8106	5.6019	1.0998	0.2806	0.3734	0.5918	0.7581	4.1341	5.2394	1.6226	24.15	1.920
1.930	0.5731	0.1425	0.2486	0.9240	0.8041	5.6435	1.1012	0.2837	0.3715	0.5899	0.7535	4.1791	5.2886	1.6314	24.43	1.930
1.940	0.5705	0.1403	0.2459	0.9268	0.7976	5.6853	1.1027	0.2868	0.3696	0.5880	0.7488	4.2242	5.3381	1.6394	24.71	1.940
1.950	0.5680	0.1381	0.2432	0.9295	0.7911	5.7273	1.1041	0.2899	0.3677	0.5862	0.7442	4.2696	5.3878	1.6473	24.99	1.950
1.960	0.5655	0.1360	0.2405	0.9322	0.7846	5.7695	1.1055	0.2929	0.3657	0.5844	0.7395	4.3152	5.4378	1.6553	25.27	1.960
1.970	0.5630	0.1339	0.2378	0.9349	0.7782	5.8118	1.1069	0.2960	0.3638	0.5826	0.7349	4.3611	5.4881	1.6633	25.55	1.970
1.980	0.5605	0.1318	0.2352	0.9375	0.7718	5.8542	1.1084	0.2990	0.3618	0.5808	0.7302	4.4071	5.5386	1.6713	25.83	1.980
1.990	0.5580	0.1298	0.2326	0.9402	0.7655	5.8969	1.1098	0.3020	0.3598	0.5791	0.7255	4.4535	5.5894	1.6794	26.10	1.990
2.000	0.5556	0.1278	0.2300	0.9428	0.7591	5.9397	1.1112	0.3050	0.3579	0.5774	0.7209	4.5000	5.6404	1.6875	26.38	2.000

$\gamma=1.400$

$M$	$\frac{T}{T_0}$	$\frac{P}{P_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A P_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A P}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4C_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{P_0}$	$M_s$	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	$\nu$	$M$
2.010	0.5531	0.1258	0.2275	0.9454	0.7528	5.9827	1.1126	0.3080	0.3559	0.5757	0.7162	4.5468	5.6918	1.6956	26.66	2.010
2.020	0.5506	0.1239	0.2250	0.9480	0.7465	6.0258	1.1140	0.3109	0.3539	0.5740	0.7115	4.5638	5.7433	1.7038	26.93	2.020
2.030	0.5482	0.1220	0.2225	0.9506	0.7403	6.0692	1.1154	0.3138	0.3518	0.5723	0.7069	4.6411	5.7952	1.7120	27.20	2.030
2.040	0.5458	0.1201	0.2200	0.9531	0.7340	6.1126	1.1167	0.3168	0.3498	0.5707	0.7022	4.6885	5.8473	1.7203	27.48	2.040
2.050	0.5433	0.1182	0.2176	0.9557	0.7279	6.1563	1.1181	0.3197	0.3478	0.5691	0.6975	4.7363	5.8996	1.7285	27.75	2.050
2.060	0.5409	0.1164	0.2152	0.9582	0.7217	6.2001	1.1195	0.3225	0.3458	0.5675	0.6928	4.7842	5.9523	1.7369	28.02	2.060
2.070	0.5385	0.1146	0.2128	0.9607	0.7156	6.2441	1.1209	0.3254	0.3437	0.5659	0.6882	4.8324	6.0051	1.7452	28.29	2.070
2.080	0.5361	0.1128	0.2104	0.9632	0.7095	6.2883	1.1222	0.3282	0.3417	0.5643	0.6835	4.8808	6.0583	1.7536	28.56	2.080
2.090	0.5337	0.1111	0.2081	0.9657	0.7034	6.3326	1.1236	0.3310	0.3396	0.5628	0.6789	4.9295	6.1117	1.7620	28.83	2.090
2.100	0.5313	0.1094	0.2058	0.9681	0.6974	6.3772	1.1250	0.3339	0.3376	0.5613	0.6742	4.9783	6.1654	1.7705	29.10	2.100
2.110	0.5290	0.1077	0.2035	0.9706	0.6914	6.4218	1.1263	0.3366	0.3355	0.5598	0.6696	5.0275	6.2193	1.7789	29.36	2.110
2.120	0.5266	0.1060	0.2013	0.9730	0.6854	6.4667	1.1276	0.3394	0.3334	0.5583	0.6649	5.0768	6.2735	1.7875	29.63	2.120
2.130	0.5243	0.1043	0.1990	0.9754	0.6795	6.5117	1.1290	0.3422	0.3314	0.5568	0.6603	5.1264	6.3280	1.7960	29.90	2.130
2.140	0.5219	0.1027	0.1968	0.9778	0.6736	6.5569	1.1303	0.3449	0.3293	0.5554	0.6557	5.1762	6.3827	1.8046	30.16	2.140
2.150	0.5196	0.1011	0.1946	0.9802	0.6677	6.6023	1.1317	0.3476	0.3272	0.5540	0.6511	5.2263	6.4377	1.8132	30.43	2.150
2.160	0.5173	0.0996	0.1925	0.9825	0.6619	6.6478	1.1330	0.3503	0.3252	0.5525	0.6464	5.2765	6.4929	1.8219	30.69	2.160
2.170	0.5150	0.0980	0.1903	0.9849	0.6561	6.6936	1.1343	0.3530	0.3231	0.5511	0.6419	5.3271	6.5484	1.8306	30.95	2.170
2.180	0.5127	0.0965	0.1882	0.9872	0.6503	6.7395	1.1356	0.3556	0.3210	0.5498	0.6373	5.3778	6.6042	1.8393	31.21	2.180
2.190	0.5104	0.0950	0.1861	0.9895	0.6446	6.7855	1.1369	0.3583	0.3189	0.5484	0.6327	5.4288	6.6602	1.8481	31.47	2.190
2.200	0.5081	0.0935	0.1841	0.9918	0.6389	6.8318	1.1382	0.3609	0.3169	0.5471	0.6281	5.4800	6.7165	1.8569	31.73	2.200
2.210	0.5059	0.0921	0.1820	0.9941	0.6333	6.8782	1.1395	0.3635	0.3148	0.5457	0.6236	5.5315	6.7730	1.8657	31.99	2.210
2.220	0.5036	0.0906	0.1800	0.9964	0.6277	6.9248	1.1408	0.3661	0.3127	0.5444	0.6191	5.5831	6.8298	1.8746	32.25	2.220
2.230	0.5014	0.0892	0.1780	0.9986	0.6221	6.9715	1.1421	0.3687	0.3106	0.5431	0.6145	5.6351	6.8869	1.8835	32.51	2.230
2.240	0.4991	0.0878	0.1760	1.0009	0.6165	7.0185	1.1434	0.3712	0.3085	0.5418	0.6100	5.6872	6.9442	1.8924	32.76	2.240
2.250	0.4969	0.0865	0.1740	1.0031	0.6110	7.0656	1.1446	0.3738	0.3065	0.5406	0.6055	5.7396	7.0018	1.9014	33.02	2.250

$\gamma=1.400$

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p}$	$\frac{F}{m \sqrt{c_p T_0}}$	$\frac{4 c_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{p_0}$	$M_s$	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	$v$	$M$
2.260	0.4947	0.0851	0.11721	1.0053	0.6656	7.1129	1.1459	0.3763	0.3044	0.5393	0.6011	5.7922	7.0597	1.9104	33.27	2.260
2.270	0.4925	0.0838	0.11702	1.0075	0.6602	7.1603	1.1472	0.3788	0.3023	0.5381	0.5966	5.8451	7.1178	1.9194	33.53	2.270
2.280	0.4903	0.0825	0.11683	1.0097	0.5648	7.2080	1.1484	0.3813	0.3003	0.5368	0.5921	5.8981	7.1762	1.9285	33.78	2.280
2.290	0.4881	0.0812	0.11664	1.0118	0.5694	7.2558	1.1497	0.3838	0.2982	0.5356	0.5877	5.9515	7.2348	1.9376	34.03	2.290
2.300	0.4859	0.0800	0.11646	1.0140	0.5641	7.3038	1.1509	0.3862	0.2961	0.5344	0.5833	6.0050	7.2937	1.9468	34.28	2.300
2.310	0.4837	0.0787	0.11628	1.0161	0.5788	7.3520	1.1521	0.3887	0.2941	0.5332	0.5789	6.0588	7.3528	1.9560	34.53	2.310
2.320	0.4816	0.0775	0.11609	1.0182	0.5736	7.4003	1.1534	0.3911	0.2920	0.5321	0.5745	6.1128	7.4122	1.9652	34.78	2.320
2.330	0.4794	0.0763	0.11592	1.0204	0.5684	7.4488	1.1546	0.3935	0.2900	0.5309	0.5702	6.1671	7.4719	1.9745	35.03	2.330
2.340	0.4773	0.0751	0.11574	1.0224	0.5632	7.4975	1.1558	0.3959	0.2879	0.5287	0.5658	6.2215	7.5319	1.9838	35.28	2.340
2.350	0.4752	0.0740	0.11556	1.0245	0.5581	7.5464	1.1570	0.3983	0.2859	0.5266	0.5615	6.2763	7.5920	1.9931	35.53	2.350
2.360	0.4731	0.0728	0.11539	1.0266	0.5530	7.5955	1.1582	0.4006	0.2839	0.5275	0.5572	6.3312	7.6525	2.0025	35.77	2.360
2.370	0.4709	0.0717	0.11522	1.0286	0.5480	7.6447	1.1595	0.4030	0.2818	0.5264	0.5529	6.3864	7.7132	2.0119	36.02	2.370
2.380	0.4688	0.0706	0.11505	1.0307	0.5430	7.6941	1.1606	0.4053	0.2798	0.5253	0.5486	6.4418	7.7742	2.0213	36.26	2.380
2.390	0.4668	0.0695	0.11488	1.0327	0.5380	7.7437	1.1618	0.4076	0.2778	0.5242	0.5444	6.4975	7.8354	2.0308	36.50	2.390
2.400	0.4647	0.0684	0.11472	1.0347	0.5331	7.7935	1.1630	0.4099	0.2758	0.5231	0.5401	6.5533	7.8969	2.0403	36.75	2.400
2.410	0.4626	0.0673	0.11456	1.0367	0.5282	7.8434	1.1642	0.4122	0.2738	0.5221	0.5359	6.6095	7.9587	2.0499	36.99	2.410
2.420	0.4606	0.0663	0.11439	1.0387	0.5233	7.8935	1.1654	0.4144	0.2718	0.5210	0.5317	6.6658	8.0207	2.0595	37.23	2.420
2.430	0.4585	0.0653	0.11424	1.0407	0.5185	7.9438	1.1665	0.4167	0.2698	0.5200	0.5276	6.7224	8.0830	2.0691	37.47	2.430
2.440	0.4565	0.0643	0.11408	1.0426	0.5137	7.9943	1.1677	0.4189	0.2678	0.5189	0.5234	6.7792	8.1455	2.0788	37.71	2.440
2.450	0.4544	0.0633	0.11392	1.0446	0.5090	8.0450	1.1689	0.4211	0.2658	0.5179	0.5193	6.8363	8.2083	2.0885	37.95	2.450
2.460	0.4524	0.0623	0.11377	1.0465	0.5043	8.0958	1.1700	0.4233	0.2639	0.5169	0.5152	6.8935	8.2713	2.0982	38.18	2.460
2.470	0.4504	0.0613	0.11362	1.0484	0.4996	8.1468	1.1712	0.4255	0.2619	0.5159	0.5111	6.9511	8.3346	2.1080	38.42	2.470
2.480	0.4484	0.0604	0.11346	1.0503	0.4950	8.1980	1.1723	0.4277	0.2599	0.5149	0.5071	7.0088	8.3982	2.1178	38.66	2.480
2.490	0.4464	0.0594	0.11332	1.0522	0.4904	8.2494	1.1734	0.4298	0.2580	0.5140	0.5030	7.0668	8.4620	2.1276	38.89	2.490
2.500	0.4444	0.0585	0.11317	1.0541	0.4858	8.3010	1.1746	0.4320	0.2561	0.5130	0.4990	7.1250	8.5261	2.1375	39.12	2.500

$\gamma=1.400$

$M$	$\frac{T}{T_0}$	$\frac{P}{P_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{1}{2} \rho V^2}{p_0}$	$M_s$	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	$\nu$	$M$
2.510	0.4425	0.0576	0.1302	1.0560	0.4813	8.3527	1.1757	0.4341	0.2541	0.5120	0.4950	7.1835	8.5905	2.1474	39.36	2.510
2.520	0.4405	0.0567	0.1288	1.0578	0.4768	8.4046	1.1768	0.4362	0.2522	0.5111	0.4911	7.2421	8.6551	2.1574	39.59	2.520
2.530	0.4386	0.0559	0.1274	1.0597	0.4724	8.4567	1.1779	0.4383	0.2503	0.5102	0.4871	7.3011	8.7200	2.1674	39.82	2.530
2.540	0.4366	0.0550	0.1260	1.0615	0.4680	8.5090	1.1790	0.4404	0.2484	0.5092	0.4832	7.3602	8.7851	2.1774	40.05	2.540
2.550	0.4347	0.0542	0.1246	1.0633	0.4636	8.5615	1.1801	0.4425	0.2465	0.5083	0.4793	7.4196	8.8505	2.1875	40.28	2.550
2.560	0.4328	0.0533	0.1232	1.0651	0.4593	8.6141	1.1812	0.4445	0.2446	0.5074	0.4754	7.4792	8.9161	2.1976	40.51	2.560
2.570	0.4309	0.0525	0.1218	1.0669	0.4550	8.6670	1.1823	0.4466	0.2427	0.5065	0.4715	7.5391	8.9820	2.2077	40.74	2.570
2.580	0.4289	0.0517	0.1205	1.0687	0.4507	8.7200	1.1834	0.4486	0.2409	0.5056	0.4677	7.5991	9.0482	2.2179	40.96	2.580
2.590	0.4271	0.0509	0.1192	1.0705	0.4465	8.7732	1.1844	0.4506	0.2390	0.5047	0.4639	7.6595	9.1146	2.2281	41.19	2.590
2.600	0.4252	0.0501	0.1179	1.0722	0.4423	8.8265	1.1855	0.4526	0.2371	0.5039	0.4601	7.7200	9.1813	2.2383	41.41	2.600
2.610	0.4233	0.0493	0.1166	1.0740	0.4382	8.8801	1.1866	0.4546	0.2353	0.5030	0.4564	7.7808	9.2483	2.2486	41.64	2.610
2.620	0.4214	0.0486	0.1153	1.0757	0.4341	8.9338	1.1876	0.4565	0.2335	0.5022	0.4526	7.8418	9.3155	2.2590	41.86	2.620
2.630	0.4196	0.0478	0.1140	1.0774	0.4300	8.9877	1.1887	0.4585	0.2317	0.5013	0.4489	7.9031	9.3829	2.2693	42.09	2.630
2.640	0.4177	0.0471	0.1128	1.0791	0.4260	9.0418	1.1897	0.4604	0.2298	0.5005	0.4452	7.9645	9.4506	2.2797	42.31	2.640
2.650	0.4159	0.0464	0.1115	1.0808	0.4220	9.0961	1.1908	0.4624	0.2280	0.4996	0.4416	8.0263	9.5186	2.2902	42.53	2.650
2.660	0.4141	0.0457	0.1103	1.0825	0.4180	9.1506	1.1918	0.4643	0.2262	0.4988	0.4379	8.0882	9.5869	2.3006	42.75	2.660
2.670	0.4122	0.0450	0.1091	1.0842	0.4141	9.2052	1.1928	0.4662	0.2245	0.4980	0.4343	8.1504	9.6554	2.3111	42.97	2.670
2.680	0.4104	0.0443	0.1079	1.0859	0.4102	9.2601	1.1939	0.4681	0.2227	0.4972	0.4307	8.2128	9.7241	2.3217	43.19	2.680
2.690	0.4086	0.0436	0.1067	1.0875	0.4063	9.3151	1.1949	0.4700	0.2209	0.4964	0.4271	8.2755	9.7931	2.3323	43.40	2.690
2.700	0.4068	0.0430	0.1056	1.0892	0.4025	9.3703	1.1959	0.4718	0.2192	0.4956	0.4236	8.3383	9.8624	2.3429	43.62	2.700
2.710	0.4051	0.0423	0.1044	1.0908	0.3987	9.4257	1.1969	0.4737	0.2174	0.4949	0.4201	8.4015	9.9319	2.3536	43.84	2.710
2.720	0.4033	0.0417	0.1033	1.0924	0.3949	9.4812	1.1979	0.4755	0.2157	0.4941	0.4166	8.4648	10.0017	2.3642	44.05	2.720
2.730	0.4015	0.0410	0.1022	1.0941	0.3912	9.5370	1.1989	0.4773	0.2140	0.4933	0.4131	8.5284	10.0718	2.3750	44.27	2.730
2.740	0.3998	0.0404	0.1010	1.0957	0.3875	9.5929	1.1999	0.4791	0.2123	0.4926	0.4097	8.5922	10.1421	2.3858	44.48	2.740
2.750	0.3980	0.0398	0.0999	1.0973	0.3838	9.6490	1.2009	0.4809	0.2106	0.4918	0.4062	8.6563	10.2127	2.3966	44.69	2.750

$\gamma=1.400$

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{A p}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4 c_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{p_0}$	$M_s$	$\frac{P_{0s}}{P_0}$	$\frac{P_s}{P}$	$\frac{P_{0s}}{P}$	$\frac{T_s}{T}$	$v$	$M$
2.760	0.3963	0.0392	0.0989	1.0988	0.3802	9.7053	1.2019	0.4827	0.2089	0.4911	0.4028	8.7205	10.2835	2.4074	44.91	2.760
2.770	0.3945	0.0386	0.0978	1.1004	0.3766	9.7618	1.2029	0.4845	0.2072	0.4903	0.3994	8.7851	10.3546	2.4183	45.12	2.770
2.780	0.3928	0.0380	0.0967	1.1020	0.3730	9.8185	1.2038	0.4863	0.2055	0.4896	0.3961	8.8498	10.4259	2.4292	45.33	2.780
2.790	0.3911	0.0374	0.0957	1.1035	0.3695	9.8753	1.2048	0.4880	0.2039	0.4889	0.3928	8.9148	10.4975	2.4402	45.54	2.790
2.800	0.3894	0.0368	0.0946	1.1051	0.3660	9.9324	1.2058	0.4898	0.2022	0.4882	0.3895	8.9800	10.5694	2.4512	45.75	2.800
2.810	0.3877	0.0363	0.0936	1.1066	0.3625	9.9896	1.2067	0.4915	0.2006	0.4875	0.3862	9.0455	10.6415	2.4622	45.95	2.810
2.820	0.3860	0.0357	0.0926	1.1081	0.3591	10.0470	1.2077	0.4932	0.1990	0.4868	0.3829	9.1111	10.7139	2.4733	46.16	2.820
2.830	0.3844	0.0352	0.0916	1.1096	0.3557	10.1046	1.2086	0.4949	0.1973	0.4861	0.3737	9.1771	10.7865	2.4844	46.37	2.830
2.840	0.3827	0.0347	0.0906	1.1111	0.3523	10.1624	1.2095	0.4966	0.1957	0.4854	0.3785	9.2432	10.8594	2.4955	46.57	2.840
2.850	0.3810	0.0341	0.0896	1.1126	0.3490	10.2204	1.2105	0.4983	0.1941	0.4847	0.3733	9.3096	10.9326	2.5067	46.78	2.850
2.860	0.3794	0.0336	0.0886	1.1141	0.3457	10.2785	1.2114	0.5000	0.1926	0.4840	0.3701	9.3762	11.0060	2.5179	46.98	2.860
2.870	0.3777	0.0331	0.0877	1.1156	0.3424	10.3368	1.2123	0.5016	0.1910	0.4833	0.3670	9.4431	11.0797	2.5292	47.19	2.870
2.880	0.3761	0.0326	0.0867	1.1171	0.3392	10.3954	1.2132	0.5033	0.1894	0.4827	0.3639	9.5101	11.1536	2.5405	47.39	2.880
2.890	0.3745	0.0321	0.0858	1.1185	0.3359	10.4541	1.2142	0.5049	0.1879	0.4820	0.3608	9.5775	11.2278	2.5518	47.59	2.890
2.900	0.3729	0.0317	0.0849	1.1199	0.3328	10.5130	1.2151	0.5065	0.1863	0.4814	0.3577	9.6450	11.3022	2.5632	47.79	2.900
2.910	0.3712	0.0312	0.0840	1.1214	0.3296	10.5720	1.2160	0.5081	0.1848	0.4807	0.3547	9.7128	11.3770	2.5746	47.99	2.910
2.920	0.3696	0.0307	0.0831	1.1228	0.3265	10.6313	1.2169	0.5097	0.1833	0.4801	0.3517	9.7898	11.4519	2.5861	48.19	2.920
2.930	0.3681	0.0302	0.0822	1.1242	0.3234	10.6908	1.2178	0.5113	0.1818	0.4795	0.3487	9.8491	11.5271	2.5976	48.39	2.930
2.940	0.3665	0.0298	0.0813	1.1256	0.3203	10.7504	1.2187	0.5129	0.1803	0.4788	0.3457	9.9175	11.6026	2.6091	48.59	2.940
2.950	0.3649	0.0293	0.0804	1.1270	0.3173	10.8102	1.2195	0.5145	0.1788	0.4782	0.3428	9.9863	11.6784	2.6206	48.78	2.950
2.960	0.3633	0.0289	0.0796	1.1284	0.3143	10.8702	1.2204	0.5160	0.1773	0.4776	0.3398	10.0552	11.7544	2.6322	48.98	2.960
2.970	0.3618	0.0285	0.0787	1.1298	0.3113	10.9304	1.2213	0.5176	0.1758	0.4770	0.3369	10.1244	11.8306	2.6439	49.18	2.970
2.980	0.3602	0.0281	0.0779	1.1312	0.3083	10.9908	1.2222	0.5191	0.1744	0.4764	0.3340	10.1938	11.9072	2.6555	49.37	2.980
2.990	0.3587	0.0276	0.0770	1.1325	0.3054	11.0514	1.2230	0.5206	0.1729	0.4758	0.3312	10.2635	11.9839	2.6673	49.56	2.990
3.000	0.3571	0.0272	0.0762	1.1339	0.3025	11.1122	1.2239	0.5222	0.1715	0.4752	0.3283	10.3333	12.0610	2.6790	49.76	3.000

# GAS FLOW TABLES ( $\gamma=1.333$ ): SUBSONIC FLOW

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{p_0}$
0.010	1.0000	0.9999	1.0000	0.0058	0.0231	0.0231	43.2958	7493.200	0.0001
0.020	0.9999	0.9997	0.9998	0.0115	0.0462	0.0462	21.6560	1868.007	0.0003
0.030	0.9999	0.9994	0.9996	0.0173	0.0693	0.0693	14.4464	826.7890	0.0006
0.040	0.9997	0.9989	0.9992	0.0231	0.0923	0.0924	10.8442	462.6179	0.0011
0.050	0.9996	0.9983	0.9988	0.0288	0.1153	0.1155	8.6851	294.2161	0.0017
0.060	0.9994	0.9976	0.9982	0.0346	0.1383	0.1386	7.2475	202.8455	0.0024
0.070	0.9992	0.9967	0.9976	0.0404	0.1612	0.1618	6.2222	147.8292	0.0033
0.080	0.9989	0.9957	0.9968	0.0461	0.1841	0.1849	5.4546	112.1800	0.0042
0.090	0.9987	0.9946	0.9960	0.0519	0.2069	0.2080	4.8587	87.7848	0.0054
0.100	0.9983	0.9934	0.9950	0.0577	0.2297	0.2312	4.3831	70.3719	0.0066
0.110	0.9980	0.9920	0.9940	0.0634	0.2523	0.2544	3.9949	57.5186	0.0080
0.120	0.9976	0.9905	0.9928	0.0692	0.2749	0.2775	3.6724	47.7680	0.0095
0.130	0.9972	0.9888	0.9916	0.0749	0.2974	0.3007	3.4003	40.2012	0.0111
0.140	0.9967	0.9870	0.9903	0.0807	0.3197	0.3239	3.1678	34.2155	0.0129
0.150	0.9963	0.9851	0.9888	0.0864	0.3420	0.3471	2.9670	29.4027	0.0148
0.160	0.9958	0.9831	0.9873	0.0921	0.3641	0.3704	2.7920	25.4777	0.0168
0.170	0.9952	0.9810	0.9857	0.0979	0.3861	0.3936	2.6383	22.2372	0.0189
0.180	0.9946	0.9787	0.9840	0.1036	0.4080	0.4169	2.5022	19.5326	0.0211
0.190	0.9940	0.9763	0.9822	0.1093	0.4298	0.4402	2.3809	17.2536	0.0235
0.200	0.9934	0.9738	0.9803	0.1150	0.4514	0.4635	2.2724	15.3166	0.0260
0.210	0.9927	0.9711	0.9783	0.1207	0.4728	0.4869	2.1747	13.6578	0.0285
0.220	0.9920	0.9684	0.9762	0.1264	0.4941	0.5102	2.0863	12.2273	0.0312
0.230	0.9913	0.9655	0.9740	0.1321	0.5152	0.5336	2.0061	10.9859	0.0340
0.240	0.9905	0.9625	0.9717	0.1378	0.5362	0.5570	1.9330	9.9026	0.0370
0.250	0.9897	0.9594	0.9694	0.1435	0.5569	0.5805	1.8662	8.9522	0.0400
0.260	0.9889	0.9562	0.9669	0.1492	0.5775	0.6040	1.8049	8.1146	0.0431
0.270	0.9880	0.9529	0.9644	0.1549	0.5979	0.6275	1.7486	7.3731	0.0463
0.280	0.9871	0.9494	0.9618	0.1605	0.6181	0.6510	1.6966	6.7140	0.0496
0.290	0.9862	0.9459	0.9591	0.1662	0.6380	0.6746	1.6486	6.1261	0.0530
0.300	0.9852	0.9422	0.9563	0.1718	0.6578	0.6982	1.6042	5.5998	0.0565
0.310	0.9843	0.9384	0.9534	0.1775	0.6774	0.7218	1.5629	5.1272	0.0601
0.320	0.9832	0.9346	0.9505	0.1831	0.6967	0.7455	1.5245	4.7016	0.0638
0.330	0.9822	0.9306	0.9475	0.1887	0.7158	0.7692	1.4888	4.3173	0.0675
0.340	0.9811	0.9265	0.9444	0.1943	0.7347	0.7929	1.4555	3.9693	0.0714
0.350	0.9800	0.9224	0.9412	0.1999	0.7533	0.8167	1.4244	3.6535	0.0753
0.360	0.9789	0.9181	0.9379	0.2055	0.7717	0.8405	1.3953	3.3663	0.0793
0.370	0.9777	0.9137	0.9346	0.2111	0.7898	0.8644	1.3680	3.1046	0.0834
0.380	0.9765	0.9093	0.9311	0.2167	0.8077	0.8883	1.3425	2.8655	0.0875
0.390	0.9753	0.9047	0.9276	0.2223	0.8253	0.9122	1.3185	2.6469	0.0917
0.400	0.9741	0.9001	0.9241	0.2278	0.8427	0.9362	1.2959	2.4466	0.0960
0.410	0.9728	0.8954	0.9204	0.2334	0.8598	0.9603	1.2747	2.2627	0.1003
0.420	0.9715	0.8906	0.9167	0.2389	0.8766	0.9843	1.2548	2.0937	0.1047
0.430	0.9701	0.8857	0.9130	0.2444	0.8932	1.0085	1.2360	1.9382	0.1091
0.440	0.9688	0.8807	0.9091	0.2499	0.9095	1.0326	1.2183	1.7949	0.1136
0.450	0.9674	0.8757	0.9052	0.2554	0.9255	1.0569	1.2016	1.6627	0.1182
0.460	0.9660	0.8706	0.9012	0.2609	0.9412	1.0811	1.1858	1.5405	0.1228
0.470	0.9645	0.8654	0.8972	0.2664	0.9567	1.1055	1.1710	1.4276	0.1274
0.480	0.9631	0.8601	0.8931	0.2718	0.9718	1.1299	1.1569	1.3231	0.1321
0.490	0.9616	0.8548	0.8890	0.2773	0.9867	1.1543	1.1436	1.2263	0.1368
0.500	0.9600	0.8494	0.8847	0.2827	1.0012	1.1788	1.1310	1.1365	0.1415

$\gamma=1.333$

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{p_0}$
0.510	0.9585	0.8439	0.8805	0.2881	1.0155	1.2033	1.1192	1.0532	0.1463
0.520	0.9569	0.8384	0.8761	0.2935	1.0295	1.2279	1.1079	0.9759	0.1511
0.530	0.9553	0.8328	0.8717	0.2989	1.0431	1.2526	1.0973	0.9041	0.1559
0.540	0.9537	0.8271	0.8673	0.3043	1.0565	1.2773	1.0872	0.8373	0.1608
0.550	0.9520	0.8214	0.8628	0.3097	1.0696	1.3021	1.0777	0.7752	0.1656
0.560	0.9504	0.8157	0.8583	0.3150	1.0823	1.3269	1.0687	0.7174	0.1705
0.570	0.9487	0.8099	0.8537	0.3204	1.0948	1.3518	1.0601	0.6636	0.1754
0.580	0.9470	0.8040	0.8490	0.3257	1.1069	1.3768	1.0520	0.6136	0.1803
0.590	0.9452	0.7981	0.8443	0.3310	1.1188	1.4018	1.0444	0.5669	0.1852
0.600	0.9434	0.7921	0.8396	0.3363	1.1303	1.4269	1.0371	0.5235	0.1901
0.610	0.9417	0.7861	0.8348	0.3416	1.1415	1.4521	1.0303	0.4830	0.1950
0.620	0.9398	0.7801	0.8300	0.3469	1.1524	1.4773	1.0238	0.4452	0.1999
0.630	0.9380	0.7740	0.8252	0.3521	1.1630	1.5026	1.0176	0.4101	0.2048
0.640	0.9362	0.7679	0.8203	0.3573	1.1733	1.5280	1.0118	0.3773	0.2096
0.650	0.9343	0.7618	0.8153	0.3626	1.1833	1.5534	1.0063	0.3467	0.2145
0.660	0.9324	0.7556	0.8104	0.3678	1.1930	1.5789	1.0011	0.3183	0.2194
0.670	0.9305	0.7494	0.8054	0.3729	1.2023	1.6045	0.9962	0.2918	0.2242
0.680	0.9285	0.7431	0.8003	0.3781	1.2114	1.6301	0.9916	0.2671	0.2290
0.690	0.9266	0.7368	0.7953	0.3833	1.2201	1.6559	0.9872	0.2441	0.2338
0.700	0.9246	0.7306	0.7902	0.3884	1.2285	1.6817	0.9831	0.2227	0.2386
0.710	0.9226	0.7242	0.7850	0.3935	1.2367	1.7075	0.9792	0.2028	0.2433
0.720	0.9205	0.7179	0.7799	0.3986	1.2445	1.7335	0.9755	0.1843	0.2480
0.730	0.9185	0.7116	0.7747	0.4037	1.2520	1.7595	0.9721	0.1671	0.2527
0.740	0.9164	0.7052	0.7695	0.4088	1.2592	1.7856	0.9688	0.1512	0.2574
0.750	0.9144	0.6988	0.7643	0.4139	1.2661	1.8118	0.9658	0.1364	0.2620
0.760	0.9123	0.6924	0.7590	0.4189	1.2727	1.8381	0.9629	0.1227	0.2666
0.770	0.9102	0.6860	0.7537	0.4239	1.2790	1.8644	0.9603	0.1100	0.2711
0.780	0.9080	0.6796	0.7484	0.4289	1.2850	1.8908	0.9578	0.0983	0.2756
0.790	0.9059	0.6732	0.7431	0.4339	1.2907	1.9174	0.9554	0.0875	0.2800
0.800	0.9037	0.6668	0.7378	0.4389	1.2961	1.9440	0.9533	0.0776	0.2844
0.810	0.9015	0.6603	0.7325	0.4438	1.3013	1.9706	0.9513	0.0685	0.2888
0.820	0.8993	0.6539	0.7271	0.4487	1.3061	1.9974	0.9494	0.0601	0.2930
0.830	0.8971	0.6475	0.7217	0.4536	1.3107	2.0243	0.9477	0.0524	0.2973
0.840	0.8949	0.6411	0.7164	0.4585	1.3149	2.0512	0.9461	0.0454	0.3015
0.850	0.8926	0.6346	0.7110	0.4634	1.3189	2.0782	0.9446	0.0391	0.3056
0.860	0.8904	0.6282	0.7056	0.4683	1.3226	2.1053	0.9433	0.0333	0.3097
0.870	0.8881	0.6218	0.7002	0.4731	1.3260	2.1326	0.9420	0.0281	0.3137
0.880	0.8858	0.6154	0.6948	0.4779	1.3292	2.1599	0.9409	0.0235	0.3176
0.890	0.8835	0.6090	0.6893	0.4827	1.3321	2.1873	0.9399	0.0193	0.3215
0.900	0.8812	0.6026	0.6839	0.4875	1.3347	2.2147	0.9390	0.0156	0.3253
0.910	0.8788	0.5963	0.6785	0.4923	1.3370	2.2423	0.9383	0.0124	0.3291
0.920	0.8765	0.5899	0.6731	0.4970	1.3391	2.2700	0.9376	0.0096	0.3328
0.930	0.8741	0.5836	0.6676	0.5018	1.3410	2.2978	0.9370	0.0072	0.3364
0.940	0.8717	0.5773	0.6622	0.5065	1.3425	2.3256	0.9365	0.0052	0.3400
0.950	0.8694	0.5710	0.6568	0.5111	1.3439	2.3536	0.9360	0.0035	0.3435
0.960	0.8670	0.5647	0.6514	0.5158	1.3449	2.3817	0.9357	0.0022	0.3469
0.970	0.8646	0.5585	0.6459	0.5205	1.3458	2.4098	0.9354	0.0012	0.3502
0.980	0.8621	0.5522	0.6405	0.5251	1.3464	2.4381	0.9353	0.0005	0.3535
0.990	0.8597	0.5460	0.6351	0.5297	1.3467	2.4664	0.9351	0.0001	0.3567
1.000	0.8573	0.5398	0.6297	0.5343	1.3468	2.4949	0.9351	0.0000	0.3598

# GAS FLOW TABLES ( $\gamma=1.333$ ): SUPERSONIC FLOW

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{p_0}$
1.010	0.8548	0.5337	0.6243	0.5389	1.3467	2.5234	0.9351	0.0001	0.3628
1.020	0.8524	0.5276	0.6189	0.5434	1.3464	2.5521	0.9352	0.0005	0.3658
1.030	0.8499	0.5215	0.6136	0.5479	1.3458	2.5809	0.9354	0.0011	0.3687
1.040	0.8474	0.5154	0.6082	0.5525	1.3450	2.6097	0.9356	0.0019	0.3715
1.050	0.8449	0.5093	0.6028	0.5569	1.3440	2.6387	0.9359	0.0029	0.3743
1.060	0.8424	0.5033	0.5975	0.5614	1.3428	2.6678	0.9363	0.0042	0.3769
1.070	0.8399	0.4974	0.5922	0.5659	1.3414	2.6970	0.9367	0.0056	0.3795
1.080	0.8374	0.4914	0.5869	0.5703	1.3397	2.7263	0.9371	0.0071	0.3820
1.090	0.8349	0.4855	0.5816	0.5747	1.3379	2.7557	0.9376	0.0089	0.3845
1.100	0.8323	0.4796	0.5763	0.5791	1.3359	2.7852	0.9381	0.0108	0.3868
1.110	0.8298	0.4738	0.5710	0.5835	1.3337	2.8148	0.9387	0.0128	0.3891
1.120	0.8272	0.4680	0.5658	0.5878	1.3313	2.8446	0.9394	0.0150	0.3913
1.130	0.8247	0.4622	0.5605	0.5922	1.3287	2.8744	0.9401	0.0173	0.3934
1.140	0.8221	0.4565	0.5553	0.5965	1.3259	2.9043	0.9408	0.0197	0.3954
1.150	0.8195	0.4508	0.5501	0.6008	1.3229	2.9344	0.9415	0.0223	0.3974
1.160	0.8170	0.4452	0.5449	0.6050	1.3198	2.9646	0.9424	0.0250	0.3993
1.170	0.8144	0.4396	0.5398	0.6093	1.3165	2.9949	0.9432	0.0277	0.4011
1.180	0.8118	0.4340	0.5347	0.6135	1.3131	3.0253	0.9441	0.0306	0.4028
1.190	0.8092	0.4285	0.5295	0.6177	1.3094	3.0558	0.9450	0.0335	0.4044
1.200	0.8066	0.4230	0.5245	0.6219	1.3057	3.0864	0.9459	0.0366	0.4060
1.210	0.8040	0.4176	0.5194	0.6261	1.3017	3.1172	0.9469	0.0397	0.4075
1.220	0.8014	0.4122	0.5143	0.6302	1.2976	3.1481	0.9479	0.0429	0.4089
1.230	0.7988	0.4068	0.5093	0.6344	1.2934	3.1791	0.9489	0.0462	0.4102
1.240	0.7962	0.4015	0.5043	0.6385	1.2890	3.2102	0.9500	0.0495	0.4115
1.250	0.7936	0.3963	0.4994	0.6426	1.2845	3.2414	0.9511	0.0529	0.4127
1.260	0.7909	0.3911	0.4944	0.6466	1.2798	3.2727	0.9522	0.0564	0.4138
1.270	0.7883	0.3859	0.4895	0.6507	1.2751	3.3042	0.9533	0.0599	0.4148
1.280	0.7857	0.3808	0.4846	0.6547	1.2701	3.3358	0.9545	0.0634	0.4158
1.290	0.7830	0.3757	0.4798	0.6587	1.2651	3.3675	0.9557	0.0670	0.4167
1.300	0.7804	0.3706	0.4749	0.6627	1.2599	3.3993	0.9569	0.0707	0.4175
1.310	0.7778	0.3657	0.4701	0.6667	1.2547	3.4313	0.9581	0.0744	0.4182
1.320	0.7751	0.3607	0.4654	0.6706	1.2493	3.4633	0.9594	0.0781	0.4189
1.330	0.7725	0.3558	0.4606	0.6746	1.2438	3.4955	0.9606	0.0819	0.4195
1.340	0.7698	0.3510	0.4559	0.6785	1.2382	3.5279	0.9619	0.0857	0.4200
1.350	0.7672	0.3462	0.4512	0.6824	1.2325	3.5603	0.9632	0.0895	0.4205
1.360	0.7646	0.3414	0.4465	0.6862	1.2266	3.5929	0.9645	0.0934	0.4209
1.370	0.7619	0.3367	0.4419	0.6901	1.2207	3.6256	0.9659	0.0973	0.4212
1.380	0.7593	0.3320	0.4373	0.6939	1.2147	3.6584	0.9672	0.1012	0.4215
1.390	0.7566	0.3274	0.4328	0.6977	1.2086	3.6914	0.9686	0.1051	0.4216
1.400	0.7540	0.3229	0.4282	0.7015	1.2025	3.7245	0.9700	0.1091	0.4218
1.410	0.7513	0.3183	0.4237	0.7053	1.1962	3.7577	0.9714	0.1130	0.4218
1.420	0.7487	0.3139	0.4192	0.7090	1.1899	3.7910	0.9728	0.1170	0.4218
1.430	0.7460	0.3094	0.4148	0.7127	1.1835	3.8245	0.9742	0.1210	0.4217
1.440	0.7434	0.3051	0.4104	0.7164	1.1770	3.8581	0.9756	0.1250	0.4216
1.450	0.7407	0.3007	0.4060	0.7201	1.1704	3.8918	0.9771	0.1290	0.4214
1.460	0.7381	0.2965	0.4017	0.7238	1.1638	3.9257	0.9785	0.1331	0.4212
1.470	0.7354	0.2922	0.3974	0.7275	1.1571	3.9597	0.9800	0.1371	0.4209
1.480	0.7328	0.2880	0.3931	0.7311	1.1504	3.9938	0.9815	0.1411	0.4205
1.490	0.7301	0.2839	0.3888	0.7347	1.1435	4.0281	0.9829	0.1452	0.4201
1.500	0.7275	0.2798	0.3846	0.7383	1.1367	4.0625	0.9844	0.1492	0.4196

$\gamma=1.333$

$M$	$\frac{T}{T_0}$	$\frac{p}{p_0}$	$\frac{\rho}{\rho_0}$	$\frac{V}{\sqrt{c_p T_0}}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap_0}$	$\frac{\dot{m} \sqrt{c_p T_0}}{Ap}$	$\frac{F}{\dot{m} \sqrt{c_p T_0}}$	$\frac{4c_f L_{\max}}{D}$	$\frac{\frac{1}{2} \rho V^2}{p_0}$
1.510	0.7248	0.2758	0.3804	0.7419	1.1298	4.0970	0.9859	0.1532	0.4191
1.520	0.7222	0.2718	0.3763	0.7454	1.1228	4.1317	0.9874	0.1573	0.4185
1.530	0.7195	0.2678	0.3722	0.7489	1.1158	4.1665	0.9889	0.1613	0.4178
1.540	0.7169	0.2639	0.3681	0.7524	1.1087	4.2014	0.9905	0.1654	0.4171
1.550	0.7143	0.2600	0.3641	0.7559	1.1016	4.2365	0.9920	0.1694	0.4164
1.560	0.7116	0.2562	0.3600	0.7594	1.0945	4.2717	0.9935	0.1734	0.4156
1.570	0.7090	0.2524	0.3561	0.7629	1.0873	4.3070	0.9950	0.1775	0.4147
1.580	0.7064	0.2487	0.3521	0.7663	1.0801	4.3425	0.9966	0.1815	0.4138
1.590	0.7038	0.2450	0.3482	0.7697	1.0729	4.3782	0.9981	0.1855	0.4129
1.600	0.7011	0.2414	0.3443	0.7731	1.0656	4.4139	0.9997	0.1895	0.4119
1.610	0.6985	0.2378	0.3405	0.7765	1.0583	4.4498	1.0012	0.1935	0.4109
1.620	0.6959	0.2343	0.3367	0.7799	1.0510	4.4859	1.0028	0.1975	0.4098
1.630	0.6933	0.2308	0.3329	0.7832	1.0436	4.5220	1.0043	0.2015	0.4087
1.640	0.6907	0.2273	0.3291	0.7865	1.0363	4.5584	1.0059	0.2055	0.4075
1.650	0.6881	0.2239	0.3254	0.7898	1.0289	4.5948	1.0075	0.2094	0.4063
1.660	0.6855	0.2206	0.3217	0.7931	1.0215	4.6314	1.0090	0.2134	0.4051
1.670	0.6829	0.2172	0.3181	0.7964	1.0141	4.6682	1.0106	0.2173	0.4038
1.680	0.6803	0.2139	0.3145	0.7996	1.0066	4.7051	1.0122	0.2213	0.4025
1.690	0.6777	0.2107	0.3109	0.8028	0.9992	4.7421	1.0137	0.2252	0.4011
1.700	0.6751	0.2075	0.3074	0.8061	0.9918	4.7793	1.0153	0.2291	0.3997
1.710	0.6726	0.2044	0.3039	0.8093	0.9843	4.8166	1.0169	0.2330	0.3983
1.720	0.6700	0.2012	0.3004	0.8124	0.9769	4.8541	1.0184	0.2369	0.3968
1.730	0.6674	0.1982	0.2969	0.8156	0.9694	4.8917	1.0200	0.2407	0.3953
1.740	0.6649	0.1951	0.2935	0.8187	0.9620	4.9294	1.0216	0.2446	0.3938
1.750	0.6623	0.1922	0.2901	0.8218	0.9545	4.9673	1.0232	0.2484	0.3922
1.760	0.6597	0.1892	0.2868	0.8249	0.9471	5.0054	1.0247	0.2522	0.3906
1.770	0.6572	0.1863	0.2835	0.8280	0.9396	5.0435	1.0263	0.2560	0.3890
1.780	0.6546	0.1834	0.2802	0.8311	0.9322	5.0819	1.0279	0.2598	0.3874
1.790	0.6521	0.1806	0.2770	0.8341	0.9248	5.1204	1.0294	0.2636	0.3857
1.800	0.6496	0.1778	0.2737	0.8372	0.9173	5.1590	1.0310	0.2673	0.3840
1.810	0.6471	0.1751	0.2706	0.8402	0.9099	5.1978	1.0326	0.2711	0.3822
1.820	0.6445	0.1723	0.2674	0.8432	0.9025	5.2367	1.0341	0.2748	0.3805
1.830	0.6420	0.1697	0.2643	0.8461	0.8951	5.2758	1.0357	0.2785	0.3787
1.840	0.6395	0.1670	0.2612	0.8491	0.8878	5.3150	1.0373	0.2822	0.3769
1.850	0.6370	0.1644	0.2581	0.8521	0.8804	5.3544	1.0388	0.2858	0.3751
1.860	0.6345	0.1619	0.2551	0.8550	0.8731	5.3939	1.0404	0.2895	0.3732
1.870	0.6320	0.1593	0.2521	0.8579	0.8658	5.4336	1.0419	0.2931	0.3714
1.880	0.6295	0.1568	0.2491	0.8608	0.8585	5.4734	1.0435	0.2967	0.3695
1.890	0.6271	0.1544	0.2462	0.8636	0.8512	5.5134	1.0450	0.3003	0.3676
1.900	0.6246	0.1520	0.2433	0.8665	0.8439	5.5535	1.0466	0.3039	0.3656
1.910	0.6221	0.1496	0.2404	0.8693	0.8367	5.5938	1.0481	0.3074	0.3637
1.920	0.6197	0.1472	0.2376	0.8722	0.8295	5.6342	1.0497	0.3110	0.3617
1.930	0.6172	0.1449	0.2348	0.8750	0.8223	5.6748	1.0512	0.3145	0.3598
1.940	0.6148	0.1426	0.2320	0.8778	0.8152	5.7155	1.0527	0.3180	0.3578
1.950	0.6123	0.1404	0.2292	0.8805	0.8081	5.7564	1.0543	0.3215	0.3558
1.960	0.6099	0.1382	0.2265	0.8833	0.8010	5.7974	1.0558	0.3249	0.3537
1.970	0.6075	0.1360	0.2238	0.8860	0.7939	5.8386	1.0573	0.3284	0.3517
1.980	0.6051	0.1338	0.2212	0.8888	0.7869	5.8800	1.0588	0.3318	0.3497
1.990	0.6026	0.1317	0.2185	0.8915	0.7799	5.9215	1.0603	0.3352	0.3476
2.000	0.6002	0.1296	0.2159	0.8942	0.7729	5.9631	1.0619	0.3386	0.3455

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$
1.05	0.558	79.937	1.0803	1.0567	1.0223	0.9845	0.99995	1.40	8.000	75.893	1.9842	1.6163	1.2276	0.8184	0.96806
1.10	1.515	76.297	1.1658	1.1157	1.0449	0.9711	0.99963		6.000	80.485	2.0575	1.6562	1.2423	0.7762	0.96286
1.15	2.000	67.003	1.1408	1.0986	1.0384	1.0434	0.99977	1.45	2.000	46.004	1.1028	1.0723	1.0284	1.3808	0.99990
	2.671	73.822	1.2565	1.1767	1.0678	0.9598	0.99879		4.000	48.679	1.2169	1.1503	1.0579	1.3091	0.99923
1.20	2.000	61.050	1.1197	1.0841	1.0329	1.1113	0.99985		6.000	51.755	1.3463	1.2357	1.0895	1.2325	0.99733
	3.944	71.977	1.3525	1.2397	1.0910	0.9502	0.99720		8.000	55.517	1.5000	1.3833	1.1250	1.1460	0.99325
1.25	2.000	56.844	1.1110	1.0780	1.0306	1.1696	0.99988		10.000	61.046	1.7114	1.4613	1.1712	1.0317	0.98440
	4.000	61.986	1.2541	1.1752	1.0672	1.0721	0.99882		10.785	67.097	1.9147	1.5779	1.2135	0.92235	0.97269
1.30	2.000	53.474	1.1065	1.0749	1.0294	1.2244	0.99989		10.000	72.994	2.0764	1.6664	1.2461	0.8386	0.96147
	4.000	57.423	1.2334	1.1613	1.0621	1.1398	0.99906		12.000	78.197	2.1836	1.7282	1.2672	0.7777	0.95324
1.35	2.000	50.634	1.0428	1.0103	0.9733	1.0274	0.99585		12.113	64.359	2.2355	1.7501	1.2774	0.7485	0.94905
	4.000	53.965	1.2238	1.1549	1.0596	1.1994	0.99916		12.000	66.589	2.0439	1.6489	1.2396	0.7332	0.94659
1.40	2.000	48.173	1.0303	1.0733	1.0287	1.2774	0.99990		12.000	46.543	1.2165	1.1030	1.0725	1.4316	0.99990
	4.000	51.117	1.2089	1.1516	1.0584	1.2553	0.99921		12.000	49.326	1.3433	1.2337	1.0578	1.3615	0.99923
1.45	2.000	45.372	1.0128	1.0442	1.0128	1.3295	0.99990		12.000	52.571	1.4887	1.3263	1.1224	1.2079	0.99739
	4.000	48.644	1.2232	1.1613	1.0621	1.1398	0.99906		12.000	56.679	1.6662	1.4345	1.1615	1.1144	0.98660
1.50	2.000	42.656	1.0000	1.0248	1.0000	1.0274	0.99585		12.000	64.359	1.9668	1.6068	1.2241	0.9607	0.96925
	4.000	46.886	1.1935	1.1386	1.0942	1.0935	0.99108		12.000	66.589	2.0439	1.6489	1.2396	0.9213	0.96385
1.55	2.000	38.861	1.0000	1.0248	1.0000	1.0274	0.99585		12.000	68.790	2.1147	1.6869	1.2537	0.8849	0.95860
	4.000	42.095	1.1935	1.1386	1.0942	1.0935	0.99108		12.000	75.995	2.3046	1.7855	1.2908	0.7854	0.94329
1.60	2.000	33.058	1.0000	1.0248	1.0000	1.0274	0.99585		12.000	79.712	2.3746	1.8207	1.3042	0.7476	0.93725
	4.000	37.311	1.1935	1.1386	1.0942	1.0935	0.99108		12.000	82.662	2.4155	1.8410	1.3121	0.7250	0.9363
1.65	2.000	28.088	1.0000	1.0248	1.0000	1.0274	0.99585		12.000	85.256	2.4404	1.8533	1.3168	0.7112	0.93141
	4.000	32.319	1.1935	1.1386	1.0942	1.0935	0.99108		12.000	87.668	2.4540	1.8599	1.3194	0.7035	0.93018
1.70	2.000	22.861	1.0000	1.0248	1.0000	1.0274	0.99585		12.000	90.000	2.5000	1.8657	1.3237	0.7000	0.92999
	4.000	27.193	1.1935	1.1386	1.0942	1.0935	0.99108		12.000	92.431	2.5345	1.8714	1.3326	0.7067	0.92999
1.75	2.000	18.412	1.0000	1.0248	1.0000	1.0274	0.99585		12.000	94.762	2.5633	1.8771	1.3414	0.7134	0.92999
	4.000	22.744	1.1935	1.1386	1.0942	1.0935	0.99108		12.000	97.193	2.6021	1.8829	1.3502	0.7207	0.93018
1.80	2.000	11.432	1.0000	1.0248	1.0000	1.0274	0.99585		12.000	99.524	2.6310	1.8887	1.3592	0.7279	0.93018
	4.000	15.764	1.1935	1.1386	1.0942	1.0935	0.99108		12.000	101.855	2.6600	1.8945	1.3680	0.7345	0.93018
1.85	2.000	6.154	1.0000	1.0248	1.0000	1.0274	0.99585		12.000	104.186	2.6888	1.9002	1.3767	0.7414	0.93018
	4.000	10.486	1.1935	1.1386	1.0942	1.0935	0.99108		12.000	106.517	2.7177	1.9060	1.3850	0.7487	0.93018
1.90	2.000	1.736	1.0000	1.0248	1.0000	1.0274	0.99585		12.000	108.848	2.7465	1.9118	1.3938	0.7557	0.93018
	4.000	6.068	1.1935	1.1386	1.0942	1.0935	0.99108		12.000	111.179	2.7753	1.9176	1.4026	0.7626	0.93018
1.95	2.000	0.408	1.0000	1.0248	1.0000	1.0274	0.99585		12.000	113.510	2.8041	1.9234	1.4114	0.7700	0.93018
	4.000	4.740	1.1935	1.1386	1.0942	1.0935	0.99108		12.000	115.841	2.8329	1.9292	1.4192	0.7779	0.93018
2.00	2.000	-0.238	1.0000	1.0248	1.0000	1.0274	0.99585		12.000	118.172	2.8617	1.9350	1.4270	0.7857	0.93018
	4.000	5.070	1.1935	1.1386	1.0942	1.0935	0.99108		12.000	120.503	2.8905	1.9408	1.4358	0.7935	0.93018
2.05	2.000	-1.736	1.0000	1.0248	1.0000	1.0274	0.99585		12.000	122.834	2.9193	1.9466	1.4436	0.7983	0.93018
	4.000	5.408	1.1935	1.1386	1.0942	1.0935	0.99108		12.000	125.165	2.9481	1.9524	1.4514	0.8061	0.93018
2.10	2.000	-3.068	1.0000	1.0248	1.0000	1.0274	0.99585		12.000	127.506	2.9769	1.9582	1.4592	0.8139	0.93018
	4.000	6.040	1.1935	1.1386	1.0942	1.0935	0.99108		12.000	129.837	3.0057	1.9640	1.4670	0.8217	0.93018
2.15	2.000	-4.738	1.0000	1.0248	1.0000	1.0274	0.99585		12.000	132.168	3.0345	1.9708	1.4748	0.8295	0.93018
	4.000	6.710	1.1935	1.1386	1.0942	1.0935	0.99108		12.000	134.500	3.0633	1.9766	1.4826	0.8373	0.93018
2.20	2.000	-3.378	1.0000	1.0248	1.0000	1.0274	0.99585		12.000	136.831	3.0921	1.9824	1.4904	0.8451	0.93018
	4.000	7.350	1.1935	1.1386	1.0942	1.0935	0.99108		12.000	139.162	3.1209	1.9882	1.4982	0.8529	0.93018
2.25	2.000	-2.110	1.0000	1.0248	1.0000	1.0274	0.99585		12.000	141.493	3.1497	1.9940	1.5060	0.8607	0.93018
	4.000	8.082	1.1935	1.1386	1.0942	1.0935	0.99108		12.000	143.824	3.1785	2.0008	1.5138	0.8685	0.93018
2.30	2.000	-0.868	1.0000	1.0248	1.0000	1.0274	0.99585		12.000	146.155	3.2073	2.0066	1.5216	0.8763	0.93018
	4.000	8.854	1.1935	1.1386	1.0942	1.0935	0.99108		12.000	148.486	3.2361	2.0124	1.5294	0.8841	0.93018
2.35	2.000	-1.508	1.0000	1.0248	1.0000	1.0274	0.99585		12.000	150.817	3.2649	2.0182	1.5372	0.8919	0.93018
	4.000	9.826	1.1935	1.1386	1.0942	1.0935	0.99108		12.000	153.148	3.2937	2.0240	1.5450	0.9007	0.93018
2.40	2.000	-0.240	1.0000	1.0248	1.0000	1.0274	0.99585		12.000	155.479	3.3225	2.0308	1.5528	0.9085	0.93018
	4.000	10.798	1.1935	1.1386	1.0942	1.0935	0.99108		12.000	157.810	3.3513	2.0366	1.5606	0.9163	0.93018
2.45	2.000	-1.878	1.0000	1.0248	1.0000	1.0274	0.99585		12.000	160.141	3.3791	2.0424	1.5684	0.9242	0.93018
	4.000	11.770	1.1935	1.1386	1.0942	1.0935	0.99108		12.000	162.472	3.4079	2.0482	1.5762	0.9320	0.93018
2.50	2.000	-0.538	1.0000	1.0248	1.0000	1.0274	0.99585		12.000	164.803	3.4367	2.0540	1.5840	0.9408	0.93018
	4.000	12.642	1.1935	1.1386	1.0942	1.0935	0.99108		12.000	167.134	3.4655	2.0608	1.5918	0.9486	0.93018
2.55	2.000	-1.298	1.0000	1.0248	1.0000	1.0274	0.99585		12.000	169.465	3.4943	2.0666	1.6096	0.9564	0.93018
	4.000	13.511	1.1935	1.1386	1.0942	1.0935	0.99108		12.000	171.796	3.5231	2.0724	1.6174	0.9642	0.93018
2.60	2.000	-0.958	1.0000	1.0248	1.0000	1.0274	0.99585		12.000	174.127	3.5519	2.0782	1.6252	0.9719	0.93018
	4.000	14.389	1.1935	1.1386	1.0942										

### Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	Oblique Shock Tables ( $\gamma = 1.4$ )			$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$		
							$M_1$	$\theta$	$\beta$						
1.60	2.000	40.724	1.1046	1.0736	1.0289	1.5323	0.99990	1.70	6.000	84.848	3.1778	2.1865	1.4534	0.6547	0.85856
	4.000	42.931	1.2189	1.1516	1.0584	1.4638	0.99921		4.000	86.619	3.1933	2.1929	1.4562	0.6467	0.85695
	6.000	45.344	1.3446	1.2346	1.0891	1.3934	0.99736		2.000	88.325	3.2021	2.1965	1.4578	0.6421	0.85602
	8.000	48.030	1.4843	1.6430	1.4207	1.3236	1.1215		1.3195	0.98376					
	10.000	51.116	1.6430	1.8320	1.5311	1.1965	1.2397		0.98766	1.027781					
	12.000	54.889	1.8320	2.0974	1.6777	1.2502	1.0232		0.95990	1.0232					
	14.000	60.537	2.0974	2.7576	2.0035	1.3764	0.7018		0.90139	14.000					
	14.652	65.828	2.3192	1.7929	1.2936	0.9136	0.94204		6.000	40.756	1.0765	1.0300	1.6816	0.6547	0.99989
	14.000	70.895	2.5000	1.8824	1.3281	1.8824	1.8320		0.92598	8.000	43.034	1.1087	1.0605	1.6133	0.99913
	12.000	75.900	2.6428	1.9504	1.3550	1.5166	1.1483		0.91256	10.000	45.531	1.0765	1.1271	1.5441	0.99713
	10.000	79.102	2.7132	1.9831	1.3682	1.7250	0.90574		12.000	48.319	1.8263	1.4266	1.733	0.99334	
	8.000	81.691	2.7576	2.0035	1.3764	0.7018	0.90139		14.000	51.547	2.0245	1.5279	1.953	0.98721	
	6.000	83.967	2.7870	2.0168	1.3879	0.6862	0.89848		16.000	55.589	2.2652	1.6384	2.357	0.97814	
	4.000	86.061	2.8059	2.0254	1.3854	0.6761	0.89660		18.000	62.944	2.6670	1.9617	3.595	0.9645	
	2.000	88.054	2.8166	2.0302	1.3873	0.6703	0.89554		18.121	65.134	2.7745	2.0112	3.795	0.91023	
									18.000	67.269	2.8728	2.0554	3.977	0.89972	
									16.000	73.757	3.1267	2.1651	4.441	0.8766	0.88991
									14.000	76.988	3.2251	2.2060	4.620	0.7735	0.86389
									12.000	79.465	3.2868	2.2312	4.731	0.7175	0.85362
									10.000	81.570	3.3295	2.2484	4.808	0.6878	0.84714
									8.000	83.451	3.3598	2.2606	4.862	0.6669	0.84266
									6.000	85.190	3.3811	2.2691	4.901	0.6409	0.83722
									4.000	86.838	3.3954	2.2748	4.926	0.6337	0.83571
									2.000	88.432	3.4036	2.2780	4.941	0.6295	0.83485
1.65	2.000	39.267	1.1058	1.0744	1.0292	1.5823	0.99990								
	4.000	41.377	1.2212	1.1531	1.0590	1.5140	0.99919								
	6.000	43.665	1.3475	1.2365	1.0898	1.4444	0.99730								
	8.000	46.181	1.4869	1.3252	1.1221	1.3720	0.99367								
	10.000	49.007	1.6429	1.4206	1.1565	1.2952	0.98766								
	12.000	52.312	1.8224	1.5257	1.1945	1.2104	0.97837								
	14.000	56.541	2.0441	1.6490	1.2396	1.1909	0.96384								
	15.855	65.547	2.46653	1.8655	1.3215	0.9184	0.92915								
	14.000	73.864	2.7642	2.0065	1.3776	0.7782	0.90073								
	12.000	77.411	2.8587	2.0491	1.3951	0.7317	0.89132								
	10.000	80.102	2.9157	2.0744	1.4056	0.7029	0.88557								
	8.000	82.389	2.9539	2.0911	1.4126	0.6833	0.88169								
	6.000	84.446	2.9798	2.1024	1.4174	0.6697	0.87904								
	4.000	86.364	2.9968	2.1097	1.4205	0.6607	0.87730								
	2.000	88.200	3.0065	2.1139	1.4222	0.6556	0.87631								
									10.000	44.057	1.6611	1.4915	1.604	1.4494	0.99998
									12.000	46.686	1.8345	1.5326	1.970	1.3725	0.99909
									14.000	49.661	2.0295	1.6411	1.2367	1.2896	0.99701
									16.000	53.198	2.2568	1.7611	1.2815	1.1958	0.94729
									18.000	57.995	2.5516	1.9072	1.3379	1.0766	0.92120
1.70	2.000	37.927	1.1072	1.0597	1.0295	1.6320	0.99989								
	4.000	39.957	1.2239	1.1550	1.0907	1.5638	0.99916								
	6.000	42.145	1.3514	1.2390	1.4946	1.4946	0.99722								
	8.000	44.528	1.4914	1.3280	1.1231	1.4232	0.99353								
	10.000	47.167	1.6466	1.4228	1.1573	1.3482	0.98750								
	12.000	50.168	1.8216	1.5252	1.1943	1.2674	0.97841								
	14.000	53.771	2.0273	1.6399	1.2362	1.1757	0.96504								
	16.000	58.794	2.2989	1.7831	1.2898	1.0569	0.94369								
	17.012	65.319	2.6171	1.9383	1.3502	0.9185	0.91502								
	16.000	71.426	2.8629	2.0510	1.3959	0.8077	0.89090								
	14.000	75.670	2.9984	2.1104	1.4208	0.7439	0.87713								
	12.000	78.555	3.0722	2.1421	1.4342	0.7080	0.86953								
	10.000	80.906	3.1208	2.1626	1.4431	0.6838	0.86450								
	8.000	82.965	3.1544	2.1767	1.4492	0.6667	0.86100								

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	
1.85	2.000	34.466	1.1121	1.0788	1.0309	0.99988	1.95	2.000	32.528	1.1160	1.0319	1.8790	0.99987	
	4.000	36.323	1.2343	1.1619	1.0623	0.99905	1.7114	4.000	34.304	1.2424	1.1674	1.8085	0.99896	
	6.000	38.302	1.3672	1.2492	1.0945	0.99689	1.6418	6.000	36.191	1.3801	1.2575	1.0975	0.99660	
	8.000	40.424	1.5123	1.3409	1.1278	0.99284	1.5711	8.000	38.204	1.5302	1.3521	1.1318	0.99221	
	10.000	42.717	1.6709	1.4373	1.1625	0.98638	1.4983	10.000	40.360	1.6938	1.4509	1.1674	0.98528	
	12.000	45.223	1.8453	1.5388	1.1992	0.97701	1.4224	12.000	42.688	1.8726	1.5542	1.2049	0.97535	
	14.000	48.014	2.0395	1.6465	1.2387	0.96417	1.3415	14.000	45.230	2.0693	1.6625	1.2446	1.4396	0.96200
	16.000	51.232	2.2607	1.7631	1.2822	0.94697	1.2524	16.000	48.059	2.2879	1.7770	1.2875	1.3553	0.94470
	18.000	55.227	2.5275	1.8956	1.3333	0.92345	1.1476	18.000	51.320	2.5368	1.9001	1.3351	1.2622	0.92258
	20.000	62.099	2.9519	2.0902	1.4123	0.88189	0.9818	20.000	55.381	2.8378	2.0397	1.3913	1.1520	0.89342
	20.198	64.872	3.1062	2.1565	1.4404	0.9205	0.9601	22.000	62.860	3.3464	2.2553	1.4938	0.9655	0.84087
	20.000	67.544	3.2437	2.2136	1.4653	0.85167	1.4653	22.092	64.716	3.4603	2.3003	1.5043	0.9229	0.82885
	18.000	73.440	3.5019	2.3165	1.5117	0.7560	0.82446	22.000	66.523	3.5685	2.3410	1.5231	0.8829	0.81774
	16.000	76.511	3.6090	2.3576	1.5308	0.7085	0.81314	20.000	72.926	3.8872	2.4601	1.5801	0.7555	0.78384
	14.000	78.861	3.6772	2.3883	1.5429	0.6773	0.80593	18.000	75.964	4.0086	2.5030	1.6015	0.7045	0.77114
	12.000	80.844	3.7252	2.4011	1.5514	0.6548	0.80088	16.000	78.253	4.0857	2.5297	1.6151	0.6710	0.76313
	10.000	82.606	3.7601	2.4140	1.5576	0.6381	0.79719	14.000	80.165	4.1401	2.5484	1.6246	0.6467	0.75750
	8.000	84.222	3.7858	2.4234	1.5622	0.6257	0.79449	12.000	81.849	4.1804	2.5620	1.6317	0.6283	0.75335
	6.000	85.740	3.8042	2.4301	1.5655	0.6166	0.79255	10.000	83.381	4.2106	2.5722	1.6370	0.6142	0.75024
	4.000	87.193	3.8167	2.4346	1.5677	0.6105	0.79124	8.000	84.808	4.2333	2.5798	1.6409	0.6036	0.74791
	2.000	88.606	3.8239	2.4373	1.5689	0.6069	0.79048	6.000	86.163	4.2497	2.5853	1.6438	0.5957	0.74623
								2.000	87.467	4.2609	2.5890	1.6458	0.5904	0.74508
								2.000	88.741	4.2674	2.5912	1.6469	0.5872	0.74441
	1.90	2.000	33.466	1.1140	1.0801	1.0314	1.8298	0.99987						
	4.000	35.279	1.2382	1.3735	1.1646	1.0633	1.7600	0.99901						
	6.000	37.209	1.3253	1.5253	1.1297	1.0959	1.6901	0.99675	2.00	2.000	31.647	1.1180	1.0829	0.99986
	8.000	39.272	1.5209	1.3463	1.1297	1.0959	1.6191	0.99254	4.000	4.000	33.390	1.2468	1.1702	1.0654
	10.000	41.490	1.6818	1.4438	1.1649	1.5464	0.98586	1.5464	6.000	35.241	1.3871	1.2620	1.0991	0.99644
	12.000	43.898	1.8582	1.5460	1.2019	1.4709	0.97624	1.4709	8.000	37.210	1.5400	1.3581	1.1339	0.99186
	14.000	46.550	2.0530	1.6538	1.2414	1.3913	0.96319	1.3913	10.000	39.314	1.7066	1.4584	1.1702	0.98464
	16.000	49.544	2.2718	1.7688	1.2844	1.3052	0.94605	1.2844	12.000	41.575	1.8884	1.5631	1.2881	0.97437
	18.000	53.095	2.5263	1.8951	1.3331	1.2077	0.92356	1.2077	14.000	44.029	2.0876	1.6724	1.2483	0.96064
	20.000	57.900	2.8557	2.0477	1.3946	1.0835	0.89162	1.3946	16.000	46.731	2.3076	1.7870	1.2913	1.4034
	21.167	64.783	3.2805	2.2286	1.4720	0.9216	0.84781	1.4720	18.000	49.785	2.5546	1.9086	1.3384	1.3131
	20.000	71.057	3.6012	2.3546	1.5012	0.81397	0.7935	0.81397	20.000	53.423	2.8429	2.0420	1.3922	0.92092
	18.000	74.861	3.7578	2.4131	1.5572	0.7274	0.79744	0.79744	22.000	58.457	3.2228	2.2051	1.4616	0.89291
	16.000	77.463	3.8466	2.4455	1.5729	0.6884	0.78810	0.78810	22.974	64.669	3.6458	2.3715	1.5373	0.9243
	14.000	79.565	3.9068	2.4671	1.5836	0.6611	0.78178	0.78178	22.000	70.332	3.9714	2.4899	1.5950	0.8017
	12.000	81.383	3.9504	2.4826	1.5913	0.6409	0.77721	0.77721	20.000	74.270	4.1570	2.5541	1.6276	0.77503
	10.000	83.020	3.9828	2.4940	1.5970	0.6257	0.77383	0.77383	18.000	76.862	4.2589	2.5883	1.6487	0.7278
	8.000	84.534	4.0068	2.5024	1.6012	0.6142	0.77133	0.77133	16.000	78.921	4.3277	2.6110	1.6574	0.6854
	6.000	85.965	4.0241	2.5084	1.6042	0.6058	0.76953	0.76953	14.000	80.684	4.3777	2.6274	1.6662	0.6558
	4.000	87.338	4.0359	2.5125	1.6063	0.6001	0.76830	0.76830	12.000	82.257	4.4153	2.6396	1.6727	0.6337
	2.000	88.677	4.0428	2.5149	1.6075	0.5967	0.76759	0.76759	10.000	83.700	4.4438	2.6487	1.6777	0.6037
									8.000	85.052	4.4653	2.6556	1.6815	0.5937
									6.000	86.339	4.4810	2.6606	1.6642	0.5864
									4.000	87.582	4.4917	2.6640	1.6861	0.5813
									2.000	88.798	4.4979	2.6660	1.6871	0.5783

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$P_1$	$\frac{P_2}{P_1}$	$\frac{M_2}{M_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{T_2}{T_1}$	$\frac{\rho_2}{\rho_1}$	$P_2$
2.05	2.000	30.816	1.1200	1.0843	1.0330	1.9771	0.99985	2.10	4.000	87.778	4.9706	2.8097
	4.000	32.532	1.2512	1.1732	1.0665	1.9050	0.99885	2.000	2.000	88.894	4.9764	2.8113
	6.000	34.350	1.3943	1.2666	1.1008	1.8330	0.99627					
	8.000	36.281	1.5502	1.3644	1.1362	1.7605	0.99148					
	10.000	38.341	1.7201	1.4664	1.1730	1.6868	0.98396					
	12.000	40.547	1.9053	1.5726	1.2116	1.6111	0.97330					
	14.000	42.928	2.1076	1.6831	1.2522	1.5326	0.95914					
	16.000	45.528	2.3300	1.7983	1.2956	1.4500	0.94112					
	18.000	48.428	2.5774	1.9195	1.3427	1.3614	0.91878					
	20.000	51.785	2.8600	2.0497	1.3953	1.2630	0.89120					
2.10	22.000	56.032	3.2057	2.1980	1.4585	1.1444	0.85565					
	23.814	64.638	3.8367	2.4419	1.5712	0.9257	0.78913					
	22.000	72.193	4.2777	2.5946	1.6487	0.7626	0.74336					
	20.000	75.324	4.4215	2.6416	1.6738	0.7056	0.72876					
	18.000	77.614	4.5107	2.6700	1.6894	0.6888	0.71981					
	16.000	79.498	4.5734	2.6898	1.7003	0.6622	0.71356					
	14.000	81.138	4.6199	2.7043	1.7084	0.6219	0.70894					
	12.000	82.617	4.6553	2.7152	1.7145	0.6062	0.70545					
	10.000	83.983	4.6824	2.7236	1.7192	0.5939	0.70278					
	8.000	85.269	4.7029	2.7299	1.7228	0.5846	0.70077					
2.15	6.000	86.497	4.7179	2.7344	1.7254	0.5776	0.69930					
	4.000	87.685	4.7283	2.7376	1.7272	0.5728	0.69829					
	2.000	88.849	4.7343	2.7394	1.7282	0.5700	0.69770					
	2.000	30.033	1.1222	1.0858	1.0335	2.0260	0.99984					
	4.000	31.723	1.2558	1.1763	1.0676	1.9530	0.99880					
	6.000	33.513	1.4017	1.2714	1.1025	1.8801	0.99609					
	8.000	35.412	1.5608	1.3709	1.1386	1.8069	0.99108					
	10.000	37.433	1.7342	1.4746	1.1760	1.7325	0.98324					
	12.000	39.592	1.9230	1.5825	1.2152	1.6564	0.97216					
	14.000	41.912	2.1290	1.6944	1.2565	1.5777	0.97570					
2.20	16.000	44.430	2.3547	1.8107	1.3004	1.4954	0.93899					
	18.000	47.210	2.6041	1.9322	1.3478	1.4078	0.91626					
	20.000	50.365	2.8848	2.0607	1.3999	1.3122	0.88870					
	22.000	54.169	3.2152	2.2019	1.4602	1.2019	0.85466					
	24.000	59.767	3.6739	2.3820	1.5424	1.0493	0.80628					
	24.614	64.621	4.0332	2.5116	1.6058	0.9273	0.76858					
	24.000	69.104	4.3238	2.6098	1.6568	0.8245	0.73867					
	22.000	73.521	4.5644	2.6870	1.6987	0.7345	0.71445					
	20.000	76.189	4.6852	2.7244	1.7197	0.6870	0.70251					
	18.000	78.257	4.7652	2.7488	1.7336	0.6543	0.69468					
2.25	16.000	80.001	4.8232	2.7662	1.7436	0.6299	0.68906					
	14.000	81.539	4.8669	2.7792	1.7512	0.6111	0.68484					
	12.000	82.938	4.9006	2.7892	1.7570	0.5964	0.68162					
	10.000	84.237	4.9264	2.7968	1.7615	0.5649	0.67914					
	8.000	85.463	4.9461	2.8025	1.7649	0.5760	0.67726					
	6.000	86.638	4.9606	2.8068	1.7664	0.5694	0.67588					

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	
2.20	20.000	77.549	5.2175	2.8799	1.8117	0.6568	0.65185	2.30	16.000	40.816	2.4701	1.8678	0.92872	
	18.000	79.308	5.2856	2.8987	1.8234	0.6296	0.64562		18.000	43.299	2.7360	1.9936	1.3724	0.90351
	16.000	80.839	5.3369	2.9127	1.8323	0.6086	0.64096		20.000	46.007	3.0276	2.1230	1.4261	0.87413
	14.000	82.216	5.3764	2.9235	1.8391	0.5921	0.63739		22.000	49.026	3.3514	2.2573	1.4847	0.84035
	12.000	83.483	5.4073	2.9318	1.8444	0.5789	0.63462		24.000	52.536	3.7216	2.3998	1.5508	0.80125
	10.000	84.670	5.4313	2.9382	1.8485	0.5686	0.63247		26.000	57.077	4.1819	2.5625	1.6319	0.75319
	8.000	85.798	5.4497	2.9431	1.8517	0.5605	0.63083		27.454	64.653	4.8739	2.7813	1.7524	0.93338
	6.000	86.883	5.4633	2.9468	1.8540	0.5545	0.62962		26.000	71.264	5.3682	2.9212	1.8377	0.68417
	4.000	87.938	5.4727	2.9493	1.8556	0.5503	0.62879		24.000	74.512	5.5649	2.9736	1.8714	0.7060
	2.000	88.973	5.4782	2.9507	1.8565	0.5479	0.62830		22.000	76.770	5.6817	3.0039	1.8915	0.66335
2.25	2.000								20.000	78.582	5.7631	3.0246	1.9054	0.6328
	4.000	29.555	1.2703	1.1859	1.0903	1.0353	2.1725		18.000	80.133	5.8238	3.0399	1.9158	0.6092
	6.000	31.277	1.4254	1.2864	1.1080	1.0712	2.0962		16.000	81.509	5.8705	3.0515	1.9238	0.59445
	8.000	33.102	1.5949	1.3916	1.1461	1.1461	2.0203		14.000	82.764	5.9071	3.0606	1.9301	0.5757
	10.000	35.034	1.7798	1.5011	1.1856	1.1856	1.9443		12.000	83.928	5.9360	3.0677	1.9350	0.5638
	12.000	37.088	1.9812	1.6147	1.2270	1.2270	1.8674		10.000	85.026	5.9586	3.0732	1.9389	0.5543
	14.000	39.277	2.2004	1.7319	1.2705	1.2705	1.7891		8.000	86.074	5.9761	3.0775	1.9419	0.5469
	16.000	41.623	2.4392	1.8527	1.3166	1.3166	1.6257		6.000	87.085	5.9890	3.0807	1.9441	0.5413
	18.000	44.161	2.7000	1.9770	1.3657	1.3657	1.5388		4.000	88.070	5.9980	3.0828	1.9456	0.5374
	20.000	46.948	2.9871	2.1055	1.4187	1.4187	1.4466		2.000	89.039	6.0033	3.0841	1.9465	0.5352
	22.000	50.091	3.3085	2.2400	1.4770	1.3464	0.84486							
	24.000	53.837	3.6830	2.3854	1.5440	1.2318	0.80532							
	26.000	59.122	4.1839	2.5632	1.6323	1.0792	0.75298							
	26.795	64.633	4.6556	2.7153	1.7145	0.9321	0.70542							
	26.000	69.627	5.0238	2.8250	1.7783	0.9115	0.66991							
	24.000	73.634	5.2707	2.8946	1.8209	0.7254	0.64698							
	22.000	76.145	5.4009	2.9301	1.8453	0.6775	0.63519							
	20.000	78.098	5.4884	2.9534	1.8583	0.6441	0.62279							
	18.000	79.744	5.5523	2.9703	1.8693	0.6189	0.62175							
	16.000	81.192	5.6011	2.9830	1.8776	0.5993	0.61749							
	14.000	82.504	5.6391	2.9929	1.8842	0.5836	0.61418							
	12.000	83.716	5.6688	3.0006	1.8893	0.5711	0.61161							
	10.000	84.856	5.6921	3.0065	1.8932	0.5612	0.60960							
	8.000	85.942	5.7100	3.0111	1.8963	0.5535	0.60806							
	6.000	86.988	5.7233	3.0145	1.8986	0.5477	0.60692							
	4.000	88.007	5.7324	3.0168	1.9002	0.5437	0.60614							
	2.000	89.008	5.7378	3.0182	1.9011	0.5413	0.60568							
2.30	2.000													
	4.000	27.294	1.1311	1.0919	1.0359	2.2212	0.9981							
	6.000	28.906	1.2753	1.1892	1.0724	2.1437	0.9954							
	8.000	30.611	1.4336	1.2916	1.1099	2.0667	0.99526							
	10.000	32.415	1.6068	1.3988	1.1487	1.9896	0.9823							
	12.000	34.326	1.7959	1.5104	1.1890	1.9117	0.97989							
	14.000	36.354	2.0019	1.6260	1.2311	1.8325	0.96684							
	14.000	38.510	2.2261	1.7452	1.2755	1.7514	0.94982							
	2.30													

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$		
2.35	6.000	87.174	6.2606	3.1453	1.9904	0.5353	0.56272	2.45	26.000	53.045	4.3053	2.6037	0.74055		
	4.000	88.129	6.2694	3.1474	1.9919	0.5315	0.56203		28.000	57.780	4.8455	2.7729	0.68691		
2.40	4.000	89.068	6.2745	3.1486	1.9928	0.5293	0.56162		29.253	64.744	5.5614	2.9727	0.62095		
	2.000	27.702	29.377	1.4505	1.1960	1.0951	1.0371	2.3184	0.99979	20.000	70.828	6.0810	3.1029	0.7837	
2.45	2.000	26.120	1.1358	1.2856	1.1960	1.0749	1.1138	2.2383	0.99839	24.000	74.185	6.3161	3.1582	0.57709	
	4.000	31.149	41.748	1.4505	1.6314	1.4137	1.540	2.0794	0.99888	18.000	81.089	6.6682	3.1891	0.55836	
2.50	10.000	33.023	12.000	35.007	12.000	1.28292	1.5295	1.1959	1.994	0.97797	16.000	82.299	6.7105	3.2464	0.54787
	20.000	44.336	47.174	1.20450	1.6495	1.23988	1.6495	1.23988	1.9181	0.96377	14.000	83.416	6.7442	3.2101	0.54076
2.55	14.000	37.112	2.2798	1.7729	1.2860	1.8350	1.2860	1.8350	0.94538	12.000	84.462	6.7710	3.2536	0.52599	
	24.000	59.351	39.351	2.5351	1.8993	1.3348	1.3348	1.3348	0.92274	10.000	85.455	6.7923	3.2640	0.6042	
2.60	18.000	41.748	2.8128	2.0285	2.0285	1.38866	1.38866	1.6613	0.89592	8.000	86.408	6.8088	3.2675	0.53555	
	28.000	59.656	64.710	5.3269	2.1604	1.4421	1.4421	1.5689	0.86505	6.000	87.331	6.8211	3.2701	0.53157	
2.65	22.000	47.174	3.4480	2.2955	1.5021	1.4709	1.5021	1.5021	0.83015	4.000	88.232	6.8296	3.2719	0.5681	
	24.000	50.371	3.8196	2.4357	1.56982	1.3644	1.56982	1.3644	0.70983	2.000	89.119	6.8346	3.2730	0.52845	
2.70	26.000	54.184	4.2521	2.5861	1.6442	1.2426	1.2426	1.2426	0.74598					0.52249	
	28.000	59.656	4.83862	2.7707	1.74662	1.0779	1.0779	0.68761	0.9370	0.64187	2.50	2.000	25.050	0.52129	
2.75	28.681	64.710	5.3269	2.9100	1.83035	1.83035	1.83035	1.83035	0.60781	0.60781	4.000	26.609	2.0774	0.52041	
	28.000	69.291	5.7130	3.0119	1.89688	0.8201	1.89688	0.8201	0.58331	6.000	28.259	1.4679	3.1333	0.51979	
2.80	26.000	73.400	6.00448	3.0845	1.94688	0.7260	0.58331	0.57121	0.6751	8.000	30.005	1.65668	1.1595	0.51943	
	24.000	75.889	6.15339	3.1203	1.9722	1.9722	1.9722	1.9722	0.57121	10.000	31.851	1.86339	1.5493	0.51943	
2.85	22.000	77.803	6.2534	3.1436	1.9892	0.6397	0.56329	0.56329	0.55758	12.000	33.802	2.0900	1.6737	0.51943	
	20.000	79.402	6.3260	3.1605	2.0016	0.6129	0.55758	0.55758	0.55326	14.000	35.866	2.3364	1.2969	0.51943	
2.90	18.000	80.800	6.3816	3.1732	2.0111	0.5919	0.55326	0.55326	0.54990	16.000	38.057	2.6042	1.9322	0.51943	
	16.000	82.059	6.4251	3.1831	2.0185	0.5751	0.54990	0.54990	0.54726	18.000	40.389	2.8949	2.0652	0.51943	
2.95	14.000	83.217	6.4596	3.1909	2.0244	0.5615	0.54726	0.54726	0.54517	20.000	42.890	3.2109	2.2002	0.51943	
	12.000	84.299	6.4870	3.1971	2.0290	0.5505	0.54517	0.54517	0.54352	22.000	45.602	3.55558	2.3373	0.51943	
3.00	10.000	85.324	6.5087	3.2019	2.0327	0.5416	0.54352	0.54352	0.53936	24.000	48.600	3.93661	2.4775	0.51943	
	8.000	86.306	6.5254	3.2057	2.0356	0.5348	0.54225	0.54225	0.54131	26.000	52.036	4.36567	2.6235	0.51943	
3.05	6.000	87.255	6.53779	3.2085	2.0377	0.5296	0.54065	0.54065	0.53932	28.000	56.335	4.8844	2.7844	0.51943	
	4.000	88.182	6.5466	3.2104	2.0392	0.5260	0.54065	0.54065	0.53892	20.979	64.782	5.8014	3.0342	0.68317	
3.10	2.000	89.094	6.5517	3.2115	2.0400	0.5238	0.54027	0.54027	0.53892	28.000	71.949	6.4249	3.1831	0.9402	
	2.45	25.572	1.1381	1.0968	1.0377	0.99978	2.3670	2.2855	0.999831	24.000	74.856	6.6273	3.2282	0.7573	
3.15	4.000	27.143	1.2908	1.1994	1.0762	1.1157	2.2048	0.99453	22.000	76.939	6.7526	3.2555	2.0742	0.51417	
	6.000	28.805	1.4591	1.3078	1.1567	1.1567	1.1567	1.1567	0.98761	20.000	80.070	6.9082	3.2885	2.1007	
3.20	8.000	30.563	1.6440	1.4212	1.241	1.0466	2.0466	1.994	0.9795	18.000	81.353	6.9602	3.2994	2.1095	
	10.000	32.422	1.8463	1.5393	1.2048	1.0466	2.0466	1.994	0.9795	16.000	82.518	7.0014	3.3080	2.1165	
3.25	12.000	34.388	2.0672	1.6615	1.2442	1.0466	1.9603	1.9603	0.96215	14.000	83.598	7.0343	3.3148	2.1221	
	14.000	36.472	2.3078	1.7871	1.2914	1.0466	1.8762	1.8762	0.94302	12.000	84.612	7.0607	3.3202	2.1266	
3.30	16.000	38.685	2.5692	1.9156	1.3412	1.0466	1.7898	1.7898	0.91955	10.000	85.576	7.0816	3.3245	2.1301	
	18.000	41.047	2.85532	2.0466	1.3941	1.0466	1.89187	1.7006	0.89187	8.000	86.502	7.0979	3.3278	2.1329	
3.35	20.000	43.588	3.1623	2.1800	1.4506	1.0466	1.6077	0.86018	0.85018	6.000	87.400	7.1100	3.3303	2.1350	
	22.000	46.358	3.5007	2.3160	1.5115	1.0466	1.5097	0.82459	0.82459	4.000	88.277	7.1184	3.3320	2.1364	
3.40	24.000	49.445	3.8759	2.4560	1.4042	1.0466	1.4042	1.4042	0.78502	2.000	89.142	7.1234	3.3330	2.1372	

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$		
2.55	2.000	24.550	1.1429	1.1001	1.0390	2.4639	0.99976	2.60	30.814	64.866	6.2972	3.1538	1.9967	0.9433	
	4.000	26.099	1.3015	1.2065	1.0788	2.3796	0.99814		30.000	69.778	6.7777	3.2609	2.0785	0.8111	
	6.000	27.739	1.4768	1.3189	1.1198	2.2961	0.99999		28.000	73.590	7.0906	3.3263	2.1317	0.7189	
	8.000	29.474	1.6699	1.4367	1.1623	2.2128	0.98642		26.000	75.955	7.2555	3.3596	2.1596	0.6673	
	10.000	31.307	1.8817	1.5593	1.2067	2.1288	0.97479		24.000	77.778	7.3665	3.3815	2.1785	0.6311	
	12.000	33.244	2.1133	1.6861	1.2534	2.0438	0.95371		22.000	79.299	7.4481	3.3974	2.1923	0.6035	
	14.000	35.293	2.3656	1.8162	1.3025	1.9573	0.93803		20.000	80.626	7.5108	3.4095	2.2029	0.5817	
	16.000	37.463	2.6399	1.9490	1.3545	1.8687	0.91283		18.000	81.815	7.5602	3.4189	2.2113	0.5641	
	18.000	39.770	2.9378	2.0840	1.4097	1.7776	0.88933		16.000	82.906	7.5997	3.4264	2.2180	0.5497	
	20.000	42.236	3.2611	2.2207	1.4685	1.6832	0.84985		14.000	83.922	7.6316	3.4324	2.2234	0.5378	
	22.000	44.899	3.6130	2.3591	1.5315	1.5845	0.81272		12.000	84.879	7.6572	3.4372	2.2277	0.5282	
	24.000	47.822	3.9995	2.4998	1.5999	1.4797	0.77209		10.000	85.792	7.6775	3.4411	2.2312	0.5204	
	26.000	51.130	4.4319	2.6449	1.6756	1.3695	0.72772		8.000	86.671	7.6934	3.4440	2.2338	0.5143	
	28.000	55.131	4.9401	2.8007	1.7638	1.2334	0.67784		6.000	87.524	7.7053	3.4462	2.2359	0.5096	
	30.000	61.449	5.6866	3.0051	1.8923	1.0385	0.61007		4.000	88.359	7.7135	3.4478	2.2372	0.5064	
	30.317	64.823	6.0466	3.0946	1.9539	0.9418	0.57989		2.000	89.183	7.7184	3.4487	2.2381	0.5045	
	30.000	67.966	6.3519	3.1664	2.0060	0.8568	0.55557								
	28.000	72.844	6.7595	3.2569	2.0754	0.7364	0.52487								
	26.000	75.440	6.9402	3.2952	2.1061	0.6793	0.51190								
	24.000	77.380	7.0575	3.3195	2.1260	0.6405	0.50368								
	22.000	78.978	7.1423	3.3368	2.1404	0.6115	0.49783								
	20.000	80.360	7.2068	3.3499	2.1514	0.5887	0.49343								
	18.000	81.594	7.2575	3.3600	2.1600	0.5703	0.49002								
	16.000	82.720	7.2978	3.3680	2.1668	0.5554	0.48732								
	14.000	83.766	7.3301	3.3744	2.1723	0.5432	0.48517								
	12.000	84.750	7.3561	3.3795	2.1767	0.5333	0.48345								
	10.000	85.688	7.3767	3.3835	2.1802	0.5253	0.48209								
	8.000	86.590	7.3927	3.3866	2.1829	0.5190	0.48104								
	6.000	87.464	7.4047	3.3890	2.1849	0.5142	0.48025								
	4.000	88.320	7.4131	3.3906	2.1864	0.5109	0.47971								
	2.000	89.163	7.4180	3.3916	2.1872	0.5090	0.47939								
	2.60	2.000	24.071	1.1454	1.1017	1.0396	0.99975	2.65	2.000	23.613	1.1479	1.1034	2.5607	0.99973	
	4.000	25.611	1.3070	1.2741	1.0801	2.4265	0.99805			25.000	25.144	1.3124	1.2136	2.4734	0.99796
	6.000	27.241	1.4831	1.3245	1.1218	2.3416	0.99371			26.000	26.766	1.4950	1.3302	2.3869	0.99341
	8.000	28.966	1.6831	1.4445	1.1651	2.2568	0.98579			28.000	28.482	1.6966	1.4525	2.3007	0.98514
	10.000	30.789	1.8998	1.5695	1.2105	2.1715	0.97665			30.000	30.295	1.9182	1.5798	2.2139	0.97247
	12.000	32.714	2.1369	2.1369	1.6986	2.0852	0.95890			32.000	32.210	2.1610	1.7113	2.1262	0.95502
	14.000	34.749	2.3985	1.8311	1.3082	1.9973	0.93541			34.000	34.232	2.4260	1.8462	2.0370	0.93270
	16.000	36.901	2.6767	1.9662	1.3613	1.9075	0.90930			36.000	36.368	2.7141	1.9835	1.9459	0.90566
	18.000	39.185	2.9817	2.1032	1.4177	1.8152	0.87584			38.000	38.632	3.0267	2.1226	1.8524	0.87423
	20.000	41.621	3.3126	2.2417	1.4778	1.7199	0.84443			40.000	41.043	3.3657	2.2630	1.84873	0.83884
	22.000	44.242	3.6723	2.3814	1.5421	1.6205	0.80645			42.000	43.627	3.7335	2.4042	1.5529	1.6559
	24.000	47.102	4.0658	2.5229	1.6116	1.5157	0.76520			44.000	46.433	4.1347	2.5465	1.6237	1.5507
	26.000	50.305	4.5028	2.6675	1.6880	1.4025	0.72060			46.000	49.549	4.5776	2.6911	1.7010	1.4380
	28.000	54.088	5.0067	2.8201	1.7754	1.2744	0.67151			48.000	53.164	5.0815	2.8416	1.7883	1.3126
	30.000	59.352	5.6706	3.0010	1.8896	1.1062	0.61145			50.000	57.877	5.7097	3.0110	1.8963	1.1576

### Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{\rho_2}{\rho_1}$	$M_2$	$\frac{T_2}{T_1}$	$\frac{P_{02}}{P_{01}}$		
2.65	4.000	88.396	8.0198	2.2891	0.5021	0.44194	2.75	24.000	45.225	4.2794	2.5951	1.6490	1.6181	0.74319		
	2.000	89.200	8.0247	2.2899	0.5003	0.44165		26.000	48.206	4.7375	2.7404	1.7288	1.5056	0.69739		
2.70	2.000	23.173	1.1503	1.1051	1.0409	2.6090	0.99972	30.000	51.579	5.2490	2.8886	1.8171	1.3832	0.64996		
	4.000	24.696	1.3179	1.2172	1.0827	2.5201	0.99786	32.000	55.674	5.8507	3.0466	1.9204	1.2416	0.59611		
	6.000	26.311	1.5042	1.3360	1.1260	2.4321	0.99311	32.173	62.549	6.7812	3.2616	2.0791	1.0209	0.52329		
	8.000	28.019	1.7102	1.4605	1.1709	2.3444	0.98446	32.000	65.002	7.0807	3.3243	2.1300	0.9476	0.50207		
	10.000	29.824	1.9369	1.5902	1.2180	2.2561	0.97125	28.000	72.678	7.8741	3.4773	2.2644	0.7401	0.45066		
	12.000	31.728	2.1855	1.7241	1.2676	2.1669	0.95309	26.000	75.285	8.0870	3.5154	2.3004	0.6789	0.43799		
	14.000	33.739	2.4569	1.8614	1.3199	2.0763	0.92991	24.000	78.766	8.3214	3.5393	2.3235	0.6378	0.43010		
	16.000	35.862	2.7523	2.0010	1.3754	1.9838	0.90191	22.000	80.110	8.3960	3.5561	2.3400	0.6071	0.42454		
	18.000	38.109	3.0727	2.1423	1.4343	1.8890	0.86948	20.000	81.303	8.4545	3.5786	2.3625	0.5634	0.41714		
	20.000	40.496	3.4200	2.2845	1.4970	1.7915	0.83311	18.000	82.386	8.5014	3.5864	2.3704	0.5474	0.41457		
	22.000	43.049	3.7964	2.4273	1.5641	1.6905	0.79337	16.000	83.387	8.5392	3.5927	2.3768	0.5343	0.41251		
	24.000	45.809	4.2059	2.5706	1.6362	1.5848	0.75072	14.000	84.324	8.5699	3.5978	2.3820	0.5234	0.41085		
	26.000	48.852	4.6560	2.7155	1.7146	1.4723	0.70538	12.000	85.212	8.5948	3.6019	2.3862	0.5145	0.40951		
	28.000	52.334	5.1626	2.8645	1.8022	1.3488	0.65692	10.000	86.062	8.6146	3.6051	2.3895	0.5072	0.40845		
	30.000	56.687	5.7730	3.0271	1.9071	1.2018	0.60288	8.000	86.882	8.6301	3.6077	2.3922	0.5015	0.40762		
	31.741	64.956	6.8143	3.2687	2.0847	0.9462	0.52090	6.000	87.680	8.6418	3.6096	2.3941	0.4972	0.40700		
	30.000	71.913	7.5186	3.4110	2.2042	0.7587	0.47286	4.000	88.462	8.6499	3.6109	2.3955	0.4942	0.40656		
	28.000	74.790	7.7529	3.4551	2.2439	0.6907	0.45808	2.000	89.234	8.6547	3.6117	2.3963	0.4924	0.40631		
	26.000	76.828	7.8997	3.4814	2.2682	0.6468	0.44930									
	24.000	78.466	7.9983	3.4997	2.2854	0.6145	0.44321									
	22.000	79.862	8.0748	3.5133	2.2994	0.5893	0.43870									
	20.000	81.095	8.1345	3.5238	2.3085	0.5691	0.43522									
	18.000	82.210	8.1821	3.5321	2.3165	0.5527	0.43247									
	16.000	83.238	8.2204	3.5388	2.3230	0.5391	0.43027									
	14.000	84.199	8.2515	3.5441	2.3282	0.5279	0.42850									
	12.000	85.109	8.2765	3.5484	2.3324	0.5188	0.42708									
	10.000	85.978	8.2965	3.5518	2.3358	0.5114	0.42595									
	8.000	86.816	8.3121	3.5545	2.3385	0.5056	0.42506									
	6.000	87.631	8.3228	3.5565	2.3404	0.5012	0.42441									
	4.000	88.430	8.3319	3.5579	2.3418	0.4981	0.42395									
	2.000	89.218	8.3367	3.5587	2.3426	0.4962	0.42368									
2.75	2.000	22.750	1.1528	1.1068	1.0415	2.6573	0.99971	28.000	54.786	5.9387	3.0683	1.9355	1.2783	0.58877		
	4.000	24.267	1.3226	1.2209	1.0841	2.5667	0.99776	32.000	60.433	6.7529	3.2555	2.0743	1.0909	0.52535		
	6.000	25.873	1.5135	1.3417	1.1280	2.4772	0.99279	32.587	65.050	7.3524	3.3788	2.1761	0.9490	0.48369		
	8.000	27.575	1.7239	1.4686	1.1738	2.3879	0.98377	32.000	69.211	7.8278	3.4689	2.2566	0.8307	0.45348		
	10.000	29.372	1.9558	1.6007	1.2219	2.2982	0.96999	30.000	73.328	8.2272	3.5399	2.3241	0.7243	0.42988		
	12.000	31.269	2.2104	1.7371	1.2724	2.2074	0.95109	28.000	75.728	8.4241	3.5735	2.3574	0.6684	0.41882		
	14.000	33.269	2.4885	1.8768	2.1153	0.92704	2.0188	2.01213	0.89806	26.000	77.543	8.5544	2.3794	0.6296		
	16.000	35.381	2.7912	2.0826	1.9219	1.4686	1.4429	2.1197	2.1622	0.86461	24.000	79.042	8.6495	2.3954	0.6002	
	18.000	37.612	3.1197	2.3063	1.4757	1.5070	1.8265	1.82724	2.3063	2.4506	0.78659	22.000	80.339	8.7224	2.4077	0.5769
	20.000	39.980	3.4757	2.4506	1.8610	1.7245	1.7245	1.7245	2.4506			81.496	8.7800	2.4174	0.5580	
	22.000	42.504													0.39971	

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$		
2.80	18.000	82.550	8.8262	3.6393	2.4252	0.5425	0.39731	2.90	6.000	24.666	1.3594	1.1344	2.6117	0.99178	
	16.000	83.525	8.8637	3.6453	2.4316	0.5297	0.39538		8.000	26.350	1.4933	1.1828	2.5175	0.98153	
	14.000	84.440	8.8942	3.6501	2.4367	0.5191	0.39382		10.000	28.129	1.6328	1.2336	2.4229	0.96597	
	12.000	85.308	8.9188	3.6540	2.4409	0.5103	0.39256		12.000	30.007	2.2873	1.7767	1.2874	2.3273	0.94475
	10.000	86.140	8.9385	3.6571	2.4442	0.5033	0.39156		14.000	31.985	2.5863	1.9238	1.3444	2.2304	0.91794
	8.000	86.943	8.9540	3.6595	2.4468	0.4977	0.39078		16.000	34.069	2.9123	2.0729	1.4050	2.1318	0.88591
	6.000	87.725	8.9656	3.6613	2.4487	0.4935	0.39019		18.000	36.264	3.2663	2.2229	1.4694	2.0313	0.84930
	4.000	88.492	8.9737	3.6626	2.4501	0.4905	0.38978		20.000	38.584	3.6496	2.3729	1.5380	1.9285	0.80886
	2.000	89.248	8.9784	3.6633	2.4509	0.4887	0.38954		22.000	41.044	4.0638	2.5222	1.6112	1.8229	0.776540
2.85	2.000	21.954	1.1579	1.1103	1.0429	2.7537	0.99968		28.000	49.655	5.5328	2.9652	1.8659	1.4788	0.62347
	4.000	23.457	1.3349	1.2283	1.0868	2.6598	0.99755		30.000	53.274	6.1364	3.1161	1.9692	1.3453	0.57262
	6.000	25.052	1.5325	1.3535	1.1323	2.5670	0.99213		32.000	57.931	6.8791	3.2824	2.0957	1.827	0.51624
	8.000	26.742	1.7520	1.4850	1.1798	2.4744	0.98230		33.363	65.145	3.4841	2.2708	0.9516	0.44840	
	10.000	28.526	1.9946	1.6220	1.2297	2.3815	0.96735		32.000	71.287	8.6350	3.6085	2.3930	0.97771	0.40736
	12.000	30.410	2.2613	1.7634	1.2824	2.2876	0.94692		30.000	74.392	8.9347	3.6565	2.4435	0.6985	0.39175
	14.000	32.394	2.5532	1.9080	1.3382	2.1923	0.92105		28.000	76.490	9.1095	3.6836	2.4730	0.6500	0.38301
	16.000	34.486	2.8712	2.0547	1.3974	2.0953	0.89006		26.000	78.142	9.2307	3.7020	2.4934	0.6149	0.37709
	18.000	36.692	3.2165	2.2025	1.4604	1.9964	0.85451		24.000	79.533	9.3212	3.7156	2.5087	0.5878	0.37275
	20.000	39.025	3.5904	2.3505	1.5275	1.8950	0.81511		22.000	80.750	9.3915	3.7260	2.5205	0.5860	0.36942
	22.000	41.505	3.9948	2.4982	1.5991	1.7900	0.77258		20.000	81.843	9.4475	3.7343	2.5300	0.5482	0.36680
	24.000	44.160	4.4325	2.6451	1.6757	1.6825	0.72766		18.000	82.845	9.4928	3.7409	2.5376	0.5335	0.36469
	26.000	47.042	4.9089	2.7916	1.7585	1.5692	0.68081		16.000	83.775	9.5296	3.7462	2.5438	0.5212	0.36299
	28.000	50.247	5.4345	2.9391	1.8490	1.4481	0.63219		14.000	84.651	9.5597	3.7506	2.5489	0.5111	0.36161
	30.000	53.992	6.0344	3.0917	1.9518	1.3127	0.58089		12.000	85.484	9.5842	3.7541	2.5530	0.5027	0.36049
	32.000	59.037	6.8013	2.62659	2.0825	1.1407	0.52283		10.000	86.283	9.6038	3.7570	2.5563	0.4959	0.35960
	32.984	65.097	7.6294	3.4320	2.2230	0.9503	0.46580		8.000	87.055	9.6191	3.7592	2.5588	0.4906	0.35890
	32.000	70.389	8.2421	3.5421	2.3266	0.8001	0.42903		6.000	87.808	9.6306	3.7608	2.5608	0.4865	0.35838
	30.000	73.893	8.5802	3.5995	2.3837	0.7107	0.41030		4.000	88.546	9.6387	3.7620	2.5621	0.4836	0.35802
	28.000	76.127	8.7648	3.6295	2.4149	0.6588	0.40050		2.000	89.275	9.6434	3.7626	2.5629	0.4819	0.35780
	26.000	77.855	8.8902	3.6495	2.43860	0.6220	0.39402								
	24.000	79.297	8.9827	3.6640	2.4516	0.5938	0.38933								
	22.000	80.552	9.0543	3.6751	2.4637	0.5713	0.38574								
	20.000	81.676	9.1110	3.6838	2.4733	0.5530	0.38294								
	18.000	82.702	9.1567	3.6908	2.4810	0.5379	0.38069								
	16.000	83.655	9.1938	3.6964	2.4872	0.5253	0.37888								
	14.000	84.549	9.2241	3.7010	2.4923	0.5150	0.37741								
	12.000	85.399	9.2486	3.7047	2.4964	0.5064	0.37523								
	10.000	86.213	9.2683	3.7077	2.4998	0.4995	0.37528								
	8.000	87.001	9.2836	3.7100	2.5023	0.4940	0.37754								
	6.000	87.768	9.2952	3.7117	2.5043	0.4899	0.37599								
	4.000	88.520	9.3033	3.7129	2.5057	0.4870	0.37580								
	2.000	89.262	9.3080	3.7136	2.5065	0.4853	0.37538								
	2.90	2.000	21.578	1.1604	1.1120	1.0435	2.8019	0.99966							
	4.000	23.076	1.3406	1.2320	1.0882	2.7062	0.99744	0.99744							

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$	Oblique Shock Tables			$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{p_{02}}{p_{01}}$	
							$M_1$	$\theta$	$\beta$					
2.95	32.000	56.997	6.9741	2.1119	1.2199	0.50950	3.00	14.000	84.837	10.2483	3.8459	2.6647	0.5038	
	33.726	65.193	8.1990	2.3194	0.9528	0.43150	12.000	85.638	10.2726	3.8491	2.6688	0.4958	0.33180	
	32.000	72.020	9.0188	3.6696	2.4577	0.7585	10.000	86.408	10.2921	3.8517	2.6721	0.4892	0.33081	
	30.000	74.838	9.2917	2.5037	0.6677	0.37416	8.000	87.154	10.3074	3.8537	2.6747	0.4841	0.33001	
	28.000	76.821	9.4585	3.7359	2.5318	0.6420	0.36628	6.000	87.881	10.3190	3.8553	2.6766	0.4801	0.32939
	26.000	78.407	9.5762	3.7530	2.5516	0.6884	0.36086	4.000	88.594	10.3270	3.8563	2.6779	0.4774	0.32892
	24.000	79.752	9.6649	3.7657	2.5666	0.5821	0.35684	2.000	89.299	10.3318	3.8569	2.6787	0.4757	0.32841
	22.000	80.935	9.7342	3.7755	2.5782	0.5610	0.35374							
	20.000	82.000	9.7896	3.7834	2.5875	0.5437	0.35128							
	18.000	82.978	9.8345	3.7896	2.5951	0.5293	0.34931							
	16.000	83.889	9.8712	3.7947	2.6013	0.5173	0.34771							
	14.000	84.747	9.9012	3.7989	2.6063	0.5074	0.34641							
	12.000	85.563	9.9255	3.8023	2.6104	0.4992	0.34536							
	10.000	86.348	9.9450	3.8050	2.6137	0.4925	0.34452							
	8.000	87.106	9.9604	3.8071	2.6163	0.4872	0.34386							
	6.000	87.845	9.9719	3.8087	2.6182	0.4832	0.34336							
	4.000	88.571	9.9799	3.8098	2.6196	0.4804	0.34302							
	2.000	89.288	9.9847	3.8104	2.6204	0.4788	0.34282							
3.00	2.000	20.867	1.0449	1.155	1.0909	2.7988	0.99721	26.000	45.110	52.2806	2.8973	1.8226	1.6874	
	4.000	22.355	1.2395	1.3714	1.1387	2.7008	0.99105	28.000	48.102	58.462	3.0455	1.9196	1.5654	
	6.000	23.936	1.5616	1.7953	1.5101	1.888	2.6031	30.000	51.455	6.4722	3.1988	2.0265	1.4345	
	8.000	25.611	2.0545	1.6546	1.2417	2.5050	0.96308	32.000	55.456	7.967	3.3478	2.1497	1.2858	
	10.000	27.383	2.2641	1.8036	1.2977	2.4060	0.94022	34.000	61.505	8.2161	3.5380	2.3222	1.7328	
	12.000	29.251	2.3404	1.9556	1.3571	2.3056	0.91148	34.407	65.288	8.7895	3.6335	2.4190	1.9552	
	14.000	31.218	2.6540	2.1095	1.4204	2.2037	0.87734	34.000	68.742	9.2596	3.7064	2.4983	0.8514	
	16.000	33.288	2.9964	3.3685	2.2641	1.4878	2.1000	0.83855	32.000	73.184	9.7779	3.7817	2.5856	0.7291
	18.000	35.467	5.7388	3.0184	2.4181	1.5596	1.9941	0.79602	30.000	75.604	10.0154	3.8146	2.6255	0.6689
	20.000	37.764	3.7713	2.42064	2.5708	1.6362	1.8858	0.75068	28.000	77.406	10.1703	3.8355	2.6516	0.6276
	22.000	40.192	5.6182	4.6761	2.7216	1.7181	1.7744	0.70340	26.000	78.880	10.2825	3.8505	2.6705	0.5965
	24.000	42.775	45.552	5.1844	2.8706	1.8060	1.6589	0.65491	24.000	80.145	10.3683	3.8617	2.6849	0.5719
	26.000	45.241	48.586	5.7388	3.0184	1.9012	1.5374	0.60560	22.000	81.267	10.4361	3.8705	2.6963	0.5518
	28.000	46.749	63.5559	3.1673	2.0067	1.4059	0.55526	20.000	82.284	10.4906	3.8775	2.7055	0.5353	0.32208
	30.000	52.014	6.9697	7.0810	3.3244	2.1300	1.2541	0.50205	18.000	83.221	10.5350	3.8833	2.7129	0.5215
	32.000	56.182	8.2682	3.5470	2.3310	1.0029	0.42755	1.0000	84.095	10.5714	3.8879	2.7190	0.5100	0.32040
	34.000	63.673	8.4917	3.5848	2.3688	0.9540	0.41510	14.000	84.921	10.6012	3.8891	2.7240	0.5005	0.31892
	34.073	65.241	8.6971	3.6186	2.4035	0.9083	0.40406	12.000	85.709	10.6255	3.8948	2.7281	0.4926	0.31777
	34.000	66.749	8.6971	3.7271	2.5217	0.7428	0.36908	10.000	86.466	10.6450	3.8973	2.7314	0.4861	0.31608
	32.000	72.642	9.3988	3.7638	2.5643	0.6779	0.35743	8.000	87.199	10.6603	3.8992	2.7340	0.4810	0.31549
	30.000	75.239	9.6517	3.7865	2.5913	0.6345	0.35029	6.000	87.914	10.6719	3.9007	2.7359	0.4772	0.31505
	28.000	77.126	9.8121	3.8024	2.6106	0.6022	0.34530	4.000	88.617	10.6799	3.9017	2.7373	0.4744	0.31474
	26.000	78.652	9.9268	3.8024	2.6253	0.5768	0.34157	2.000	89.310	10.6847	3.9023	2.7381	0.4728	0.31456
	24.000	79.956	10.0139	3.8144	2.6368	0.5563	0.33868							
	22.000	81.106	10.0824	3.8237	2.6460	0.5394	0.33638							
	20.000	82.147	10.1373	3.8311	2.6536	0.5253	0.33453	3.10	2.000	20.205	1.1707	1.1190	1.0462	0.99942
	18.000	83.103	10.1819	3.8371	2.6840	0.5136	0.33302	4.000	21.684	1.3640	1.0397	1.2471	2.8911	0.99696

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$
3.10	6.000	23.258	1.5815	1.1431	2.7894	0.99027	3.15	28.000	47.216	6.0688	3.1000	1.9577	1.6194
	8.000	24.927	1.8249	1.5271	2.6881	0.97822		30.000	50.449	6.7158	3.2475	2.0680	1.4886
	10.000	26.692	2.0956	1.6767	2.5864	0.96004		32.000	54.201	7.4487	3.3975	2.1924	1.3441
	12.000	28.554	2.3949	1.8308	1.3081	2.4837		34.000	59.196	8.3736	3.5650	2.3489	1.1632
	14.000	30.513	2.7236	1.9879	1.3701	2.3798		35.033	65.382	9.4008	3.7274	2.5221	0.9575
	16.000	32.574	3.0831	2.1467	1.4362	2.2743		34.000	70.719	10.1474	3.8325	2.6478	0.9369
	18.000	34.739	3.4740	2.3057	1.5067	2.1672		32.000	74.089	10.5396	3.8839	2.7137	0.7064
	20.000	37.017	3.8973	2.4637	1.5819	2.0581		30.000	76.244	10.7550	3.9111	2.7499	0.6531
	22.000	39.421	4.3543	2.6198	1.6621	1.9468		28.000	77.906	10.9014	3.9292	2.7745	0.6152
	24.000	41.968	4.8470	2.7733	1.7477	1.8329		26.000	79.289	11.0097	3.9424	2.7927	0.5860
	26.000	44.692	5.3788	2.9241	1.8395	1.7154		24.000	80.490	11.0936	3.9524	2.8068	0.5627
	28.000	47.646	5.9563	3.0727	1.9385	1.5928		22.000	81.560	11.1602	3.9604	2.8180	0.5436
	30.000	50.935	6.5922	3.2205	2.0470	1.4620		20.000	82.535	11.2142	3.9668	2.8270	0.5278
	32.000	54.800	7.3197	3.3723	2.1705	1.3157		18.000	83.436	12.583	3.9720	2.8344	0.5145
	34.000	60.205	8.2768	3.5485	2.3325	1.1241		16.000	84.279	11.2945	3.9762	2.8405	0.5035
	34.726	65.335	9.0925	3.6810	2.4701	0.9564		14.000	85.076	11.3243	3.9797	2.8455	0.4942
	34.000	69.872	9.7174	3.7732	2.5754	0.8203		12.000	85.838	11.3486	3.9825	2.8496	0.4865
	32.000	73.661	10.1577	3.8339	2.6495	0.7171		10.000	86.571	11.3682	3.9848	2.8529	0.4803
	30.000	75.938	10.3881	3.8631	2.6874	0.6607		8.000	87.281	11.3835	3.9866	2.8555	0.4754
	28.000	77.666	10.5334	3.8834	2.7126	0.6212		6.000	87.976	11.3951	3.9879	2.8574	0.4716
	26.000	79.091	10.6435	3.8971	2.7311	0.5911		4.000	88.657	11.4032	3.9889	2.8588	0.4690
	24.000	80.324	10.7282	3.9077	2.7454	0.5671		2.000	89.330	11.4080	3.9894	2.8596	0.4674
	22.000	81.419	10.7954	3.9161	2.7557	0.5476							
	20.000	82.413	10.8496	3.9228	2.7658	0.5314							
	18.000	83.331	10.8938	3.9282	2.7732	0.5179							
	16.000	84.189	10.9301	3.9327	2.7793	0.5067							
	14.000	85.001	10.9599	3.9363	2.7843	0.4973							
	12.000	85.775	10.9842	3.9393	2.7884	0.4895							
	10.000	86.520	11.0037	3.9416	2.7917	0.4832							
	8.000	87.242	11.0190	3.9435	2.7942	0.4781							
	6.000	87.945	11.0306	3.9449	2.7962	0.4743							
	4.000	88.637	11.0387	3.9458	2.7975	0.4716							
	2.000	89.321	11.0434	3.9464	2.7983	0.4701							
	3.15	2.000	19.891	1.1734	1.1208	1.0469							
	4.000	21.366	1.3699	1.2510	1.0951	2.9371							
	6.000	22.937	1.5915	1.3896	1.1453	2.8336							
	8.000	24.603	1.8399	1.5357	1.1981	2.7304							
	10.000	26.366	2.1166	1.6878	1.2540	2.6267							
	12.000	28.225	2.4226	1.8445	1.3134	2.5222							
	14.000	30.181	2.7592	2.0042	1.3767	2.4165							
	16.000	32.238	3.1273	2.1654	1.4443	2.3092							
	18.000	34.398	3.5279	2.3266	1.5163	2.2003							
	20.000	36.668	3.9617	2.4866	1.5933	2.0895							
	22.000	39.061	4.4302	2.6444	1.6753	1.9767							
	24.000	41.594	4.9349	2.7992	1.7629	1.8613							
	26.000	44.296	5.4793	2.9510	1.8567	1.7427							

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{\rho_2}{\rho_1}$	$M_2$	$\frac{T_2}{T_1}$	$P_{02}$	$P_{01}$
3.20	22.000	81.694	11.5307	4.0035	2.8802	0.5398	0.28438	3.30	2.000	19.009	1.1812	1.1262	1.0489	3.1858	0.99953
	20.000	82.649	11.5844	4.0096	2.8882	0.5243	0.28280		4.000	20.475	1.3880	1.2626	1.0993	3.0748	0.99642
	18.000	83.533	11.6285	4.0146	2.8996	0.5113	0.28115		6.000	22.039	1.6222	1.4082	1.1520	2.9653	0.98853
	16.000	84.363	11.6647	4.0187	2.9026	0.5004	0.27996		8.000	23.699	1.8859	1.5617	1.2076	2.8563	0.97453
	14.000	85.147	11.6945	4.0220	2.9076	0.4913	0.27899		10.000	25.457	2.1807	1.7216	1.2666	2.7468	0.95347
	12.000	85.897	11.7188	4.0247	2.9117	0.4837	0.27820		12.000	27.310	2.5078	1.8861	1.3296	2.6364	0.92526
	10.000	86.619	11.7385	4.0269	2.9150	0.4776	0.27757		14.000	29.261	2.8688	2.0536	1.3970	2.5248	0.89031
	8.000	87.320	11.7539	4.0286	2.9196	0.4727	0.27707		16.000	31.308	3.2640	2.2219	1.4690	2.4118	0.84954
	6.000	88.003	11.7655	4.0299	2.9209	0.4690	0.27689		18.000	33.456	3.6947	2.3898	1.5460	2.2974	0.80409
	4.000	88.675	11.7736	4.0308	2.9217	0.4664	0.27643		20.000	35.710	4.1617	2.5557	1.6284	2.1813	0.75527
	2.000	89.340	11.7784	4.0313		0.4649	0.27628		22.000	38.077	4.6655	2.7184	1.7163	2.0636	0.70444
									24.000	40.573	5.2081	2.8773	1.8101	1.9439	0.65272
3.25	2.000	19.293	1.1786	1.1244	1.0482	3.1380	0.99955	28.000	46.062	6.4212	3.1822	2.0178	1.6955	0.55020	
	4.000	20.762	1.3818	1.2586	1.0979	3.0290	0.99656	30.000	49.163	52.667	3.3294	2.1342	1.5638	0.50034	
	6.000	22.328	1.6119	1.4019	1.1498	2.9215	0.98902	32.000	56.963	7.8658	3.4758	2.2630	1.4218	0.45116	
	8.000	23.990	1.8704	1.5530	1.2044	2.8145	0.97545	34.000	56.963	8.7622	3.6291	2.4144	1.2575	0.40064	
	10.000	25.749	2.1590	1.7103	1.2624	2.7070	0.95518	35.882	65.518	10.3564	3.8602	2.6829	0.9606	0.32741	
	12.000	27.604	2.4791	1.8722	1.3242	2.5986	0.92789	34.000	72.501	11.3896	3.9873	2.8565	0.7502	0.28914	
	14.000	29.556	2.8318	2.0370	1.3901	2.4889	0.89402	32.000	75.148	11.7036	4.0230	2.9092	0.6797	0.27869	
	16.000	31.606	3.2179	2.2030	1.4607	2.3779	0.85437	30.000	77.029	11.8983	4.0445	2.9418	0.6336	0.27247	
	18.000	33.757	3.63984	2.3687	1.5360	2.2653	0.81004	28.000	78.535	12.0364	4.0595	2.9650	0.5993	0.26817	
	20.000	36.016	4.0940	2.53226	1.6165	2.1511	0.76227	26.000	79.812	12.1408	4.0706	2.9825	0.5725	0.26497	
	22.000	38.390	4.5858	2.6937	1.7024	2.0350	0.71232	24.000	80.932	12.2227	4.0793	2.9963	0.5507	0.26251	
	24.000	40.898	5.1156	2.8513	1.7941	1.9168	0.66129	22.000	81.938	12.2884	4.0862	3.0073	0.5328	0.26055	
	26.000	43.563	5.6858	3.0049	1.8922	1.7958	0.61015	20.000	82.859	12.3420	4.0918	3.0163	0.5178	0.25896	
	28.000	46.426	6.3015	3.1548	2.53226	2.1511	0.76227	18.000	83.714	12.3860	4.0964	3.0236	0.5052	0.25767	
	30.000	49.566	6.9727	3.3020	2.1116	1.5394	0.50960	16.000	84.517	12.4223	4.1001	3.0297	0.4946	0.25662	
	32.000	53.141	7.7223	3.4494	2.2387	1.3970	0.45998	14.000	85.278	12.4523	4.1032	3.0348	0.4858	0.25575	
	34.000	57.616	8.6213	3.6062	2.3907	1.2287	0.40809	12.000	86.007	12.4767	4.1057	3.0389	0.4785	0.25504	
	35.610	65.473	10.0327	3.8170	2.62917	0.9596	0.34078	10.000	86.708	12.4964	4.1077	3.0422	0.4725	0.25448	
	34.000	71.993	10.9786	3.9386	2.7875	0.7636	0.30361	8.000	87.390	12.5120	4.1093	3.0448	0.4677	0.25403	
	32.000	74.827	11.3120	3.9783	2.8434	0.6878	0.29180	6.000	88.056	12.5237	4.1105	3.0467	0.4641	0.25369	
	30.000	76.787	11.5124	4.0014	2.8771	0.6396	0.28489	4.000	88.710	12.5319	4.1114	3.0481	0.4616	0.25346	
	28.000	78.339	11.6529	4.0173	2.9007	0.6043	0.28035	2.000	89.357	12.5367	4.1119	3.0489	0.4601	0.25332	
	26.000	79.649	11.7584	4.0291	2.9184	0.5767	0.27692								
	24.000	80.793	11.8408	4.0382	2.9322	0.5545	0.27429								
	22.000	81.819	11.9067	4.0454	2.9433	0.5362	0.27220								
	20.000	82.757	11.9604	4.0513	2.9523	0.5210	0.27052								
	18.000	83.626	12.0044	4.0560	2.9596	0.5082	0.26916								
	16.000	84.442	12.0407	4.0599	2.9657	0.4974	0.26804								
	14.000	85.214	12.0705	4.0631	2.9707	0.4885	0.26712								
	12.000	85.953	12.0949	4.0658	2.9748	0.4810	0.266637								
	10.000	86.665	12.1145	4.0679	2.9781	0.4750	0.26577								
	8.000	87.356	12.1300	4.0695	2.9807	0.4702	0.26530								
	6.000	88.030	12.1417	4.0707	2.9827	0.4665	0.26495								
	4.000	88.693	12.1498	4.0716	2.9840	0.4639	0.26470								
	2.000	89.348	12.1547	4.0721	2.9848	0.4624	0.26455								

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$
3.35	24.00	40.264	5.3024	2.9033	1.8263	1.9704	0.64409	3.40	34.000	73.352	12.2131	4.0783	0.7279
	26.00	42.898	5.8998	3.0588	1.9288	1.8468	0.59200		32.000	75.717	12.4992	4.1080	0.6426
	28.00	45.716	6.5433	3.2097	2.0386	1.7198	0.54090		30.000	77.467	12.6849	4.1268	0.6225
	30.00	48.782	7.2416	3.3568	2.1573	1.5874	0.49109		28.000	78.891	12.8193	4.1402	0.5963
	32.00	52.225	8.0134	3.5024	2.2880	1.4458	0.44232		26.000	80.110	12.9221	4.1503	0.5135
	34.00	56.375	8.9114	3.6528	2.4396	1.2844	0.39294		24.000	81.185	13.0033	4.1582	0.5437
	36.00	63.380	10.2976	3.8524	2.6730	1.0339	0.32979		22.000	82.156	13.0688	4.1645	0.5264
	36.143	65.562	10.6853	3.9023	2.7382	0.9616	0.31454		20.000	83.047	13.1224	4.1697	0.5119
	36.00	67.623	11.0286	3.9446	2.7958	0.8957	0.30180		18.000	83.876	13.1665	4.1739	0.4997
	34.00	72.950	11.8006	4.0338	2.9255	0.7384	0.27557		16.000	84.656	13.2030	4.1774	0.4894
	32.00	75.444	12.0992	4.0662	2.9755	0.6723	0.26624		14.000	85.396	13.2331	4.1802	0.4808
	30.00	77.255	12.2891	4.0863	3.0074	0.6279	0.28053		12.000	86.105	13.2578	4.1826	0.4736
	28.00	78.719	12.4252	4.1004	3.0302	0.5946	0.25653		10.000	86.789	13.2777	4.1844	0.4678
	26.00	79.965	12.5287	4.1110	3.0476	0.5684	0.25355		8.000	87.453	13.2934	4.1859	0.4557
	24.00	81.062	12.6102	4.1193	3.0612	0.5471	0.25124		6.000	88.103	13.3052	4.1870	0.4596
	22.00	82.050	12.6758	4.1259	3.0722	0.5295	0.24939		4.000	88.741	13.3135	4.1878	0.4572
	20.00	82.956	12.7293	4.1313	3.0812	0.5148	0.24790		2.000	89.372	13.3184	4.1883	0.4557
	18.00	83.798	12.7734	4.1357	3.0886	0.5024	0.24668					1.1892	0.4557
	16.00	84.588	12.8098	4.1393	3.0947	0.4920	0.24568					1.1316	0.4557
	14.00	85.339	12.8398	4.1422	3.0997	0.4832	0.24486					1.0509	0.4557
	12.00	86.057	12.8644	4.1446	3.1038	0.4760	0.24420					1.0036	0.4557
	10.00	86.750	12.8842	4.1466	3.1072	0.4701	0.24366					1.0588	0.4557
	8.00	87.422	12.8998	4.1481	3.1098	0.4654	0.24324					1.0274	0.4557
	6.00	88.080	12.9116	4.1493	3.1118	0.4618	0.24292					1.0755	0.4557
	4.00	88.726	12.9198	4.1501	3.1131	0.4593	0.24270					1.0962	0.4557
	2.00	89.365	12.9246	4.1506	3.1140	0.4578	0.24256					1.1036	0.4557
	3.40	2.00	18.467	1.1298	1.1298	1.1298	0.99949	2.000	18.209	19.668	21.226	1.6536	1.0509
	6.00	19.928	1.4001	1.2704	1.1022	1.1662	0.99613	4.000	18.000	22.884	1.9331	1.5881	1.0962
	8.00	21.488	1.6430	1.4207	1.1565	1.2077	0.98766	6.000	18.000	24.639	2.2468	1.7559	1.1218
	10.00	23.147	2.2245	1.7444	1.2140	2.9395	0.97253	20.000	18.000	26.491	2.5962	1.9284	1.1606
	12.00	26.755	2.5664	1.9143	1.3407	2.7115	0.91981	22.000	18.000	28.438	2.9823	2.1035	1.1883
	14.00	28.702	2.9440	2.0868	1.4108	2.5958	0.88269	24.000	18.000	30.481	3.4063	2.2791	1.2218
	16.00	30.746	3.3583	2.2600	1.4860	2.4788	0.83962	26.000	18.000	32.621	3.8688	2.4535	1.2698
	18.00	32.789	3.8100	2.4322	1.5665	2.3604	0.79194	28.000	18.000	34.863	4.3706	2.6251	1.3115
	20.00	35.133	4.2998	2.6019	1.6526	2.2407	0.74110	30.000	18.000	48.080	7.5215	3.4115	1.4168
	22.00	37.489	4.8289	2.7679	1.7446	2.1195	0.68851	32.000	18.000	51.420	8.3194	3.5558	1.4788
	24.00	39.967	5.3980	2.9293	1.8428	1.9966	0.63546	34.000	18.000	55.344	9.2294	3.7018	1.5339
	26.00	42.588	6.0096	3.0857	1.9476	1.8716	0.58292	36.000	18.000	56.903	10.4358	3.8705	1.6104
	28.00	45.386	6.6675	3.2370	2.0598	1.7435	0.53162	38.000	18.000	65.647	11.3584	3.9837	1.6622
	30.00	48.422	7.3802	3.3842	2.1808	1.6105	0.48186	40.000	18.000	69.850	12.0718	4.0633	1.7267
	32.00	51.810	8.1645	3.5290	2.3135	1.4690	0.43348	42.000	18.000	73.716	12.6278	4.1211	1.7802
	34.00	55.838	9.0673	3.6771	2.4659	1.3098	0.38509	44.000	18.000	75.970	12.9035	4.1485	1.8399
	36.00	61.914	10.3308	3.8568	2.6786	1.0874	0.32845	46.000	18.000	81.302	13.4020	4.1961	1.9044
	36.393	65.605	11.0193	3.9435	2.7943	0.9625	0.30214	48.000	18.000	82.256	13.4675	4.2021	1.9477
	36.00	68.960	11.5817	4.0093	2.8887	0.8566	0.28269	50.000	18.000	83.951	13.5654	4.2111	2.02213

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{\rho_2}{\rho_1}$	$M_2$	$\frac{T_2}{T_1}$	$\frac{P_{02}}{P_{01}}$
3.45	16.000	84.720	13.6020	4.2145	3.2275	0.4869	0.22521	3.55	2.000	17.715	1.1947	1.1353	3.4246	0.99943
	14.000	85.451	13.6322	4.2172	3.2325	0.4794	0.22448		4.000	19.170	1.4187	1.2822	3.3029	0.99566
	12.000	86.151	13.6570	4.2195	3.2367	0.4774	0.22388		6.000	20.726	1.6748	1.4396	3.1829	0.98619
	10.000	86.826	13.6770	4.2213	3.2400	0.4656	0.22340		8.000	22.383	1.9653	1.6059	3.1238	0.96935
	8.000	87.482	13.6928	4.2227	3.2427	0.4610	0.22302		10.000	24.138	2.2920	1.7791	3.0633	0.94435
	6.000	88.125	13.7047	4.2238	3.2447	0.4575	0.22273		12.000	25.989	2.6566	1.9569	3.2883	0.91123
	4.000	88.756	13.7130	4.2245	3.2461	0.4551	0.22253		14.000	27.936	3.0603	2.1370	3.8224	0.87077
	2.000	89.379	13.7180	4.2250	3.2469	0.4536	0.22241		16.000	29.977	3.5040	2.3174	4.5121	0.82424
3.50	2.000	17.958	1.9415	1.1920	1.1335	1.0516	3.2769	0.9945	24.000	39.149	5.6937	3.0069	1.8935	0.82424
	4.000	20.972	1.6642	1.2783	1.1050	1.1611	3.2574	0.99582	26.000	41.738	6.3495	3.1659	2.0056	0.82424
	6.000	22.629	1.9491	1.5970	1.2205	1.2205	3.0222	0.97044	28.000	44.488	7.0535	3.3187	2.1254	0.77322
	10.000	24.384	2.2693	1.7675	1.2839	1.2839	2.9044	0.94626	30.000	47.447	7.8120	3.4660	2.2539	0.71939
	12.000	26.236	2.6262	1.9426	1.3519	1.3519	2.7856	0.91415	32.000	50.705	8.6392	3.6092	2.3937	0.66437
	14.000	28.182	3.0211	2.1202	1.4249	1.4249	2.6657	0.87481	34.000	54.463	9.5691	3.7520	2.5504	0.60946
	16.000	30.225	3.4549	2.2982	1.5033	1.5033	2.5445	0.82942	36.000	59.399	10.7262	3.9075	2.7450	0.55575
	18.000	32.363	3.9283	2.4747	1.5874	1.5874	2.4222	0.77952	37.091	65.729	12.0520	4.0612	2.9676	0.50395
	20.000	34.602	4.4421	2.6482	1.6774	1.6774	2.2986	0.72668	36.000	71.121	12.9969	4.1576	3.1261	0.45445
	22.000	36.947	4.9969	2.8173	1.7737	1.7737	2.1739	0.67245	34.000	74.353	13.4667	4.2021	3.2021	0.40714
	24.000	39.410	5.5936	2.9811	1.8764	1.8764	2.0478	0.61813	32.000	76.427	13.7265	4.2257	3.2504	0.36118
	26.000	42.009	6.2345	3.1392	1.9860	1.9860	1.9199	0.56478	30.000	78.025	13.9033	4.2415	3.1885	0.31299
	28.000	44.774	6.9227	3.2916	2.1032	2.1032	1.7894	0.51313	28.000	79.351	14.0342	4.2530	3.2779	0.26768
	30.000	47.755	7.6654	3.4388	2.2291	1.6549	0.46353	0.25353	26.000	80.497	14.1355	4.2618	3.2998	0.21803
	32.000	51.053	8.4777	3.5625	2.3664	1.5131	0.41586	0.21271	24.000	81.517	14.2163	4.2687	3.3168	0.21271
	34.000	54.888	9.3968	3.7268	2.5214	1.3570	0.36917	0.21271	22.000	82.442	14.2819	4.2743	3.3303	0.21090
	36.000	60.090	10.5715	3.8879	2.7191	1.1594	0.31891	0.21271	20.000	83.294	14.3333	4.2789	3.3413	0.22221
	36.867	65.689	11.7027	4.0229	2.9090	0.9643	0.27872	0.21271	18.000	84.090	14.3804	4.2827	3.3503	0.20944
	36.000	70.545	12.5396	4.1121	3.0494	0.8105	0.25324	0.21271	16.000	84.839	14.4173	4.2858	3.3578	0.20826
	34.000	74.048	13.0455	4.1623	3.1342	0.7098	0.23934	0.21271	14.000	85.552	14.4478	4.2883	3.3640	0.20729
	32.000	76.207	13.3126	4.1877	3.1790	0.6529	0.23241	0.21271	12.000	86.235	14.4729	4.2904	3.3733	0.20529
	30.000	77.851	13.4920	4.2044	3.2094	0.6128	0.22791	0.21271	10.000	86.895	14.4931	4.2921	3.3767	0.20485
	28.000	79.207	13.6238	4.2165	3.2311	0.5820	0.22468	0.21271	8.000	87.537	14.5091	4.2934	3.3794	0.20451
	26.000	80.375	13.7255	4.2256	3.2481	0.5574	0.22223	0.21271	6.000	88.165	14.5212	4.2944	3.3814	0.20425
	24.000	81.413	13.8064	4.2329	3.2617	0.5373	0.22031	0.21271	4.000	88.782	14.5296	4.2951	3.3828	0.20407
	22.000	82.352	13.8719	4.2387	3.2727	0.5205	0.21877	0.21271	2.000	89.392	14.5346	4.2956	3.3836	0.20397
	20.000	83.216	13.9256	4.2435	3.2817	0.5065	0.21751	0.21271						
	18.000	84.022	13.9700	4.2474	3.2891	0.4946	0.21649	0.21271						
	16.000	84.781	14.0067	4.2506	3.2952	0.4846	0.21564	0.21271	3.60	2.000	17.479	1.1973	1.1371	0.09940
	14.000	85.503	14.0371	4.2532	3.3003	0.4762	0.21494	0.21271	4.000	18.932	1.4250	1.2862	1.1079	0.3482
	12.000	86.194	14.0620	4.2554	3.3045	0.4692	0.21438	0.21271	6.000	20.488	1.6857	1.4461	1.1657	0.99549
	10.000	86.862	14.0822	4.2572	3.3079	0.4635	0.21392	0.21271	8.000	22.144	1.9816	1.6149	1.2271	0.98567
	8.000	87.510	14.0980	4.2585	3.3105	0.4590	0.21356	0.21271	10.000	23.899	2.3149	1.7907	1.2927	0.9821
	6.000	88.145	14.1100	4.2596	3.3125	0.4555	0.21329	0.21271	12.000	25.751	2.6873	1.9711	1.3633	0.8590
	4.000	88.769	14.1184	4.2603	3.3139	0.4531	0.21310	0.21271	14.000	27.698	3.0999	2.1538	2.7347	0.86667
	2.000	89.386	14.1234	4.2607	3.3148	0.4516	0.21298	0.21271	16.000	29.740	3.5540	2.3366	1.5210	0.81895
									18.000	31.876	4.0498	2.5174	2.6088	0.76685

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$M_2$	$\frac{T_2}{T_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{P_{02}}{P_{01}}$
3.60	20.000	34.110	4.5883	2.6945	1.7029	2.3552	0.71207	3.65	37.513	65.808	12.7662	4.1349	3.0874	0.9668
	22.000	36.448	5.1699	2.8666	1.8035	2.2267	0.65625		36.000	72.054	13.9006	4.2413	3.2775	0.7684
	24.000	38.898	5.7953	3.0327	1.9109	2.0973	0.60079		34.000	74.894	14.3206	4.2776	3.3478	0.6877
	26.000	41.478	6.4663	3.1924	2.0255	1.9664	0.54674		32.000	76.827	14.5690	4.2984	3.3894	0.6371
	28.000	44.215	7.1862	3.3457	2.1479	1.8335	0.49483		30.000	78.345	14.7420	4.3126	3.4183	0.6000
	30.000	47.153	7.9610	3.4930	2.2791	1.6971	0.44543		28.000	79.617	14.8713	4.3231	3.4400	0.5712
	32.000	50.376	8.8038	3.6357	2.4215	1.5547	0.39847		26.000	80.723	14.9723	4.3311	3.4569	0.5480
	34.000	54.066	9.7460	3.7772	2.5802	1.4002	0.35321		24.000	81.712	15.0533	4.3376	3.4705	0.5287
	36.000	58.793	10.8943	3.9283	2.7733	1.2149	0.30670		22.000	82.610	15.1191	4.3427	3.4815	0.5127
	37.306	65.769	12.4065	4.0985	3.0271	0.9660	0.25708		20.000	83.440	15.1734	4.3470	3.4906	0.4992
	36.000	71.617	13.4496	4.2005	3.2019	0.7805	0.22897		18.000	84.215	15.2184	4.3505	3.4981	0.4877
	34.000	74.634	13.8916	4.2405	3.2760	0.6945	0.21831		16.000	84.947	15.2557	4.3534	3.5043	0.4781
	32.000	76.633	14.1452	4.2626	3.3184	0.6420	0.21249		14.000	85.644	15.2866	4.3558	3.5095	0.4689
	30.000	78.190	14.3199	4.2776	3.3477	0.6041	0.20861		12.000	86.313	15.3120	4.3577	3.5137	0.4632
	28.000	79.487	14.4500	4.2885	3.3695	0.5746	0.20578		10.000	86.959	15.3325	4.3593	3.5172	0.4576
	26.000	80.614	14.5512	4.2969	3.3864	0.5510	0.20362		8.000	87.587	15.3487	4.3606	3.5199	0.4532
	24.000	81.617	14.6320	4.3036	3.3999	0.5315	0.20191		6.000	88.201	15.3609	4.3615	3.5219	0.4499
	22.000	82.528	14.6976	4.3090	3.4109	0.5152	0.20054		4.000	88.807	15.3695	4.3622	3.5234	0.4475
	20.000	83.369	14.7517	4.3134	3.4200	0.5015	0.19942		2.000	89.405	15.3746	4.3625	3.5242	0.4461
	18.000	84.154	14.7965	4.3170	3.4275	0.4899	0.19849							
	16.000	84.894	14.8336	4.3200	3.4337	0.4801	0.19774							
	14.000	85.599	14.8643	4.3225	3.4388	0.4719	0.19711							
	12.000	86.275	14.8895	4.3245	3.4430	0.4651	0.19660							
	10.000	86.928	14.9099	4.3262	3.4465	0.4595	0.19619							
	8.000	87.562	14.9260	4.3274	3.4491	0.4551	0.19586							
	6.000	88.184	14.9381	4.3284	3.4512	0.4517	0.19562							
	4.000	88.794	14.9466	4.3291	3.4529	0.4493	0.19545							
	2.000	89.398	14.9517	4.3295	3.4534	0.4479	0.19535							
3.65	2.000			17.250	1.2001	1.0537	3.5198	0.99938	20.000	33.653	4.7382	2.7406	1.7289	2.4105
	4.000	18.701	1.4312	1.2902	1.1094	3.3936	0.99932	22.000	35.985	5.3474	2.9156	1.8341	2.2783	0.64001
	6.000	20.256	1.6964	1.4524	1.1680	3.2691	0.98515	24.000	38.426	6.0027	3.0840	1.9464	2.453	0.58349
	8.000	21.913	1.9980	1.6239	1.2304	3.1451	0.96710	26.000	40.991	6.7053	3.2452	2.0662	2.0114	0.52883
	10.000	23.668	2.3381	1.8024	1.2972	3.0207	0.94042	28.000	43.704	7.4580	3.3993	2.1940	1.8758	0.47677
	12.000	25.520	2.7183	1.9854	1.3691	2.8953	0.90525	30.000	46.605	8.2664	3.5467	2.3307	1.7375	0.42765
	14.000	27.468	3.1402	2.1707	1.4466	2.7688	0.86248	32.000	49.768	9.1422	3.6886	2.4785	1.5940	0.38140
	16.000	29.509	3.6043	2.3558	1.5300	2.6412	0.81364	34.000	53.344	10.1123	3.8277	3.0840	2.6418	1.4404
	18.000	31.645	4.1117	2.5387	1.6196	2.5125	0.76044	36.000	57.760	11.2596	3.9721	2.8346	2.6223	1.29362
	20.000	33.878	4.6628	2.7176	2.3830	2.07470	0.70470	37.713	65.847	13.1309	4.1705	3.1485	2.9675	0.23710
	22.000	36.212	5.2580	2.8911	2.8911	2.2527	0.64814	36.000	72.443	14.3517	4.2802	3.3530	0.7577	0.20791
	24.000	38.658	5.8984	3.0584	2.9286	2.1215	0.59212	34.000	75.135	14.7539	4.3136	3.4203	0.6814	0.19937
	26.000	41.230	6.5849	3.2189	2.0457	1.9891	0.53777	32.000	77.009	14.9979	4.3332	3.4612	0.6324	0.19442
	28.000	43.954	7.3210	3.3726	2.1707	1.8549	0.48578	30.000	78.492	15.1693	4.3467	3.4899	0.5962	0.19104
	30.000	46.873	8.1124	3.5199	2.3047	2.1716	0.43650	28.000	79.740	15.2983	4.3567	3.5115	0.5680	0.18855
	32.000	50.064	8.9714	3.6622	2.4497	1.5746	0.38990	26.000	80.828	15.3992	4.3664	3.5283	0.5451	0.18664
	34.000	53.694	9.9271	3.8025	2.6107	1.4207	0.34529	24.000	81.802	15.4802	4.3706	3.5419	0.5261	0.18512
	36.000	58.251	11.0727	3.9499	2.8033	1.2394	0.30022	22.000	82.688	15.5463	4.3756	3.5530	0.5103	0.18389

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{\rho_2}{\rho_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$
3.70	20.000	83.507	15.6008	4.3797	3.5621	0.4969	0.18289	3.75	2.000	89.416	16.2379	4.4261	3.6687	0.4428
	18.000	84.274	15.6460	4.3831	3.5656	0.4856	0.18206						1.0558	3.6624
	16.000	84.998	15.6836	4.3859	3.5759	0.4760	0.18138						1.3022	0.99931
	14.000	85.687	15.7147	4.3882	3.5811	0.4680	0.18082	3.80	2.000	89.416	16.2379	4.4261	1.1137	0.99479
	12.000	86.348	15.7402	4.3901	3.5854	0.4613	0.18035						1.4503	0.98349
	10.000	86.988	15.7609	4.3916	3.5889	0.4558	0.17998						1.7294	0.96355
	8.000	87.610	15.7772	4.3928	3.5916	0.4515	0.17989						2.0480	0.93423
	6.000	88.219	15.7896	4.3937	3.5937	0.4481	0.17947	10.000	2.000	89.416	16.2379	4.4261	1.2404	0.89566
	4.000	88.817	15.7982	4.3944	3.5951	0.4458	0.17932	12.000	2.000	89.416	16.2379	4.4261	1.6511	0.84963
	2.000	89.411	15.8033	4.3947	3.5960	0.4444	0.17922						2.0288	0.89697
3.75	2.000		16.810	1.2055	1.1426	1.0551	0.99933	20.000	2.000	89.416	16.2379	4.4261	1.3108	0.79728
	4.000	18.260	1.4440	1.2982	1.1123	3.4840	0.999497	22.000	2.000	89.416	16.2379	4.4261	1.3867	0.79728
	6.000	19.814	1.7184	1.4654	1.1727	3.3550	0.98405	24.000	2.000	89.416	16.2379	4.4261	1.4718	0.79728
	8.000	21.470	2.0312	1.6420	1.2370	3.2264	0.96476	26.000	2.000	89.416	16.2379	4.4261	1.7294	0.79728
	10.000	23.227	2.3849	1.8258	1.3062	3.0974	0.93634	28.000	2.000	89.416	16.2379	4.4261	2.0480	0.79728
	12.000	25.081	2.7813	2.0142	1.3808	2.9674	0.89905	30.000	2.000	89.416	16.2379	4.4261	2.4088	0.79728
	14.000	27.030	3.2217	2.2046	1.4614	2.8363	0.85397	32.000	2.000	89.416	16.2379	4.4261	2.8134	0.79728
	16.000	29.072	3.7069	2.3943	1.5482	2.7042	0.80280	34.000	2.000	89.416	16.2379	4.4261	3.2631	0.79728
	18.000	31.207	4.2379	2.5813	1.6417	2.5712	0.74744	36.000	2.000	89.416	16.2379	4.4261	3.7592	0.79728
	20.000	33.438	4.8148	2.7637	1.7422	2.4376	0.68987	38.000	2.000	89.416	16.2379	4.4261	4.2137	0.79728
	22.000	35.767	5.4382	2.9401	1.8497	2.3034	0.63185	38.092	2.000	89.416	16.2379	4.4261	4.6517	0.79728
	24.000	38.204	6.1086	3.1095	1.9645	2.1688	0.57486	38.000	2.000	89.416	16.2379	4.4261	5.1080	0.79728
	26.000	40.762	6.8272	3.2714	2.0869	2.0333	0.51996	36.000	2.000	89.416	16.2379	4.4261	5.5972	0.79728
	28.000	43.464	7.5969	3.4259	2.2175	1.8964	0.46786	34.000	2.000	89.416	16.2379	4.4261	6.0943	0.79728
	30.000	46.350	8.4228	3.5733	2.3572	1.7570	0.41888	32.000	2.000	89.416	16.2379	4.4261	6.6341	0.79728
	32.000	49.486	9.3159	3.7148	2.5078	1.6129	0.37500	30.000	2.000	89.416	16.2379	4.4261	7.2373	0.79728
	34.000	53.014	10.0313	3.8529	2.6736	1.4594	0.32964	28.000	2.000	89.416	16.2379	4.4261	7.8700	0.79728
	36.000	57.310	11.4538	3.9947	2.8672	1.2839	0.28896	26.000	2.000	89.416	16.2379	4.4261	8.5356	0.79728
	37.906	65.884	13.5007	4.2052	3.2105	0.9683	0.22770	24.000	2.000	89.416	16.2379	4.4261	9.2822	0.79728
	36.000	72.794	14.8041	4.3176	3.4287	0.7481	0.19834	22.000	2.000	89.416	16.2379	4.4261	9.9397	0.79728
	34.000	75.361	15.1917	4.3484	3.4936	0.6755	0.19061	20.000	2.000	89.416	16.2379	4.4261	10.60402	0.79728
	32.000	77.180	15.4318	4.3669	3.5338	0.6280	0.18602	18.000	2.000	89.416	16.2379	4.4261	11.4221	0.79728
	30.000	78.631	15.6021	4.3798	3.5623	0.5926	0.18286	16.000	2.000	89.416	16.2379	4.4261	12.3571	0.79728
	28.000	79.856	15.7307	4.3894	3.5838	0.5649	0.18053	14.000	2.000	89.416	16.2379	4.4261	13.3512	0.79728
	26.000	80.927	15.8316	4.3968	3.6007	0.5423	0.17872	12.000	2.000	89.416	16.2379	4.4261	14.3997	0.79728
	24.000	81.887	15.9128	4.4028	3.6143	0.5237	0.17728	10.000	2.000	89.416	16.2379	4.4261	15.4729	0.79728
	22.000	82.762	15.9792	4.4076	3.6254	0.5080	0.17612	8.000	2.000	89.416	16.2379	4.4261	16.3532	0.79728
	20.000	83.572	16.0339	4.4115	3.6345	0.4948	0.17517	6.000	2.000	89.416	16.2379	4.4261	17.1586	0.79728
	18.000	84.330	16.0794	4.4148	3.6422	0.4836	0.17439	4.000	2.000	89.416	16.2379	4.4261	17.9633	0.79728
	16.000	85.045	16.1172	4.4175	3.6485	0.4741	0.17374	2.000	2.000	89.416	16.2379	4.4261	18.7457	0.79728
	14.000	85.727	16.1485	4.4198	3.6537	0.4662	0.17321						19.5633	0.79728
	12.000	86.382	16.1743	4.4216	3.6580	0.4595	0.17277						20.3633	0.79728
	10.000	87.016	16.1951	4.4231	3.6615	0.4541	0.17242	3.85	2.000	89.416	16.2379	4.4261	21.110	0.79728
	8.000	87.632	16.2116	4.4242	3.6643	0.4498	0.17214						21.9457	0.79728
	6.000	88.235	16.2240	4.4251	3.6663	0.4465	0.17193						22.7797	0.79728
4.000	88.829	16.2327	4.4257	3.6678	0.4441	0.17178						23.6137	0.79728	

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{P_2}{P_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	
3.85	10.000	22.812	1.8495	1.3153	3.1734	0.93209	3.90	24.000	37.584	6.4345	3.1853	2.0201	2.2371	
	12.000	24.668	2.8456	2.0432	1.3927	3.0386	0.89264	26.000	40.126	7.2035	3.3492	2.1568	2.0968	
	14.000	26.619	3.3050	2.2386	1.4764	2.9028	0.84523	28.000	42.802	8.0258	3.5046	2.2901	1.9558	
	16.000	28.664	3.8121	2.4330	1.5668	2.7661	0.79172	30.000	45.646	9.8059	3.6519	2.4387	1.8131	
	18.000	30.799	4.3670	2.6239	1.6643	2.6287	0.73428	32.000	48.716	9.8536	3.7923	2.5983	1.6668	
	20.000	33.028	4.9706	2.8097	1.7691	2.4909	0.67493	34.000	52.126	10.8901	3.9278	2.7726	1.5130	
	22.000	35.353	5.6230	2.9887	1.8814	2.3529	0.61558	36.000	56.149	12.0723	4.0633	2.9710	1.3425	
	24.000	37.783	6.3245	3.1601	2.0013	2.2146	0.55770	38.000	62.087	13.6897	4.2224	3.2421	1.1106	
	26.000	40.330	7.0764	3.3234	2.1293	2.0760	0.50236	38.445	65.991	14.6407	4.3043	3.4014	0.9704	
	28.000	43.014	7.8808	3.4785	2.2656	1.9364	0.45026	38.000	69.501	15.4023	4.3647	3.5289	0.8527	
	30.000	45.871	8.7425	3.6259	2.4111	1.7948	0.40167	36.000	73.678	16.1768	4.4218	3.6584	0.7240	
	32.000	48.961	9.6715	3.7666	2.5677	1.6493	0.35654	34.000	75.956	16.5334	4.4468	3.7181	0.6600	
	34.000	52.407	10.6904	3.9030	2.7390	1.4957	0.31434	32.000	77.640	16.7653	4.4626	3.7589	0.6160	
	36.000	56.508	11.8605	4.0404	2.9355	1.3239	0.27366	30.000	79.006	16.9330	4.4738	3.7849	0.5828	
	38.000	62.939	13.5472	4.2095	3.2183	1.0767	0.22655	28.000	80.172	17.0613	4.4823	3.8064	0.5633	
	38.272	65.956	14.2556	4.2721	3.3369	0.9697	0.211003	26.000	81.199	17.1629	4.4890	3.8234	0.5347	
	38.000	68.733	14.8512	4.3214	3.4366	0.8764	0.19738	24.000	82.121	17.2449	4.4943	3.8371	0.5168	
	36.000	73.407	15.7160	4.3883	3.5814	0.7314	0.18079	22.000	82.966	17.3122	4.4986	3.8483	0.5016	
	34.000	75.770	16.0813	4.4150	3.6425	0.6649	0.17436	20.000	83.749	17.3680	4.5022	3.8576	0.4888	
	32.000	77.495	16.3155	4.4316	3.6816	0.6198	0.17039	18.000	84.483	17.4143	4.5052	3.8654	0.4780	
	30.000	78.888	16.4839	4.4433	3.7098	0.5859	0.16762	16.000	85.177	17.4529	4.5076	3.8778	0.4688	
	28.000	80.072	16.6122	4.4522	3.7313	0.5591	0.16555	14.000	85.840	17.4850	4.5097	3.8772	0.4610	
	26.000	81.112	16.7135	4.4591	3.7482	0.5372	0.16394	12.000	86.477	17.5113	4.5114	3.8816	0.4545	
	24.000	82.047	16.7952	4.4646	3.7619	0.5190	0.16266	10.000	87.093	17.5327	4.5127	3.8852	0.4492	
	22.000	82.901	16.8622	4.4691	3.7731	0.5037	0.16162	8.000	87.693	17.5496	4.5138	3.8880	0.4450	
	20.000	83.692	16.9175	4.4728	3.7823	0.4907	0.16076	6.000	88.280	17.5623	4.5146	3.8901	0.4418	
	18.000	84.434	16.9636	4.4758	3.7900	0.4798	0.16006	4.000	88.858	17.5713	4.5151	3.8916	0.4395	
	16.000	85.136	17.0019	4.4784	3.7964	0.4705	0.15947	2.000	89.430	17.5786	4.5155	3.8925	0.4382	
	14.000	85.804	17.0337	4.4805	3.8017	0.46227	0.15899							
	12.000	86.447	17.0598	4.4822	3.8061	0.4561	0.15859							
	10.000	87.068	17.0810	4.4836	3.8097	0.4508	0.15827	3.95	2.000	16.001	1.2166	1.1500	1.0578	
	8.000	87.674	17.0978	4.4847	3.8125	0.4465	0.15802	4.000	17.447	1.4697	1.3144	1.1182	3.6641	
	6.000	88.266	17.1104	4.4855	3.8146	0.4433	0.15783	6.000	19.001	1.7630	1.4915	1.1821	3.5255	
	4.000	88.849	17.1193	4.4861	3.8161	0.4410	0.15770	8.000	20.660	2.0992	1.6786	1.2506	3.3874	
	2.000	89.426	17.1245	4.4865	3.8169	0.4397	0.15762	10.000	22.422	2.4815	1.8734	1.3246	3.2486	
								12.000	24.280	2.9112	2.0724	1.4048	3.1090	0.88602
3.90	2.000	16.196	1.2138	1.1482	3.7573	0.99926	14.000	26.234	3.3902	2.2727	1.4917	2.9684	0.99923	
	4.000	17.642	1.4633	1.3104	1.1167	0.99441	16.000	28.281	3.9194	2.4716	1.5858	2.8270	0.88626	
	6.000	19.196	1.7517	1.4849	1.1797	0.98322	20.000	32.646	5.1304	2.8554	1.7967	2.5430	0.65992	
	8.000	20.854	2.0821	1.6694	1.2472	3.3473	0.96105	22.000	34.969	5.8125	3.0370	1.9139	2.4010	0.59933
	10.000	22.614	2.4570	1.8614	1.3200	3.2111	0.9290	24.000	37.393	6.5462	3.2103	2.0391	2.2591	0.54068
	12.000	24.472	2.8783	2.0578	1.3987	3.0739	0.88935	26.000	39.929	7.3323	3.3748	2.1727	2.1172	0.48503
	14.000	26.424	3.3474	2.2557	1.4840	2.9857	0.84077	28.000	42.598	8.1726	3.5304	2.3149	1.9748	0.43302
	16.000	28.469	3.8655	2.4523	1.5763	2.7967	0.78611	30.000	45.431	9.0717	3.6778	2.4666	1.8310	0.38488
	18.000	30.605	4.4329	2.6492	1.6758	2.6570	0.72761	32.000	48.483	10.0386	3.8178	2.6294	1.6838	0.34053
	20.000	32.834	5.0501	2.8326	1.7828	2.5171	0.66743	34.000	51.859	11.0931	3.9524	2.8067	1.5299	0.29849
	22.000	35.157	5.7171	3.0129	1.8975	2.3771	0.60746	36.000	55.812	12.2888	4.0863	3.0073	1.3604	0.26054

## Oblique Shock Tables ( $\gamma = 1.4$ )

$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{T_2}{T_1}$	$M_2$	$\frac{P_{02}}{P_{01}}$	$M_1$	$\theta$	$\beta$	$\frac{p_2}{p_1}$	$\frac{\rho_2}{\rho_1}$	$M_2$	$\frac{T_2}{T_1}$	$\frac{P_{02}}{P_{01}}$
3.95	38.000	61.406	13.8667	4.2383	3.2718	1.1389	0.21889	4.00	28.000	80.359	17.9765	4.5402	3.9594	0.5513
	38.612	66.026	15.0309	4.3358	3.4667	0.9711	0.19376		26.000	81.359	18.0767	4.5464	3.9765	0.5302
	38.000	70.101	15.9275	4.4038	3.6167	0.8345	0.17703		24.000	82.261	18.1615	4.5514	3.9803	0.5126
	36.000	73.928	16.6412	4.4541	3.7361	0.7172	0.16509		22.000	83.087	18.2296	4.5555	4.0017	4.4978
	34.000	76.131	6.9904	4.4776	3.7945	0.6554	0.15965		20.000	83.854	18.2861	4.5588	4.0111	4.4852
	32.000	77.777	17.2203	4.4927	3.8330	0.6125	0.15620		18.000	84.574	18.3331	4.5616	4.0190	4.4746
	30.000	79.120	17.3877	4.5035	3.8609	0.5798	0.15375		16.000	85.256	18.3723	4.5639	4.0255	4.4655
	28.000	80.268	17.5161	4.5117	3.8824	0.5537	0.15191		14.000	85.907	18.4049	4.5659	4.0310	4.4579
	26.000	81.281	17.6179	4.5181	3.8994	0.5324	0.15047		12.000	86.533	18.4317	4.5674	4.0355	4.4515
	24.000	82.192	17.7003	4.5232	3.9132	0.5147	0.14932		10.000	87.139	18.4535	4.5687	4.0391	4.4463
	22.000	83.028	17.7680	4.5274	3.9245	0.4997	0.14838		8.000	87.730	18.4707	4.5697	4.0420	4.4421
	20.000	83.803	17.8241	4.5309	3.9339	0.4870	0.14761		6.000	88.307	18.4837	4.5705	4.0442	4.4390
	18.000	84.529	17.8708	4.5338	3.9417	0.4762	0.14698		4.000	88.876	18.4928	4.5710	4.0457	4.4367
	16.000	85.218	17.9097	4.5362	3.9482	0.4671	0.14645		2.000	89.439	18.4982	4.5713	4.0466	4.4354
	14.000	85.874	17.9420	4.5381	3.9536	0.4594	0.14601							
	12.000	86.505	17.9686	4.5398	3.9581	0.4530	0.14566							
	10.000	87.116	17.9902	4.5411	3.9617	0.4477	0.14537							
	8.000	87.711	18.0072	4.5421	3.9645	0.4435	0.14514							
	6.000	88.294	18.0201	4.5429	3.9667	0.4404	0.14497							
	4.000	88.868	18.0291	4.5434	3.9682	0.4381	0.14485							
	2.000	89.435	18.0345	4.5438	3.9691	0.4368	0.14478							
4.00	2.000	15.813	1.2194	1.1519	1.0586	1.1196	3.7089	0.99920						
	4.000	17.258	1.4763	1.3185	1.3421	1.1844	3.5679	0.99401						
	6.000	18.812	1.7743	1.4980	1.2540	1.34273	0.95845							
	8.000	20.471	2.1166	1.6879	1.3293	3.2860	0.92542							
	10.000	22.234	2.5061	1.8853	2.0870	1.4109	3.1439	0.88264						
	12.000	24.095	2.9445	2.0870	2.2898	1.4994	3.0009	0.83170						
	14.000	26.050	3.4334	3.0611	2.4909	1.5954	2.8570	0.77474						
	16.000	28.098	3.9741	2.4909	2.6877	1.6991	2.7128	0.71422						
	18.000	30.236	4.5667	3.5561	2.8782	1.8107	2.5686	0.65240						
	20.000	32.464	5.2116	3.0611	3.9304	2.4246	0.59123							
	22.000	34.786	5.9090	3.2352	6.6592	2.0583	2.2809	0.53224						
	24.000	37.208	6.6592	3.9768	7.4625	3.4002	2.1947	0.47648						
	26.000	39.740	7.4625	4.1091	3.5561	2.3401	1.9935	0.42453						
	28.000	42.402	8.3215	3.7034	3.4949	1.8485	1.8485	0.37666						
	30.000	45.224	9.2397	3.8430	2.6609	1.7006	0.33272							
	32.000	48.258	10.2259	4.4403	3.9768	2.8413	1.5463	0.29223						
	34.000	51.605	11.2995	4.4855	4.1091	3.0444	1.3776	0.25409						
	36.000	55.495	12.5100	4.2556	4.1091	3.3049	1.1637	0.21432						
	38.000	60.827	14.0647	4.3665	4.4403	3.5329	0.9717	0.18613						
	38.774	66.059	15.4261	4.4855	4.4855	3.8144	0.8196	0.16833						
	38.000	70.601	16.4407	4.5076	4.5220	3.8718	0.6511	0.15282						
	36.000	74.161	17.1095	4.5324	4.5324	3.9099	0.6090	0.14959						
	34.000	76.297	17.4525	4.5324	4.5324	3.9379	0.5769	0.14729						
	32.000	77.908	17.6808	4.5324	4.5324	3.9379								
	30.000	79.227	17.8479	4.5324	4.5324	3.9379								