

ENGINEERING TRIPOS PART IIB

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Friday 6 May 2011 9 to 10.30

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Module 4A12

TURBULENCE AND VORTEX DYNAMICS

*Answer not more than **three** questions.*

*All questions carry the same number of marks.*

*The **approximate** percentage of marks allocated to each part of a question is indicated in the right margin.*

*Attachments:*

*4A12 Data Card: (i) Vortex Dynamics (1 page);  
(ii) Turbulence (2 pages)*

STATIONERY REQUIREMENTS

Single-sided script paper

SPECIAL REQUIREMENTS

Engineering Data Book

CUED approved calculator allowed

**You may not start to read the questions  
printed on the subsequent pages of this  
question paper until instructed that you  
may do so by the Invigilator**

1 (a) State the Prandtl-Batchelor theorem. [10%]

(b) Consider a steady, two-dimensional flow with closed streamlines. It has a steady temperature field  $T$ , thermal diffusivity  $\alpha$ , and stream-function  $\psi$ .

(i) Starting with the advection-diffusion equation:

$$\mathbf{u} \cdot \nabla T = \alpha \nabla^2 T$$

show that  $T$  takes the form  $T = T(\psi)$  when  $\alpha = 0$ .

(ii) Hence show that, when  $\alpha$  is small but finite:

$$\mathbf{u} \cdot \nabla T = \alpha \left\{ \nabla \cdot [T'(\psi) \nabla \psi] + (\text{small correction}) \right\}$$

where the correction tends to zero as  $\alpha \rightarrow 0$ .

(iii) Use Gauss' theorem to integrate this equation over the area  $A$  which is bounded by a streamline  $C$ . Confirm that:

$$\alpha \left\{ T'(\psi) \int \nabla \psi \cdot d\mathbf{S} + (\text{small correction}) \right\} = 0$$

where  $d\mathbf{S} = \mathbf{n} d\ell$  is part of the surface that encircles the area  $A$ ,  $\mathbf{n}$  is a unit normal to the streamline and  $d\ell$  is part of that streamline.

(iv) Use the fact that  $\int \nabla \psi \cdot d\mathbf{S} = -\oint_C \mathbf{u} \cdot d\mathbf{r}$ , where  $d\mathbf{r}$  is a short element of the streamline,  $|d\mathbf{r}| = d\ell$ , to deduce that  $T'(\psi) = 0$  in the limit of  $\alpha \rightarrow 0$ . What is the physical interpretation of this result? [50%]

(c) Discuss how the proof of  $T'(\psi) = 0$  in part (b) may be adapted to prove the Prandtl-Batchelor theorem. [30%]

(d) What is the physical interpretation of the Prandtl-Batchelor theorem? [10%]

2 (a) State Kelvin's circulation theorem and Helmholtz's two laws of vortex dynamics. [20%]

(b) Consider two adjacent fluid particles,  $A$  and  $B$ , in an inviscid fluid which are linked by the displacement vector  $d\mathbf{r}(t) = \mathbf{x}_B(t) - \mathbf{x}_A(t)$  and which lie on the same vortex line at  $t = 0$ . Since  $d\mathbf{r}$  and  $\boldsymbol{\omega}(\mathbf{x}_A)$  are parallel at  $t = 0$ , we may write  $d\mathbf{r}(t = 0) = \lambda \boldsymbol{\omega}(\mathbf{x}_A, t = 0)$ , where  $\lambda$  is some scalar. We give  $\lambda$  the property  $D\lambda/Dt = 0$ , so that it is a constant of the fluid element  $AB$  for  $t > 0$ , and introduce the vector:

$$\mathbf{C}(\mathbf{x}_A, t) = d\mathbf{r}(t) - \lambda \boldsymbol{\omega}(\mathbf{x}_A, t)$$

which is zero at  $t = 0$ .

(i) Use the fact that a short material line element obeys:

$$\frac{D}{Dt}(d\mathbf{r}) = (d\mathbf{r} \cdot \nabla) \mathbf{u}$$

to show that, in an inviscid fluid:

$$\frac{D\mathbf{C}}{Dt} = \mathbf{C} \cdot \nabla \mathbf{u}$$

(ii) Hence show that, by virtue of the initial conditions, we have  $\mathbf{C}(\mathbf{x}_A, t) = 0$  for  $t > 0$ , and that consequently:

$$\frac{d\mathbf{r}(t)}{|d\mathbf{r}(t=0)|} = \frac{\boldsymbol{\omega}(\mathbf{x}_A, t)}{|\boldsymbol{\omega}(\mathbf{x}_A, t=0)|}$$

(iii) Deduce that  $A$  and  $B$  always lie on the same vortex line and that consequently vortex lines in an inviscid fluid are convected like material lines. [50%]

(c) An alternative proof that vortex lines are frozen into an inviscid fluid can be constructed with the help of Kelvin's theorem. Consider a thin, isolated vortex tube and a closed curve  $C$  that encircles the tube at  $t = 0$ . Use Kelvin's theorem to show that, if  $C$  is a material curve always composed of the same fluid particles, then  $C$  must encircle the vortex tube for all time. Deduce that vortex lines must move with the fluid, like dye lines. [30%]

PAD03

(TURN OVER)

3 A hot wire is to be used at the centreline of a turbulent axisymmetric air jet originating from a round nozzle with diameter  $d = 10 \text{ mm}$  and with a uniform nozzle velocity  $U_j = 50 \text{ ms}^{-1}$ . Both the jet fluid and the ambient air are at 1 bar and 300 K. The mean centreline velocity  $U_0$  at a distance  $x$  from the nozzle obeys  $U_0/U_j = 6.4(x/d)^{-1}$ .

(a) By making reasonable guesses for the turbulent integral length-scale and the r.m.s. velocity fluctuations at  $x = 50d$ , estimate the highest frequency of motion seen by the hot wire at this location. [70%]

(b) For how long should one collect data if we require the mean measured velocity to be within 1% of its true average at this location? [30%]

4 In a dense rain, raindrops of diameter  $d = 2$  mm and at a number density of  $n = 10^4$  drops/m<sup>3</sup> fall at their terminal velocity  $u = 2$  ms<sup>-1</sup> in air that has no mean flow. Assume that the wakes behind the drops eventually produce stationary homogeneous isotropic turbulence with integral length-scale  $L$  and kinetic energy  $k$ .

(a) Discuss why we may expect  $d < L < n^{-1/3}$ . [20%]

(b) Using the extreme values for  $L$  from part (a), provide upper and lower bounds for the characteristic turbulent velocity  $k^{1/2}$ . Find  $L$  if  $k^{1/2} = u$ . [60%]

(c) In a more detailed view of the turbulence in this problem, the wakes behind the drops can be thought of as self-preserving axisymmetric wakes. Given that the half-width  $\delta_{1/2}$  of an axisymmetric wake grows as  $\delta_{1/2}/d \approx 0.6(x/d)^{1/3}$ , where  $x$  is the distance behind the drop, provide an alternative estimate for  $L$ . [20%]

**END OF PAPER**



## Vortex Dynamics Data Card

### Grad, Div and Curl in Cartesian Coordinates

$$\nabla f = \left( \frac{\partial f}{\partial x}, \frac{\partial f}{\partial y}, \frac{\partial f}{\partial z} \right)$$

$$\nabla \cdot A = \frac{\partial A_x}{\partial x} + \frac{\partial A_y}{\partial y} + \frac{\partial A_z}{\partial z}$$

$$\nabla \times A = \begin{vmatrix} i & j & k \\ \frac{\partial}{\partial x} & \frac{\partial}{\partial y} & \frac{\partial}{\partial z} \\ A_x & A_y & A_z \end{vmatrix} = \left( \frac{\partial A_z}{\partial y} - \frac{\partial A_y}{\partial z}, \frac{\partial A_x}{\partial z} - \frac{\partial A_z}{\partial x}, \frac{\partial A_y}{\partial x} - \frac{\partial A_x}{\partial y} \right)$$

### Integral Theorems

$$\text{Gauss : } \int (\nabla \cdot A) dV = \oint A \cdot dS$$

$$\text{Stokes : } \int (\nabla \times A) \cdot dS = \oint A \cdot dl$$

### Vector Identities

$$\nabla(A \cdot B) = (A \cdot \nabla)B + (B \cdot \nabla)A + A \times (\nabla \times B) + B \times (\nabla \times A)$$

$$\nabla \cdot (fA) = f(\nabla \cdot A) + A \cdot \nabla f$$

$$\nabla \times (A \times B) = A(\nabla \cdot B) - B(\nabla \cdot A) + (B \cdot \nabla)A - (A \cdot \nabla)B$$

$$\nabla \times (\nabla \times A) = \nabla(\nabla \cdot A) - \nabla^2 A$$

$$\nabla \times (\nabla f) = 0$$

$$\nabla \cdot (\nabla \times A) = 0$$

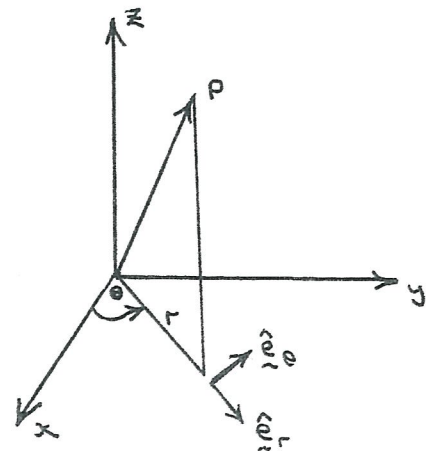
### Cylindrical Coordinates (r, θ, z)

$$\nabla f = \left( \frac{\partial f}{\partial r}, \frac{1}{r} \frac{\partial f}{\partial \theta}, \frac{\partial f}{\partial z} \right)$$

$$\nabla \cdot A = \frac{1}{r} \frac{\partial}{\partial r} (rA_r) + \frac{1}{r} \frac{\partial A_\theta}{\partial \theta} + \frac{\partial A_z}{\partial z}$$

$$\nabla \times A = \frac{1}{r} \begin{vmatrix} \hat{e}_r & r\hat{e}_\theta & \hat{e}_z \\ \frac{\partial}{\partial r} & \frac{\partial}{\partial \theta} & \frac{\partial}{\partial z} \\ A_r & rA_\theta & A_z \end{vmatrix}$$

$$\nabla \times A = \left( \frac{1}{r} \frac{\partial A_z}{\partial \theta} - \frac{\partial A_\theta}{\partial z}, \frac{\partial A_r}{\partial z} - \frac{\partial A_z}{\partial r}, \frac{1}{r} \frac{\partial}{\partial r} (rA_\theta) - \frac{1}{r} \frac{\partial A_r}{\partial \theta} \right)$$



# Cambridge University Engineering Department

## 4A12: Turbulence

### Data Card

Assume incompressible fluid with constant properties.

Continuity:

$$\frac{\partial \bar{u}_i}{\partial x_i} = 0$$

Mean momentum:

$$\frac{\partial \bar{u}_i}{\partial t} + \bar{u}_j \frac{\partial \bar{u}_i}{\partial x_j} = -\frac{1}{\rho} \frac{\partial \bar{p}}{\partial x_i} + \nu \partial^2 \bar{u}_i / \partial x_j^2 - \frac{\partial \overline{u'_i u'_j}}{\partial x_j} + \bar{g}_i$$

Mean scalar:

$$\frac{\partial \bar{\phi}}{\partial t} + \bar{u}_i \frac{\partial \bar{\phi}}{\partial x_i} = D \frac{\partial^2 \bar{\phi}}{\partial x_i^2} - \frac{\partial \overline{u'_i \phi'}}{\partial x_i}$$

Turbulent kinetic energy ( $k = \overline{u'_i u'_i} / 2$ ):

$$\begin{aligned} \frac{\partial k}{\partial t} + \bar{u}_j \frac{\partial k}{\partial x_j} &= -\frac{1}{\rho} \frac{\partial \overline{u'_j p'}}{\partial x_j} - \frac{1}{2} \frac{\partial \overline{u'_j u'_i u'_i}}{\partial x_j} + \nu \frac{\partial^2 k}{\partial x_j^2} \\ &\quad - \overline{u'_i u'_j} \frac{\partial \bar{u}_i}{\partial x_j} - \nu \overline{\left( \frac{\partial u'_i}{\partial x_j} \right)^2} + \bar{g}'_i u'_i \end{aligned}$$

Scalar fluctuations ( $\sigma^2 = \overline{\phi' \phi'}$ ):

$$\frac{\partial \sigma^2}{\partial t} + \bar{u}_j \frac{\partial \sigma^2}{\partial x_j} = D \frac{\partial^2 \sigma^2}{\partial x_j^2} - 2 \overline{\phi' u'_j} \frac{\partial \phi'}{\partial x_j} - 2 \overline{\phi' u'_j} \frac{\partial \bar{\phi}}{\partial x_j} - 2D \overline{\left( \frac{\partial \phi'}{\partial x_j} \right)^2}$$

Energy dissipation:

$$\varepsilon = \nu \overline{\left( \frac{\partial u'_i}{\partial x_j} \right)^2} \approx \frac{u^3}{L_{turb}}$$



Scalar dissipation:

$$2\bar{N} = 2D \overline{\left(\frac{\partial \phi'}{\partial x_j}\right)^2} \approx 2\frac{\varepsilon}{k}\sigma^2$$

Scaling rule for shear flow, flow dominant in direction  $x_1$ :

$$\frac{u}{L_{turb}} \sim \frac{\partial \bar{u}_1}{\partial x_2}$$

Kolmogorov scales:

$$\begin{aligned}\eta_K &= (\nu^3/\varepsilon)^{1/4} \\ \tau_K &= (\nu/\varepsilon)^{1/2} \\ \nu_K &= (\nu\varepsilon)^{1/4}\end{aligned}$$

Taylor microscale:

$$\varepsilon = 15\nu \frac{u^2}{\lambda^2}$$

Eddy viscosity (general):

$$\begin{aligned}\overline{u'_i u'_j} &= -\nu_{turb} \left( \frac{\partial \bar{u}_i}{\partial x_j} + \frac{\partial \bar{u}_j}{\partial x_i} \right) + \frac{2}{3} k \delta_{ij} \\ \overline{u'_j \phi'} &= -D_{turb} \frac{\partial \bar{\phi}}{\partial x_j}\end{aligned}$$

Eddy viscosity (for simple shear):

$$\overline{u'_1 u'_2} = -\nu_{turb} \frac{\partial \bar{u}_1}{\partial x_2}$$



**ENGINEERING TRIPOS PART IIB 2011**

**MODULE 4A12: TURBULENCE AND VORTEX DYNAMICS.**

**NUMERICAL ANSWERS**

Q1. –

Q2. –

Q3. (a) 80kHz, (b) 4.9 s

Q4. (a) -, (b)  $L = 10.7$  m, (c)  $L = 3.4$  mm