

ENGINEERING TRIPOS PART IIB

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Friday 6 May 2011 2.30 to 4

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Module 4C7

RANDOM AND NON-LINEAR VIBRATIONS

*Answer not more than **three** questions.*

*All questions carry the same number of marks.*

*Candidates may bring their notebooks to the examination.*

*The **approximate** percentage of marks allocated to each part of a question is indicated in the right margin.*

*There are no attachments.*

STATIONERY REQUIREMENTS

Single-sided script paper

SPECIAL REQUIREMENTS

CUED approved calculator allowed

Engineering Data Book

**You may not start to read the questions printed on the subsequent pages of this question paper until instructed that you may do so by the Invigilator**

- 1 (a) The autocorrelation function  $R_{xx}(\tau)$  of a stationary random process  $x(t)$  is given by

$$R_{xx}(\tau) = \left( \frac{\pi S_0}{\gamma} \right) e^{-\gamma|\tau|}$$

where  $S_0$  and  $\gamma$  are constants.

- (i) Find the mean squared value of  $x(t)$ , and derive an expression for the double-sided spectral density  $S_{xx}(\omega)$ . [20%]

- (ii) The random process  $x(t)$  is the output of a linear system which is excited by a white noise force  $F(t)$  which has double sided spectral density  $S_0$ . Deduce that the governing differential equation must be

$$A\dot{x} + Bx = F(t)$$

- and find the constants  $A$  and  $B$  in terms of  $\gamma$ . [20%]

- (iii) Calculate the mean squared velocity  $E[\dot{x}^2]$  and comment on the physical relevance of your result. [20%]

- (b) A linear system has the equation of motion

$$a_N \frac{d^N x}{dt^N} + a_{N-1} \frac{d^{N-1} x}{dt^{N-1}} + a_{N-2} \frac{d^{N-2} x}{dt^{N-2}} + \dots + a_0 x = F(t)$$

where  $F(t)$  is white noise excitation. Consider the mean squared value of the  $m$ th derivative of  $x(t)$  :

$$E \left[ \left( \frac{d^m x}{dt^m} \right)^2 \right]$$

For what range of values of  $m$  does this quantity have a finite value? Comment on the practical implications of this result. [40%]

2 An item of equipment of mass  $M$  is mounted inside a spacecraft launch vehicle via a spring-damper system of stiffness  $K$  and damper rate  $C$ . The equation of motion of the system can be written as

$$M\ddot{x} + C\dot{x} + Kx = F(t)$$

where  $x(t)$  is the displacement of the equipment and the force  $F(t)$  arises from acoustic excitation of the vehicle and can be approximated as white noise with double-sided spectral density  $S_0$ .

(a) The equipment will suffer impact damage if the response  $x(t)$  exceeds the clearance  $b$  between the equipment and the spacecraft sidewall. Derive an expression for the probability that impact damage will occur for a launch of duration  $T$  seconds. [35%]

(b) For a given design, the impact probability is found to be too high. The design can be improved by increasing the clearance  $b$ , or by modifying the spring-damper system. The effective “cost” of the design is found to be proportional to the product  $b^3CK$ . Explain how the design should be modified to reduce the impact probability in the most cost efficient way. [30%]

(c) The stress in the equipment mount is given by  $S = Kx$ , and the S-N law governing fatigue failure of the mount has the form

$$N = \alpha S^{-\beta}$$

where  $\alpha$  and  $\beta$  are constants with  $\beta > 1$ . Derive an expression for the fatigue damage accumulated by the equipment mount for a launch of duration  $T$ . If the predicted fatigue damage is too great by a factor of 2, explain how you would modify the design of the mount to produce an acceptable result. [35%]

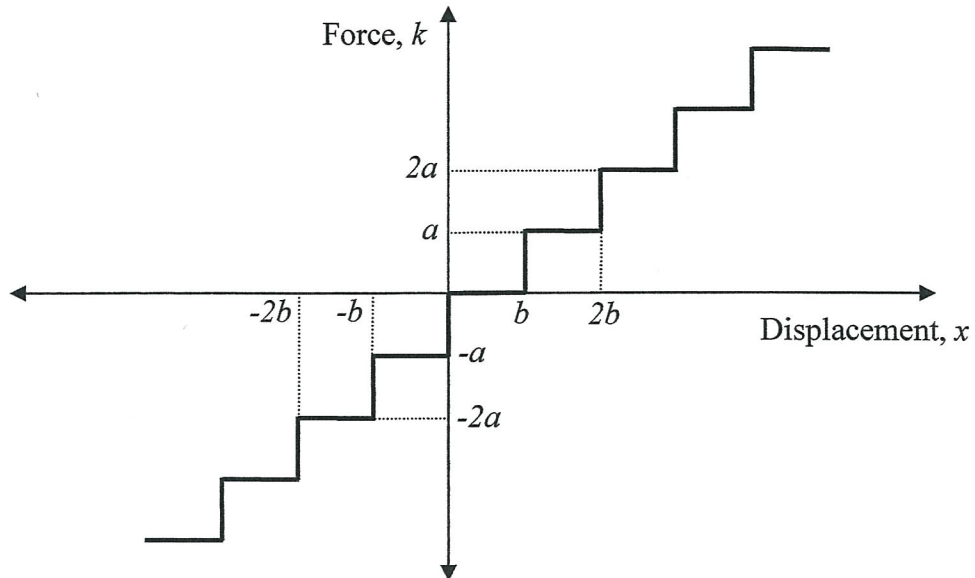
3 The free vibration behaviour of a particle is described by the following non-linear equation

$$\ddot{x} + k_0x + k_1x^2 - k_2x^3 = 0$$

where  $k_0$ ,  $k_1$  and  $k_2$  are real, positive constants.

- (a) Identify the equilibrium or singular points of the system. [20%]
- (b) Derive the type and stability of each singular point. [40%]
- (c) Sketch the behaviour of the system in the phase plane. [30%]
- (d) Sketch the restoring force as a function of vibration amplitude and show that this sketch agrees with the behaviour described in the phase plane. [10%]

4 An undamped non-linear system of unit mass has a force-displacement characteristic given by a staircase function as shown below



- (a) Sketch the input and output waveforms when this system is sinusoidally driven with amplitude  $\alpha$ , where  $b < \alpha < 2b$ . [20%]
- (b) Derive the Describing Function for this system for input amplitude  $\alpha < 2b$ . [50%]
- (c) If the system is driven by a force  $A \cos \omega t$ , determine an approximate relation between response amplitude  $\alpha$  and  $\omega$  for  $\alpha < 2b$ . [30%]

**END OF PAPER**

